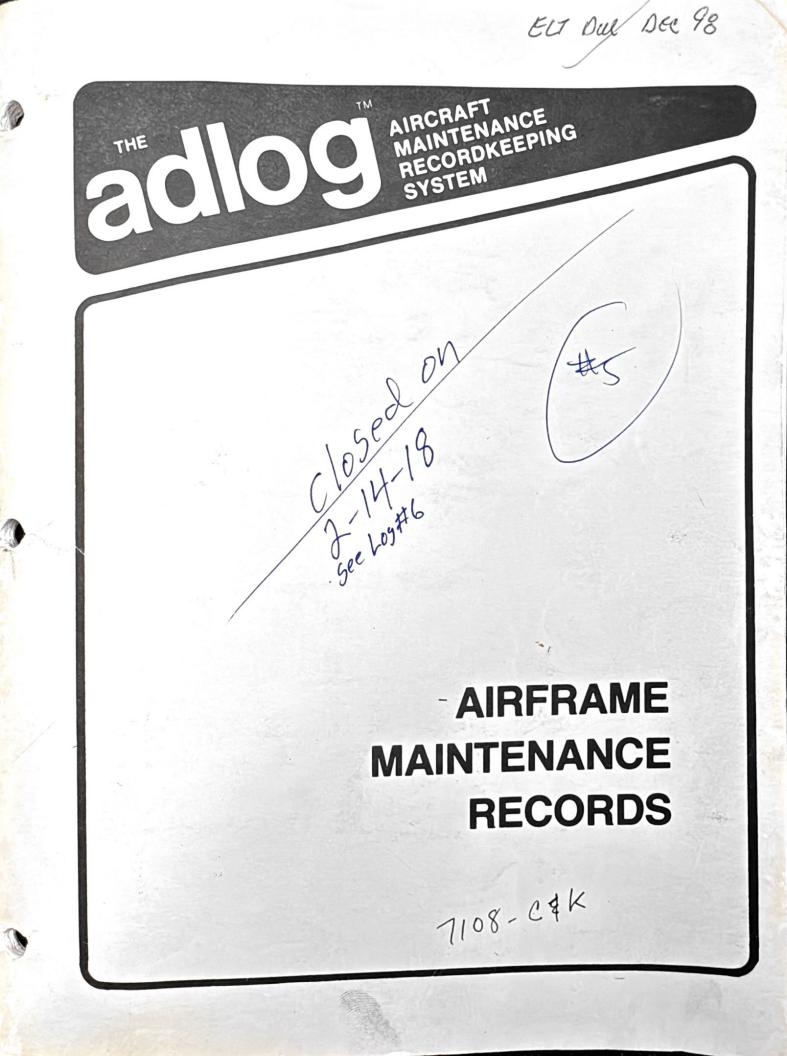
# **N5288J** 1976 Cessna 421C

# **Airframe Logbook**#5 of 6 1997 - 2018

## MSN: 421C-0158



Prepared by the worldwide aviation specialists at RidgeAire, Inc.





# AIRFRAME MAINTENANCE RECORDS

Log No. 5 Aircraft Registration No.\_\_ 5288ン Aircraft Mfg. CESSNA Model 421C Serial No. 4216-0158 Engine Mfg. CONTINENTAL Model GTSIO - SZOL Serial No. 6/4 245802-R Engine Mfg. CONTINENTAL Model GTSIG - SZO-L Serial No. 8/4 608108 Propeller Mfg. MC CAULEY Model 3FF32C501-A Serial No. 4/H- 764541 Hub Serial No.\_ Hub Design No. Blade Design No. 904MB-0 Blade Serial No's. Propeller Mfg. MCCAULEY Model 3FF 326561-A Serial No. PH 820823 Hub Serial No.\_\_\_ Hub Design No.\_ 90 UMB-0 Blade Serial No's. Blade D<del>esign N</del>o.

(All applicable information must be filled in)



Page No. DESCRIPTION OF WORK PERFORMED-TACH OR TOTAL RECORDING SIGNATURE & CERTIFICATE NO. OF PERSON PERFORMING WORK TIME METER DATE IN SERVICE TIME TOTAL brought forward from previous page 12/24/97 Permanent airframe Log entry for C421 N5288J Hourmeter: 748.6 ELT s/n 161844 inspected and tested per FAR 91.207d. Battery expires 12/98 "G" switch and signal strength satisfactory. William C. Kunder A&P 552949948 0 W/O 1024 6133 Freeport Blvd. Executive Air Maintenance 916\428-6345 Sacramento, CA 95822 12-29-97 4847.3 7.49.7 RT MAIN TIRE + REPALLED BEARINGS ROTATER RT NEN TUBE MAIN TIRE INSTALLED INSTALLED BODOVEAR 650×1078 DLY TIRE LT MAN SERVICED 111 NOSE aur AP2020357 1-72-98 760,3 VR286 FAA/PMA VOLTAGE REGULATORS 1) INSTALLED 2 AD SUSSED TO AND SPEC AP 202085 3-6-98 4878.4 780.8 RESECUCED AT WING FUEL TANK DANEL SCREWS REMOVED + REINSTALLED BOTH Pase FOR OVERHAUL REPAIRED LT COWL FASTENERS AS NSEDED INSTALLED NEW LT PROP DEICER BRUSHES (5 INSTALLED OUBRHAURED HEATER FUEL VALUE CLEARNED UNDER CABIN FLOOR 6 D REMOVED + REINSTOLLED CABIN FIRE EXTINGUISHER FOR LERTIFICATION & RECHARGE (NOTE UNIT HAD LOW PRESSURE AND SUB-TOTAL this page WAS NOT DISCHARGED IN ACET TOTAL-Carry forward to next page CONT

Page No. \_\_\_ TOTAL TIME TACH OR RECORDING DATE DESCRIPTION OF WORK PERFORMED 194 METER SERVICE SIGNATURE & CERTIFICATE NO. OF PERSON PERFORMING WORK THAT TOTAL brought forward from previous page 3-6-98 4878-4 780,8 3 SUUNIS GEAR & REREGEMED EMERGENCY BLOW DOWN OPS CHELL NORMAL SERVICED WITH 102 ( MSTALLED 2 B3-5-1 VALVUM RELIEF FICTERS I CERTICY THAT THIS AIRCRAFT HAS BEEN INSPECTED IN ACCORDANCE WITH A CESSINA PROCEESSIVE CARE OPERATION # 2 AND IS PPPROKED FOR DETURN TO SERVICE Warnit fall AP 50 50 57 74 2-13-98 785.4 INSTALLY NEW TALK GENERATOR LT ENGINE P/N 22, A667 Wan's fiftely AP 2020557 INSTALLED OVER HAULED LT ALTERNATOR 796.7 4-1-98 R/N ALV 95103 Dand fipal AP 2020557 61-98 4917,2 819,6 () "JALLED ACRT TO FACILITATE LANDING CEAR WORK 3 INSTALLED NEW 650-10-10RY FUGHT CUSTON TIRE RT MAIN CLEANED + GREASED WHEEL BEARINES INSTALLED ONE NEW NESE WHEEL BEARING CLEANIED OF REGREASED NOSE BEARINGS RELINED RT BRAKE & RESEALED SLAVE CYLINDER E David Atalf AP2020357 SUB-TOTAL this page TOTAL-Carry forward to next page

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| 7-10-98        | 4925.0  | 827.4                                 | 2.6.431 (19,3814 780,8 )  |
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| ar enable      | 1948 - S. | Ð                                     | INSTALLED NEW AUTO DILOT DISCOUNCET SWITCH  |
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|                |   | F                                     | PLEANED, GAPDED & TESTED SPARE PLUGS  |
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|                | 1.  | ~                                     | PROGRESSIVE CARE OPERATION #3 AND IS  |
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|   | RIGHT                          | MAIN (                                | EAR ACTUATOR, NOSE GEAR UPLOCK ACTUATOR   |
|   | AND                            | RIGHT                                 | MAIN UPLOCK ACTUATOR.   |
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| 2. <sup>1</sup> 63.   | GEAR                           | Bron                                  | DOWN TEST & SERVICED SYSTEM   |
| B   | INSTAL                         | LED I                                 | VEN BEARING RUDDER TRIM HORN  |
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| Ô   | REMOV                          | 02                                    | REINSTALLED FIRE EXTINGUISHER FOR SERVICE   |
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Page No. TOTAL TACH OR TIME RECORDING DATE DESCRIPTION OF WORK PERFORMED-IN SERVICE SIGNATURE & CERTIFICATE NO. OF PERSON PERFORMING WORK TIME TOTAL brought forward from previous page 11-2299 5094.9 996.6 6-00 5116.0 1618 SERVICED TIRES BRAKES, + BATFERY LUSED CONTROLS + Kitor CI 18ATERS LIGHTS AP2020957 12-10-99 51 05.7 1008,1 ( INSTALLED FALTORY NEN FUEL TRANSFER LT WING LOCAER P/N 476411 Pump 25 CD 5196 9 (2) CHECKED 9 OPS Y NO LEACS OPERATION TOR NORMOL APZOZOG 12-29-99 5618402-3 BOOT DEICER JIMER FACTORI NEW INSTALLED DEICER PRESSURE HOSE FROM 3 ENGINE INSTALLED 27 PUMP TO VALVE CONTROL DEICER (3 VALVY CLEANED 17 For HRORER OPERATION CHECEED ISTEM AP2020957 (MASSARD) HINGS 1-13-00 RESEALED NOSE SERVICED WITH STRUT ALFT JAURED  $\bigcirc$ FROM JACT REMODED ALFT MIL 5606 AP2020 1712 I REPUCTALE SUB-TOTAL this page TOTAL-Carry forward to next page

