# N20RD 2001 Maule MT-7-260

## Airplane Flight Manual

Aircraft S/N: 27008C



Prepared by the worldwide aviation specialists at RidgeAire, Inc.

### Revised Weight and Balance Data Sheet and Equipment Supplement

Date

May 26, 2004

Make & Model

**Maule MT-7-260** 

Registration

N20RD

Date

Serial Number 27008C

#### Superceded Weight & Balance

Old CG

Old Moment

Old Weight

	· · · ·			
February 9, 2001	1701.0	11.82	20	105.82
<b>Equipment Removed</b>	Weight	Arm	Moment	S/N
Garmin GNC-250XL GPS/COM	-2.10	-2.3	4.83	83105425
PM 1000II Intercom	-0.80	-2.8	2.24	FA02732-E02732
Garmin GA-56 GPS Antenna	-0.50	42.0	-21.00	59049590
Equipment Added	Weight	Arm	Moment	S/N
Garmin GMA-340 Audio Panel	1.70	-3.0	-5.10	96269284
Garmin GNS-430 GPS/Nav/Com	6.50	-5.0	-32.50	97120758
Garmin GI-106A Indicator	1.25	-3.0	-3.75	E04-10216
King KX-155 Nav/Com	5.00	-5.0	-25.00	66516
King KI-208 Indicator	1.20	-3.0	-3.60	24781
CI-121 Comm Antenna	1.00	42.0	42.00	4044656
CI-102 Marker Beacon Antenna	1.00	4.0	4.00	4022567
Garmin GA-56 GPS Antenna	0.50	42.0	21.00	59093197
CI-505 Diplexer	0.20	-9.0	-1.80	9411

#### Current Weight & Balance Data

New Empty Weight

Empty Weight C. G.

**New Moment** 

1715.95

11.71

20087.14

Johnny C. Blain

Avionics Unlimited

FAA CRS #VUER010Y

10015 MILITARY DR.

Conroe, Texas 77303



#### **FAA APPROVED**

#### AIRPLANE FLIGHT MANUAL

FOR **MAULE MT-7-260** 

Airplane Serial No	
	•
Registration No	

THIS DOCUMENT MUST BE KEPT IN THE AIRPLANE AT ALL TIMES.

FAA APPROVED:

Manager, Aircraft Certification Office Federal Aviation Administration Atlanta, Georgia USA

DATE: APR 1 7 1998



EXPERIENCE THE PERFORMANCE

2099 Georgia Hwy. 133 South~Moultrie, GA 31768

Tel: 912-985-2045~Fax: 912-890-2402

(229) 985 -2045

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#### **LOG OF REVISIONS**

REV.	TO PAGES	DESCRIPTION	APPROVAL AND DATE
A	5, 6 12 19	Added and revised Placards.  Corrected Crosswind Component.  Corrected omission.	Marger, Aircraft Certification Office, FAA Atlanta, Georgia
В	<b>4</b>	Vendor part number change.	Eugene X. Bollin Manger, Aircraft Certification Office, FAA Atlanta, Georgia Date: 6-23-99
С	12	Corrected (fully retracted) to read (first notch) in Item F.	Eugene K. Bollin  Manger, Aircraft Certification  Office, FAA  Atlanta, Georgia  Date: 9-10-99
D	2 4 13	Corrected unusable fuel to read 2.3 gallons.  Added McCauley Propellers B2D37C224-[]/[]-90RA-10.5 and B3D32C414-[]/[]-82NDA-2.	Manager, Atlanta Certification Office, FAA Atlanta, Georgia  Date: 4-17-00
		,	

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#### **LOG OF SUPPLEMENTS**

SUPP. NO.	NO. OF PAGES	DESCRIPTION	APPROVAL DATE
1	21	Installation of <b>Wipline 3000 Amphibious Floats</b> per Maule Drawing <b>9186A</b> .	03/12/99
2	8	Installation of <b>Aqua 2400 Floats</b> per Maule Drawing <b>9135A</b> , Rev. <b>E</b> or later revision.	11/19/99
3	6	English to <b>Metric Conversion Charts</b> - required in aircraft when registered <b>in Canada</b> .	09/03/99

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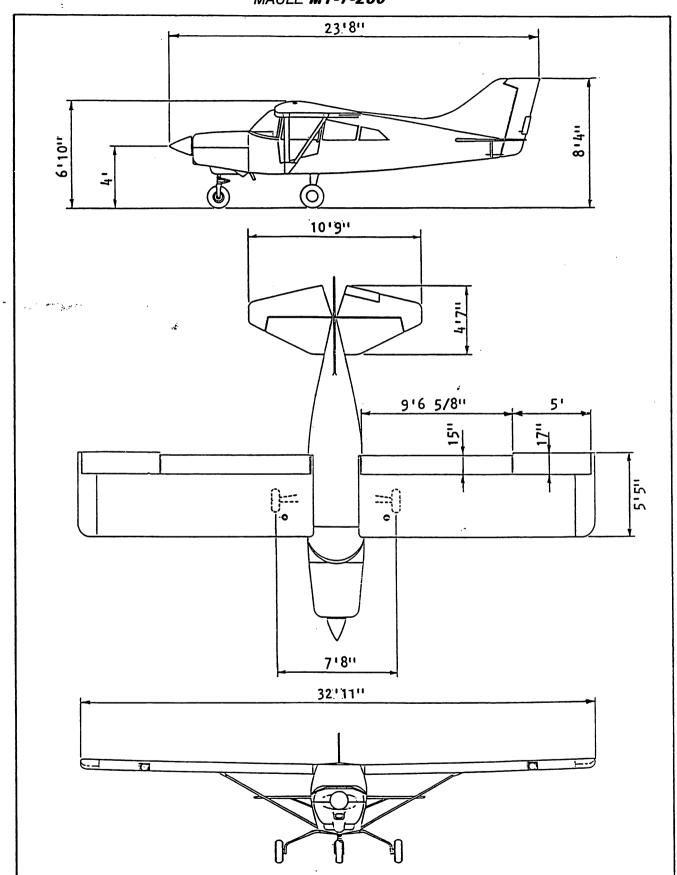
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**SECTION I GENERAL** 

SECTION I

**GENERAL: NORMAL CATEGORY OPERATION** 

1.1 **MAXIMUM WEIGHT: 2500 Pounds** 

**CENTER OF GRAVITY LIMITS:** +15.0 to +20.0 @ 2500 lbs.

+12.5 to +20.0 @ 1700 lbs. or less

Straight line variation between points given

Datum: Wing Leading Edge

NOTE: It is the responsibility of the pilot to assure that the airplane is property loaded.

Refer to the Weight and Balance Data for baggage/cargo loading recommen-

dations and loading graphs.

*IIIICAUTIONIIII* 

CHECK WEIGHT AND BALANCE CAREFULLY, ESPECIALLY WHEN USING THE 5<sup>TH</sup> SEAT OR WHEN CARGO OR BAGGAGE IS CARRIED IN THE REAR CABIN AREA. ALSO, FLIGHT PLANNING SHOULD INCLUDE

ALLOWANCE FOR FORWARD C.G. SHIFT WITH FUEL BURN.

MANEUVERS: Only Normal Category Maneuvers including Stalls, Lazy Eights, Chandelles and steep turns involving bank angles not greater than 60° are approved in this airplane.

*IIIICAUTIONIIII* 

AEROBATICS AND INTENTIONAL SPINS PROHIBITED.

1.4 FUEL CAPACITY:

Usable Fuel: MAIN TANKS - 21.5 Gal. each

OPTIONAL AUXILIARY TANKS - 15.0 Gal. or 21.0 Gal. each\*

Unusable Fuel: 2.3 Gallons per Main Tank

\*Fuel Capacity - See Instrument Panel Placard for Auxiliary Tank configuration installed in this aircraft.

**IIIICAUTIONIIII** 

FUEL REMAINING IN TANK WHEN INDICATOR READS EMPTY CANNOT

BE USED SAFELY IN FLIGHT.

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Rev. D dated: APR 1 7 2000

#### SECTION II

#### **LIMITATIONS**

- 2.1 <u>AIRSPEED LIMITS</u>: All airspeeds are Indicated Airspeeds (IAS).
  - A. AIRSPEED INDICATOR MARKINGS:

Red Radial, (V<sub>NF</sub>) - 182 mph (158K)

Yellow Arc, Caution Range - 147 - 182 mph (128 - 158K)

Green Arc, Normal Operating Range - 62 - 147 mph (54 - 128K)

White Arc, Flap Operating Range - 50 - 95 mph (43 - 83K)

- B. EXPLANATION OF AIRSPEED INDICATOR MARKINGS:
  - Red Radial Line Never Exceed Speed ( $V_{\rm NE}$ ) 182 mph (158K): Maximum safe airspeed in smooth air.
  - Yellow Arc Caution Range, 147-182 mph (128-158K): Operation in this speed range should be conducted only in smooth air and control movements should not be large or abrupt.
  - Green Arc Normal Operating Range, 62-147 mph (54-128K): Extends from flaps up, power off stall speed at 2500 lbs. ( $V_{\rm S1}$ ) to design cruise speed ( $V_{\rm C}$ ).
  - White Arc Flap Operating Range, 50-95 mph (43-83K): Extends from full flap, power off minimum stall speed at 2500 lbs. ( $V_{SO}$ ) to the Maximum flaps extended speed ( $V_{FE}$ ).

SECTION II LIMITATIONS

#### 2.2 POWER PLANT LIMITS:

Engine:

Lycoming IO-540-V4A5

**Engine Limits:** 

260 hp @ 2700 RPM, Full Throttle Continuous

Propeller:

MT-Propeller:

MTV-14-B/190-17

Hartzell:

HC-C2YR-1BF/F8477D-6

HC-C3YR-1RF/F7693(F)-( )\*

McCauley:

B2D37C224-[ ]/[ ]-90RA-10.5 or -12

B3D32C414-[]/[]-82NDA-2 or -4

Fuel:

100/100LL Minimum Grade Aviation Gasoline

**Engine Instrument Markings:** 

**Cylinder Head Temperature:** 

Green Arc - Normal Operating Range,

200°F - 435°F

Red Radial - Operating Limit, 500°F

Oil Temperature:

Green Arc - Normal Operating Range,

140°F - 245°F

Red Radial - Operating Limit, 245°F

Oil Pressure:

Green Arc - Normal Operating Range,

55 to 95 psi

Yellow Arc - Caution Range, 25 to 55

and 95 to 115 psi

Red Radial - Minimum Operating Pressure,

25 psi, Maximum Operating

Pressure, 115 psi

Manifold Pressure

Green Arc - Normal Operating Range,

14.5 to 29 ins. of Mercury

**Fuel Flow** 

Red Radial - Maximum, 8.9 psi or 26.9 gph

Tachometer:

Green Arc - Normal Operating Range,

2000 - 2700 RPM

Red Radial - Maximum RPM, 2700 RPM

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Rev. D dated: APR 1 7 2000

<sup>\*</sup> Limited to no dash number (78" diameter) to -2 (76" diameter)

SECTION II
LIMITATIONS

2.3 FLIGHT LOAD FACTORS:

Flaps Fully Retracted: 3.8g Positive to 1.5g Negative

Flaps Extended: 1.9g Positive to 0g Negative

NOTE: DESIGN MANEUVERING SPEED: The maximum safe airspeed at which full aerodynamic controls can be applied (V<sub>A</sub>) is 125 mph (109K). This airspeed is not marked on the airspeed indicator.

#### 2.4 PLACARDS:

4,945

The following placards are in the cockpit in clear view of the pilot:

"THIS AIRPLANE MUST BE OPERATED AS A NORMAL CATEGORY AIRPLANE IN COMPLIANCE WITH THE OPERATING LIMITATIONS STATED IN THE FLIGHT MANUAL AND IN THE FORM OF PLACARDS AND MARKINGS."

"NO AEROBATIC MANEUVERS INCLUDING SPINS, APPROVED."

"ROUGH AIR OR MANEUVERING SPEED: 125 MPH (109K) I.A.S."

"THIS AIRPLANE APPROVED FOR DAY OR NIGHT IFR NON-ICING FLIGHT WHEN EQUIPPED IN ACCORDANCE WITH FAR 91 OR FAR 135."

"DO NOT TURN OFF ALTERNATOR IN FLIGHT EXCEPT IN CASE OF EMERGENCY."

"FUEL REMAINING IN TANK WHEN INDICATOR READS ZERO CANNOT BE USED SAFELY IN FLIGHT.

"SEE LOADING INSTRUCTIONS IN WEIGHT AND BALANCE SECTION OF AIRPLANE FLIGHT MANUAL."

"DEMONSTRATED CROSSWIND 14 MPH"

On the instrument panel at the auxiliary tank transfer switches:

**FUEL TRANSFER PUMPS** 

PUSH FOR AUX. QUANT.

PUSH FOR AUX. QUANT.

LEFT

**RIGHT** 

FUEL CAPACITY: MAIN TANKS 21.5 GAL. USABLE EACH, AUX. TANKS \* GAL. USABLE EACH.

\* 15 Gal. or 21 Gal. Instrument Panel Placard will show capacity of the tanks installed in this aircraft.

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SECTION II **LIMITATIONS** 

2.4 PLACARDS: (Cont'd)

At the main fuel tank selector valve on the left kick panel:

**FUEL SELECTOR VALVE** 

**LEFT: 21.5 GAL.** 

OFF

**BOTH** 

RIGHT: 21.5 GAL.

Located on flap control handle:

"FLAPS / PULL ON / 2<sup>ND</sup> NOTCH / TAKEOFF / 3<sup>RD</sup> NOTCH / LANDING."

In rear cabin area:

"CHECK WEIGHT AND BALANCE CAREFULLY WHEN USING  $5^{\text{TH}}$  SEAT OR LOADING REAR/CARGO/BAGGAGE." "MAX. REAR SEAT LOADING IS 170 LBS."

**"CARGO OR BAGGAGE LIMITATIONS** MAX. LOAD AREA "A" 170 LBS. MAX. LOAD AREA "B" 350 LBS. MAX. LOAD AREA "C" 250 LBS.

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SECTION III NORMAL PROCEDURES

SECTION III

#### **NORMAL PROCEDURES:**

#### 3.1 PREFLIGHT INSPECTION:

Α.	INT	ER	IOR:
Α.	117		IUN.

O VERIFY
15
<del>1</del> )

EXTERIOR: Begin at the left front door, proceed around the left wing to the nose area, then around the right wing and back to the fuselage, then around the tail section.

