1984 Citation SII

Damage History Entries

MSN: S550-0004



1984 Citation SII

Nose Incident

MSN: S550-0004



Office DEPOT



Maintenance Transaction Record

This Maintenance Report is To Be Used Solely For (Check One)

mobile income security and the properties of the

MTR ID # 2813957 Page 1 of 1

CAMP Systems International 8200 E. 34TH STREET BUILDING 1600, SUITE 1607 Wichita, KS 67226

						Alleran Re	entification and Stat				- 174 Mary 1940	Table I	APU Events	Eroon Hrs
A/C Serial #	AC Unit #	A/C Registration #	Date	City ID	Total A/C Hours		s Engine 1 Ttl Hrs		Hrs Engine 1 Ttl Cycle 7539		Ttl Cycles	APU Hrs	APU EVEILS	0
\$550-0004	0004	N178DA	14-Dec-2017	KSUS	7512	7812	7409,1	7409.1						-
0000			THE PARTY OF	Comp	onent Changes In	spections Servic	e Bulletins or Airwo	orthiness Direc	ctives Accomplished					
	Transac		No.	ANNY	Position	Part Number /	10 TO	Mod Level	Part Serial	Removal Reason	Status	nstalled Part TSN/TSO/TS	Materia R Costs	Man Hours
ltom _	No. T	ype	Item Name	AND RES	Position				N/A	N	N	OMTHOHR	S	
B13957001	1 1 Commen	RADOME ts: Removed and repl ed the new radome to	aced the rador	ne with nev	w in reference to Ce	6591101-7 ssna Model S550 strip (p/n S2800-3	MM chapter 53-50-0	1. Fitted and d	rilled the new radome to	fit and applie	d aerodyr	namic filler. San	ded	
ALC: 198	THE RESERVE		maton die die	Mario. Iriott		5513090-10			N/A	N	N	O MTH O H	RS	
13957	2 1	FRAME	IN PARTIES H		NOSE COME		reference to Cassn	Model 500 S	eries SRM chapters 51-	30-03 and 53	-11-00.			
13957	Comment 3 1	BULKHEAD	iced the forwa	rd frame fo	FWD FUSELAGE	5513092-2	Tibletence to Cassin		N/A	N	N	0 MTH 0 H	IRS	
	Comments	· Removed and repla	ced the forwar	d bulkhead	EDAME	rd fuselage frame	with new in referen	ce to Cessna f	Model 500 Series SRM	chapters 51-3	0-03 and	53-11-00.		
	-	ONIN DEDAID	CONTRACTOR OF THE PARTY OF THE	7-2		WEST TO STATE OF			ace Drawing No72G-70			Town WA		
957004	Comments:	Repaired the left sld	e skin on the fi	orward par	I DI DIO INSCIMBO OF							0 MTH 0		

Removal Reasons: WO - Worn to Limits, SC - Scheduled, UN - Unscheduled, CO - Convenience, N - Other(note in comments)

Installed Part Status: N - New, R - Repaired / Rebuilt, S - Serviceable, O - Overhauled, T - Tested, M - Modified, I - Inspected

						togradim meaning to michael		THE PERSON NAMED IN STREET
				Work	N. Committee of the com		Mainten	ince Log
Repair Facility	KCAC AVIATION	Certified Repair Station Number	M9TR576Y	Order No	MS-	17-03868	Section	Page
Work Performed By			Certificate No.	M9TR576Y	Date	14-Dec-2017		
I certify that the a	above stated maintenance and/	or inspection was performed in acc	ordance with the current regulation	ons of the				
Federal Aviat	ion Administration	Other (Specify				and		

THE AIRCRAFT, AIRFRAME, ENGINE, PROPELLER, OR APPLIANCE IDENTIFIED ABOVE WAS REPAIRED AND INSPECTED IN ACCORDANCE WITH CURRENT REGULATIONS OF THE FEDERAL AVAILATION ADMINISTRATION AND IS APPROVED FOR RETURN TO SERVICE WITH RESPECT TO THE WORK PERFORMED. PERTINENT DETAILS ARE ON

1984 Citation SII

Bird Strike Repair

MSN: S550-0004



US Department of Transportation Federal Aviation

MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

Form Approved OMB No. 2120-0020

For FAA Use Only

Office Identification

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43. Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in civil penalty not to exceed \$1,000 for each such violation (Section 901 Federal Aviation Act of 1958).