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DESCRIPTION OF WORK PERFORMED-TACH OR RECORDING TOTAL SIGNATURE & CERTIFICATE NO. OF PERSON PERFORMING WORK TIME DATE METER IN SERVICE TIME TOTAL brought forward from previous page 2-6-00 5116.0 1018.4 DINSTRUCED & NEW 650X10 FLIGHT CUSTOM TIRES BEARING REPAURED WHEEL INSTALLED NEW SEAL & REDALLED NOSE WHEEL BEARINGS (2)SWUND LANDING GEAR Dand PAtel AP 2020357 9-25-00 5146,9 1049:3 REMOVED ORIGINAL ELT AND INSTALLED AK450 ELT (D) FABRICATE WEN MOUNT FOR BUT AND INSTALLED REMOTE SWITCH & INDRATOR. WIRING FOR REMOTE ROUTED THRU AN EXISTING HOLE & RESEALED SWITCH WAS AT REAR PRESSURE BURK HEAD 2) RESECURED LT BAGGAGE DOOR STAY JALLED ACRT, SWUNG LANDING BEAR, DERFORMED (3)BLOW DOWN TEST, INSTALLED 2 NEW EMERGENCY SYSTEM FILTERS, SERVICED WITH MIL 5606 & NE (4) RT MAIN GRAD, AFT OF GRAD, SKIN PATCH NORN AND LOOSE. INSTALLED LARGER PATCH (5) INSTALLED NEW LT GEAR DOOR HINGE PIN ( IN STALLED NEW AN SOF- 8R-8 SCREWS LT GEAR DOOR INBOARD HINGE HALF DINSTALLED NEW FAA | PMA WHEEL BEARING LT MAIN P/N LM29749-20629 + LM29710-20629 (2) RESECURED SEVERAL RIVETS LT FLAD OR REPLACED 19 RESECURED LT WING TIP TRAILING EDGE ABOUT I NEAR AILERON SUB-TOTAL this page CONT 7 TOTAL-Carry forward to next page

Page No. \_ TOTAL TACH OR TIME DATE RECORDING DESCRIPTION OF WORK PERFORMED-IN METER SERVICE SIGNATURE & CERTIFICATE NO. OF PERSON PERFORMING WORK TIME 5146.9 TOTAL brought forward from previous page (10) RESECURED REAR CABIN DOOR CABLE INSPECTED ALLARON, RUDDER, & ELEVATOR TRINK TAB FREERAY PER BERUILE MANUAL ALL ARE WITHIN LIMITS REMOVED BUT FLOW VALUE AND CLEANED PER INSTRUCTION (12) ON VALVE IN STALLEO 113 NEW RIGHT SIDE OVERBOARD EXHAUST TAIL PIPZ WITH NEW BELLOWS & REIBOW (ID REMOLEO AT TURBO W/E REPAR BY FOR KNISLEY SEE TAG (15 INSTALLED 2 NEW OVERBOARD BELLOWS CLAMPS (16) INSTALLED NEW WASTE 24TE DRAIN HOSE LT BUG 17 INSTALLED NEW HOSE LT ENGINE FUEL POMP TO METERINC BODY (18 INSTALLED OVERHAULED WASTE BATE ACTUATOR LT ENG 119 INSTALLED NEW HOSE ENG TO OIL FILTER ADAPTER RIGHT ENGINE 2 NEW BRACKETT AIR CLEANER ECEMENTS 20) INSTALLED HEATER FUEL FILTER & SPARK PLUG 121 CLEANED BOTH ENGINE FUEL STRAINERS 22 CLEANED NEW EGT PROBE IT ENGINE INSTALLED SEZ Æ4 2 DA AD LISTING FOR I CERTIFY THAT THIS AIRCRAFT SILL CERTIFY THAT THIS ELT HAS BEEN INSPECTED IN ACCORDANCE WITH HAS BEEN INSPECTED IN ACCORDANCE FAR91.207(D) WITH AN AMMALINSPECTION AND WAS NEXT DUE DETERMINED TO BE IN AIRWORTHY CONDITION SUB-TOTAL this page TOTAL-Carry forward to next page

Page No. TACH OR DESCRIPTION OF WORK PERFORMED 10TAL TIME RECORDING SIGNATURE & CERTIFICATE NO. OF PERSON PERFORMING WORK DATE METER IN TIME SERVICE DATE: 08-23-02 N5288J S/N: 421C-0158 W.O. #: 12332 MODEL: 421C TOTAL TIME: 5154.4 **TACH TIME: 1056.8** CPAFT MAINTENA AIRFRAME LOG FAA APPROVED REPAIR STATION # JVR016L 2980 AIRWAY AV,, COSTA MESA, CA 92626 CIRUS DESIGN AUTHORIZED REPAIR STATION TEL: 714-433-2275 AND 714-433-2276 THE AIRCRAFT AND / OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED AND INSPECTED IN ACCORDANCE WITH CURRENT FEDERAL AIR REGULATIONS AND WAS FOUND AIRWORTHY FOR RETURN TO SERVICE, PERTINENT DETAILS OF THE REPAIR / REPCACEMENT ARE ON FILE AT THIS AGENCY. SIGNED: in 08-23-02 DATE: C/W an ANNUAL INSPECTION as per Cessna 421C Annual Inspection Guide. C/W Yearly Inspection of ELT as per 91-207(d) battery replacement due at 03-04. Installed a Ram Overhauled engine into the Left side of this aircraft on this date. C/W AD200-01-16 Exhaust Cracks, no cracks found at this time, AD next due at 1106 Tach. R & R Oxygen Bottle for Hydrostatic check done by Fire Stop. Installed JPI Engine Monitoring System P/N EDM760 as per STC# SA00729SE. FAA form 337 made out this date. Installed new Instrument Filter P/N AM1035351A. Installed new Air Filter on Right Engine P/N BA3. END 9-16-02 Seridar 250 3 3 7 SUB-TOTAL this page TOTAL-Carry forward to next page

| DATE  | TOTAL  | TACH OR   |  | -                         |
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|       | S/N: 421C-   | 0158  |  |                           |
| 200   | MODEL: A   | 121C  | DATE: 12-18-02   |                           |
|       | inch m   | TE: 1108.0  | W.O. #: 12600<br>TOTAL TIME: 5205.6  |                           |
|       |  |   | AIRFRAME LOG   |                           |
| 100   | 2980 AIR<br>TEL: 7   | WAY AV,. COS<br>14-433-2275 AN  | STA MESA, CA 92626       FAA APPROVED REPAIR STATION # JVR016L         SD 714-433-2276       CIRUS DESIGN AUTHORIZED REPAIR STATION  |                           |
|       | THE<br>ACCORD/<br>TO SEE   | CAIRCRAFT AN<br>ANCE WITH CU<br>ANCE. PERTIN  | ND / OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED AND INSPECTED IN<br>URRENT FEDERAL AIR REGULATIONS AND WAS FOUND AIRWORTHY FOR RETURN<br>NENT DETAILS OF THE REPAIR / REPLACEMENT ARE ON FUE AT THIS AGENCY.   |                           |
|       | DATE:  | 12-18-02  | SIGNED: Aren and Anin and  |                           |
|       |  |   | The and the and the  |                           |
|       | R & R Attitud  | de indicator for re   | pair done by Global Tech-under their WO#43033. END   |                           |
|       |  |   |  | -                         |
|       | -  |   |  |                           |
|       | 5288J  |   |  |                           |
| N     | /N: 421C-0158<br>IODEL: 421C   |   | DATE: 02-11-03   | -                         |
|       | ACH TIME: 11   | 24.9  | W.O. #: 12739  |                           |
|       |  |   | TOTAL TIME: 5222.50  |                           |
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| A     | TEL: 714-43<br>THE AIR<br>CCORDANCE<br>TO SERVICE  | 3-2275 AND 714<br>CRAFT AND / O<br>WITH CURRE!<br>. PERTINENT   | A433-2276 CIRUS DESIGN AUTHORIZED REPAIR STATION # JVR016L<br>CIRUS DESIGN AUTHORIZED REPAIR STATION —<br>OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED AND INSPECTED IN<br>NT FEDERAL AIR REGULATIONS AND WAS FOUND AIRWORTHY FOR RETURN<br>DETAILS OF THE REPAIR / REPLACEMENT ARE ON FILE AT THIS AGENCY.  |                           |
| A     | TEL: 714-43<br>THE AIRC<br>CCORDANCE<br>TO SERVICE   | 3-2275 AND 714<br>CRAFT AND / O<br>WITH CUPPE   | I-433-2276 CIRUS DESIGN AUTHORIZED REPAIR STATION  |                           |
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| A     | TEL: 714-43<br>THE AIR<br>CCORDANCE<br>TO SERVICE<br>DATE: 02-<br>R Vacuum pum                   | 3-2275 AND 714<br>CRAFT AND / O<br>WITH CURREN<br>. PERTINENT<br>11-03<br>p with Overhauk   | 1-433-2276       CIRUS DESIGN AUTHORIZED REPAIR STATION         OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED AND INSPECTED IN         NT FEDERAL AIR REGULATIONS AND WAS FOUND AIRWORTHY FOR RETURN         DETAILS OF THE REPAIR / REPLACEMENT ARE ON FILE AT THIS AGENCY.  |                           |
| A     | TEL: 714-43<br>THE AIR<br>CCORDANCE<br>TO SERVICE<br>DATE: 02-<br>R Vacuum pum                   | 3-2275 AND 714<br>CRAFT AND / O<br>WITH CURREN<br>PERTINENT<br>11-03<br>p with Overhauk   | 1-433-2276       CIRUS DESIGN AUTHORIZED REPAIR STATION         OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED AND INSPECTED IN         NT FEDERAL AIR REGULATIONS AND WAS FOUND AIRWORTHY FOR RETURN         DETAILS OF THE REPAIR / REPLACEMENT ARE ON FILE AT THIS AGENCY.  |                           |
| A     | TEL: 714-43<br>THE AIR<br>CCORDANCE<br>TO SERVICE<br>DATE: 02-<br>R Vacuum pum                   | 2-2275 AND 714<br>CRAFT AND / O<br>WITH CURREN<br>. PERTINENT<br>11-03<br>p with Overhauk   | 1-433-2276       CIRUS DESIGN AUTHORIZED REPAIR STATION         OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED AND INSPECTED IN         NT FEDERAL AIR REGULATIONS AND WAS FOUND AIRWORTHY FOR RETURN         DETAILS OF THE REPAIR / REPLACEMENT ARE ON FILE AT THIS AGENCY.  |                           |
| A     | TEL: 714-43<br>THE AIR<br>CCORDANCE<br>TO SERVICE<br>DATE: 02-<br>R Vacuum pum                   | 2-2275 AND 714<br>CRAFT AND / O<br>WITH CURREN<br>. PERTINENT<br>11-03<br>p with Overhauk   | 1-433-2276       CIRUS DESIGN AUTHORIZED REPAIR STATION         OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED AND INSPECTED IN         NT FEDERAL AIR REGULATIONS AND WAS FOUND AIRWORTHY FOR RETURN         DETAILS OF THE REPAIR / REPLACEMENT ARE ON FILE AT THIS AGENCY.  |                           |
| A     | N5288J<br>S/N: 421C-G<br>MODEL: 4<br>TACH TIM  | 2-2275 AND 714<br>CRAFT AND / O<br>WITH CURREN<br>. PERTINENT<br>11-03<br>p with Overhauk<br>p with Overhauk<br>0158<br>21C<br>[E: 1156.3   | I-433-2276       CIRUS DESIGN AUTHORIZED REPAIR STATION         OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED AND INSPECTED IN         NT FEDERAL AIR REGULATIONS AND WAS FOUND AIRWORTHY FOR RETURN         DETAILS OF THE REPAIR / REPLACEMENT ARE ON FIRE AT THIS AGENCY.  |                           |
| A     | N5288J<br>S/N: 421C-0<br>MODEL: 4<br>2980 AIR<br>2980 AIR<br>2980 AIR                            | 23-2275 AND 714<br>CRAFT AND / O<br>WITH CURREE<br>. PERTINENT<br>11-03<br>p with Overhauk<br>p with Overhauk<br>21C<br>E: 1156.3<br>WAY AV,. COS<br>14-433-2275 AN   | I-433-2276       CIRUS DESIGN AUTHORIZED REPAIR STATION         OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED AND INSPECTED IN         NT FEDERAL AIR REGULATIONS AND WAS FOUND AIRWORTHY FOR RETURN         DETAILS OF THE REPAIR / REPLACEMENT ARE ON FILE AT THIS AGENCY.         SIGNED:       Image: Im  |                           |
| A     | N5288J<br>S/N: 421C-0<br>MODEL: 4<br>2980 AIR<br>2980 AIR<br>72980 AIR<br>72980 AIR<br>72980 AIR | 2-2275 AND 714<br>CRAFT AND / O<br>WITH CURREL<br>. PERTINENT<br>11-03<br>p with Overhauk<br>p w | I-433-2276       CIRUS DESIGN AUTHORIZED REPAIR STATION         OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED AND INSPECTED IN       NT FEDERAL AIR REGULATIONS AND WAS FOUND AIRWORTHY FOR RETURN         DETAILS OF THE REPAIR / REPLACEMENT ARE ON FILE AT THIS AGENCY.       SIGNED:  | N                         |
| A     | N5288J<br>S/N: 421C-0<br>MODEL: 4<br>2980 AIR<br>2980 AIR<br>72980 AIR<br>72980 AIR<br>72980 AIR | 2-2275 AND 714<br>CRAFT AND / O<br>WITH CURREL<br>. PERTINENT<br>11-03<br>p with Overhauk<br>p w | I-433-2276       CIRUS DESIGN AUTHORIZED REPAIR STATION         OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED AND INSPECTED IN       NT FEDERAL AIR REGULATIONS AND WAS FOUND AIRWORTHY FOR RETURN         DETAILS OF THE REPAIR / REPLACEMENT ARE ON FILE AT THIS ACENCY.  | N                         |
| A     | N5288J<br>S/N: 421C-0<br>MODEL: 4<br>2980 AIR<br>2980 AIR<br>72980 AIR<br>72980 AIR<br>72980 AIR | 2-2275 AND 714<br>CRAFT AND / O<br>WITH CURREL<br>. PERTINENT<br>11-03<br>p with Overhauk<br>p w | I-433-2276       CIRUS DESIGN AUTHORIZED REPAIR STATION         OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED AND INSPECTED IN       NT FEDERAL AIR REGULATIONS AND WAS FOUND AIRWORTHY FOR RETURN         DETAILS OF THE REPAIR / REPLACEMENT ARE ON FILE AT THIS AGENCY.       SIGNED:  | N                         |
| A     | N5288J<br>S/N: 421C-C<br>MODEL: 4<br>TO SERVICE  | 2-2275 AND 714<br>CRAFT AND / O<br>WITH CURRENT<br>- PERTINENT<br>11-03<br>   | Image: Additional and the state of the |                           |