1. 2.	Fuel drains behind stepLeft Flap	DRAIN (2) CHECK HINGES & CONTROL
۷.	Leit Flap	ATTACHMENTS
3.	Aileron	
3.	Alleion	ATTACHMENTS
4.	Left Wing Top	
₹.	Left Willig Top	INDICATION OF INTERNAL
		DAMAGE
5.	Left Wing Main & Aux Fuel Tank Drain	DRAIN (2)
6.	Left Wing Tip & Nav Light	CHECK FOR DAMAGE
7.	Auxiliary Fuel Tank	VISUALLY CHECK QUANTITY
8.	Landing Light	CHECK FOR DAMAGE
9.	Left Wing Tiedown	. REMOVE
10.	Pitot Tube	. REMOVE COVER
11.	Stall Warning Switch	. CHECK FOR FREEDOM OF
	•	MOVEMENT
12.	Main Fuel Tank	. VISUALLY CHECK QUANTITY
13.	Left Landing Gear	CHECK TIRE INFLATION AND
		BRAKE LINE SECURITY
14.	Bottom left side of Cowl	. DRAIN GASCOLATOR (1)
15.	Top Cowl, Oil Access Door	CHECK OIL QUANTITY
16.	Propeller	CHECK LEADING EDGE FOR
		DAMAGE. CHECK SPINNER
		FOR SECURITY
17.	Air Inlets	INSPECT VISIBLE CONNECTIONS
		AND COMPONENTS
		AND COMPONENTS

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Altro Tube Switch

## MAULE AEROSPACE TECHNOLOGY, INC. SECTION III AIRPLANE FLIGHT MANUAL NORMAL PR MAULE MT-7-260

SECTION III NORMAL PROCEDURES

#### 3.1 PREFLIGHT INSPECTION: (Cont'd)

	18.	Nose Gear	. CHECK TIRE INFLATION & STRUT
			EXTENSION, CLEAN STRUT
	19.	Right Landing Gear	. CHECK TIRE INFLATION &
			BRAKE LINE SECURITY
	20.	Right Wing & Controls	. INSPECT SAME AS LEFT WING
	21.	Wing Main & Aux Fuel Tank Drain	. DRAIN (2)
	22.		INSPECT FOR WRINKLES
			AS INDICATION OF
			INTERNAL DAMAGE
	23.	Right Side Static Port	CLEAR
	24.	Right Stabilizer	CHECK ATTACHMENT POINTS
		- <b></b>	& FLYING WIRES
	25.	Right Elevator	CHECK HINGE POINTS
	26.		CHECK HINGE POINTS,
			CONTROL ATTACHMENT
			& NAV. LIGHT
	27.	Tail Skid	CHECK FOR DAMAGE,
			REMOVE TIEDOWN
	28.	Left Elevator	CHECK TAB CONTROLS
			& ALL HINGE POINTS
	29.	Left Stabilizer	CHECK ATTACHMENT POINTS
			& STRUT
	30.	Left Fuselage, Side, Top & Bottom	CHECK FOR WRINKLES AS
			INDICATION OF INTERNAL
			DAMAGE
	31.	Left Side Static Port	CLEAR
3.2	<u>OPER</u>	ATING CHECK LISTS:	
	A. BE	EFORE STARTING:	
	·		FI A OTENED
	1.		FASTENED
	2.		RETRACTED
	3.	Circuit Breakers	CHECK
	B. S	TARTING:	
	4	Dading or Too Brokes	ON
	1.	<u> </u>	ON FULLEST TANK OR
	2.	ruei Selector valve	BOTH IF SAME QUANTITY
	^	Throttle	
	3.	· · · · · · · · · · · · · · · · · · ·	FULL INCREASE RPM
	4.		RICH (SEE NOTE NEXT
	5.		PAGE FOR HOT START)
	_	. Anti-Collision Light	
	6.	. AIII-COIIISION LIGHT	🔾 11

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SECTION III NORMAL PROCEDURES

#### 3.2 OPERATING CHECK LISTS: (Cont'd)

7.	BAT and ALT Switch	ON
8.	Prime	AS REQUIRED USING BOOST PUMP
9.	Mixture Control	FULL LEAN
10.	Starter Switch	TWIST FULL RIGHT TO ENGAGE
11.	Mixture Control	FULL RICH WHEN ENGINE
		STARTS TO FIRE

NOTE: FOR A HOT START, DO NOT PRIME. A HOT ENGINE MAY FLOOD ON A START ATTEMPT. TO CLEAR A FLOODED ENGINE, PULL MIXTURE FULL LEAN AND OPEN THROTTLE, CRANK WITH STARTER. WHEN ENGINE STARTS, PULL THROTTLE TO IDLE AND EASE MIXTURE TO FULL RICH.

NOTE: FOR A COLD ENGINE (FIRST START OF THE DAY), PLACE MIXTURE TO FULL RICH, THROTTLE 1/4" OPEN. PRIME WITH BOOST PUMP FOR 3 TO 5 SECONDS. IF ENGINE DOES NOT START, PRIME FOR ANOTHER 3 TO 5 SECONDS. OVER-PRIME CAN BE NOTED BY FUEL COMING FROM DRAIN IN CENTER OF BOTTOM COWL.

IN EVENT OF ENGINE FIRE, CONTINUE CRANKING. PULL MIXTURE TO FULL LEAN. IF ENGINE FAILS TO START AFTER SEVERAL REVOLUTIONS, AND FIRE CONTINUES, SECURE IGNITION, BAT. AND ALT. SWITCHES, TURN FUEL VALVE OFF AND EXIT AIRCRAFT.

12. After Starting..... CHECK OIL PRESSURE

IF OIL PRESSURE DOES NOT EXCEED 25 PSI WITHIN 30 SECONDS, SHUT DOWN ENGINE.

13.	Alternator	CHECK CHARGING
	Radios & other electrical switches	
15	Parking Brake	OFF

#### C. ENGINE CHECK:

1	Parking Brake	ON, IF DESIRED
2	Engine Instruments	CHECK, IN GREEN ARCS
3	Throttle	INCREASE TO 2000 RPM
4	Magnetos	SWITCH TO RIGHT, LEFT,
٦.	Magnotoo	BOTH, CHECKING RPM DROPS

A RPM DROP OF MORE THAN 175 RPM OR A DIFFERENCE BETWEEN LEFT AND RIGHT OF MORE THAN 50 RPM IS UNACCEPTABLE.

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SECTION III NORMAL PROCEDURES

#### 3.2 OPERATING CHECK LISTS: (Cont'd)

		Propeller Control	OF 500 RPM DROP IS NOTED. RETURN TO FULL RPM. REPEAT. SET FULL INCREASE RPM
	6.	Alternate Air Control	NORMAL RPM DROP WITH ALTERNATE AIR IS APPROXIMATELY 50 RPM
	7.	Alternate Air Control	. PUSH IN AND TURN RIGHT TO LOCK
	8.	Vacuum Gauge	. CHECK IN GREEN
		Alternator	
			RPM
	10.	Throttle	
		Parking Brake	
		Taking Diake	. 011
D.	REF	FORE TAKEOFF:	
D.	יייי	ONE PAREOTT.	
	1.	Fuel Selector	ON FULLEST TANK OR BOTH
	2.	Flaps	
	2. 3.	Trim Controls	
	4.	Flight Controls	
	_		PROPER TRAVEL
	5.	Mixture Control	FULL RICH
	6.	Propeller Control	FULL INCREASE RPM
	7.	Alternate Air Control	
	8.	Engine Instruments	RECHECK IN NORMAL RANGE
	9.	Radios	AS DESIRED
	10.	Altimeter	SET
	11.	Attitude Indicator	CHECK ERECT
	12.	Directional Indicator	
	13.	Seat Belts & Shoulder Harnesses	
	14.	Doors	
	15.	Passengers	BELTS & HARNESSES SECURED.
	13.	rassengers	BRIEFED ON OPENING DOORS.
	16	Parking Brake	
	10.	Parking Drake	Orr
E.	BE	FORE LANDING:	
	1.	Seat Belts & Shoulder Harnesses	FASTENED
	2.	Fuel Selector Valve	
	3.	Mixture Control	
	4.	Propeller Control	FULL INCREASE RPM
	5.	Flaps	AS REQUIRED
	5. 6.	Alternate Air Control	
	O.	AILEITIALE AII COILLOI	III AND LOOKLD

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SECTION III NORMAL PROCEDURES

#### 3.2 OPERATING CHECK LISTS: (Cont'd)

#### F. ENGINE SHUT DOWN:

1.	Parking Brakes	. ON, IF DESIRED
2.	Radios	.OFF
3.	All other electrical switches	. AS DESIRED
4.	Flaps	AS DESIRED
5.	Magneto Grounding Check	. PERFORM BELOW 1000 RPM
6.	Mixture Control	
7.	Magneto Switch	.OFF
8.	Anti-Collision Light	. OFF
9.	BAT & ALT Switch	. OFF
10.	Parking Brake	. OFF (AS DESIRED)

#### 3.3 NORMAL FLIGHT OPERATIONS:

#### A. NOTE: FLAP SETTINGS:

The following Flap Settings are available:

Flap Configuration	Flap Handle Position	Flap Position
Cruise	Handle Full Down	-7°
Flaps Up	First Notch	0°
Takeoff	Second Notch	<b>24°</b>
Landing	Third Notch	40°

#### **B. RECOMMENDED FLAP SETTINGS:**

Flap settings are given in number of notches above the fully retracted position which is handle full down (Normal -7°).

NOTE: The airplane meets CAR 3 takeoff climb requirements at 90 mph IAS with the flaps selected in any of the following three positions: (a) Fully Retracted, Handle full down (-7°), (b) First Notch (0°), and (c) Second Notch (24°).

Normal Takeoff - Second Notch (24°)

Normal Climb - First Notch (0°)

Best Angle of Climb - Second Notch (24°)

Cruise - Fully retracted (-7°/no notches or 0°/1st notch)

Landing - Normally Third Notch (40°/full flaps) - other positions optional

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SECTION III NORMAL PROCEDURES

#### 3.3 NORMAL FLIGHT OPERATIONS: (Cont'd)

#### C. CLIMBING:

Best Rate of Climb - 90 mph IAS, flaps @ First Notch (0°)

Best Angle of Climb - 75 mph IAS with flaps set @ Second Notch (24°)

FOR TAKEOFF OR LANDING UNDER GUSTY CROSSWIND CONDITIONS, FLAP SETTING OF 0° (one notch) IS RECOMMENDED. -7° OPTIONAL.

USE CLIMB AIRSPEED BELOW 90 MPH ONLY AS NECESSARY AND CHECK CYLINDER HEAD TEMPERATURE FREQUENTLY WHEN

DOING SO.

#### D. RUDDER TRIM:

NOTE: To assure full effectiveness of the Right Rudder Trim:

Unlock "T" handle (1/2 turn left), depress right rudder as you pull "T" handle full out. Lock "T" handle ½ turn right before releasing right rudder pressure. If too much trim, move handle in until trim is correct and then lock.

#### E. STALLS:

Stalls are preceded by mild buffet that can be felt through the rudder pedals. The red stall warning light on the instrument panel will illuminate at 5 to 10 mph above the stall speed. Loss of altitude prior to recovery from a stall may be as much as 300 feet.

////CAUTION/// ////CAUTION///

THE STALL WARNING LIGHT IS INOPERATIVE WHEN THE BATTERY SWITCH IS OFF.

#### F. CROSSWIND LANDINGS & TAKEOFFS:

Maximum demonstrated crosswind component is 14 mph and flap extension should be limited to 0° (first notch) with such crosswind or higher. 14 mph is the maximum demonstrated for certification of the airplane and is not considered limiting with flaps at 0°.

Fuel is fed to the engine from the main (inboard) tanks and is controlled by the selector valve on the left kick panel. Auxiliary (outboard) tanks feed their respective main tanks via transfer pumps that are controlled by switches on the instrument panel.

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Rev. C dated: SEP 1 0 1999

**SECTION III NORMAL PROCEDURES** 

#### G. FUEL SYSTEM MANAGEMENT:

These transfer pumps transfer fuel at a rate of 0.4 gallons per minute or approximately 45 minutes for a full auxiliary tank. Since overfilling a main tank from an auxiliary tank will force excess fuel overboard, it is recommended that the transfer pumps not be activated until their respective main tanks are slightly more than one quarter full. If the tank being transferred to is feeding the engine, however, transfer can be initiated when the main tank is down to approximately one half. Confirm fuel transfer by illumination of the transfer pump switch, an increase in the respective main tank fuel gauge indicator, and a decrease on the respective auxiliary tank indicator.

#### DOOR-OFF OPERATION: 3.4

This aircraft may be operated with either one (not both) of the front doors removed, or when both front doors are installed, with the rear passenger door or rear passenger and baggage doors off. When doing so, observe the following additional limitations:

82.9 dBA

- 1. Maximum airspeed 125 mph
  - Maximum bank angle 30° 2.
  - Maximum yaw angle 10°
  - No Smoking permitted
  - Limit flight to VFR conditions

#### 3.5 NOISE LEVEL:

1

1

The noise level obtained during certification per FAR 36, Appendix G was:

with MT propeller	4 blade	74.8 dBA	
with Hartzell (78")	2 blade	78.0 dBA	
with McCauley (78")	2 blade	78.8 dBA	(79.5"): 80.1 dBA
with Hartzell	3 blade	78.0 dBA	
with McCauley (78")	3 blade	78.0 dBA	(80"): 78.2 dBA
The noise level obtained du	ring certification pe	r ICAO Anne	x 16 Chapter 10 was:
with MT propeller	4 blade	77.9 dBA	
with Hartzell (78")	2 blade	79.7 dBA	
with McCauley (78")	2 blade	80.2 dBA	(79.5"): 81.5 dBA
with Hartzell	3 blade	79.7 dBA	
with McCauley (78")	3 blade	79.7 dBA	(80"): 79.9 dBA

No determination has been made by the Federal Aviation Administration that the noise level of this airplane is or should be acceptable for operation at, into, or out of any airport.

#### **ANTI-COLLISION LIGHT:** 3.6

with Hartzell (81")

	ANTI-COLLISION LIGHT MAY CAUSE ADVERSE EFFECT ON PILOT
////WARNING////	WHEN FLYING IN VISIBLE MOISTURE OVERCAST OR HAZE. IT IS
///////////////////////////////////////	RECOMMENDED THAT IT BE TURNED OFF SHOULD PILOT
	DISCOMFORT BE NOTICED.

2 blade

FAA APPROVED: 4/17/98 Rev. D dated: APR 1 7 2000

### MAULE AEROSPACE TECHNOLOGY, INC. SECTION IV AIRPLANE FLIGHT MANUAL EMERGENC MAULE MT-7-260

SECTION IV EMERGENCY PROCEDURES

#### SECTION IV

#### **EMERGENCY PROCEDURES**

#### 4.1 <u>EMERGENCY BASIC RULES:</u>

To assist the pilot when an emergency occurs, three basic rules are established which apply to most emergencies occurring while airborne. Each aircrew member should remember them.

- 1. Maintain aircraft control
- 2. Analyze the situation and take proper action
- 3. Land as soon as conditions permit

#### 4.2 ENGINE EMERGENCY SHUT DOWN:

- 1. Mixture Full lean
- 2. Fuel Selector Off
- 3. Ignition Switch Off

#### 4.3 ENGINE FIRE DURING STARTING:

- 1. Mixture Full lean
- 2. Throttle Open
- 3. Continue cranking for several revolutions. Attempt to draw fire inside engine.
- 4. Accomplish ENGINE EMERGENCY SHUT DOWN if fire continues.

#### 4.4 ENGINE FIRE AFTER STARTING:

- 1. Accomplish ENGINE EMERGENCY SHUT DOWN
- 2. Master Switch Off

#### 4.5 EMERGENCY EXIT ON THE GROUND:

- 1. Accomplish ENGINE EMERGENCY SHUT DOWN
- 2. Master Switch Off
- 3. Leave aircraft by either door or kick out side window panels or baggage door.