, for eacl	h such	violation (Section 9	901 Federal Aviation	n Act of	1958	3) .					.
		Make CESSNA					Model S-	-550			
1. Aircra	ift	Serial No. 0004									
2. Owne	r	Name (As shown o J. R. TOMKINSON	on registration cen N INC.	rificate)	,		1401 DO	(As shown on regis OVE STREET RT BEACH, CALIFC			
		•			3. Fc	or FAA Use O	nly				
,							.,				nade in
•	•		• "		4. Uı	nit Identificati	on			5. Type	
Uni	t	Ma	ake		,	Model		Serial No).	Repair	Alteration
AIRFRAMI	E			(As desc	cribed	d in Item 1 abo	ve)——			. ×	
POWERPI	LANT	programme Personal		4		· Fine g	.417 (C)				
PROPELLI	1		garter		i ne	ers con quest su	41 - 1987 		· .		
APPLIANO	50	Type Manufacturer	The Company of the Co								
, .		<u>. </u>		6.	Con	formity State	ment	<u> </u>			<u> </u>
A. Agend	cy's Na	ame and Address				and of Agency			C. Certif	icate No.	
AIRCRA	FT M	AINTENANCE SE	RVICES			U.S. Certificate	d Mechanic	;			
3950 ME						Foreign Certific	ated Mecha	inic	CRSF	11R638N	
KISSIMIN	/IEE, I	FLORIDA 34741			X	Certificated Re	pair Station		-		
D. I cer	rtify the	at the repair and/or alte	ration made to the uni	t(s) identif	ied in	Manufacturer item 4 above and	described	on the reverse or attack	l hments here	to have been m	ade in
accordance	ce with	the requirements of Pa	art 43 of the U.S. Fede	ral Aviatio	n Regi	ulations and that t	he informati	ion furnished herin is tru	ıe and сотте	ct to the best of a	my knowledge.
Date		03/09/2007			Sigr	nature of Author	rized Indiv	vidual			
				7. App	rova	for Return/T	o Service	•			
Pursuar Adminis	nt to th strator	ne authority given pe of the Federal Avia	ersons specified b tion Administration	elow, the	e unit [identified in√ite ☑ APPROVED		nspected in the mar REJECTED	nner presc	ribed by the	
ву	FAA Inspe	Flt. Standards ector	Manufacturer		Insp	ection Authoria	zation	Other (Specify	()		
		Designee X	Repair Station		Can	son Approved ada Airworthir	ess Grou	P	••	: •••••	
Date of A		al or Rejection 09/2007	Certificate or Designation No. CRS R1JR6		Sigr	nature of Autho	orized Indiv	vidual			• :
FAA Fo	orm 3	37 (12-88)		Elec	tronic V	/ersion (Adobe)					

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

N72AM, S-II 550, SERIAL NUMBER 0004

- 01. REMOVED HORIZONTAL STABILIZER ASSEMBLY P/N 5532000-48, S/N 509 PER CITATION S-550 M.M. CHAPTER 55-10-01 PAR. 3 A, FOR STRUCTURAL REPAIR DUE TO L/H LEADING EDGE BIRD STRIKE DAMAGE.
- 02. PERFORMED A DETAILED STRUCTURAL INSPECTION PER CITATION S-550 M.M. CHAPTER 55-10-01-212, PAR. 3, (HORIZONTAL STABILIZER INTERNAL INSPECTION), AND PAR. 4, (HORIZONTAL STABILIZER FORWARD, AFT AND AUXILIARY SPARS INSPECTION).
- 03. REMOVED AND REPLACED THE FOLLOWING STRUCTURAL MEMBERS PER CITATION S-550 STRUCTURAL REPAIR MANUAL 55-10-09, PAR 2 :
 - 1. SKIN ASSEMBLY 124A, P/N 5532000-53, 2. SKIN ASSEMBLY LOWER 4A, P/N 5532000-45
 - 3. SKIN LEAD, EDGE L/H, P/N 5532000-65 4. RIB 24A, P/N 5532024-16
 - 5. RIB P/N 5532023-23, 5532023-24.
- 6. TIP AY 124A, P/N 5532030-3, L/H AND R/H