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|      | and here i                                |   | A subscription of the last second  |  |
|------|---|---|---|--|
| DATE | E TOTAL<br>TIME<br>IN<br>SERVICE          | TACH OR<br>RECORDING<br>METER<br>TIME   | SIGNATURE & CERTIFIC  | N OF WORK PERFORMED—<br>ATE NO. OF PERSON PERFORMING WORK  |
|      |   | TOTAL brought f   | forward from previous page  | 12124 Construction of the second s  |
|      |   |   | 10.   | A  |
| _    | N5288J                                    | 0   |   | DATE: 00 20 02   |
|      | SN: 421C-0158                             |   | -   | DATE: 09-29-03<br>W.O. #: 13205  |
|      | MODEL: 4210                               | 20  | the state of the state  | TOTAL TIME: 5290.4   |
|      | TACH TIME: 119                            | 2   | AIRFRAME LOG  |  |
|      | $\cup$                                    |   |   |  |
|      | TEL: 714-433                              | AV., COSTA MESA,<br>-2275 AND 714-433-  | -2276 CIRUS D   | PROVED REPAIR STATION # JVR016L<br>ESIGN AUTHORIZED REPAIR STATION   |
| 2    | THE AIRC<br>ACCORDANCE<br>TO SERVICE      | CRAFT AND / OR CO<br>WITH CURRENT F<br>, PERTINENT DET  | AILS OF THE BETAR / REPLAC  | WAS REPAIRED AND INSPECTED IN<br>D WAS FOUND AIRWORTHY FOR RETURN<br>EMENT ARE ON FILE AT THIS AGENCY.   |
|      | DATE <u>: 09-</u> 2                       | 29-0.3  | SIGNED: Jores 0   | husor  |
|      |   | IN SOME OF LODI   | Corona 121C Annual Inspection Gu  | 8  |
| Ē    | CONTRACT Income                           | tion of FIT as ner 91   | r Cessna 421C Annual Inspection Gui<br>-207(d) battery replacement due at 03  | -04.   |
| -    | C/W AD200-01-16                           | Exhaust Cracks, no c  | cracks found at this time, AD next due  | at 1292 Tach.  |
|      | R & R Nitrogen Bo                         | ottie for repair done by  | / S.J. Obrien.  |  |
|      | R&R RH brake lin                          | ings (RA66-66)<br>ium Filters P/N RAB3  | -5-1 END  |  |
| L    | Installed new vacu                        | ium riters ray RABS   | -5-1, 1, 0  |  |
|      |   |   |   | and the second of the  |
|      |   |   |   | the second se  |
| 102  |   |   |   |  |
|      | N5288J<br>S/N: 421C-0158                  |   |   | DATE: 08-11-04   |
|      | MODEL: 421C                               |   |   | W.O. #: 13949<br>TOTAL TIME: 5389.2  |
| 85   | HOBBS TIME: 1                             | 291.6   | TACPAFT MAINTENANS  | TOTAL MALL SCORE   |
|      |   |   | AIRFRAME LOG  |  |
|      | 2980 AIRWAY<br>TEL: 714-43                | AV., COSTA MESA<br>3-2275 AND 714-433   |   | PROVED REPAIR STATION # JVYR016L<br>DESIGN AUTHORIZED REPAIR STATION   |
|      | THE AIR                                   | CRAFT AND / OR C  | COMPONENT IDENTIFIED ABOV<br>FEDERAL AIR REGULATIONS A<br>TAILS OF THE REPAIR / REPLA   | E WAS REPAIRED AND INSPECTED IN<br>NO WAS FOUND AIRWORTHY FOR RETURN<br>CENTEXT ARE ON FILM AT THIS AGENCY.  |
| 100  | TO SERVIC                                 | E. PERTINENT DE   |   |  |
|      | TO SERVIC                                 | E. PERTINENT DE   | SIGNED: Aamen   | I Show you -   |
|      | TO SERVIC                                 | E. PERTINENT DE<br>3-11-04  | SIGNED:   | I Thomas your  |
|      | TO SERVIC<br>DATE: 08<br>R&R Right main t | E. PERTINENT DE<br>3-11-04<br>tire and tube with new.   | SIGNED:   | A Shor your  |
|      | TO SERVIC<br>DATE: 08<br>R&R Right main t | E. PERTINENT DE<br>3-11-04<br>ire and tube with new.<br>ased bearings. Ops ch<br>Hobbs  | SIGNED: Americanov<br>. P/N# 6.50-10-10FLCII<br>neck ok. END<br>5: 1304, 0  | A Shorsyaw   |
|      | TO SERVIC<br>DATE: 08<br>R&R Right main t | E. PERTINENT DE<br>3-11-04<br>ire and tube with new<br>ased bearings. Ops ch<br>Hobbs<br>CERTIFIED  | SIGNED: American<br>  | ATIC SYSTEMS   |
|      | TO SERVIC<br>DATE: 08<br>R&R Right main t | E. PERTINENT DE<br>3-11-04<br>ire and tube with new.<br>ased bearings. Ops ch<br>Hobbs<br>CERTIFIED<br>REQUIRED   | SIGNED:<br>   | A.R. PART 43,  |
|      | TO SERVIC<br>DATE: 08<br>R&R Right main t | E. PERTINENT DE<br>3-11-04<br>ire and tube with new<br>ased bearings. Ops ch<br>Hobbs<br>CERTIFIED<br>REQUIRED<br>APPENDIX  | SIGNED:<br>   | PERFORMED  |
|      | TO SERVIC<br>DATE: 08<br>R&R Right main t | E. PERTINENT DE<br>3-11-04<br>ire and tube with new.<br>ased bearings. Ops ch<br>Hobbs<br>CERTIFIED<br>REQUIRED<br>APPENDIX<br>A.T.C. TRA   | SIGNED:<br>   | PERFORMED IN TRED BY F.A.R.  |
|      | TO SERVIC<br>DATE: 08<br>R&R Right main t | E. PERTINENT DE<br>a-11-04<br>ire and tube with new.<br>ased bearings. Ops ch<br>Hobbs<br>CERTIFIED<br>REQUIRED<br>APPENDIX<br>A.T.C. TRA<br>91.413 AC                                    | SIGNED:<br>P/N# 6.50-10-HOFLCII<br>ALTIMETER(S) AND ST/<br>BY F.A.R. 91.411 I.A.W. F<br>E TO <u>30,000</u> FEET<br>NSPONDER TESTS REQU<br>COMPLISHED I.A.W. F   | A.R. PART 43,<br>PERFORMED<br>IRED BY F.A.R.<br>A.R. PART 43,  |
|      | TO SERVIC<br>DATE: 08<br>R&R Right main t | E. PERTINENT DE<br>3-11-04<br>irre and tube with new.<br>ased bearings. Ops ch<br>Hobbs<br>CERTIFIED<br>REQUIRED<br>APPENDIX<br>A.T.C. TRA<br>91.413 AC<br>APPENDIX                       | SIGNED:<br>P/N# 6.50-105-00FLCII<br>ALTIMETER(S) AND STA<br>BY F.A.R. 91.411 I.A.W. F<br>E TO <u>30,000</u> FEET<br>NSPONDER TESTS REQU<br>COMPLISHED I.A.W. F.<br>F.<br>W.O.# 44   | A.R. PART 43,<br>PERFORMED<br>IRED BY F.A.R.<br>A.R. PART 43,<br>A.R. PART 43,<br>A.R. PART 43,  |
|      | TO SERVIC<br>DATE: 08<br>R&R Right main t | E. PERTINENT DE<br>a-11-04<br>ire and tube with new.<br>ased bearings. Ops ch<br><i>Hobbs</i><br>CERTIFIED<br>REQUIRED<br>APPENDIX<br>A.T.C. TRA<br>91.413 AC<br>APPENDIX<br>LT. ALT. SN_ | SIGNED:<br>P/N# 6.50-10510FLCII<br>Neck ok. END<br>ALTIMETER(S) AND STA<br>BY F.A.R. 91.411 I.A.W. F<br>E TO 30 000 FEET<br>NSPONDER TESTS REQU<br>CCOMPLISHED I.A.W. F.<br>F. W.O.# 4<br>SALSTAL #1XPDR SM   | A.R. PART 43,<br>PERFORMED<br>IRED BY F.A.R.<br>A.R. PART 43,<br>A.R. PART 43,<br>A.R. PART 43,<br>A.R. PART 43,   |
|      | TO SERVIC<br>DATE: 08<br>R&R Right main t | E. PERTINENT DE<br>irre and tube with new.<br>ased bearings. Ops ch<br>Hobbs<br>CERTIFIED<br>REQUIRED<br>APPENDIX<br>A.T.C. TRA<br>91.413 AC<br>APPENDIX<br>LT. ALT. SN<br>RT. ALT. SN    | SIGNED:<br>P/N# 6.50-10510FLCII<br>neck ok. END<br>ALTIMETER(S) AND ST/<br>BY F.A.R. 91.411 I.A.W. F<br>E TO <u>30 000</u> FEET<br>INSPONDER TESTS REQU<br>CCOMPLISHED I.A.W. F.<br>F. W.O.# 4<br>XISPON #1XPDR SN<br>XISPON #1XPDR SN<br>XISPON SI | A.R. PART 43,<br>PERFORMED<br>IRED BY F.A.R.<br>A.R. PART 43,<br>A.R. PART 43,<br>A.R. PART 43,<br>A.R. PART 43,   |
|      | TO SERVIC<br>DATE: 08<br>R&R Right main t | E. PERTINENT DE<br>irre and tube with new.<br>ased bearings. Ops ch<br>Hobbs<br>CERTIFIED<br>REQUIRED<br>APPENDIX<br>A.T.C. TRA<br>91.413 AC<br>APPENDIX<br>LT. ALT. SN<br>RT. ALT. SN    | SIGNED:<br>P/N# 6.50-10510FLCII<br>neck ok. END<br>ALTIMETER(S) AND ST/<br>BY F.A.R. 91.411 I.A.W. F<br>E TO <u>30,000</u> FEET<br>INSPONDER TESTS REQU<br>COMPLISHED I.A.W. F.<br>W.O.# 4<br>ST + ST + 122PDR SI<br>DED FAILS SN                   | A.R. PART 43,<br>PERFORMED<br>IRED BY F.A.R.<br>A.R. PART 43,<br>A.R. PART 44,<br>A.R. PART |
|      | TO SERVIC<br>DATE: 08<br>R&R Right main t | E. PERTINENT DE<br>irre and tube with new.<br>ased bearings. Ops ch<br>Hobbs<br>CERTIFIED<br>REQUIRED<br>APPENDIX<br>A.T.C. TRA<br>91.413 AC<br>APPENDIX<br>LT. ALT. SN<br>RT. ALT. SN    | SIGNED:<br>P/N# 6.50-10510FLCII<br>neck ok. END<br>ALTIMETER(S) AND ST/<br>BY F.A.R. 91.411 I.A.W. F<br>E TO <u>30,000</u> FEET<br>INSPONDER TESTS REQU<br>COMPLISHED I.A.W. F.<br>W.O.# 4<br>ST + ST + 122PDR SI<br>DED FAILS SN                   | A.R. PART 43,<br>PERFORMED<br>IRED BY F.A.R.<br>A.R. PART 43,<br>A.R. PART 43,<br>A.R. PART 43,<br>A.R. PART 43,   |