#### 4.6 TAKEOFF ABORT: (BEFORE LIFT-OFF)

- 1. Throttle Closed
- 2. Brakes As Required

#### 4.7 ENGINE FAILURE AFTER TAKEOFF OR FORCED LANDING:

- 1. Glide Establish 80 mph IAS with flaps at 0°
- 2. Switch Fuel Selector to fullest tank

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DATE: APR 17 1998

## MAULE AEROSPACE TECHNOLOGY, INC. SECTION IV AIRPLANE FLIGHT MANUAL EMERGENC MAULE MT-7-260

SECTION IV EMERGENCY PROCEDURES

#### 4.7 ENGINE FAILURE AFTER TAKEOFF OR FORCED LANDING: (Cont'd)

- 3. Electric Fuel Pump On
- 4. Mixture Rich, Ignition On
- 5. Alternate Air Control Pull On
- 6. If engine does not restart, accomplish EMERGENCY SHUT DOWN
- 7. Wing Flaps As Required
- 8. Master Switch Off

#### 4.8 PARTIAL POWER FAILURE DURING FLIGHT OR AFTER TAKEOFF:

- 1. Mixture Rich
- 2. Alternate Air Control Pull On
- 3. Airspeed Glide at 80 mph IAS if unable to maintain level flight
- 4. Fuel Selector Both
- 5. Electric Fuel Pump On
- 6. Ignition Switch Both
- 7. Master Switch On

#### 4.9 COMPLETE POWER FAILURE DURING FLIGHT:

- 1. Glide Establish 80 mph IAS
- 2. Attempt engine airstart if warranted

#### 4.10 ENGINE AIRSTART:

- 1. Fuel Selector Both
- 2. Electric Fuel Pump On
- 3. Mixture Rich
- 4. Ignition Switch Both (start if propeller is not turning)
- 5. If engine does not start, try flooded engine clearing procedure with throttle wide open and mixture full lean.
- 6. If no start, make forced landing

NOTE: PROPELLER WILL NOT WINDMILL BELOW 70 MPH.

NOTE: AT ALTITUDES OVER 8000 FEET, A LEANER MIXTURE MAY BE REQUIRED.

#### 4.11 ELECTRICAL FIRE:

1. Master Switch - Off

#### 4.12 ENGINE FIRE DURING FLIGHT:

- 1. Accomplish ENGINE EMERGENCY SHUT DOWN
- 2. Make forced landing

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DATE: APR 17 1998

SECTION IV EMERGENCY PROCEDURES

#### 4.13 SMOKE AND FUME ELIMINATION:

- 1. Cabin Heat Knob In
- 2. Cabin Air Knob In
- 3. Upper Air Vents Open
- 4. Pilot's Window Open (below 120 mph)

#### 4.14 STRUCTURAL DAMAGE:

- 1. On Takeoff Abort
- 2. In flight, maintain controllable airspeed
- 3. Climb to safe stall recovery altitude
- 4. Notify appropriate controlling agency, if appropriate.
- 5. Determine control difficulty airspeed by slowing down while flying straight ahead. Do not allow the aircraft to stall.
- 6. Make full stop landing using 5-10 mph above difficulty airspeed or above normal approach speed, whichever is higher.

#### 4.15 RECOVERY FROM INADVERTENT SPINS:

Intentional spins are prohibited. If the aircraft inadvertently enters a spin, simultaneously apply full rudder opposite to the direction of rotation and full nose down elevator with ailerons neutral and reduce power to idle. When the rotation stops, neutralize the rudder and elevator, and ease back on the control wheel as required to smoothly regain level flight. Wing flaps should be retracted to avoid exceeding the maximum flap speeds during recovery.

#### 4.16 ALTERNATOR FAILURE:

Alternator output should be monitored by reference to the ammeter located on the right side of the engine instrument cluster. Should the ammeter indicate a minus deflection when engine RPM is above 900 and/or red "ALTERNATOR OFF WARNING" light is illuminated, push ALT switch OFF then ON. Repeat two times as necessary to reset. If system will not reset, reduce the electrical load as much as possible, land as soon as practical and investigate the electrical system malfunction before further flight.

FAA APPROVED DATE: 100

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**SECTION V** WEIGHT AND BALANCE

**SECTION V** 

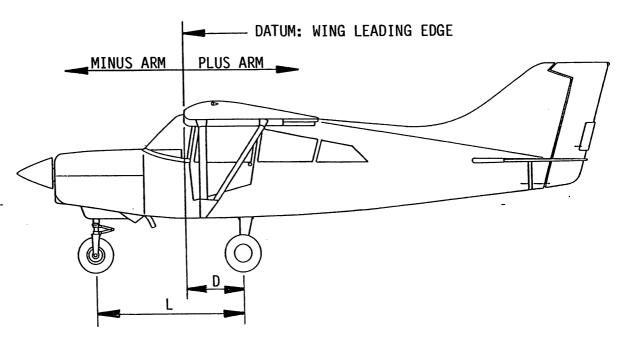
5 1	WEIGHT	AND	BAL	ANCE:
J. I	VVLIGITI	$\Delta$ IND		$\neg$ 11 $\cup$ $\vdash$ .

Serial Nu	ımber <u>27008C</u>	Registration Number	N20RD
loaded p	roperly. The empty weigh r this airplane as reweigh	e airplane owner and the pilot to nt, empty weight center of gravit ed after engine change. If the a raft records for this information.	y and useful load are listed
WE	IGHT AND BALANCE DA	TA SUMMARY AS DELIVERED	FROM THE FACTORY:
	Basic Empty Weight (inclu	ding engine oil)	<u>/70/</u> Lbs.
(	Gross Weight		<u>2500</u> Lbs.
1	Useful Load		<u>799</u> Lbs.
1	Empty Center of Gravity		Inches
	CENTER OF GRAVITY R	ANGE:	
	At Weight of	Center of Gravity Range	
	2500 lbs.	+15.0 to +20.0 inches	
	1700 lbs.	+12.5 to +20.0 inches	
	NOTE: Straight line varia DATUM: Wing leading ed	tion between given points dge	

**PAGE 17** 

FORM 71

DETAILED CALCULATIONS OF EMPTY WEIGHT AND EMPTY WEIGHT CENTER OF GRAVITY AS DELIVERED FROM FACTORY:



#### PROCEDURE:

- 1. Place each of the wheels on a scale with the airplane in approximate level flight attitude.
- Place a level on the leveling mark and leveling lug on the bottom of the right wing near the root. Adjust the extension of the nosewheel oleo until the aircraft is level, or flatten the tire if necessary.
- 3. Measure the following distances:
  - a. Wheel base (L) the <u>horizontal</u> distance from the nosewheel weight point (center of axle) to the main wheel weight point (center of axle).

b. Main Wheel Station (D) - the horizontal distance from the main wheel weight point (center of axle) to the datum line.

- 4. Measure the weights at the following points:
  - a. Right Main Wheel..... Lbs.
  - b. **Left Main Wheel.....** = <u>625</u> Lbs.
  - c. Nosewheel, (N)....= 466 Lbs.

Total Weight as Weighted (W) = \_\_\_\_/70/\_\_\_Lbs.

APK 17 1998

SECTION V WEIGHT AND BALANCE

#### 5.1 WEIGHT AND BALANCE: (Cont'd)

The above empty weight includes unusable fuel of 27.6 lbs. at 24 inches and 8 quarts of oil at minus 34 inches, plus all items of equipment as marked on the accompanying Equipment Lists. The Certificated empty weight is the above weight less 16 lbs. drainable oil at

a minus arm of 34 inches, and for this airplane is 1685 lbs. The corresponding empty weight center of gravity is 12-28 inches.

- 5. Calculations for determining weight, C.G. and moment:
  - a. Center of Gravity (inches) =  $D \underbrace{N \times L}_{W}$

b. Moment (inch pounds) = W x C.G.

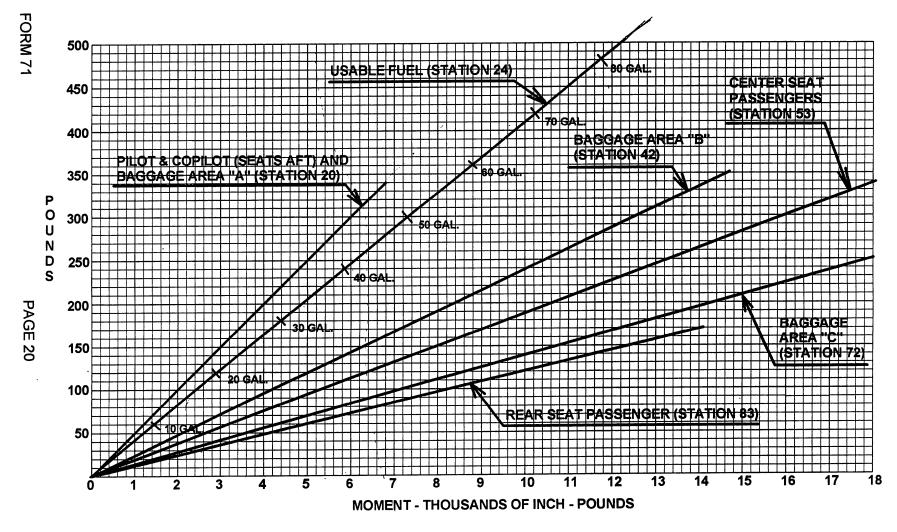
#### **EXAMPLE OF WEIGHT AND BALANCE CALCULATION FOR LOADED AIRCRAFT:**

An airplane with an empty weight of 1549 lbs. and empty weight C.G. location of 11.2 inches is loaded with a pilot and front seat passenger, fuel and baggage.

Item	Weight, lbs.	C.G. Location	Moment, In.Ibs
Empty Weight (including engine oil)	1549	11.2	17,349
Pilot and Front Passenger Fuel - 40 gal. in Mains	340	*	6,800
plus 30 gal. in Auxiliary Tanks	420	*	10,080
Baggage (Area "C")	150	*	<u> 10,800</u>
,	2459	18.3	45,029

<sup>\*</sup>Moments can be read directly from the loading graph.

By locating the point corresponding to 2459 lb. aircraft weight and a C.G. Location of 18.3 inches on the Center of Gravity envelope graph, you can see that this point falls within the envelope, signifying the loading is acceptable.

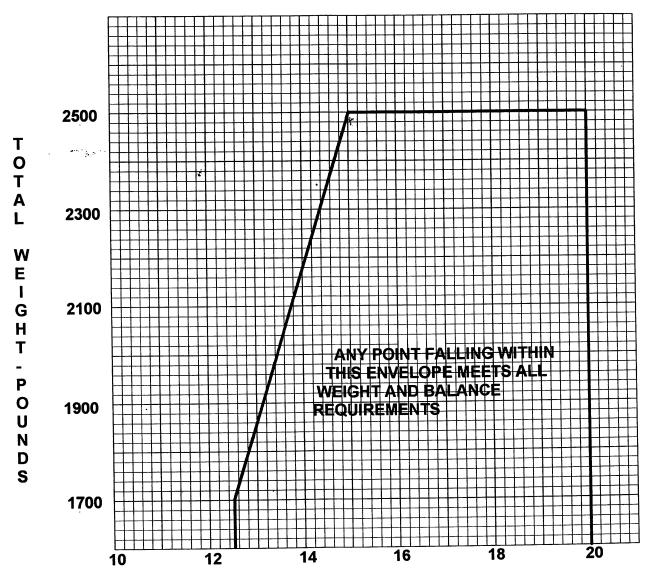


#### PROCEDURE FOR DETERMINING WEIGHT AND CENTER OF GRAVITY

#### **LOADING GRAPH**

- 1. Add weight of items to be carried to the licensed empty weight of the aircraft.
- 2. Find moments of items to be carried by using the above loading graph and add these moments to the empty moment of the aircraft. Divide total moment by total weight for aircraft C.G. location.
- 3. Using the C.G. location from Step 2, find the point on the Weight and Balance Envelope.

#### WEIGHT AND BALANCE ENVELOPE



C. G. LOCATION, INCHES AFT OF DATUM (DATUM: WING LEADING EDGE)

**MAULE MT-7-260** 

SECTION V WEIGHT AND BALANCE

#### STRUCTURAL CAPACITY CHART

ſ		DATUM - WING LEADING EDGE
PILOT'S SEAT	BAGGAGE AREA "A" (REF. STA. 20) MAX. LOAD 170 LBS.	
		FRONT SEAT BELT AND BAGGAGE STRAP ATTACHMENT, STATION 28
BAGGAGE (REF. S <u>MAXIMUM LC</u>		
		REAR SEAT BELT AND BAGGAGE STRAP ATTACHMENT, STATION 56 3/4
	E AREA "C" STA. 72)	
MAXIMUM L	OAD 250 LBS.	
		REAR SEAT BELT & BAGGAGE STRAP ATTACHMENT, STA. 84
		REAR OF CABIN, STATION 88 1/4

#### MAULE AEROSPACE TECHNOLOGY, INC. SECTION VI AIRPLANE FLIGHT MANUAL AIRPLANE S MAULE **MT-7-260** HANDLING 8

SECTION VI AIRPLANE SERVICING, HANDLING & MAINTENANCE

SECTION VI

A Maria

#### AIRCRAFT SERVICING, HANDLING AND MAINTENANCE

#### 6.1 INTRODUCTION:

Our dealers and distributors are anxious to serve you and will gladly furnish advice as to proper servicing methods. You may also address request for information on any items not covered in the manual to the Service Department of Maule Air, Inc. In correspondence, please be certain to give complete information on Serial Number, engine make and model, etc.

The aircraft Type Data Plate can be found on the left side of the vertical fin just above the horizontal stabilizer. Also, pertinent engine and propeller data is in the aircraft Log Book.

A Service Manual is furnished with each aircraft. Extra copies and a Parts Manual can be obtained by contacting the Service Dept. of Maule Air, Inc.

#### 6.2 AIRPLANE INSPECTION PERIOD:

The airplane must be maintained as outlined in FAR 43. Recommended inspections are outlined in the airplane Maintenance Manual. The owner/operator is responsible for Airworthiness Directives (AD's) that may be issued from time to time. Reference should be made to FAR 91 and FAR 43 requirements for properly certified agency or personnel to accomplish the required FAA inspection and most of the manufacturer's recommended inspections. It is recommended that owner's name and address along with aircraft serial number be registered with Maule Air for any Maule Service Letters or Bulletins released affecting their aircraft.

#### 6.3 PREVENTIVE MAINTENANCE THAT MAY BE ACCOMPLISHED BY A CERTIFIED PILOT:

- A. A certified pilot who owns or operates an airplane not used as an air carrier is authorized by FAR Part 43 to perform limited preventive maintenance on his airplane. Refer to FAR Part 43 for list of things the pilot may do. Pilots operating aircraft of other than U.S. registry should refer to the regulations of the country of certification for information on preventive maintenance that may be performed by pilots. All other maintenance required on airplane is to be accomplished by appropriately licensed personnel and that airplane dealer or service station should be contacted for further information.
- B. Preventive maintenance should be accomplished in accordance with the appropriate airplane Maintenance Manual. Manual should be obtained prior to performing preventive maintenance to be sure that proper procedures are followed.

#### 6.4 ALTERATIONS OR REPAIRS TO AIRPLANE:

Alterations or repairs to airplane must be accomplished by licensed personnel. The FAA should be contacted prior to any alterations on airplane to insure that Airworthiness of the airplane is not violated.