	7. L/E PANEL, D3715				
04	I. INSTALLED HORIZONTAL STA PER CITATION S-550 M.M. CH			6/N 509 AFTER REPAIR	
. 05	5. INSTALLED L/H AND R/H ELEV VERIFIED ELEVATOR NEUTR PAR. 3 A. RIG ELEVATOR TRIM 27-31-00 PAR. 3 B AND 4 A.	AL ALIGNMENT PE	R CITATION M.M. CH	IAPTER 27-31-00,	
06	5. INSTALLED AIRFOIL ANTI-ICE SYSTEM TEST AND LEAK CHI 30-10-00, PAR. 2,12,14,16, 20-	ECK, PER CITATION			
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	· · · · · · · · · · · · · · · · · · ·	[] Additional She	ets Are Attached	·	
					•

1984 Citation SII

Rudder Repair

MSN: S550-0004



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US Departmen of fransportant Federal Aviats Administration	on .	(A	MAJOR RE	PAIR AND ALTERATION AND PROPERTY OF A	TION Appliar	nce)		o. 2120-0020 For FAA Us	
INSTRUCT and disposi for each su	IONS: Pri tion of thi ch violatio	ntor slori on (S	type all entries. See m. This report is requ ection 901 Federal A	FAR 43.9, FAR 43 Appendix E ired by law (49 U.S.C. 1421). viation Act of 1958).	, and AC Failure t	43.9-1 (or subseque o report can result ir		entification	FIL FSBC
1. Aircraft	Make Serial	Ces	sna		Mode	S550			xceed \$1,000
2. Owner	Name (/	As st	0-0004 nown on registration ne Supply Com		Addre 10	nality and Registrat N830CB PSS (As shown on re 1 Meatte St. rtageville, M	gistration		
				3. For FAA Use (<u>`</u>	
	· · -			•					;
Unit				4. Unit Identificat	on			F Tune	
			Make	Model		Serial No	 D.	5. Type	7
IRFRAME		~~		(As described in Item 1 abo	(B)			Х	Alteration
OWERPLANT	·		\$ ⁶ ,	S (_		DEC 11	. / /	,	
PLIANCE L	ype	rer	• 2		* "	BIL ESO	DE S		
Agency's Nam	e and Add	tress	7	6. Conformity Statem	ent				
Premier A 18 Termin E. Alton,	ir Cen al Dri IL 62	ter ve 024	, Inc.	B. Kind of Agency U.S. Certificated Foreign, Certificat X. Certificated Repa	ed Mecha ir Staiion	inic (C. Certific	068н	
have been maked her	ne repair a ade in acc rein is true	eng/ ords	or elteration made to ince with the require I correct to the best i	the unit(s) identified in item ments of Part 43 of the U.S. of my knowledge.	above a Federal	and described on the Aviation Regulation	reverse or	attachment	s hereto
- 12	2-8-89			Signature of Authori	red Indiv	vidual	Q	···o tillolina	lion.
irsuant to the i	authority the Federi	give:	n persons specified	7. Approval for Return To S below, the unit identified in and is X APPROVED			Inspe	ctor	7
Inspector			Manufacturer	Inspection Authorization		JECTED Other (Specify)	ie mannei	prescribed	by the
FAA Desi		x	Repair Station	Person Approved by Tr. Canada Airworthiness (auor				
12-8-89			Pasignation No. PAZRO68H	Signature of Authorize	a Individ Leck	<u></u>	Insp	ector	

TIN WAR WILL

Weight and milance or opera ing limitation changes shall be entered in the appropriate aircraft record. An alteration must be computable with all previous alterations to say ire continued conformity with the applicable airworthiness requirements.

Description of Work A compile ed

(if more space is requires, attac | additional snowle identity with aircraft nationality and registration mark and date work completed.) Removed rudder from aircrass to removed the lower skin from rudder. Removed sixth interspar rib from bottom of rudder a ...67.1 to to WL165.0. Installed new rib in same location, P/N 5533110-82 (rlb supplied by Cessna Aircraft). Fabricated new LH lower skin from .025 2024T3 alclad aluminum and installed Same number, type and size fasteners used as original P/N NAS1097AD4; NAS1097AD5 & CR3242-4. Refinished rudder, checked balance & reinstalled on aircraft with new lower bearing P/N MS20200KP4 & new banding jumper P/N MS25083-2AB4. Checked travels & rigging of. rudder & rudder trim tab per Cessna S550 maintenance manual. All work was performed in accordance with AC43.13 paragraph "B" and Cessna Citation structural repair manual

Additional Sheets Are Attached