TOTAL—Carry forward to next page

Page No. TOTAL TACH OR TIME DATE RECORDING DESCRIPTION OF WORK PERFORMED-IN SERVICE METER SIGNATURE & CERTIFICATE NO. OF PERSON PERFORMING WORK TIME TOTAL brought forward from previous page N5288J S/N: 421C-0 58 2 MODEL: 42 C TACH TIME: 1192.8 DATE: 10-05-04 W.O. #: 14014 TOTAL TIME: 5401.6 AIRFRAME LOG 2980 AIRWAY AV., COSTA MESA, CA 92626 TEL: 714-433-2275 AND 714-433-2276 FAA APPROVED REPAIR STATION # JVYR016L CIRRUS DESIGN AUTHORIZED REPAIR STATION THE AIRCRAFT AND / OR COMPONENT IDENTIFIED ABOVE WAS RETAIRED AND INSPECTED IN ACCORDANCE WITH CURRENT FEDERAL AIR REGULATIONS AND WAS FOUND AIRWORTHY FOR RETURN TO SERVICE - PERTINENT DETAILS OF THE REGULATIONS AND WAS FOUND AIRWORTHY FOR RETURN TO SERVICE. PERTINENT DETAILS OF THE REPAIR / REPLACE MENT ARE ON FILE AT THIS AGENCY. DATE: SIGNED: Gema han x 20 C/W an ANNUAL INSPECTION as per Cessna 421C Annual Inspection Guide. C/W Yearly Inspection of ELT as per 91-207(d) R&R Batteries. C/W AD 2060-01-16 Exhaust Inspection, as per AD. No cracks found. p R&R #1 JPI Probe with new. R&R RH and LH brake linings (RA66-66) R&R Vacuum filter with new. P/N# B351 R&R Taxi light bulb with new. P/N# GE4594. Installed new gears in right and left landing light. P/N# A6580, A6582 R&R Left main tire and tube with new \_P/N# C650-10-8TT, M650-10. Cowl light installed as per STC SA4270SW. Ground run up inspection performed, ops check ok. END 2 N5288J DATE: 12-15-04 S/N: 421C-0158 W.O. #: 14180 MODEL: 421C TOTAL TIME: 5513.6 HOBBS TIME: 1304.8 AIRFRAME LOG FAA APPROVED REPAIR STATION # JVYR016L 2980 AIRWAY AV., COSTA MESA, CA 92626 CIRRUS DESIGN AUTHORIZED REPAIR STATION TEL: 714-433-2275 AND 714-433-2276 THE AIRCRAFT AND / OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED AND INSPECTED IN ACCORDANCE WITH CURRENT FEDERAL AIR REGULATIONS AND WASYOUND AIRWORTHY FOR RETURN TO SERVICE. PERTINENT DETAILS OF THE REPAIR / REPLACEMENT ARE ON FILE AT THIS AGENCY. 0-Alla amaly SIGNED: 12-15-04 DATE: R&R Fuel Drain Valve with new. P/N# 50C4. Ground run up inspection performed, ops check ok. END SUB-TOTAL this page TOTAL-Carry forward to next page

N5288J S/N: 421C-0158 MODEL: 421C HOBBS TIME: 1327.7



DATE: 04-06-05 W.O. #: 14386 TOTAL TIME: 5536.5

### 2980 AIRWAY AV,. COSTA MESA, CA 92626 TEL: 714-433-2275

### FAA APPROVED REPAIR STATION # JVYR016L

THE AIRCRAFT AND / OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED AND INSPECTED IN ACCORDANCE WITH CURRENT FEDERAL AIR REGULATIONS AND WAS FOUND AIRWORTHY FOR RETURN TO SERVICE. PERTINENT DETAILS OF THE REPAIR / REPLACEMENT ARE ON FILE AT THIS AGENCY.

DATE: 04-06-05

SIGNED:\_\_\_\_

InstalledNew Light Bulb. P/N 4594 InstalledNew Light Bulb GE. P/N 305 Ground un up inspection performed, ops check ok. END

N5288J S/N: 421C-0158 MODEL: 421C HOBBS TIME: 1328.3



DATE: 04-26-05 W.O. #: 14420 TOTAL TIME: 5537.1

AIRFRAME LOG

2980 AIRWAY AV,. COSTA MESA, CA 92626 TEL: 714-433-2275

### FAA APPROVED REPAIR STATION # JVYR016L

THE AIRCRAFT AND / OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED AND INSPECTED IN ACCORDANCE WITH CURRENT FEDERAL AIR REGULATIONS AND WAS FOUND AIRWORTHY FOR RETURN TO SERVICE. PERTINENT DETAILS OF THE REPAIR / REPLACEMENT ARE ON FILE AT THIS AGENCY.

DATE: 04-26-05

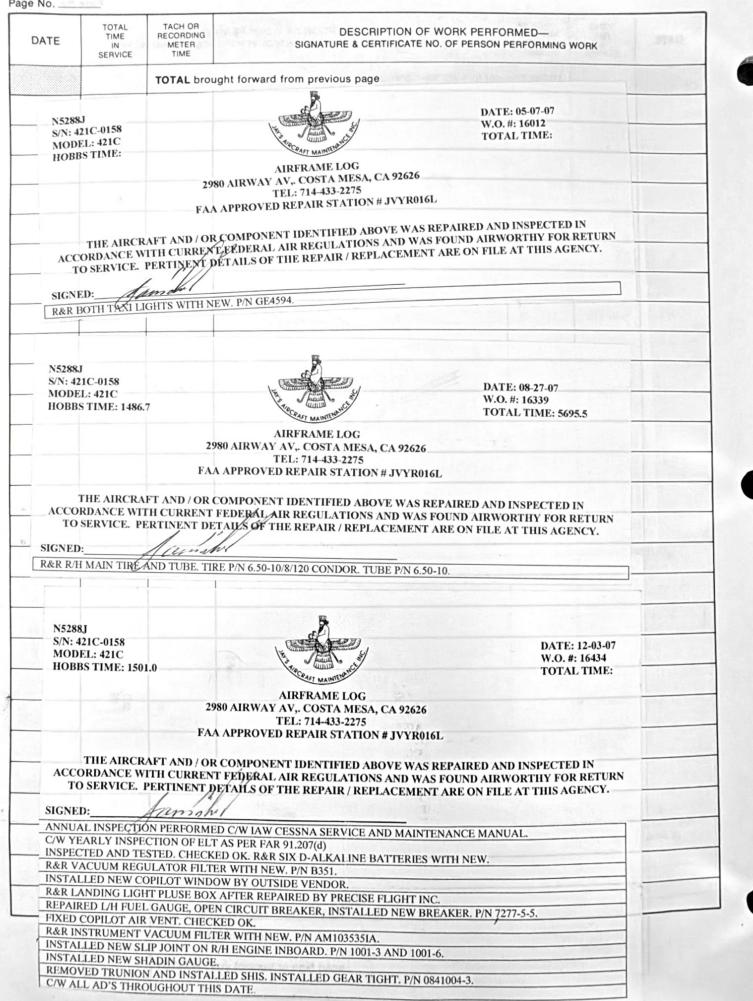
\_\_\_SIGNED:\_\_

Installed STC SA 4005NM KIT per Pulselite Installation Manual Ground run up inspection performed, ops check ok. END