FAA APPROVED: DATE:APR 1 7 1998

## APR 17 1

#### MAULE AEROSPACE TECHNOLOGY, INC. AIRPLANE FLIGHT MANUAL MAULE MT-7-260

SECTION V WEIGHT AND BALANCE

REG.NO	MODEL						
EQUIPMENT CHANGE - WEIGHT AND BALANCE							
WEIGHT	ARM	MOMENTS					
lb	S.						
vityin:	S.						
Momentin	. lbs.						
lk	os.						
Supersedes all previous weight and balance data. For aircraft loading see instructions in original weight and balance forms.							
	DATE						
	HT AND BALANCE WEIGHT	WEIGHT ARM					

#### MAULE AEROSPACE TECHNOLOGY, INC.

M5-235C/180C/M6-235/180/MT7-235/260/M7-235/235A/B/C/260/C M8-235/MX7-235/160/160C/180/180A/B/C/AC/MXT7-160/180/180A

PAGE **1** of 9 REV. <u>53</u> DATE: 11/07/00

#### **REQUIRED EQUIPMENT LIST**

SERIAL NO. 27008C REG. NO. N20RD PROD. NO. 2299		
NOTE: Equipment items installed below are designated by circle		
	HT (Lbs)	ARM (Ins)
PROPELLER AND ACCESSORIES		
1. PROPELLER		
A. Hartzell		
1. HC-C2YR-1BF/F7666A (M5/6/MX7/MXT7180/MX7180B/C)	56.0	-61.3
2. HC-C2YR-1BF/F8468A-6R (M5/6/8/MX7/MT7235/M7235/A/B/C)	50.0	<del>-61.5</del>
3. HC-C2YR-1BF/F8468A-3R (M5/6/8/MX7/MT7235/M7235/A/B/C)	51.0	<del>-61.5</del>
4. HC-C2YR-1BF/F8477D-6* (M5/6/8/MX7235/M7235/A/B/C/MT7260/M7260/C)	50.0	<del>-61.5</del>
5. HC-C3YR-1RF/F7693(F)-( )* (M5/6/8/MX7235/M7235/A/B/C/MT7260/M7260/C)	73.0	<u>-61.5</u>
(-3R with 7:00 min. tires/26 psi min.) (*For –235, use with O-540-J3A5/B4B5 engines only)	10.0	<del>-01.0</del>
( Or with 7.50 thin. diesizo partitin.) ( 1 of -200, dae with 0-040-00/00-400 engines only)		
B. McCauley	-	
1. B3D32C414-[]/[]-82NDA-4 (M6/M7/M8/MX7/MT7/235/260/A/B/C)	66.5	-61.5
2. B3D32C414-[]/[]-82NDA-2 (M5/6/8/MX7/MT7235/260/M7235/A/B/C/260/C)	67.0	<u>-61.5</u>
3. B2D37C224-B/G-90RA-9* (M5/6T8/MX7/MT7235/A/B/C)	48.5	<u>-61.5</u>
4. B2D37C224-[]/[]-90RA-12 (MT7260/M7260/C)	48.5	<del>-61.5</del>
5. B2D37C224-[]/[]-90RA-10.5 (MT7260/M7260/C)	48.5	-61.5
(-9 with 7:00 min. tires/26 psi min.; * foreign use only with -260)	40.5	-01.5
(-5 with 7.00 min. thes/20 psi min., Toleigh use only with -200)		
C. Sensenich		
1. 74DM7S5-0-52 (MXT7/MX7-160/160C only)	36.3	-60.5
2. 74DM7S5-0-54 (MXT7/MX7-160/160C only)	36.3	-60.5
3. 74DM7S5-0-56 (MXT7/MX7-160/160C only)	36.3	-60.5
4. 74DM7S8-0-52 (MXT7/MX7-160/160C only)	36.3	-60.5
5. 74DM7S8-0-54 (MXT7/MX7-160/160C only)	36.3	-60.5
6. 74DM7S8-0-56 (MXT7/MX7-160/160C only)	36.3	-60.5
53 7. 76EM8S5-0-56 (MXT7-180A/MX7-180A/AC only)	36.4	-60.5
	36.4	-60.5
<u>/53</u> 8. 76EM8S8-0-56 (MXT7-180A/MX7-180A/AC only)	30.4	-00.5
D. MT		
1. MTV-14-B/190-17 (MT7260/M7260/C)	49.2	-61.5
1. WITV-14-D/190-17 (MIT/260/WI/260/C)	49.2	-01.5
2. SPINNER & BULKHEAD ASSEMBLY		
2. SPINNER & BULKHEAD ASSEMBLY		
A. Hartzell		
1. A2298-2 (Use with propeller A.1,2,3,4 only)	4.0	-62.2
2. 82C-3535-1P (Use with propeller A.5 only)	5.0	-62.2
2. 020-000-11 (Ose with properlet A.0 only)		<u> </u>
B. McCauley		
$\sqrt{1. D-6240}$ (Use with propeller B.1,2 only)	4.2	-62.3
2. D-6195 (Use with propeller B.3,4,5 only)	5.5	-62.3
1 2. D 0 100 (Ooc with proposition b.o., 7,0 offin)		<u> </u>
C. Sensenich		
1. S74DM (Use with propeller C.1-6 only)	4.8	-61.5
2. S76EM (Use with propeller C.7,8 only)	4.6	-61.5
12. Orocin (ood man propositor o.r., o oriny)	- 1.0	31.3
D. MT		
1. P-238-A-1 (Use with propeller D.1 only)	3.8	-61.5

#### MAULE AEROSPACE TECHNOLOGY, INC.

M5-235C/180C/M6-235/180/MT7-235/260/M7-235/235A/B/C/260/C M8-235/MX7-235/160/160C/180/180A/B/C/AC/MXT7-160/180/180A

PAGE **2** of 9 REV. <u>53</u> DATE: 11/07/00

#### **REQUIRED EQUIPMENT LIST**

SERIAL NO. 27008C REG. NO. NO. NO. 2299  NOTE: Equipment items installed below are designated by circle										
<u>CA</u>	TEG	ORY, MANU	<u>JFACTUF</u>	RER AND	PART OR MOD	EL NO.		· · ·	WEIGHT (Lbs)	ARM (Ins)
3. GOVERNOR (Not used with Fixed Pitch Propellers)										
	A.	Woodward								
		1. H2106	381	(O-360 c	nly)				4.5	-52.5
		2. B2107	<del>'</del> 61	(O-540-J	I/IO-540-W only)				4.5	-52.5
		3. F2106	81*	(O-540-J	I/IO-540-W only)	1			4.5	-52.5
		4. E2107	<b>′61</b>	(O-540-E	3 only)				4.5	-52.5
	(*Refer to AD #81-25-01 for eligibility information.)  B. McCauley									
		1. C290	D3(X)/T30	) (	(O-540-J/IO-540	-W only)			2.8	-52.5
		2. C290I	D3(X)/T31	l	(O-540-B only)				2.8	-52.5
		3. C290I	D3(X)/T29	)	(O-360 only)	_		•	2.8	<u>-52.5</u>
		4. DC29	0D1(X)/T	14	(O-540-J/IO-540	-W only)			2.8	-52.5
		5. DC29	0D1(X)/T	15	(O-540-B only)				2.8	-52.5
		6. DC29	0D1(X)/T	12	(O-360 only)				2.8	-52.5
	V	7. DC29	0D1(X)/T	33	(IO-540-V only)				2.8	-52.5

#### **ENGINE AND ACCESSORIES**

#### 4. ENGINE

٨	A.	Tex	tron/Lycoming			
<b>/53</b> \		1.	O-360-C1F	(M5/6/MX7/MXT7-180/MX7/MXT7-180A/MX7-180B/C/AC)	288.2	-42.2
		2.	O-540-J1A5D	(M5/6/8/MX7-235/M7-235/A/B/C)	390.1	-39.6
		3.	IO-540-W1A5D	(M5/6/8/MX7/MT7-235/M7-235/A/B/C)	395.5	-39.6
		4.	O-540-B4B5	(M5/6/8/MX7-235/M7-235/A/B/C)	395.6	-39.5
		5.	O-320-B2D	(MX7-160/C/MXT7-160 only)	282.5	-42.2
		6.	O-540-J3A5	(M5 6/8/MX7-235/M7-235/A/B/C)	395.2	-39.6
٨		7.	IO-540-W1A5	(M5/6/8/MX7/MT7-235/M7-235/A/B/C)	394.0	-39.6
<u>/53</u>		8.	O-360-C4F	(MX7/MXT7-180A/MX7-180AC)	293.0	-42.2
	マ	9.	IO-540-V4A5	(MT7-260/M7-260/C)	402.5	-39.6
<u></u>	<b>J</b>	4. 5. 6. 7.	O-540-B4B5 O-320-B2D O-540-J3A5 IO-540-W1A5 O-360-C4F	(M5/6/8/MX7-235/M7-235/A/B/C) (MX7-160/C/MXT7-160 only) (M5 6/8/MX7-235/M7-235/A/B/C) (M5/6/8/MX7/MT7-235/M7-235/A/B/C) (MX7/MXT7-180A/MX7-180AC)	395.6 282.5 395.2 394.0 293.0	

#### 5. OIL COOLER

A. Harrison

	1.	8529245	O-320/ O-360/IO-540 only)	3.5	-26.0/-25.0
	2.	8534108	O-540-J only)	3.5	-45.0
В.	Nia	gara			
	1.	20003A	(O-360/O-320 only)	3.5	-26.0
	2.	20006A, 20003A	(O-540 only)	3.5	-45.0
	3.	20003A	(IO-540 -235 only)	3.5	-25.0
V	4.	20006A	(IO-540 -260 only)	3.5	-25.0

#### **ELECTRICAL**

#### 6. BATTERY

**ΠΑULE ΔΕΡΟΣΡΑCE TECHNOLOGY, INC.** M5-235C/180C/M6-235/180/MT7-235/260/M7-235/235A/B/C/260/C M8-235/MX7-235/160/160C/180/180A/B/C/AC/MXT7-160/180/180A

PAGE **3** of 9 REV. <u>53</u> DATE: 11/07/00

#### **REQUIRED EQUIPMENT LIST**

	ALGOIALD LGOI MLITT LIGI		
SEF	RIAL NO. <u>27008C</u> REG. NO. <u>N20RD</u> PROD. NO. <u>2299</u>		
NO	TE: Equipment items installed below are designated by circle		
CAT	FEGORY, MANUFACTURER AND PART OR MODEL NO. WE	IGHT (Lbs)	ARM (Ins)
	A = 11		
Г	A. Exide (42 ::24)	27.0	**
Ĺ	1. AC78M (12 volt)	27.0	
	B. Rebat		
• • Г	1. R-35M (12 volt)	26.4	**
	1. (12 toly		
	C. Willard		
	1. W-78M (12 Volt)	27.0	**
_	D. Concorde		**
	1. CB 35M (12 volt)	26.5	**
}	2. RG-35A (12 volt) √ 3. RG-35AXC (12 volt)	29.5 32.0	**
L	✓ 3. RG-35AXC (12 volt)	32.0	
	E. Teledyne/Gil		
[	1. G-35M (12 volt)	27.0	**
t	2. G-243 (28 volt)	28.0	***
ļ	3. G-35S (12 volt)	27.0	**
<b>53 7</b> .	Battery Arm: 14" for MX7-160/C/180A/B/C/AC/MXT7-160/180/A; 96.4" (lt. side) or 95 (28 volt) M7/MT7; 101.1" for M8/MX7-235.  BATTERY SOLENOID	5.6" (rt. side);	for
	A NA/hita Dadgara/DDN//Casay		
	A. White Rodgers/RBM/Essex  ✓ 1. 71-111221-5 (14 volt) (M5/6/MX7-180)	.9	12.5
53	(MX7-160/C/180A/B/C/AC/MXT7-160/180A/A)	.9	-18.0
700 1	(M7/MT7, It. side)	.9	80.0
	(M7/MT7, rt. side)	.9	106.5
	(M5/6/8/MX7-235)	.9	101.5
Λ			
<u>/53\</u>	2. 71-117226-5 (28 volt) (MX7-160/C/180A/B/C/AC/MXT7-160/180/A)	8_	12.5
	(M7/MT7, It. side)	.9	80.0
	(M7/MT7, rt. side) (M8/MX7-235)	.8 .8	106.5 101.5
	(IVIO/IVIA/7-233)	.0	101.5
8.	STARTER SOLENOID		
	A. Prestolite		_
	1. SAZ-4201E	.7	-18.0
		· · · · · · · · · · · · · · · · · · ·	
	B. White Rodgers/RBM Controls/Essex		40.0
	√ 1. 70-112225-5 (14 volt)	.7 .7	-18.0 -18.0
	2. 70-118225-5 (28 volt)		-10.0

**MAULE AEROSPACE TECHNOLOGY, INC.** M5-235C/180C/M6-235/180/MT7-235/260/M7-235/235A/B/C/260/C M8-235/MX7-235/160/160C/180/180A/B/C/AC/MXT7-160/180/180A

PAGE 4 of 9 REV. <u>53</u> DATE: 11/07/00

#### **REQUIRED EQUIPMENT LIST**

			CREG. NO. NZOF		.299	
			installed below are design		MEIOUT (LES)	A DA4 ((no)
<u>UA</u>	IEC	BURT, MANUFACTU	JRER AND PART OR MO	DEL NO.	WEIGHT (Lbs)	ARM (Ins)
9.	EL	ECTRICAL FUEL P				
	Α.					
		1. 1184-00-1	(O-320/-360/-540 only)	(14 volt)	2.1	-20.0
		2. 1184-00-3	(O-320/-360/-540 only)	(14 volt)	2.1	-20.0
		3. 1471-00-1	(IO-540 only)	(14 volt)	2.1	-20.0
	V	4. 1471-00-3	(IO-540 only)	(14 volt)	2.1	-20.0
	<u></u>	5. 4140-00-400		(28 volt)	2.1	-20.0
	L	6. 4140-00-401	(IO-540)	(28 volt)	2.2	-20.0
<b>10.</b> 	,,,	1. B-00371-8	e Corporation/Lamar, Inc. (14 volt)			-16.0
		2. B-00368-12	(28 volt)		.3	-16.0
16	A.	1. B-00396-1 (1	e Corporation/Lamar, Inc. 14/28 volt)	IENT MARKINGS TO BE PER	.3 R AFM/AFMS.	-16.0
	J	1. 6016F			.1	1.5
13	. C	OMPASS	ents			
		1. HI-400			.6	-2.0
	_B.	. Airpath				
		1. C-2300			.8	-2.0
		2. C-2300-DL			.8	-2.0
		3. C-2400-LP	·		.8	-2.0
	V	4. C-2400-L4P			.8	-2.0
14		LTIMETER . Kollsman				
		1. C-12			1.0	-2.0
		2. C-13			1.0	-2.0
	-	3. B-11			1.0	-2.0

**ΠΑULE ΔΕΡΟΣΡΑCE TECHNOLOGY, INC.** M5-235C/180C/M6-235/180/MT7-235/260/M7-235/235A/B/C/260/C M8-235/MX7-235/160/160C/180/180A/B/C/AC/MXT7-160/180/180A

PAGE **5** of 9 REV. <u>53</u> DATE: 11/07/00

SERIAL	NO. 27008C REG. NO. N	PROD. NO. 22	79	
CATEG	Equipment items installed below are ORY, MANUFACTURER AND PART O	designated by circle	WEIGHT (Lbs)	ARM (Ins)
3	STATE OF THE PROPERTY OF THE P	K MODEL NO.	WEIGHT (EDS)	AINI (IIIS)
<b>B</b> .	Aerosonic			
	1. 101720-01545		1.0	-2.0
	2. 101720-01546 (Mb)		1.0	-2.0
	3. 12003		1.0	-2.0
	4. 101720-01694		1.0	-2.0
C	Aeritalia			
	1. 16030SK		1.2	-2.0
<b>D</b>	I looke of to observe and/B.64-b			
	United Instrument/Mitchell		4.0	
$\vdash$	1. 5934-1 code A68 (InHg) 2. 5934P-1 code A83 (InHg w/	Jordon mole)	1.0	-2.0
		Barber pole)	1.0	-2.0
-	4. 5934PM 1 code A64 (MD W/B)	arber pole)	1.0	-2.0
		/InHg w/Barber pole)	1.0 1.0	-2.0
	6. 5934M-3 code A73 (Mb)	inng w/barber pole)	1.0	-2.0
		Barber pole)	1.0	<u>-2.0</u> -2.0
	7. 3934F-3 Code A63 (IIIII) W/	barber pole)	1.0	-2.0
	Reporting Altimeters per Section 5.C on RSPEED  Aerosonic	. Optional Equipment List.		
	1. 230-217-56-1210		.8	-2.0
	2. 230-220-56-1210		.8	-2.0
<u> </u>	3. 230-40-201-2		.8	-2.0
	4. 230-217-56-1410		.8	-2.0
	5. S15KAW		1.0	-2.0
	6. 20025-01200		.8	-2.0
	7. 20025-01208		.8	-2.0
	8. S18KAW-1		.8	-2.0
R	EDO-Aire/Sigma Tek, Inc.			
	1. EA5171-01	TO NO.	.7	-1.0
	2. EA5171-02 MAL		.6	-1.0
	3. EA5171-03 MAL		.6	-1.0
	4. EA5171-04 MAL		.6	-1.0
	5. EA5171-05 MAL	- W. J	.6	-1.0
		80/MXT-7-180A)	.6	-1.0
	7. EA5171-07 MAL (M-6/M)	(-7-235)	.6	-1.0
	8. EA5171-08 MAL (M-7-23	5)	.6	-1.0
abla	9. EA5171-09 MAL (MXT7-	180/M7-235B/C/260/C/MT7-235/260) (N	/IPH/K) .6	-1.0
	10. EA5171-010 MAL (MXT7-	180/M7-235B/C/260/C/MT7-235/260) (		-1.0
	11. EA5174-01		.7	-1.0
	12. EA5174-02 PT-MAL		.7	-1.0
	13. EA5174-03 PT-MAL (TAS)		.7	-1.0
		180/M7-235B/C/260/C/MT7-235/260) (		-1.0
	15. EA5174-01 PTL-MAL		.7	-1.0

**MAULE AEROSPACE TECHNOLOGY, INC.** M5-235C/180C/M6-235/180/MT7-235/260/M7-235/235A/B/C/260/C M8-235/MX7-235/160/160C/180/180A/B/C/AC/MXT7-160/180/180A

PAGE **6** of 9 REV. <u>53</u> DATE: 11/07/00

SERIAL NO. 27008C REG. NO. NZORD PROD. NO. 2	299	
NOTE: Equipment items installed below are designated by circle	<u> </u>	<del></del>
CATEGORY, MANUFACTURER AND PART OR MODEL NO.	WEIGHT (Lbs)	ARM (Ins)
C. United Instruments, Inc.		
1. UI8000-B690	.7	-1.1
16. INSTRUMENT CLUSTER		
A . MAALII E		
A. MAULE √ 1. 7040B	1.0	-0.6
¥ 1. 1040B	1.0	-0.0
17. MANIFOLD PRESSURE/FUEL PRESSURE (0-360/0-540 only)	•	
A. Aerosonic		
1. 735-102	1.0	-1.5
B. EDO-Aire/Sigma Tek	•	
B. EDO-Aire/Sigma Tek  1. IU028-005-38	1.0	-1.0
1. 10020-000-00	1.0	-1.0
C. Castleberry Instruments & Avionics, Inc.		
1. 6331 H 14	1.0	-1.0
D. United Instrument Inc.		
D. United Instrument, Inc.  1. UI6333-H.170	1.0	-1.0
1. 010000-11.170	1.0	-1.0
18. FUEL PRESSURE GAUGE (O-320/O-360 w/fixed pitch only)		
A. United Instruments  1. 6213		4.0
1. 6213	.5	-1.0
19. MANIFOLD PRESSURE/FUEL FLOW (IO-540 only)		
•		
A. A.I.D.		
1. 21-1003-1	1.0	-1.5
B. United Instruments, Inc.		
√ 1. UI6333-H.151	1.0	-1.5
	· · · · · ·	
C. Sigma-Tek		
1. 1U028-005-65	1.3	-1.5
20. TACHOMETER		
A. Stewart-Warner		
1. 551-AWB (O-360/O-320 only)	.8	-1.1
2. 551-ACR	.8	-1.1
3. 551-ASJ	.8	-1.1
B. AC		
1. RT-7	.8	-1.1
2. AC6412611	.8	-1.1

**TAULE AEROSPACE TECHNOLOGY, INC.** M5-235C/180C/M6-235/180/MT7-235/260/M7-235/235A/B/C/260/C M8-235/MX7-235/160/160C/180/180A/B/C/AC/MXT7-160/180/180A

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TECO	Equipment items	IDED AND DAD	T OR MODEL NO.	MEIOUT (LL.)	A DA 4 (1)
LEGOI	KT, WANDFACT	WEIGHT (Lbs)	ARM (Ins)		
C. A	erosonic				
1.				.8	-1.1
	litchell				
1.		(O-540/IO-54		.8	<u>-1.1</u>
2.		(O-360/O-32		.8	-1.1
<b>√</b> 3.			O-360-C4F (Fixed Pitch), IO-540-V4A5)	.8	-1.1
4.				.8	-1.1
5. 6.			- Constant Speed)	.8	-1.1
7.			D/J3A5, IO-540-W1A5D/W1A5)	.8	-1.1
	. CD-122-5008	3 (O-540-B4B5	0)	.8	-1.1
E. E	:lectronic Internati	ional	g		
1.			(O-320-B2D,O-360-C4F)	.6	-1.8
2.	. R-1-4-G30R2	215G235R270	(O-360-C1F - Fixed Pitch)	.6	-1.8
3.	. R-1-4-G30R2	200G225R270	(O-360-C1F - Constant Speed)	.6	-1.8
4.	. R-1-6-G30R2	240	(O-540-J1A5D/J3A5)	.6	-1.8
5.	. R-1-6-G30R2	240	(IO-540-W1A5D/W1A5)	.6	-1.8
6		257	(O-540-B4B5)	.6	-1.8
7.	. R-1-6-G30R2	170			
	NG GEAR		(IO-540-V4A5) PLY TO TSO C62b	.6	-1.8
TIRE	NG GEAR	a.) TYPE III 4/6 F	PLY TO TSO C62b		
TIRE	NG GEAR S w/TUBES (2 ea A. 6:00 x 6	a.) TYPE III 4/6 F	PLY TO TSO C62b	8.3	-41.0
TIRE	NG GEAR  S w/TUBES (2 each	a.) TYPE III 4/6 F (Nosewheel - 1 (For Trigear M	PLY TO TSO C62b 1 each) ains)	8.3 16.6	-41.0 30.4
TIRE	NG GEAR  S w/TUBES (2 each	a.) TYPE III 4/6 F (Nosewheel - 1 (For Trigear M (For Tailwheel	PLY TO TSO C62b  1 each) ains) Models)	8.3 16.6 23.6	-41.0 30.4 -2.5
TIRE	NG GEAR  S w/TUBES (2 ea  A. 6:00 x 6  B. 7:00 x 6  C. 7:00 x 6  D. 8:00 x 6	a.) TYPE III 4/6 F  (Nosewheel - 1  (For Trigear M  (For Tailwheel  (For Tailwheel	PLY TO TSO C62b  1 each) ains) Models) Models only)	8.3 16.6 23.6 25.2	-41.0 30.4 -2.5 -2.5
TIRE  V A V B C C C C C C MAIN	NG GEAR  S w/TUBES (2 each	a.) TYPE III 4/6 F  (Nosewheel - 1/2)  (For Trigear M  (For Tailwheel  (For Tailwheel  (For Tailwheel	PLY TO TSO C62b  1 each) ains) Models) Models only)	8.3 16.6 23.6	-41.0 30.4 -2.5 -2.5
TIRE  V A  V B  C  C  C  C  MAIN	NG GEAR  S w/TUBES (2 each	a.) TYPE III 4/6 F  (Nosewheel - 1/2)  (For Trigear M  (For Tailwheel  (For Tailwheel  (For Tailwheel	PLY TO TSO C62b  1 each) ains) Models) Models only)	8.3 16.6 23.6 25.2 38.2	-41.0 30.4 -2.5 -2.5 -2.5
TIRE  V A V B C C C C C C C C C C C C C C C C C C C	A. 6:00 x 6 B. 7:00 x 6 C. 7:00 x 6 C. 8:00 x 6 E. 8:50 x 6 NWHEEL (2 each	a.) TYPE III 4/6 F  (Nosewheel - 1  (For Trigear M  (For Tailwheel  (For Tailwheel  (For Tailwheel  a.)	PLY TO TSO C62b  1 each) ains) Models) Models only) Models only)	8.3 16.6 23.6 25.2 38.2	-41.0 30.4 -2.5 -2.5 -2.5
TIRE  V A  V B  C  C  C  C  MAIN	A. 6:00 x 6 B. 7:00 x 6 C. 7:00 x 6 E. 8:50 x 6  NWHEEL (2 each of the color of the	a.) TYPE III 4/6 F  (Nosewheel - 1/2)  (For Trigear M  (For Tailwheel  (For Tailwheel  (For Tailwheel	PLY TO TSO C62b  1 each) ains) Models) Models only) Models only)	8.3 16.6 23.6 25.2 38.2	-41.0 30.4 -2.5 -2.5 -2.5 -2.5
TIRE  V A V B C C C C C C C C C C C C C C C C C C C	A. 6:00 x 6 B. 7:00 x 6 C. 7:00 x 6 D. 8:00 x 6 E. 8:50 x 6  NWHEEL (2 each of the color of the	a.) TYPE III 4/6 F  (Nosewheel - 1  (For Trigear M  (For Tailwheel  (For Tailwheel  (For Tailwheel  a.)	PLY TO TSO C62b  1 each) ains) Models) Models only) Models only)	8.3 16.6 23.6 25.2 38.2	-41.0 30.4 -2.5 -2.5 -2.5 -2.5 -2.5
TIRE	NG GEAR  S w/TUBES (2 each	a.) TYPE III 4/6 F  (Nosewheel - ' (For Trigear M (For Tailwheel (For Tailwheel (For Tailwheel a.)	PLY TO TSO C62b  1 each) ains) Models) Models only) Models only)	8.3 16.6 23.6 25.2 38.2	-41.0 30.4 -2.5 -2.5 -2.5 -2.5 -2.5
TIRE    A   B     B     A   C     A     A   C     A     A   C     A     A   C     A	A. 6:00 x 6 B. 7:00 x 6 C. 7:00 x 6 C. 8:00 x 6 E. 8:50 x 6 E. 8:50 x 6 E. 40-97D C. 40-75D (Dual B. 40-28 E. C38500HA E. C38500HA E. CIeveland	a.) TYPE III 4/6 F  (Nosewheel - ' (For Trigear M (For Tailwheel (For Tailwheel a.)	PLY TO TSO C62b  1 each) ains) Models) Models only) Models only)	8.3 16.6 23.6 25.2 38.2 12.4 12.8 10.2 10.2	-41.0 30.4 -2.5 -2.5 -2.5 -2.5 -2.5 -2.5
TIRE    V   A     V   B     C	A. 6:00 x 6 B. 7:00 x 6 C. 7:00 x 6 C. 8:00 x 6 E. 8:50 x 6 E. 8:50 x 6 C. 40-97D C. 40-75D (Dual 3. 40-28 E. C38500HA E. KES - MAIN WHILE Cleveland E. 30-63E	a.) TYPE III 4/6 F  (Nosewheel -  (For Trigear M  (For Tailwheel  (For Tailwheel  a.)  II) (For-Tailwheel  EEL (2 ea.)	PLY TO TSO C62b  1 each) ains) Models) Models only) Models only)	8.3 16.6 23.6 25.2 38.2 12.4 12.8 10.2 10.2	-41.0 30.4 -2.5 -2.5 -2.5 -2.5 -2.5 -2.5
TIRE    A   C     B     A   C     A	A. 6:00 x 6 B. 7:00 x 6 C. 7:00 x 6 C. 8:00 x 6 E. 8:50 x 6 E. 8:50 x 6 E. 40-97D C. 40-75D (Dual B. 40-28 E. C38500HA E. C38500HA E. CIeveland	a.) TYPE III 4/6 F  (Nosewheel -  (For Trigear M  (For Tailwheel  (For Tailwheel  a.)  II) (For-Tailwheel  EEL (2 ea.)	PLY TO TSO C62b  1 each) ains) Models) Models only) Models only)	8.3 16.6 23.6 25.2 38.2 12.4 12.8 10.2 10.2	-41.0 30.4 -2.5 -2.5 -2.5 -2.5 -2.5 -2.5 -2.5
TIRE    A   C     C	A. 6:00 x 6 B. 7:00 x 6 C. 7:00 x 6 C. 8:00 x 6 E. 8:50 x 6  N WHEEL (2 each of the color of the	a.) TYPE III 4/6 F  (Nosewheel -  (For Trigear M  (For Tailwheel  (For Tailwheel  a.)  II) (For-Tailwheel  EEL (2 ea.)	PLY TO TSO C62b  1 each) ains) Models) Models only) Models only)	8.3 16.6 23.6 25.2 38.2 12.4 12.8 10.2 10.2	-41.0 30.4 -2.5 -2.5 -2.5 -2.5 -2.5 -2.5

**TAULE AEROSPACE TECHNOLOGY, INC.** M5-235C/180C/M6-235/180/MT7-235/260/M7-235/235A/B/C/260/C M8-235/MX7-235/160/160C/180/180A/B/C/AC/MXT7-160/180/180A

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17.5

1.8

### **REQUIRED EQUIPMENT LIST**

<u>TEGORY, MANUFACTURE</u>	R AND PART OR MODEL NO.	WEIGHT (Lbs)	ARM (Ins
TAILWHEEL (Tailwheel M	Models only)		
A. MAULE			
1. SFS-P8A-1-2		7.0	185.0
2. SFS-P8B-1-2		8.0	185.0
FETY BELTS			
FRONT SEAT BELT & SH	IOULDER HARNESS ASSEMBLIES (2 ea	a.)	
A. AM-Safe, Inc.			00.0
<u> </u>		2.6	36.0
2. 500661-405 3. 500661-413	<del></del>	2.6	36.0
	(x) /-417 (L) (Rotary Buckle)	2.6 2.6	36.0 37.0
	t) /-417 (L) (Rotary Buckle) t) /-425 (L) (Rotary Buckle)	2.5	37.0 37.0
		2.5	37.0
B. Pacific Scientific (Rotal 1. 1101020-7 (L)/-9		1.6	36.0
	Shoulder Harness)	.9	38.0
	,		
C. Davis  1. FDC-6400-568-2	14	2.6	36.0
2. FDC-8300-64R-0		2.0	36.0
3. FDC-8300-64L-0			36.0
REAR SEAT BELT (1 ea.)  A. Rupert	) & SHOULDER HARNESS/LAP BELT AS	SEMBLIES (2 ea.)	
	AC (Lap Belt - Double Occupant)	.6	52.0
B. AM-Safe. Inc.			
1. 500915-3	(Lap Belt - Double Occupant)	.6	52.0
2. 500661-409	(Shoulder Harness & Lap Belt)	2.5	55.0/80.0
C. Davis			55.0100
✓ 1. FDC-6400-568-1	//3	2.6	55.0/80.0
. CARGO & LUGGAGE ST	TRAPS		
A. Kine-Dyne			
✓ 1. 3101003GO72-0		.8	43. 71.
✓ 2. 3101003GO72-0	022 (Rear)	.8	71.
GHTS			
B. NAVIGATION LIGHTS - N	WING		
A. Whelen Engineering	Co		