| DATE     | TOTAL<br>TIME<br>IN<br>SERVICE | TACH OR<br>RECORDING<br>METER<br>TIME  |   | DESCRIF<br>GNATURE & CERT                      | PTION OF WOR                                    | K PERFORMED—<br>PERSON PERFORM                 | ING WORK                                    | - 47.80                        |
|----------|--------------------------------|--|---|--|---|--|---|--------------------------------|
|          |                                | TOTAL bro                              | ught forward from                                     | previous page                                  |   |  |   |                                |
|          |                                |  |   | cf.  | B   |  |   |                                |
|          | N528<br>S/N:                   | 421C-0158                              |   | 1-   | S.  |  | DATE: 05-03-05<br>W.O. #: 14435             |                                |
|          | MOL                            | DEL: 421C                              | 0.3   | 13.5.10  | Cuntin Car                                      |  | TOTAL TIME:                                 | 5539.1                         |
|          | HOB                            | BS TIME: 133                           | 0.5   | CRAFT  | MAINTEN   |  |   |                                |
|          |                                |  |   | AIRFRAM  | ,, COSTA MESA                                   | CA 92626                                       |   |                                |
|          |                                |  |   | TEL  | : 714-433-2275                                  |  |   |                                |
|          |                                |  | FAA   | APPROVED R                                     | EPAIR STATION                                   | N # JVYR016L                                   | 1.  |                                |
| 5        | ACC                            | THE AIRCH<br>CORDANCE W<br>TO SERVICE. | RAFT AND / OR CO<br>VITH CURRENT FI<br>PERTINENT DET/ | MPONENT IDE<br>EDERAL AIR RI<br>AILS OF THE RI | INTIFIED ABOV<br>EGULATIONS A<br>EPAIR / REPLAC | E WAS REPAIRE<br>ND WAS FOUND<br>CEMENT AME OF | AND INSPECT<br>ARWORTHY F<br>FILE AT THIS   | TED IN<br>OR RETURN<br>AGENCY. |
| )        |                                | TE <u>: 05-0</u>                       |   | SIGNED:  | Au  | min  |   |                                |
|          |                                |  |   | avad Vinke                                     |   |  |   |                                |
| 4        | Re-ad                          | justed Turbo In                        | let Temp. Probe, Rem                                  | loved KINKS.                                   |   |  |   |                                |
|          | 1                              | 1                                      |   |  | *   |  |   |                                |
|          | N5288                          | 3J<br>121C-0158                        |   |  |   |  | DATE: 10-10-05                              | 5                              |
|          |                                | EL: 421C                               |   | The of   |   |  | W.O. #: 14766                               | 5502 2                         |
|          |                                | BS TIME: 138-                          | 1.5   | AIRCRAFT                                       | MAINTENANC                                      |  | TOTAL TIME:                                 | 5593.3                         |
|          |                                |  |   | AIRFRAM  | E LOG   |  |   |                                |
|          | 1                              |  | 29  |  | COSTA MESA<br>: 714-433-2275                    | , CA 92626                                     |   |                                |
|          | 24                             |  | FAA   | APPROVED RE                                    | EPAIR STATION                                   | # JVYR016L                                     |   |                                |
|          |                                | THE AIRCR<br>CORDANCE W<br>O SERVICE.  | AFT AND / OR CO<br>ITH CURRENT FE<br>PERTINENT DETA   | MPONENT IDE<br>DERAL AIR RE<br>AILS OF THE RE  | NTIFIED ABOV<br>CGULATIONS AN<br>EPAIR / REPLAC | E WAS REPAIR<br>ND WAS FOUND<br>CEMENY ARE ON  | D AND INSPECT<br>ARWORTHY F<br>FILE AT THIS | TED IN<br>OR RETURN<br>AGENCY. |
|          | DA                             | TE <u>: 10-10</u>                      | 0-05  | SIGNED:  | Aa  | maky   |   |                                |
| 2 Mars   |                                | (0) (                                  |   | with   | 11  |  | 24  |                                |
|          | R&R                            | LIH Landing Ligh                       | ght Bulb. P/N 1524 b<br>at with new. P/N 4594         | 4 lamp   | 0   |  |   |                                |
| 1        |                                |  |   |  | -   |  |   |                                |
| <u>.</u> | N528                           | 8J                                     |   | da.  | A THE STATE                                     |  | DATE: 01-18-0                               | 6                              |
|          |                                | 421C-0158                              |   | 12   | Y   |  | W.O. #: 14897<br>TOTAL TIME                 |                                |
|          |                                | EL: 421C<br>BS TIME: 140               | 0.6 .   | S ALACRAFT                                     | MAINTENAME                                      |  | TOTAL IIME.                                 |                                |
|          |                                | " Acq                                  |   | AIRFRAM  | IE LOG  |  | 1   | 1 2                            |
|          |                                |  | 25  | 980 AIRWAY AV<br>TEL                           | . COSTA MESA                                    | A, CA 92626                                    |   |                                |
| 1.1.1.1  |                                |  |   | APPROVED R                                     | EPAIR STATIO                                    | N # JVYR016L                                   | 19.2  |                                |
| 1        |                                |  |   |  |   |  | IND INSPEC                                  | TED IN                         |
| 1        |                                | THE AIRCI<br>CORDANCE V<br>TO SERVICE. | RAFT AND / OR CO<br>VITH CURRENT FI<br>PERTINENT DET/ | OMPONENT IDE<br>EDERAL AIR RI<br>AILS OF THE R | ENTIFIED ABOV<br>EGULATIONS A<br>EPAIR / REPLA  | ND WAS YOUND<br>CENTRY ARE OF                  | AIRWORTHY F<br>N FILE AT THIS               | OR RETURN<br>AGENCY.           |
|          |                                |  | 8-06  | SIGNED:  | for   | where the                                      | 3   |                                |
|          | Instal                         | led New Gill A                         | cid Lead Battery. P/N                                 | 11-01680 MODI                                  | G-246   | 4  |   |                                |
|          |                                |  | L. this page  |  | - Alf   |  |   |                                |
|          | Г                              |  | arry forward to ne                                    |  |   |  |   |                                |
|          |                                | I TAL-C                                | any lorward to ne                                     | xt page  |   |  |   |                                |

| ATE |   | TACH OR   |   |  |  |  |  |  |  |
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|     | TOTAL<br>TIME<br>IN<br>SERVICE  | RECORDING<br>METER<br>TIME  | DESCRIPTION OF WORK PERFORM<br>SIGNATURE & CERTIFICATE NO. OF PERSON PERI   | IED-<br>FORMING WORK   |  |  |  |  |  |
|     |   | TOTAL brough  | t forward from previous page  | LATOT  |  |  |  |  |  |
|     |   |   |   |  |  |  |  |  |  |
|     | N5288J<br>S/N: 421C   |   |   |  |  |  |  |  |  |
|     | MODEL:  | -0158<br>421C   |   | DATE: 11-07-06   |  |  |  |  |  |
|     | HOBBS TI  | ME: 1429.1  |   | W.O. #: 15432  |  |  |  |  |  |
|     |   |   |   | TOTAL TIME: 5637.9   |  |  |  |  |  |
|     |   |   | AIRFRAME LOG  | AND -  |  |  |  |  |  |
|     |   |   |   |  |  |  |  |  |  |
| 1   |   |   | 2980 AIRWAY AV,. COSTA MESA, CA 92626<br>TEL: 714-433-2275  |  |  |  |  |  |  |
|     |   |   |   | 1  |  |  |  |  |  |
| 100 |   |   | FAA APPROVED REPAIR STATION # JVYR016L  | 1  |  |  |  |  |  |
|     | THE   | AIRCRAFT AND  | OR COMPONENT IDENTIFICATION OF  |  |  |  |  |  |  |
|     | ACCORDA<br>TO SER   | NCE WITH CUR<br>VICE. PERTINE   | ) / OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED A<br>RENT FEDERAL AIR REGULATIONS AND WAS FOUND AIR<br>NT DETAILS OF THE REPAIR / REPLACEMENT ARE ON FI  | ND INSPECTED IN<br>RWORTHY FOR RETURN<br>LE AT THIS AGENCY.  |  |  |  |  |  |
|     | DATE <u>:</u>   |   | SIGNED:   |  |  |  |  |  |  |
|     | C/W an ANNI   | AL INSPECTION   | N in second se |  |  |  |  |  |  |
| 14  |   | · and Cooning Sint  | N, in accordance with Cessna 421C Service and Maintenance Manual.<br>pud as per STC SA3721SW. P/N 1221-1  |  |  |  |  |  |  |
|     | - C/W Tearly In   | spection of ELT as  | per FAR 91.207 (d) R&R Batteries with name size "D"   |  |  |  |  |  |  |
|     | Thoi Static Ira   | insponder Test Perfo  | ormed by: Western Avionics  |  |  |  |  |  |  |
|     | C/W AD95-09   | -13 Fuel Float Valve  | e Functional Test, by Inspection.   |  |  |  |  |  |  |
|     | Elbow Exhaust   | P/N K9910300-01   | ection, in accordance with Cessna SB MEB 99-6-9-12. Replaced L/H  | Tail Pipe Overboard and  |  |  |  |  |  |
|     | C/W AD2005-2  | 20-26 Avionics Bus  | Switch Inspection/ Replacement Circuit Breaker in accordance with   | Cesena ME.S.B. MED.05.1  |  |  |  |  |  |
|     | Rock Cheult D   | caker Switch with r   | new.  | Cessia ME-3.D. MED 03-1,   |  |  |  |  |  |
| H   | C/w with all A  | D's Throughout 200  | 06-22 AD Compliance Record.   |  |  |  |  |  |  |
| — F | Installed Out   | R&R R/H and L/H Brake Linings with new, P/N RA66-66   |   |  |  |  |  |  |  |
|     | Installed Overhauled L/H wing Locker Fuel Cell By: Aero Tech, under W.O.# 23484-2. P/N 99-10241, S/N 76-3597<br>R&R Battery with new. P/N G-246   |   |   |  |  |  |  |  |  |
| E   | R&R Battery w   | th new. P/N G-246   |   | 8/N 76-3597  |  |  |  |  |  |
|     | R&R Battery will<br>R&R Fuel Line   | th new. P/N G-246<br>Divider Line with n  | ew. P/N 5100108-73  | S/N 76-3597  |  |  |  |  |  |
|     | R&R Battery w<br>R&R Fuel Line<br>R&R Hydraulic   | th new. P/N G-246<br>Divider Line with n<br>Pump Gasket with 1  | ew. P/N 5100108-73<br>new. P/N 5154056-2  | 8/N 76-3597  |  |  |  |  |  |
|     | R&R Battery w<br>R&R Fuel Line<br>R&R Hydraulic   | th new. P/N G-246<br>Divider Line with n<br>Pump Gasket with 1  | ew. P/N 5100108-73  | S/N 76-3597  |  |  |  |  |  |
|     | R&R Battery w<br>R&R Fuel Line<br>R&R Hydraulic<br>R&R Alternator   | th new. P/N G-246<br>Divider Line with n<br>Pump Gasket with 1  | ew. P/N 5100108-73<br>new. P/N 5154056-2  | S/N 76-3597  |  |  |  |  |  |
|     | R&R Battery w<br>R&R Fuel Line<br>R&R Hydraulic<br>R&R Alternator   | th new. P/N G-246<br>Divider Line with n<br>Pump Gasket with 1<br>Following after Out   | ew. P/N 5100108-73<br>new. P/N 5154056-2<br>tside repair, performed by: Aero Accessories.   | S/N 76-3597<br>DATE: 01-19-07  |  |  |  |  |  |
|     | R&R Battery w<br>R&R Fuel Line<br>R&R Hydraulic<br>R&R Alternator<br>N5288J<br>S/N: 421C-01   | th new. P/N G-246<br>Divider Line with n<br>Pump Gasket with n<br>Following after Out   | ew. P/N 5100108-73<br>new. P/N 5154056-2<br>tside repair, performed by: Aero Accessories.   | DATE: 01-19-07<br>V.O. #: 15768  |  |  |  |  |  |
|     | R&R Battery w<br>R&R Fuel Line<br>R&R Hydraulic<br>R&R Alternator   | th new. P/N G-246<br>Divider Line with n<br>Pump Gasket with n<br>Following after Out   | ew. P/N 5100108-73<br>new. P/N 5154056-2<br>tside repair, performed by: Aero Accessories.   | DATE: 01-19-07   |  |  |  |  |  |
|     | R&R Battery w<br>R&R Fuel Line<br>R&R Hydraulic<br>R&R Alternator<br>N5288J<br>S/N: 421C-01<br>MODEL: 42  | th new. P/N G-246<br>Divider Line with n<br>Pump Gasket with n<br>Following after Out   | ew. P/N 5100108-73<br>new. P/N 5154056-2<br>tside repair, performed by: Aero Accessories.   | DATE: 01-19-07<br>V.O. #: 15768  |  |  |  |  |  |
|     | R&R Battery w<br>R&R Fuel Line<br>R&R Hydraulic<br>R&R Alternator<br>N5288J<br>S/N: 421C-01<br>MODEL: 42  | th new. P/N G-246<br>Divider Line with n<br>Pump Gasket with n<br>Following after Out   | ew. P/N 5100108-73<br>new. P/N 5154056-2<br>tside repair, performed by: Aero Accessories.   | DATE: 01-19-07<br>V.O. #: 15768  |  |  |  |  |  |
|     | R&R Battery w<br>R&R Fuel Line<br>R&R Hydraulic<br>R&R Alternator<br>N5288J<br>S/N: 421C-01<br>MODEL: 42  | th new. P/N G-246<br>Divider Line with n<br>Pump Gasket with n<br>Following after Out   | ew. P/N 5100108-73<br>new. P/N 5154056-2<br>tside repair, performed by: Aero Accessories.   | DATE: 01-19-07<br>V.O. #: 15768  |  |  |  |  |  |
|     | R&R Battery w<br>R&R Fuel Line<br>R&R Hydraulic<br>R&R Alternator<br>N5288J<br>S/N: 421C-01<br>MODEL: 42<br>HOBBS TIM   | th new. P/N G-246<br>Divider Line with n<br>Pump Gasket with n<br>Following after Out   | ew. P/N 5100108-73<br>new. P/N 5154056-2<br>tside repair, performed by: Aero Accessories.   | DATE: 01-19-07<br>V.O. #: 15768<br>TOTAL TIME: 5651.2  |  |  |  |  |  |
|     | R&R Battery w<br>R&R Fuel Line<br>R&R Hydraulic<br>R&R Alternator<br>N5288J<br>S/N: 421C-01<br>MODEL: 42<br>HOBBS TIM   | th new. P/N G-246<br>Divider Line with n<br>Pump Gasket with n<br>Following after Out   | ew. P/N 5100108-73<br>new. P/N 5154056-2<br>tside repair, performed by: Aero Accessories.   | DATE: 01-19-07<br>N.O. #: 15768<br>FOTAL TIME: 5651.2  |  |  |  |  |  |
|     | R&R Battery w<br>R&R Fuel Line<br>R&R Hydraulic<br>R&R Alternator<br>N5288J<br>S/N: 421C-01<br>MODEL: 42<br>HOBBS TIM   | th new. P/N G-246<br>Divider Line with n<br>Pump Gasket with n<br>Following after Out<br>58<br>1C<br>E: 1442.4  | ew. P/N 5100108-73<br>new. P/N 5154056-2<br>tside repair, performed by: Aero Accessories.<br>AirFRAME LOG<br>2980 AIRWAY AV., COSTA MESA, CA 92626<br>TEL: 714-433-2275<br>FAA APPROVED REPAIR STATION # JVYR016L<br>OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED A   | DATE: 01-19-07<br>N.O. #: 15768<br>FOTAL TIME: 5651.2<br>ND INSPECTED IN   |  |  |  |  |  |
|     | R&R Battery w<br>R&R Fuel Line<br>R&R Hydraulic<br>R&R Alternator<br>N5288J<br>S/N: 421C-01<br>MODEL: 42<br>HOBBS TIM   | th new. P/N G-246<br>Divider Line with n<br>Pump Gasket with n<br>Following after Out<br>58<br>1C<br>E: 1442.4  | ew. P/N 5100108-73<br>new. P/N 5154056-2<br>tside repair, performed by: Aero Accessories.<br>AirFRAME LOG<br>2980 AIRWAY AV., COSTA MESA, CA 92626<br>TEL: 714-433-2275<br>FAA APPROVED REPAIR STATION # JVYR016L<br>OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED A   | DATE: 01-19-07<br>N.O. #: 15768<br>FOTAL TIME: 5651.2<br>ND INSPECTED IN   |  |  |  |  |  |
|     | R&R Battery w<br>R&R Fuel Line<br>R&R Hydraulic<br>R&R Alternator<br>S/N: 421C-01<br>MODEL: 42<br>HOBBS TIM<br>THE A<br>ACCORDAN<br>TO SERV   | th new. P/N G-246<br>Divider Line with n<br>Pump Gasket with n<br>Following after Out<br>58<br>1C<br>E: 1442.4<br>AIRCRAFT AND /<br>NCE WITH CURR<br>TCE. PERTINEN  | ew. P/N 5100108-73<br>new. P/N 5154056-2<br>tside repair, performed by: Aero Accessories.   | DATE: 01-19-07<br>N.O. #: 15768<br>FOTAL TIME: 5651.2<br>ND INSPECTED IN   |  |  |  |  |  |
|     | R&R Battery wi<br>R&R Fuel Line<br>R&R Hydraulic<br>R&R Alternator<br>N5288J<br>S/N: 421C-01<br>MODEL: 42<br>HOBBS TIM<br>THE A<br>ACCORDAN<br>TO SERV<br>DATE:                             | th new. P/N G-246<br>Divider Line with n<br>Pump Gasket with n<br>Following after Out<br>158<br>IC<br>E: 1442.4<br>AIRCRAFT AND /<br>CE WITH CURR<br>ICE. PERTINEN<br>01-19-07                                | ew. P/N 5100108-73<br>new. P/N 5154056-2<br>tside repair, performed by: Aero Accessories.<br>Aireframe LOG<br>2980 AIRWAY AV, COSTA MESA, CA 92626<br>TEL: 714-433-2275<br>FAA APPROVED REPAIR STATION # JVYR016L<br>OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED A<br>EENT FEDERAL AIR REGULATIONS AND WAS FOUND AIR<br>T DETAILS OF THE REPAIR / REPLACEMENT ARE ON FI<br>SIGNED:   | DATE: 01-19-07<br>N.O. #: 15768<br>FOTAL TIME: 5651.2<br>ND INSPECTED IN   |  |  |  |  |  |
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|     | R&R Battery w<br>R&R Fuel Line<br>R&R Hydraulic<br>R&R Alternator<br>N5288J<br>S/N: 421C-01<br>MODEL: 42<br>HOBBS TIM<br>THE A<br>ACCORDAN<br>TO SERV<br>DATE:<br>Installed New S           | th new. P/N G-246<br>Divider Line with n<br>Pump Gasket with n<br>Following after Out<br>158<br>1C<br>E: 1442.4<br>AIRCRAFT AND /<br>CE WITH CURR<br>TCE. PERTINEN<br>01-19-07<br>eals on R/H Wasteg          | ew. P/N 5100108-73<br>new. P/N 5154056-2<br>tside repair, performed by: Aero Accessories.   | DATE: 01-19-07<br>N.O. #: 15768<br>FOTAL TIME: 5651.2<br>ND INSPECTED IN   |  |  |  |  |  |
|     | R&R Battery wi<br>R&R Fuel Line<br>R&R Hydraulic<br>R&R Alternator<br>N5288J<br>S/N: 421C-01<br>MODEL: 42<br>HOBBS TIM<br>THE A<br>ACCORDAN<br>TO SERV<br>DATE:                             | th new. P/N G-246<br>Divider Line with n<br>Pump Gasket with n<br>Following after Out<br>158<br>1C<br>E: 1442.4<br>AIRCRAFT AND /<br>CE WITH CURR<br>TCE. PERTINEN<br>01-19-07<br>eals on R/H Wasteg          | ew. P/N 5100108-73<br>new. P/N 5154056-2<br>tside repair, performed by: Aero Accessories.<br>Aireframe LOG<br>2980 AIRWAY AV, COSTA MESA, CA 92626<br>TEL: 714-433-2275<br>FAA APPROVED REPAIR STATION # JVYR016L<br>OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED A<br>EENT FEDERAL AIR REGULATIONS AND WAS FOUND AIR<br>T DETAILS OF THE REPAIR / REPLACEMENT ARE ON FI<br>SIGNED:   | DATE: 01-19-07<br>N.O. #: 15768<br>FOTAL TIME: 5651.2<br>ND INSPECTED IN   |  |  |  |  |  |
|     | R&R Battery w<br>R&R Fuel Line<br>R&R Hydraulic<br>R&R Alternator<br>N5288J<br>S/N: 421C-01<br>MODEL: 42<br>HOBBS TIM<br>THE A<br>ACCORDAN<br>TO SERV<br>DATE:<br>Installed New S           | th new. P/N G-246<br>Divider Line with n<br>Pump Gasket with n<br>Following after Out<br>158<br>1C<br>E: 1442.4<br>AIRCRAFT AND /<br>CE WITH CURR<br>TCE. PERTINEN<br>01-19-07<br>eals on R/H Wasteg          | ew. P/N 5100108-73<br>new. P/N 5154056-2<br>tside repair, performed by: Aero Accessories.<br>Aireframe LOG<br>2980 AIRWAY AV, COSTA MESA, CA 92626<br>TEL: 714-433-2275<br>FAA APPROVED REPAIR STATION # JVYR016L<br>OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED A<br>EENT FEDERAL AIR REGULATIONS AND WAS FOUND AIR<br>T DETAILS OF THE REPAIR / REPLACEMENT ARE ON FI<br>SIGNED:   | DATE: 01-19-07<br>N.O. #: 15768<br>FOTAL TIME: 5651.2<br>ND INSPECTED IN   |  |  |  |  |  |
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|     | R&R Battery w<br>R&R Fuel Line<br>R&R Hydraulic<br>R&R Alternator<br>N5288J<br>S/N: 421C-01<br>MODEL: 42<br>HOBBS TIM<br>THE A<br>ACCORDAN<br>TO SERV<br>DATE:<br>Installed New S           | th new. P/N G-246<br>Divider Line with n<br>Pump Gasket with n<br>Following after Out<br>158<br>1C<br>E: 1442.4<br>AIRCRAFT AND /<br>CE WITH CURR<br>TCE. PERTINEN<br>01-19-07<br>eals on R/H Wasteg          | ew. P/N 5100108-73<br>new. P/N 5154056-2<br>tside repair, performed by: Aero Accessories.<br>Aireframe LOG<br>2980 AIRWAY AV, COSTA MESA, CA 92626<br>TEL: 714-433-2275<br>FAA APPROVED REPAIR STATION # JVYR016L<br>OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED A<br>EENT FEDERAL AIR REGULATIONS AND WAS FOUND AIR<br>T DETAILS OF THE REPAIR / REPLACEMENT ARE ON FI<br>SIGNED:   | DATE: 01-19-07<br>N.O. #: 15768<br>FOTAL TIME: 5651.2<br>ND INSPECTED IN   |  |  |  |  |  |
|     | R&R Battery w<br>R&R Fuel Line<br>R&R Hydraulic<br>R&R Alternator<br>S/N: 421C-01<br>MODEL: 42<br>HOBBS TIM<br>THE A<br>ACCORDAN<br>TO SERV<br>DATE:<br>Installed New S<br>Installed Sun Vi | th new. P/N G-246<br>Divider Line with n<br>Pump Gasket with n<br>Following after Out<br>158<br>1C<br>E: 1442.4<br>AIRCRAFT AND /<br>ICE WITH CURR<br>TCE. PERTINEN<br>01-19-07<br>eals on R/H Wasteg<br>sor. | ew. P/N 5100108-73<br>new. P/N 5154056-2<br>tside repair, performed by: Aero Accessories.<br>AIRFRAME LOG<br>2980 AIRWAY AV, COSTA MESA, CA 92626<br>TEL: 714-433-2275<br>FAA APPROVED REPAIR STATION # JVYR016L<br>OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED A<br>ENT FEDERAL AIR REGULATIONS AND WAS FOUND AFF<br>T DETAILS OF THE REPAIR / REPLACEMENT ARE ON FI<br>  | DATE: 01-19-07<br>N.O. #: 15768<br>FOTAL TIME: 5651.2<br>ND INSPECTED IN<br>RWORTHY FOR RETURN<br>LE AT THIS AGENCY. |  |  |  |  |  |
|     | R&R Battery w<br>R&R Fuel Line<br>R&R Hydraulic<br>R&R Alternator<br>S/N: 421C-01<br>MODEL: 42<br>HOBBS TIM<br>THE A<br>ACCORDAN<br>TO SERV<br>DATE:<br>Installed New S<br>Installed Sun Vi | th new. P/N G-246<br>Divider Line with n<br>Pump Gasket with n<br>Following after Out<br>158<br>1C<br>E: 1442.4<br>AIRCRAFT AND /<br>CE WITH CURR<br>TCE. PERTINEN<br>01-19-07<br>eals on R/H Wasteg          | ew. P/N 5100108-73<br>new. P/N 5154056-2<br>tside repair, performed by: Aero Accessories.<br>AIRFRAME LOG<br>2980 AIRWAY AV, COSTA MESA, CA 92626<br>TEL: 714-433-2275<br>FAA APPROVED REPAIR STATION # JVYR016L<br>OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED A<br>ENT FEDERAL AIR REGULATIONS AND WAS FOUND AFF<br>T DETAILS OF THE REPAIR / REPLACEMENT ARE ON FI<br>  | DATE: 01-19-07<br>N.O. #: 15768<br>FOTAL TIME: 5651.2<br>ND INSPECTED IN<br>RWORTHY FOR RETURN<br>LE AT THIS AGENCY. |  |  |  |  |  |