(Power Supply A490-T-DF-14)

A650 PR (Red)

**V** 1.

**TAULE AEROSPACE TECHNOLOGY, INC.** M5-235C/180C/M6-235/180/MT7-235/260/M7-235/235A/B/C/260/C M8-235/MX7-235/160/160C/180/180A/B/C/AC/MXT7-160/180/180A

PAGE **9** of 9 REV. <u>53</u> DATE: 11/07/00

SERIAL NO. 27008C REG. NO. N20RD PROD. NO. 2299		
NOTE: Equipment items installed below are designated by circle	-	
CATEGORY, MANUFACTURER AND PART OR MODEL NO.	WEIGHT (Lbs)	ARM (Ins)
✓ 2. A650 PG (Green) (Power Supply A490-T-DF-14)	1.8	17.5
3. 62-A470-W (Power Supply A490-T-DF-14)	.5	37.9
4. A650 PR (Red) (Power Supply A490-T-DF-28) (28 V)	1.8	17.5
5. A650 PG (Green) (Power Supply A490-T-DF-28) (28 V)	.8	17.5
<ul><li>B. See Optional Equipment List, Section 1.A, Items 1, 2 and 5.</li><li>29. NAVIGATION LIGHTS - TAIL</li></ul>		
A. Grimes		
1. Type A-2064 (14 or 28 volt)	.2	199.0
B. Luminator Aircraft	<del>-</del>	
√ 1. 5107 C14B	.2	199.0
EMERGENCY LOCATION TRANSMITTERS  30. ELT		
A. Leigh		
1. 7H-2-190 Mod. # Type AF/AP/P	2.3	18.8
B. Narco		
1. ELT-10	3.5	18.8
C. Emergency Beacon Corp.		
1. EBC-102A	1.2	36.7
✓ 2. EBC-502	2.8	36.7
D. Merl, Inc.		
1. Model #79007-P	2.5	20.0

M5-180C/200/210C/235C/MX7-160/C/180/180A/C/AC/235/420/M6-180/235/ M8-235/ MT-235/260/M7-235/A/B/C/260/C/420A/AC/MXT7-160/180/180A/420/M7-235/A per STC SA2661SO DATE: 11/07/00

SERI	AL NO	. 27008C REG. NO. N20RD	_PROD. NO2299	
CATE	GORY	pment items installed below are designated by check marks.  7. MANUFACTURER AND PART OR MODEL NO.	WEIGHT (Lbs)	ARM (Ins)
1.	AN	TI-COLLISION LIGHTS		
	A.	STROBES		
		1. Grimes		
		a. 30-0550-2-14 (Dual Red) b. 30-0550-16-14 (Dual White)	4.1 4.1	11.8
		b. 30-0550-16-14 (Dual White) c. 30-0550-19-14 (Nav/Strobes)	2.6	11.8 16.5
		2. SDI Hoskins		
		a. 701750-215	1.3	9.0
		b. 701144-2	1.5	10.0
		c. 701144	1.3	9.0
		3. Scientifico		
		a. Model 1400 (Top)	»·· .4	37.9
		b. Model 1400 (Bottom)	.4	-24.0
		4. Aeroflash a. X2FS (Top)	5	07.0
		a. X2FS (Top) b. X2FS (Bottom)	.5 .5	37.9 -24.0
		, ,	.0	21.0
		5. Grimes a. Type 1285, Red (Left) (14 or 28 volt)	2	16.0
		b. Type 1285, Red (Left) (14 of 28 volt)	.3 .3	16.0
	В.	LANDING LIGHT PULSER		
	D.	1. Precise Flight, Inc.		
		a. 1210/2405-2	.8	-9.0
		Note: Flasher Unit Mode 902-12 to be installed with 4.a & 4.b.		
		Trade. Tradition of the Mode one 12 to be indicated with 4.2 d. 4.5.		
2.	CO	MMUNICATION AND NAVIGATION EQUIPMENT		
	A.	COMM/NAV TRANSCEIVERS		
	,	1. EDO-Aire		
		a. RT-553/553A	5.5	-5.5
		b. RT-563/563A	7.9	-5.5
		2. King		
		a. KX 125 b. KX 145	4.0 3.8	-5.0
		b. KX 145 c. KX 155	5.0	-3.9 -5.0
		d. KX 165	5.0	-5.0
		e. KX 170B	7.1	-5.5
		f. KX 175B	7.1	-5.5
		3. Genave		
		a. Alpha/200B		
		b. Alpha/360 c. Alpha/500		
		d. Alpha/600		
		4. Narco		
		a. Escort 110	4.1	-6.0
		b. MK-12D w/GS	5.2	-5.0
		c. MK-12D w/o GS	4.8	-5.0

 MAULE AEROSPACE TECHNOLOGY, INC.
 PAGE 2 of 18

 M5-180C/200/210C/210TC/235C/MX7-160/C/180/180A/C/AC/235/420/M6-180/235/ M8-235/
 REV. 53

 MT-235/260/M7-235/A/B/C/260/C/420A/AC/MXT7-160/180/180A/420/M7-235/A per STC SA2661SO
 DATE: 11/07/00

SERIAL NO	0. <u>27008C</u> REG. NO. <u>NZORD</u>	PROD. NO. <u>2299</u>	
NOTE: Equ	ipment items installed below are designated by check marks. Y, MANUFACTURER AND PART OR MODEL NO.		
CATEGOR	Y, MANUFACTURER AND PART OR MODEL NO.	WEIGHT (Lbs)	ARM (Ins)
	5. Тепта		
	a. TXN 920	3.0	-5.2
	b. TXN 960	4.6	-5.2
			0.2
В.	COMMUNICATION TRANSCEIVER	•	
	1. Bendix		
	a. RT-241B		
	2. Collins		
	a. VHF-251	3.8	-6.0
		0.0	-0.0
	3. EDO-Aire		
	a. RT-551/551A	3.3	
	b. RT-661/661A		_
_	c. RT-771 d. RT-773		
-	d. RT-773 "		-
	4. Genave	•	
	a. Alpha/10		
	b. Alpha/100		
	c. Alpha/100-360		
	5. King		
	a. KY-92		
	b. KT-96		
	c. KY-195B		
	d. KY-197		
	O. News		
	6. Narco a. Com 10/10A	4.2	<i></i>
	b. Com 11A/11B	4.3 3.3	-5.5 -5.5
	c. Com 111/111B	3.3	-5.5
	d. Com 120		
	7. Terra		
	a. TX 720	1.3	-5.3
C.	HIGH FREQUENCY TRANSCEIVERS		
	1. Pantronics		
	a. Pan Com DX10-DA		
	b. Pan Com DX10-RA		
	c. Pan Com PT10-A		
	d. Pan Com SB-10 e. Pan Com TSB-10		
	e. Fail Colli 130-10		
D.	RADIO POWER SUPPLY		
	1. Narco		
	a. T-12-MP-12A1	4.0	16.5
	2. King		
	a.	3.3	16.5
		,	-
	3. Collins		
	a. PWC-150	1.5	
E.	NAVIGATION (VOR) RECEIVERS		
_ <del></del>	_1 Bendix		
	a. RN-242A		

**TAULE AEROSPACE TECHNOLOGY, INC.**M5-180C/200/210C/210TC/235C/MX7-160/C/180/180A/C/AC/235/420/M6-180/235/ M8-235/

MT-235/260/M7-235/A/B/C/260/C/420A/AC/MXT7-160/180/180A/420/M7-235/A per STC SA2661SO

DATE: 11/07/00

SERIAL NO	2700	08C REG. NO. N2ORD	PROD. NO. 2299	
CATEGORY	<u>PMENT ITEMS</u>	installed below are designated by check marks. TURER AND PART OR MODEL NO.	MEIOLIT (LES)	ADM (I==)
ONILOON	I INITIACI AC	TORER AND PART OR MODEL NO.	WEIGHT (Lbs)	ARM (Ins)
	_2 Col	lins		
	a.	VIR-351	3.1	-5.5
	b.	VIR-350		
	3. ED	O-Aire		
	a.	R-552		
	b.	R-554		
	C.	R-662		
	d.	R-664		
	4 No.			
	4. Nar	rco NAV 11	0.5	
	a. b.	NAV 11 NAV 12	2.5	-5.5 5.5
	C.	NAV 12	2.5 7.2	-5.5 5.3
	d.	NAV 111	1.2	-5.3
٠٠ هو	e.	NAV 112 -		
	f.	NAV 114		
	g.	NAV 121		
	h.	NAV 122		
	1.	NAV 122A	•	
4	j.	NAV 824 (VOR/LOC Rec.)	3.1	-5.5
	k.	NAV 825 (VOR/LOC/GS Rec.)	3.3	-5.5
	5. Ter	<del></del>		
	3. 1ei	TN 200	1.6	-5.3
	<del>"</del>	111 200	1.0	-5.5
F.	VOR INDI			
	_1 ED	O-Aire		
	a.	CID-552A		
	b.	CID-554A		
	C.	CID-662		
	d.	CID-664		
	e. f.	R-772 CID-774		
	L '·	CID-114		
	_2 Kin	ng		
	a.	KI-203		
	b.	KI-204		
	C.	KI-205		
	<b>d</b> .	KI-206		
	e.	KI-207		
	f.	KI-208 KI-209		
	g. h.	KI-209 KI-201C		
		KI-214		
	j.	KI-226		
	k.	KNI-520-01		
	1.	KI-521		
	m.	KI-211		
	n.	KI-212		
	2 D-	ndiv		
	3. Be	ndix IN-244A		
	b.	IN-245A		
	c.	RN-242		
	d.	GM-247		
	4 -	w		
		illins IND-350		
	a.	טטט-טאוו		

**TAULE AEROSPACE TECHNOLOGY, INC.**M5-180C/200/210C/210TC/235C/MX7-160/C/180/180A/C/AC/235/420/M6-180/235/ M8-235/
MT-235/260/M7-235/A/B/C/260/C/AZ0A/AC/MXT7-160/180/180A/420/M7-235/A per STC SA2661SO

DATE: 11/07/00

SERIAL NO	. 27008C REG. NO. NZORD	PROD. NO. <u><b>2299</b></u>	
CATEGORY	pment items installed below are designated by check marks.  , MANUFACTURER AND PART OR MODEL NO.	WEIGHT (Lbs)	ARM (Ins)
		***************************************	,, <u>,</u>
	b. IND-351		
	c. IND-351C		
	5. Genave		
	a. Theta/300		
	b. Theta/400		
	6. Narco		
	a. NAV 10	1.7	-5.5
	b. DGO 9/9A		
	c. DGO 10		
	d. VOA-40	1.8	-3.8
	e. VOA-40M	2.8	-3.8
	f. VOA-50	1.8	-3.8
	g. VOA-50M	2.9	-3.8
	h. VOA-8	- 2.8	<b>-4</b> .0 "
	I. VOA-9	2.9	-4.0
	j. NAV/111		-3.3
	k. NAV/11	2.5	-5.5
	I. NAV/12	2.5	-5.5
	m. NAV/14	7.2	-5.3
	n. CLC-60		
	o. NAV/112		
	p. NAV/114		
	q. NAV/124		
	r. ID 824	.9	-2.0
	s. ID 825	1.0	-2.0
G.	VOR/LOC CONVERTER		
<b>J</b> .	1. King		
	a. KN 72		
	•		
	2. Narco		
	a. OC 110		
Н.	AUTOMATIC DIRECTION FINDERS		
	1. King		
	a. KR 85	5.1	-3.2
	b. KR 86	3.9	-4.0
	c. KR 87		
	2 Norse		
	2. Narco a. ADF 141	6.3	36.7
	b. PDF-35	4.3	-4.7
	c. ADF 140	4.3	-4.7
	d. RMI 35		
	e. RMI 140		
	e. 1/(a) 140		
	3. EDO-Aire		
	a. R-556D	8.5	
	b. R-556E		
	4. Bendix		
	a. ADF-T-12 C/D		
	b. 20IC/201D/201F		
	5. Collins		
	a. RCR-650	2.3	-4.2

# MAULE AEROSPACE TECHNOLOGY, inc.

**PAGE 5** of 18

M5-180C/200/210C/210TC/235C/MX7-160/C/180/180A/C/AC/235/420/M6-180/235/ M8-235/ MT-235/260/M7-235/A/B/C/260/C/420A/AC/MXT7-160/180/180A/420/M7-235/A per STC SA2661SO DATE: 11/07/00

SERIAL N	0. <b>27008C</b> RE	EG. NO. NZORD	PRC	DD. NO. 2299	
NOTE: Eq	uipment items installed below are d RY, MANUFACTURER AND PART	esignated by check marks. OR MODEL NO		WEIGHT (Lbs)	ARM (Ins)
		OK MODEL NO.		WEIGHT (EDG)	- Altin (mo)
	6. Genave a. Sigma/1500				
I.	ADF INDICATORS				
	1. Bendix				
	a. 551A b. 55IRL				
	2. Collins				
	a. IND-650			.6	-1.7
	3. King				
	a. KI 225 b. KI 227				
			بو		
	4. Narco a. ADF 101	<u>-</u>			
J.	GLIDE SLOPE AND MARKER	RECEIVERS			
	1. King a. KR 21				
	b. KN 73				
	c. KN 75				
	d. KA 40 e. KR 22				
	2. Bendix				
	a. GM-247A				
	3. Narco				
	a. UGR 2 & 2A b. UGR 3				
	c. MBT				
	4. Genave		·		
	a. Delta/300 b. Delta/303/303R				
	c. PHI/20				
	5. EDO-Aire				
	a. R-775/775R				
	6. Collins			7	2.0
	a. MKR-350 b. GLS-350			.7 .3	-2.0 -5.3
	Terra				
	a. TGS 40			.3	-5.3
ĸ					
	1. Bendix/King a. KT 76/76A			3.0	-5.0
Δ	b. KT 78/78A	<b>44 . D. -0.</b> 444		3.0	-5.0
<u>/53</u>		(Maule Dwg 7211A)		2.4	-5.0
	2. Narco a. AT-50A				
	b. AT-150				

**RAULE AEROSPACE TECHNOLOGY, INC.**M5-180C/200/210C/210TC/235C/MX7-160/C/180/180A/C/AC/235/420/M6-180/235/ M8-235/
MT-235/260/M7-235/A/B/C/260/C/420A/AC/MXT7-160/180/180A/420/M7-235/A per STC SA2661SO