Page No.\_



Page No. TOTAL TACH OR TIME DATE RECORDING DESCRIPTION OF WORK PERFORMED-IN METER SIGNATURE & CERTIFICATE NO. OF PERSON PERFORMING WORK SERVICE TIME N5288J S/N: 421C-0158 DATE: 09-10-08 MODEL: 421C W.O. #: 17153 HOBBS TIME: 1551.3 TOTAL TIME: AIRFRAME LOG 2980 AIRWAY AV,, COSTA MESA, CA 92626 TEL: 714-433-2275 FAA APPROVED REPAIR STATION # JVYR016L THE AIRCRAFT AND / OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED AND INSPECTED IN ACCORDANCE WITH CURRENT FOR REFURAL AIR REGULATIONS AND WAS FOUND AIRWORTHY FOR RETURN TO SERVICE. PERTINENT DETAILS OF THE REPAIR / REPLACEMENT ARE ON FILE AT THIS AGENCY. SIGNED: amo SECURED WIRDING AND INSPECTED FOR FULL TRAVEL OF YOKE. TRAVEL CLEARANCE CHECKED OK. N5288J S/N: 421C-0158 DATE: 10-01-08 W.O. #: 17086 MODEL: 421C TOTAL TIME: HOBBS TIME: 1574.3 AIRFRAME LOG 2980 AIRWAY AV,. COSTA MESA, CA 92626 TEL: 714-433-2275 FAA APPROVED REPAIR STATION # JVYR016L THE AIRCRAFT AND / OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED AND INSPECTED IN ACCORDANCE WITH CURRENT FEDERAL AIR REGULATIONS AND WAS FOUND AIRWORTHY FOR RETURN TO SERVICE. PERTINENT DETAILS OF THE REPAIR / REPLACEMENT ARE ON FILE AT THIS AGENCY. lon SIGNED: 21 SERVICED AIR CONDITIONING AS PER AMM. P/N R 12 FREON. R&R RIGHT SIDE TACH GENERATOR WITH NEW. P/N 22A667EX R&R R/H PULSE LIGHT WITH NEW. P/N GE4594. R&R LEFT MAIN TIRE AND TUBE WITH NEW. P/N 650-10. N5288J DATE: 04-25-09 S/N: 421C-0158 W.O. #: 17438 MODEL: 421C TOTAL TIME: HOBBS TIME: 1580.6 AIRFRAME LOG 2980 AIRWAY AV,. COSTA MESA, CA 92626 TEL: 714-433-2275 FAA APPROVED REPAIR STATION # JVYR016L THE AIRCRAFT AND / OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED AND INSPECTED IN ACCORDANCE WITH CURRENT FEDERAL AIR REGULATIONS AND WAS FOUND AIRWORTHY FOR RETURN TO SERVICE. PERTINENT BETAILS OF THE REPAIR / REPLACEMENT ARE ON FILE AT THIS AGENCY. SIGNED: C/W ANNUAC INSPECTION IAW CESSNA SERVICE AND MAINTENANCE MANUAL. FAR 91-207(#) COMPLIED WITH YEARLY INSPECTION OF ELT BATTERY. BY PHYSICAL TEST OPS. CHECKED OK. REMOVED AND REINSTALLED OXYGEN BOTTLE AFTER HYDROSTATIC TEST PERFORMED BY AERO ACCESSORIES UNDER INV # 09-9474. SERVICED W/OXYGEN. REMOVED AND REINSTALLED BLOWDOWN BOTTLE AFTER HYDROSTATIC TEST PERFORMED BY AERO ACCESSORIES UNDER INV # 09-9474. SERVICED W/NITROGEN R&R RH RECOGNITION LIGHT W/NEW. P/N GE4594. INSTALLED SERVICEABLE NOSE CABIN BAGGAGE LATCH R&R VACUUM REGULATOR FILTERS W/NEW. P/N B3-5-1. AD95-09-13 FUEL FLOAT VALVE FUNCTIONAL TEST DUE AT HOBBS TIME 2029.1 C/W ALL ADs THROUGHOUT THIS DATE.