DATE: 11/07/00

SERI	AL NO	27008C	REG. NO. NZORD	PROD. NO. <u>2299</u>	
NOTE	: Equi	pment items installed below are MANUFACTURER AND PAR	e designated by check marks.		ADM (Inc)
OAIL	.GON	, IMANUFACTURER AND PAR	I OR MODEL NO.	WEIGHT (Lbs)	ARM (Ins)
		3. EDO-Aire			
		a. RT-667/667A		3.9	-5.0
		b. RT-777 c. RT-787		3.3	-5.0
		d. RT-887			-5.0
					0.0
		4. Genave			
		a. Beta/4096 b. Beta/500			
		c. Beta/5000			
		5. Bendix			
		a. TPR-660 b. TR-661A			
		b. 11001A			
	-	6. Collins	<i>y</i>		-
		a. TDR-950		2.0	-5.0
		7. Terra			
		a. TRT 250		1.7	-5.2
<u></u>		8. Garmin	(Manda Dun 7007A)	2.2	0.0
<u>/53\</u>		a. GTX-327	(Maule Dwg 7227A)	2.2	-3.6
	L.	DISTANCE MEASURING E	QUIPMENT		
		1. King			
		a. KN 61 b. KN 62/62A/64		2.6	-5.3
		c. KN 65/65A		2.6	-5.3
		d. KN 63			
		O. Name			
		2. Narco a. DME 190			
		b. DME 195			
		c. DME 890		3.9	-5.0
		d. IDME 891		2.6	<b>-4</b> .6
	M.	MARKER LIGHTS			
		1. King			
		a. KA 40			
		2. Collins			
		a. MKL-350		.3	-0.2
		<u> </u>			
	N.	AUDIO CONTROL PANELS	3		
		1. King a. KMA 20-03		2.3	
		b. KMA 20-04		2.3	
		c. KMA 24		1.7	-2.6
		d. KA 134			
		2. Narco			
		a. CP 125			
		b. CP 126			
		c. CP 127			
		d. CP 135/135M e CP 136/136M			

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**MAULE AEROSPACE TECHNOLOGY, INC.**M5-180C/200/210C/210TC/235C/MX7-160/C/180/180A/C/AC/235/420/M6-180/235/ M8-235/ MT-235/260/M7-235/A/B/C/260/C/420A/AC/MXT7-160/180/180A/420/M7-235/A per STC SA2661SO DATE: 11/07/00

REV. <u>53</u>

SERI	AL NO	270C	)8C RE	G. NO. NZORD	PROD. NO. 2299	
CATE	⊑qu GOR	Y, MANUFAC	TURER AND PART C	esignated by check marks. DR MODEL NO.	WEIGHT (Lbs)	ARM (Ins)
•		3. Ben				
		a.	AS-248A			
		4. Coll	lins			
		a.	AUD-250/250D		1.5	-2.8
		b.	AMR-350/350H		1.8	-2.9
			O-Aire			
		a. b.	A/AM-550 AM-660			
		c.	A-770		2.0	
		6. Ger	nave			
		a.	TAU/200			
٠٠ ټو		_7 Ten		-		
		a.	TMA 230		1.1	-2.8
٨		_8PS	Engineering			
<u>/53\</u>		<b>√</b> a.	PM199H- 100011	(Maule Dwg 7217A)	.8	-2.8
		b.	PMA6000M	(Maule Dwg 7221A)	.8	-3.4
<b>/53</b>		c. d.	PMA7000S (M-S) PMA6000M-S	(Maule Dwg 7223A) (Maule Dwg 7233A)	1.5 .8	-3.4 -3.4
700					.0	0.1
	Ο.	1. Kin	ON AUDIO AMPLIFIEI Ia	K		
		a.	KA 25A		1.3	
	P.	ANTENNA	A			
		1. Na				
		a. b.	Loop ADF 01073-1 UDA-3 DME (for ID		.5	
			·	WIL 091)	.0	
			ndix			
		a. b.	2321E AT-662A			
			llins ANT-251		o	
		a. b.	ANT-650		.8 2.3	
		4. ED	O-Aire			
		a.	AT-551A			
		<b>b</b> .	AT-556A			
		C.	AT-774A			
		d.	AT-775A			
		5. Me	eridian Electronics MB-5		.9	95.0
		b.	MB-7		.9 .9	95.0
		c.	BB-9		.4	7.0
		d.	BB-16		.3	7.0
•		e.	354		.6	.0 160.0
		f. g.	M809 VHF-1		.4 .2	160.0 7.0
		g.	VIII *1		.2	7.0

**TAULE AEROSPACE TECHNOLOGY, INC.**M5-180C/200/210C/235C/MX7-160/C/180/180A/C/AC/235/420/M6-180/235/ M8-235/
MT-235/260/M7-235/A/B/C/260/C/420A/AC/MXT7-160/180/180A/420/M7-235/A per STC SA2661SO
DATE: 11/07/00

SERIAL NO. 27008 C REG. NO. N20RD	PROD. NO. 2299	
NOTE: Equipment items installed below are designated by check marks.  CATEGORY, MANUFACTURER AND PART OR MODEL NO.	WEIGHT (Lbs)	ARM (Ins)
6. Maule a. Sense Antenna 7030B	.2	7.0
7. Genave a. Broad Band Lamba/100 b. Marker Beacon Lamba/75 c. Lamba/1000	.3 .8	7.0 95.0
8. King a. KA 22 b. KA 23 c. KA 32 d. KA 42 e. KA 42B f. KA 44B g. KA 48 h. KA 60		
9. Pantronics Deluxe Ant. Loading Unit SB10-RL		
10. Pantronics Deluxe Ant. Loading Unit DX10-RL	-	••
11. Pantronics Manual Load Unit DX10ML		. •
12. Pantronics Manual Reel Ant. Kit DX10-MR		
13. Pantronics Electric Reel Ant. Kit DX-EA-12		
14. Comant a. CI 121 SP (For Apollo I Rec.)	.9	43.0
15. SRD Labs a. LW-209 (For L-NAV 25 Rec.)	.7	43.0
16. Texas Instrument a. 2480191-0001 (For TI 9100 Rec.)	.9	43.0
17. ARNAV Systems a. 455-6054 (For ARNAV 20 Rec.)	.8	43.0
18. Foster Airdata Systems a. NY154 (For LRN500/F4 Rec.)	.5	43.0
19. Trimble a. 16248-20 (For TNL 2000 GPS)	.4	44.6
Q. MICROPHONES  1. EDO-Aire  a. M-551	.6	13.5
2. Electro Voice a. 205-STC		
3. Telex a. 66CRA b. 66T c. 5 by 5 MARK II (Headset)		
R. SPEAKERS  1. Jensen		

**MAULE AEROSPACE TECHNOLOGY, INC.**M5-180C/200/210C/210TC/235C/MX7-160/C/180/180A/C/AC/235/420/M6-180/235/ M8-235/
MT-235/260/M7-235/A/B/C/260/C/420A/AC/MXT7-160/180/180A/420/M7-235/A per STC SA2661SO

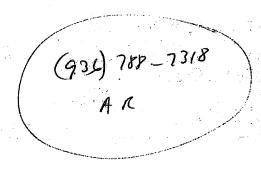
DATE: 11/07/00

	AL NO. 27008C REG. NO. N20RD	PROD. NO. 2299	
CATE	E: Equipment items installed below are designated by check marks. EGORY, MANUFACTURER AND PART OR MODEL NO.	WEIGHT (Lbs)	ARM (Ins)
21.11.		VVEICHT (EDS)	74 (11) (11)07
	b. C7788-4	•	
	2. Utah Electronics		
	a. A5CI15		
	b. A46FC		
	3. Quam-Nichols		
	→ a. 46C5FR	1.0	28.5
	S. AREA NAVIGATION SYSTEMS		
	_1 Bendix/King		
	a. KN 74		
$\wedge$	b. KNS-80	6.0	-6.0
<u>/53\</u>	c. KLX-135A (GPS/COMM) (Maule Dwg 7219A)	5.0	-5.2
	2. Collins		
	a. ANS 351	,	
	3. SRD Labs		
	a. L-NAV 25 Loran C Nav System	3.2	-5.0
	_4 Texas Instrument		
	a. TI 9100 Loran C Nav System	8.2	-5.0
	5. Il Morrow		
	a. Apollo I Nav System	3.7	-5.6
	b. Apollo II Nav System	3.7	-5.6
	6. Northstar Avionics		
	a. Northstar MI Loran C Nav System	4.2	-5.3
	7. Foster Airdata Systems		
	a. F4 Phoenix	4.5	-6.0
	b. LRN500	3.7	-6.0
	8. Trimble		
	a. TNL 2000 GPS Navigator	3.4	-6.0
	9. Garmin		
$\wedge$	a. GNC 300	2.4	-2.8
<u> </u>	b. GNS-430 (GPS/NAV/COMM) (Maule Dwg 7		-5.1
Λ	c. GNC-300XL (GPS/COMM) (Maule Dwg 7		-2.3
<u> </u>	d. GNC-250XL (GPS/COMM) (Maule Dwg 7	(209A) 2.1	-2.3
	T. PUSH TO TALK SWITCHES		
	1. Telex a. PT-200		
	U. TRANSCOM SYSTEMS  1 David Clark		
	a. ISOCOM Voice Activated	0.7	-3.0
	2. Sigtronics		
	a. Transcom SPA-400	0.6	-2.0

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M5-180C/200/210C/210TC/235C/MX7-160/C/180/180A/C/AC/235/420/M6-180/235/ M8-235/ MT-235/260/M7-235/A/B/C/260/C/420A/AC/MXT7-160/180/180A/420/M7-235/A per STC SA2661SO DATE: 11/07/00

SERIA	AL NO	. 27008C REG. NO. NZORD	_PROD. NO. <u>2299</u>	
NOTE	: Equi	pment items installed below are designated by check marks.  /, MANUFACTURER AND PART OR MODEL NO.	WEIGHT (Lbs)	ARM (Ins)
JATE	OOK	THE TAIL OF MODEL NO.	VVEIGHT (LDS)	VIVIAL (1119)
	V.	STEREOCOM SYSTEMS		
		1. Sigtronics'	0.5	0.0
		a. Stereocom-400	0.5	-9.0
	W.	COURSE DEVIATION INDICATOR (CDI)		
		1. Terra		
		a. TRI NAV	2.3	-2.3
		b. TRI NAV C	1.3	-2.3
٨		2. Garmin		
<u>/53</u>		a. GI-106A (per Maule Dwg 7215A)	1.4	-1.9
	X.	NAV CONVERTER 1. Century		
		a. 1C707-1	0.7	-15.1
_			_ 0	
	Y.	GPS/NAV SW/ANNUNICIATOR		
<b>/</b> 53		1. Garmin (144) (155) 45 (144) (155) 45 (155)	0.0	0.0
733		a. 013-00029-10 (28v) or –11 (14v) (per Maule Dwg 7213A)	0.8	-2.0
	Z.	HORIZONTAL SITUATION INDICATOR (HIS)		
		1. Century		
		a. NSD360A-15 (per Maule Dwg 7225A)	4.6	-4.3
	Aa.	VOLTAGE CONVERTERS		
Λ	,	1. Garmin		
<u>/53</u>		a. 011-00181-00 (28v/14v) (per Maule Dwg 7229A)	4.5	18.0
<u> </u>		b. 013-00051-00 (14v/28v) (per Maule Dwg 7231A)	2.3	18.0
<b>3.</b>	VA	CUUM SYSTEM		
<b>.</b>	174	OOOM OTOTEM		
	A.	DRY AIR PUMP		
		1. Airborne Mechanisms (Reciprocating engines only)		
		a. 200 cc b. 205 cc	3.5 2.3	
		c. 211 cc	2.3 1.8	
		d. 215 cc	1.9	
		e. 225 cc		
		O Att. H. I		
		Airborne Mechanisms (Turbine engines only)     a. 212 cw	1.8	-39.5
		d. 212 0W	1.0	-59.5
		3. Sigma-Tek (previously EDO-Aire)		
		a. IU128-002 (Reciprocating engines only)		
		b. IU128-003 (Reciprocating engines only) c. IU128-006 (Reciprocating and Turbine engine)		
		c. IU128-006 (Reciprocating and Turbine engine)		
	В.	VACUUM REGULATOR		
		1. Airborne Mechanisms		
		a. 133A4	.6	-15.5 16.3
		✓ b. 2H3-12	.4	-16.3
	C.	AIR FILTER		
		1. RC Allen		
		a. J4161-01		



# MAULE AEROSPACE TECHNOLOGY, inc.

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M5-180C/200/210C/210TC/235C/MX7-160/C/180/180A/C/AC/235/420/M6-180/235/ M8-235/ MT-235/260/M7-235/A/B/C/260/C/420A/AC/MXT7-160/180/180A/420/M7-235/A per STC SA2661SO DATE: 11/07/00

SERIA	AL NO		PROD. NO. 2299	
CATE	GOR'	ipment items installed below are designated by check marks. Y, MANUFACTURER AND PART OR MODEL NO.		ADM (Inc.)
OMIL	<u>oon</u>	I, MANOI ACTORER AND PART OR MODEL NO.	WEIGHT (Lbs)	ARM (Ins)
		2. Airborne Mechanisms		
		✓ a. 1J7-1	.5	-14.2
	D.	VACUUM GAUGE		
		1 Airborne Mechanisms		
		a. 1G3-1	.7	-0.8
		b. 1G3-4	.7	-0.8
		O A		
		2. Aerosonic	<u>_</u>	
		a. 09007-0117	.7	-0.8
		3. UMA		
		a. 3-200-1	1.0	-0.8
		b. 3-200-12	1.0	-0.8 -0.8
		J. 0 250 12	1.0	-0.0
4.	VA	CUUM INSTRUMENTS		
	A.	DIRECTIONAL GYRO		
		1. Aeritalia		
		a. 31100C	1.8	-2.8
		•		
		2. Aerosonic		
		a. ANS-20	2.5	-2.8
		3. RC Allen		
		a. RCA 11A/-10	3.3	-2.8
		b. RCA 11A-8	3.3	-2.8 -2.8
		c. 15AK1-14V (14 Volt)	2.3	-2.0 -3.3
		d. 15BK1-28V (28 Volt)	2.3	-3.3
		(20 7014)	2.0	-5.5
		4. EDO-Aire/Sigma Tek		
		a. IU262-001-9/4000 B-8	2.6	-3.0
		b. IU262-001-39/4000 B-30	2.6	-3.0
		5. Sperry		
		a. A5737-1	3.3	-2.8
		6. Jack & Heintz		
		a. JH5500		
		a. JH3300		
	В.	ARTIFICIAL HORIZON		
		1. Aerosonic		
		a. ANS-30	2.5	-2.0
		_2. RC Allen		
		a. RCA-20	3.3	-2.3
		b. RCA-21	2.2	-2.3
		c. RCA-21-23	2.2	-2.3
		✓ d. RCA 22-7	2.2	-2.3
		e. 26AK2-14V (14 Volt)	2.3	-3.4
		f. 26BK2-28V (28 Volt)	2.3	-3.4
		3. Jack & Heintz		
		a JH6500		

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**RAULE AEROSPACE TECHNOLOGY, INC.** M5-180C/200/210C/210TC/235C/MX7-160/C/180/180A/C/AC/235/420/M6-180/235/ M8-235/ M5-180C/200/210C/210TC/235C/MX7-160/C/180/180A/C/AC/235/420/M6-180/235/ M8-235/ MT-235/260/M7-235/A/B/C/260/C/420A/AC/MXT7-160/180/180A/420/M7-235/A per STC SA2661SO DATE: 11/07/00

		OF HORAL EQUIPMENT LIST		
		. 27008C REG. NO. N20RD	PROD. NO. 2299	
NOTE	: Equ	pment items installed below are designated by check marks.		
CATE	GOR	, MANUFACTURER AND PART OR MODEL NO.	WEIGHT (Lbs)	ARM (Ins)
		4. Sperry		•
		a. A5736-1	4.4	-2.5
		5. Southwestern Industries	0.5	4.0
		a. AO-10-2	2.5	-4.0
		6. EDO-Aire/Sigma Tek		
		a. 23-501-06-9/5000B-20 (Blue/black display)	2.0	-3.0
		b. 23-501-06-16/5000B-36 (Blue/brown display)	2.0	-3.0
		c. 23-501-031-1/5000B-38 (Has warning flag)	2.0	-3.0
5.	FLI	<u>GHT INSTRUMENTS</u>		
<b>.</b>				
	A.	RATE OF CLIMB (For all models with reciprocating engines only)		
	_	1. Aeritalia a. 21030S	<sup>-</sup> 1.3	-2.0
		a. 210305	1.3	-2.0
		2. Aerosonic		•
		a. 340-50	1.0	-2.0
		b. 340(340)-56-1210		
		c. RC30V10		
		d. 30840-0169		
		3. United Instruments		
		<b>√</b> a. 7030M	1.0	-2.0
		b. 7040M	1.0	-2.0
		4 EDO Airo		
		4. EDO-Aire a. EA-1409-3Z	.8	-1.5
		u. Ut 1400 02	.0	-1.0
		5. Eclipse Pioneer		
		a. 1636-B-1		
		6. Garwin		
		a. 22-204-01-A		
		7. Aircraft Instrument & Dev.		
		a. 32-1007-1		
	В.	RATE OF CLIMB (For all models with turbine engines only)		
		1. Aircraft Spruce & Specialty		
		a. 10-05400	1.0	-2.0
	C.	TURN AND RANKTURN COORDINATOR		
	C.	TURN AND BANK/TURN COORDINATOR  1. Aeritalia		
		a. 26220CA	1.4	-3.0
		b. 27221	1.4	-3.0
		2 Agragania		
		2. Aerosonic a. ANS-50	2.0	-3.0
		b. ANS-51 (28 volt)	2.0	-3.0
		3. Brittain	2.2	2.0
		a. 1677A b. 600-009-900	2.3 2.3	-3.0 -3.0
		D. 000-003-300	2.3	-3.0
		4. RC Allen		
		a. A2475	2.4	-3.0
		b. A2670	2.4	-3.0

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M5-180C/200/210C/210TC/235C/MX7-160/C/180/180A/C/AC/235/420/M6-180/235/ M8-235/
MT-235/260/M7-235/A/B/C/260/C/420A/AC/MXT7-160/180/180A/420/M7-235/A per STC SA2661SO

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REV. <u>53</u>
DATE: 11/07/00

SERIAL I	10. 27008C REG. NO. NZORD	PROD. NO. 2299	
CATEGO	uipment items installed below are designated by check marks.  RY, MANUFACTURER AND PART OR MODEL NO.	WEIGHT (Lbs)	ARM (Ins)
OMILOO	THE MINING! ACTORER AND PART OR MODEL NO.	WEIGHT (LDS)	ANIVI (IIIS)
	c. RCA80A		
	d. 21500		
	e. RCA82A-1	1.3	-3.0
	f. RCA14W2D2A (14 volt)	1.0	0.0
	g. RCA28W2D2A (28 volt)		
	g. 1107 20172521 (20 1017)		
	5. Garwin		
	a. 23-324-01A		
	6. Whittaker	•	
	a. D-24		
	7. Astronautics		
	a. 303770-113MSN		
	8. Mid-Continent Instrument Co.		
	a. 1394TIOO-7Z (Turn Coordinator)	1.2	-2.0
	b. 1394TIOO-7TZ (Turn and Bank)	1.2	-2.0
	5	1.4	2.0
D	REPORTING ALTIMETER		
_	1. Aerosonic		
	a. 101420-(01249)	1.9	
	b. 101627-01340	1.4	
	c. 101627-01344	1.4	
	d. 101627-01696	1.4	
	e. 102200-01812		
	C. 102200-01012		
	2. Smiths Industries		
	a. 01-200-104	1.8	
	d. 01 200 104	1.0	
	3. King		
	a. KEA 125-13	1.9	
	b. KEA 125-14	1.9	
	c. KEA 126-17	1.9	
	d. KEA 126-18	1.9	
	e. KEA 127	1.0	
	f. KEA 128	1.8	-2.5
	- VEA 400	1.8	-2.5
	g. KEA 129	1.0	-20
	4. United Instruments		
	a. 5035P2-P39	1.8	-2.5
	a. 30001 2-1 03	1.0	2.0
	5. Terra		
	a. AT 3000	.5	-12.0
	4. /// 5555	.0	
	6. Narco		
	√ a. AR 850	.8	-3.1
E	ALTERNATE STATIC SYSTEM		
	1. Maule		
	a. 6075B		
	<del></del>		
6.	ACCESSORY INSTRUMENTS		
4	A. EXHAUST GAS TEMPERATURE KIT (EGT) (For Reciprocating engines of	nnlv)	
•	1. Alcor Aviation	,,	
	a. 394-37-935 (EA 35 4PB) (-180/160)		
	b. 394-37-937 (EA 35 6PB) (-235/260)		
	c. 394-37-946 (EGT-225)		
	· · · · · · · · · · · · · · · · · · ·		

# MAULE AEROSPACE TECHNOLOGY, INC.

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M5-180C/200/210C/210TC/235C/MX7-160/C/180/180A/C/AC/235/420/M6-180/235/ M8-235/ MT-235/260/M7-235/A/B/C/260/C/420A/AC/MXT7-160/180/180A/420/M7-235/A per STC SA2661SO DATE: 11/07/00

REV. <u>53</u>

			OFTIO	MAL EQ	DIFINIENT LIST		_	
SERIAL NO. 25	1008	<u>8C</u>	REG. NO	NRC.	PRD	PROD. I	NO. 2299	
NOTE: Equipment CATEGORY, MAN	<u>I ITEMS IN</u> JUIFACTI	ISTAILED DEIOV	v are designate	ed by chec	k marks.		WEIGHT (Lbs)	ARM (Ins)
	d. e. f.	46150 211-140-0 211-160-0 211-110-0	74.Ç. GIKIMOL	(-180/160 (-235)				/ www (me)
2.	a. b.		(14/28 volts) (14/28 volts)				.3	-0.5
3.		nt Instrument GEM-602		Pr	Instrumen obes & Wiring Harnes		.8 1.3	-3.3 -40.0
	J. P. ] a.	Instruments E.G.T. 701				_	.9	-3.5
B. CLC	DCK Borg a. b.	CA 7212 CA 7286				•	.3 .3	-0.5 -0.5
2.	Elgin a.	A-11					.3	-0.5
3.	Revie	ew Type B4.0					.3	-0.5
4.	Shin: a. b.	s American 46G 53G					.3 .3	-0.5 -0.5
5.	Wak a. b.	mann A-11 W33 7510-	101				.3 .3	-0.5 -0.5
6.	Walt a.	ham 22840					.3	-0.5
7.	Aero a. b. c. d. e.	86200-0129 86200-0129 86200-0114 86200-0119 Type C-134	5 1 5				.3 .3 .3 .3	-0.5 -0.5 -0.5 -0.5 -0.5
8.	Lonç a.	gines-Whittna A-11-90	auer				.3	-0.5
9.	Mitc a. b. c.	hell Aircraft I 98470 035-370-02 99500-ELT			(12/28 volts) (12/28 volts)		.3 .5	-1.0 -1.0
10	. Mid- a. b.	Continent In: MD-88 MD-90( )	strument		(12/28 volts)		.4 .4	-1.0 -1.0

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M5-180C/200/210C/210TC/235C/MX7-160/C/180/180A/C/AC/235/420/M6-180/235/ M8-235/ MT-235/260/M7-235/A/B/C/260/C/420A/AC/MXT7-160/180/180A/420/M7-235/A per STC SA2661SO DATE: 11/07/00

SERIAL NO	D. 27008C REG. NO. N20RD	PROD. NO. 2299	
CATEGOR	uipment items installed below are designated by check marks. Y, MANUFACTURER AND PART OR MODEL NO.	WEIGHT (Lbs)	ARM (Ins)
	1. Scott a. 2716	.1	12.0
	2. Ashcroft a. NHM70	.1	12.0
	3. Rochester  a. 1592-27	.1	12.0
	4. Piper a. 550-541		
	5. Vanmark a. 1592-30		
	6. Dresser a. 20 B160 R 000		
D.	CARBURETOR AIR TEMPERATURE GAUGE (For Carburetor engines onl 1. Aircraft Instrument & Development a. 29-208	1.0	-1.5
	2. Aerosonic a. 651250 (w/Kit B-5 Probe)		
	3. Mid-Continent a. MD-11-6		
E.	HOURMETER  1. Maule a. 6072B		
F.	ANGLE OF ATTACK (M-6-235 s/n 7249C-7465C only)  1. Safe Flight		
G.	a. SC-150  FUEL COMPUTER (For Turbine engines only)	1.0	-1.0
	1. ARNAV a. FC-10	1.0	-1.3
Н.	FUEL FLOW METER  1. Shadin Co., Inc. a. Miniflow-L	14.0	-2.5
	b. Microflow-L	14.0	-2.5
7. <u>A</u> (	<u>CCESSORIES</u>		
A.	HEATED PITOT  1. Aero Instruments  a. PH-502-12 (14 volt)  b. PST-305-12 (14 volt)  c. PH-502 (28 volt)	1.0 1.0 1.0	43.0 43.0 43.0
В.	c. PH-502 (28 volt)  FIRE EXTINGUISHER	1.0	<b>43.</b> U
	1. Balkamp a. 4-2740 b. 770-7007	4.5	8.1

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M5-180C/200/210C/210TC/235C/MX7-160/C/180/180A/C/AC/235/420/M6-180/235/ M8-235/ MT-235/260/M7-235/A/B/C/260/C/420A/AC/MXT7-160/180/180A/420/M7-235/A per STC SA2661SO DATE: 11/07/00

SERIAL NO. 27008C REG. NO. N20RD	PROD. NO. <u>2299</u>		
NOTE: Equipment items installed below are designated by check marks.  CATEGORY, MANUFACTURER AND PART OR MODEL NO.	WEIGHT (Lbs)	ARM (Ins)	
2. General a. CP-234 b. CP-2 1/2J c. TCP-2 1/2J d. GH-2 1/2J	4.5 4.5 4.5 5.4	8.1 8.1 8.1 8.1	
3. Kidde a. Type BC Size 1  4. C.O. Two Fyre Fighter	4.5	8.1	
5. Amerex a. 403423 6. Pemall	5.0	8.1	
6. Pemall a. PA27ABC	4.5	8.1	
7. Dual-Halon a Model RT-A600	2.0	8.1	
C. LANDING LIGHT  1. Maule  a. 9030E	1.0	5.0	
D. FLOAT PLANE REINFORCEMENT  1. Maule  a. 9001F, Sht:1  b. 9001F, Sht:2 (includes forward reinforcement)			
E. AUTO FLIGHT SYSTEMS (AUTOPILOTS)  1. Century Flight Systems  a. Century IIB/AK 513  b. Century 21/AK 932	10.6 13.4	7.8 7.8	
2. S-Tec Corporation a. 9193A System 50 (14v) b. 9195A System 55 (14v) c. 9196A System 55 (14v) d. 9197A System 20/30 (14v) e. 9200A System 50 (28v) f. 9201A System 20 (28v)			
F. RADIO COUPLE  1. EDO-Aire (For E.1.a.)  a. IC 388M			
G. GLIDER TOW HITCH  1. Maule  a. 3196F (For Tailwheel models only) (Schweizer & Tost)  b. 9160B (For Nosewheel models only) (Schweizer)  c. 9207E (For Nosewheel models only) (Tost)	<b>4</b> .5 6.3	123.5 176.0	
H. WHEEL FAIRINGS  1. Maule  a. 4034B			
I. SIREN/PA SYSTEM			

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M5-180C/200/210C/210TC/235C/MX7-160/C/180/180A/C/AC/235/420/M6-180/235/ M8-235/ MT-235/260/M7-235/A/B/C/260/C/420A/AC/MXT7-160/180/180A/420/M7-235/A per STC SA2661SO DATE: 11/07/00

IAL NO.	27008C REG. NO. N20RD oment items installed below are designated by check marks.	_PROD. NO. 2299	
EGORY	, MANUFACTURER AND PART OR MODEL NO.	WEIGHT (Lbs)	ARM (Ins
	a. PA 100 Amplifier	7.5	5.0
	b. PA 200 Amplifier	5.0	-3.0
	c. TS-100 Speaker	8.0	95.0
J.	RADAR ALTIMETER		
	1. King a. KRA-10 Radar Alt	2.0	
K.	PORTABLE OXYGEN SYSTEM		
IX.	Rajay Industries		
	a. SK-9	10.0	38.0
	b. SK-10	12.0	38.0
	2. Scott Aviation Products		
	a. Executive Mark I	9.0	38.0
	b. Executive Mark II-	14.0	38.0
	D. Executive Mark II-	14.0	30.0
	3. Puritan-Bennett Aero System		20.0
	a. ZP 202		38.0
	b. ZP 204		38.0
L.	LONG RANGE PORTABLE OXYGEN SYSTEM		
	1. Rajay Industries	47.0	00.0
	a. SK-9-30	17.0	60.0
	b. SK-9-40	19.0	60.0
	c. SK-9-48	25.0	60.0
	d. SK-10-30	17.0	60.0
	e. SK-10-40 f. SK-10-48	19.0 25.0	60.0 60.0
	2 Duritor Borrott Assa Contain		
	2. Puritan Bennett Aero System	17 E	60.0
	a. ZP 400 b. ZP 401	17.5	60.0
	0. ZP 401	17.5	60.0
M.	AUXILIARY HEATER		
	Maule (TW Models w/Reciprocating engines only)     a. 5310E	2.1	18.1
		2.1	10.1
N.	OPTIONAL JUMP SEAT  1. Maule		
	1. Maule a. 1216B (M6/8-235/MX7-180/B/C/AC/235/420/ MXT7-180/420)	8.8	70.0
		0.0	70.0
Ο.	STORM SCOPE  1. B.F. Goodrich		
	a. WX900	1.6	-3.7
_			
P.	STRIKE FINDER		
	1. Insight a. SF 2000	1.2	-4.7
_			
Q.	AVIONICS COOLING KIT  1. King		
	a. KA-33	1.1	-12.0
	2. Narco		
	a. 03312-502	.2	-12.0
R.	DIPLEXER/TRIPLEXER		
11.	1. Meriden		

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**ΠΑULE ΔΕΤΟΣΡΑCE TECHNOLOGY, INC.**M5-180C/200/210C/210TC/235C/MX7-160/C/180/180A/C/AC/235/420/M6-180/235/ M8-235/

MT-235/260/M7-235/A/B/C/260/C/420A/AC/MXT7-160/180/180A/420/M7-235/A per STC SA2661SO

DATE: 11/07/00

SERIAL NO. 27008C REG. NO. N20RD	PROD. NO. 2299	
NOTE: Equipment items installed below are designated by check marks.		
CATEGORY, MANUFACTURER AND PART OR MODEL NO.	WEIGHT (Lbs)	ARM (Ins)
a. NT-1A (Triplexer)	.2	-8.0
b. NC-4 (Diplexer)	.3	-8.0
S. VORTEX GENERATORS		
_1. Micro Aerodynamics		
a.	(Negligible)	
T. AFT FLOAT ATTACHMENT		
1. Maule		
a. 1348E Plate (weld-on)	1.7	55.2
b. 1350B Installation (not incl. 1348E) Float Attachment	3.0	55.2