CERTIFIED ALTIMETER(S) AND STATIC SYSTEMS REQUIRED BY FAR 91.411 I.A.W. FAR PART 43, APPENDIX E TO SOUD FEET. -PERFORMED A.T.C. TRANSPONDER TESTS REQUIRED BY FAR 91.413 ACCOMPLISHED I.A.W. FAR PART 43, APPENDIX F.

| TAIL NUMBER: N52885       | WO#        | 222417             |
|---------------------------|------------|--------------------|
| LT. ALT S/N:              | #1 XPDR _  | <u> </u>           |
| RT. ALT S/N:              | #2 XPDR _  | NA                 |
| ADC / SDC / BLD. ENC S/N: | ~          |                    |
| INSPECTOR                 | DATE: _    | 1/14/09            |
| WESTERN AVIONICS, INC.    | FAA REPAIR | STATION # YH3R997L |
| Hobbs 4580.6              |            |                    |

| DATE   | TOTAL<br>TIME<br>IN<br>SERVICE | TACH OR<br>RECORDING<br>METER<br>TIME | DESCRIPTION OF WORK PERFORMED—<br>SIGNATURE & CERTIFICATE NO. OF PERSON PERFORMING WORK                                       |            |
|--|--------------------------------|---------------------------------------|---|------------|
|  |                                | TOTAL bro                             | ught forward from previous page   |            |
| N  | 5288J                          |                                       |   | The lose   |
| S  | N: 421C-0158                   |                                       | DATK: 08-06-10  |            |
|  | ODEL: 421C<br>OBBS TIME:       |                                       | W.O. #: 18298   |            |
|  | Obto mana                      | 10100                                 | TOTAL TIME: 5822.1  | 1 1.00     |
|  |                                |                                       | AIRFRAME LOG  |            |
|  |                                |                                       | 2980 AIRWAY AV,, COSTA MESA, CA 92626<br>TEL: 714-433-2275  |            |
|  |                                |                                       | FAA APPROVED REPAIR STATION # JVYR016L  |            |
|  |                                |                                       |   |            |
|  | CORDANCE                       | CRAFT AND /<br>S WITH CURR            | OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED AND INSPECTED IN<br>ENT FEDERAL AIR REGULATIONS AND WAS FOUND AIRWORTHY FOR RETURN |            |
|  | TO SERVIC                      | E. PERTINEN                           | T DETAILS OF THE REPAIR / REPLACEMENT ARE ON FILE AT THIS AGENCY.   | 1.1.1      |
| SIG  | NED:                           | F                                     |   |            |
|  |                                | SPECTION IAV                          | V CESSNA SERVICE AND MAINTENANCE MANUAL   |            |
| REM  | IOVED & REF                    | PLACED ELT B                          | BATTERIES W/NEW. P/N D-CELL   |            |
| - FAR  | 91-207(d) CO                   | MPLIED WITH                           | YEARLY INSPECTION OF ELT BATTERY BY PHYSICAL TEST. OPS CHECKED OK   | 1          |
| EL1  | BATTERIES I                    | EXPIRE JUNE 2                         | 2010.<br>TEST DUE DATE APRIL 2011.  | 1 1 1 1 1  |
| R&R  | VACUUM RI                      | EGULATOR FI                           | TEST DUE DATE APRIL 2011.<br>LTERS W/NEW, P/N B3-5-1,   | 1          |
| HEA  | TER INSPECT                    | FED AND TEST                          | TED OK.   |            |
| REM  | OVED AND F                     | REINSTALLED                           | LH ALTERNATOR AFTER BENCH TESTED BY OUTSIDE VENDOR, TESTED OK.  | 1111111111 |
| 1851   | ALLED I WO                     | STRIPS AT AF                          | T BULKHEAD.<br>NEW. P/N 600-6 8 PLY CO & TU 600-6 ACL (RESPECTIVELY).   |            |
| DISA   | SSEMBLED A                     | AND ROTATED                           | D RH MAIN TIRE,   |            |
| SERV   | ICED STRUT                     | S WITH NITRO                          | OGEN.   |            |
| TIGH   | ALTERNATO                      | DRAIN FITTI                           | NG.<br>S W/NEW, P/N MS25171-1S.   |            |
| - C/W  | ALL ADS THR                    | ROUGHOUT TH                           | IS DATE   | 1.1        |
|  |                                |                                       |   |            |
|  |                                | Hobbs                                 | 1634.2  | Sugar was  |
|  |                                | CERTIFI                               | ED ALTIMETER(S) AND STATIC SYSTEMS  |            |
|  |                                |                                       | ED BY F.A.R. 91.411 I.A.W. F.A.R. PART 43,  |            |
|  |                                |                                       | RANSPONDER TESTS REQUIRED BY F.A.R.   |            |
| 100  |                                |                                       | ACCOMPLISHED I.A.W. F.A.R. PART 43,   |            |
|  |                                | APPEND                                |   |            |
|  |                                |                                       |   |            |
|  | al and a second                | LT. ALT.                              |   |            |
|  |                                | RT. ALT.                              | SN #2XPDR SN N14  |            |
|  | 2. 1 2 5 3                     | ADCIE                                 | D.C.BLDENC BN   |            |
|  | and the second                 |                                       |   |            |
|  |                                | INSPECT                               | DATE 2/18/11  |            |
|  |                                | WESTERN                               | AVIONICS, INC. N 52 885 F.A.A. YH3R997L   |            |
|  |                                | WESTERN                               | AVIONICS, INC. N 52835 F.A.A. YH3R997L  |            |
|  |                                | 1000                                  |   |            |
|  |                                |                                       |   |            |
|  |                                |                                       |   |            |
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| the second s |                                |                                       |   |            |
| 1997.3   |                                |                                       |   |            |
|  |                                |                                       |   |            |
|  |                                |                                       |   |            |
|  |                                | SUB-TOTAL                             | this page   | i da       |

|   | The construction of the second  |   |   |   |
|---|---|---|---|---|
| DATE  | TOTAL<br>TIME<br>IN<br>SERVICE  | TACH OR<br>RECORDING<br>METER<br>TIME   | DESCRIPTION OF WORK PERFORME<br>SIGNATURE & CERTIFICATE NO. OF PERSON PERFO   |   |
|   |   | AA  | Aircraft Maintenance  |   |
|   | ach<br>TE 09-08-1   | 7070<br>Ho  | Aircraft Maintenance<br>Summit Valley Rd. Hesperia, Ca. 92345<br>(760) 947-0807<br>Dbbs 1647.7<br>5288J S/N 421C-0158 MODEL 421C  |   |
| Ser   | viced A/C sy<br>ertify this Air   | d airframe and i<br>stem. Inspected<br>rcraft has been<br>nined to be in a  | attery NEXT DUE 09-2013. Opened aircraft for annual flight controls, Replaced both main tires and tubes.<br>d cabin components. Post service engine run checks good.<br>inspected in Accordance With Annual/100 hr inspection n Airworthy condition.  |   |
|   | Garry Abbo  | ott   | AP564570639IA   |   |
| Cle   | Coil filter<br>an/ Gap plug<br>filter x   | gs x Check  | x Exhaust system x Oil service x<br>x fuel x Lube x<br>checked x  |   |
| MAKE: CES<br>MODEL: 42<br>S/N: 421C0<br>REG. NO: N<br>WORK ORI  | 1C<br>158   | AVIONICE  | Hangar One Avionics      Repair Station No. 1HGR058B 2026 Palomar Airport Road Carlsbad, CA 92011 Phone: 760-929-2270   | DATE: 4/<br>A/C TSN:<br>HOBBS:  |
| Avionics  | s Entries   |   | Filone. 100-929-2210  |   |
| Removed<br>antenna a  | Ameriking AK-45<br>nd coax supplied   | 50 121.5 / 243.0 MH   | Iz ELT unit S/N 453168 and installed new Ameriking AK450 121.5 / 243.0 MH   |   |
|   | ing open opphed   | with new AK-450 u   | init.   | Iz ELT unit with S/N 500619. Replaced   |
| I certify the   |   | een inspected in acc  | cordance with the requirements of FAR Section 91.207(d) and was found to b  |   |
| I certify the expiration  | at the ELT has be<br>date = Dec 2017<br>at the altimeter a  | een inspected in acc<br>7.<br>nd static system tes  | Init.   | e in satisfactory condition. Battery  |
| I certify the<br>expiration<br>I certify the<br>91.413 ha<br>Certified to<br>#1 Altim<br>#2 Altim   | at the ELT has b<br>date = Dec 2017<br>at the altimeter a<br>ve been perform<br>o 30,000 feet:<br>heter P/N 5934PA  | een inspected in acc<br>,<br>nd static system tes<br>ed and found to con<br>A-3 (A.86) S/N 4264<br>A-1 (A.69) S/N V294  | cordance with the requirements of FAR Section 91.207(d) and was found to b<br>ots and inspections required by 14 CFR 91.411 and transponder tests includin<br>nply with 14 CFR Part 43 Appendices E and F.<br>98  | e in satisfactory condition. Battery  |
| I certify the<br>expiration<br>I certify the<br>91.413 ha<br>Certified to<br>#1 Altim<br>#2 Altim<br>Garmin<br>During FA<br>Pilot and 0<br>condition 1<br>new MS20<br>by Cessna   | at the ELT has bi<br>date = Dec 2017<br>at the altimeter a<br>ve been perform<br>o 30,000 feet:<br>heter P/N 5934PA<br>GTX 327 Transp<br>R 91.411 inspec<br>Copilot system at<br>LS Lower Static I<br>0822-4D elbow fit<br>a Aircraft Compar  | een inspected in acc<br>and static system tes<br>ed and found to con<br>A-3 (A.86) S/N 4264<br>A-1 (A.69) S/N V294<br>wonder P/N 011-0049<br>tion, found the Pilot<br>s follows: LS Lower<br>Port P/N CM2933-6<br>tting, and two new c<br>ny under WO#'s 113  | cordance with the requirements of FAR Section 91.207(d) and was found to b<br>ets and inspections required by 14 CFR 91.411 and transponder tests includin<br>inply with 14 CFR Part 43 Appendices E and F.<br>98<br>1<br>90-00 S/N 83711103<br>and Copilot Static system to show excessive leak rate. Troubleshot problem in<br>Heated Static Port to be cracked and leaking, both #1 and #2 system Drain S<br>S/N N/A (ref. FAA form 8130-3 dated Feb. 22, 2013 completed by Cessna Ai<br>condition Drain Sumps P/N S1258-1 S/N N/A (ref. FAA form 8130-3 dated Sep<br>355245 and 1246177), removed and inspected and re-sealed the outer static.   | e in satisfactory condition. Battery<br>g data correspondence required by 14 C<br>and determined the source of leaks for t<br>umps defective and leaking. Installed Ni<br>rcraft Company under WO# 1259473) w<br>0. 20, 2011 and Dec. 27, 2012 complete<br>port plate and re-installed per 4C-43 13   |
| l certify the<br>expiration<br>l certify the<br>91.413 ha<br>Certified to<br>#1 Altim<br>#2 Altim<br>#2 Altim<br>Garmin<br>During FA<br>Pilot and 0<br>condition<br>new MS20<br>by Cessna<br>standards<br>During FA<br>evaluation<br>5934PA-1   | at the ELT has bi<br>date = Dec 2017<br>at the altimeter a<br>we been perform<br>o 30,000 feet:<br>leter P/N 5934PA<br>GTX 327 Transp<br>R 91.411 inspect<br>Copilot system as<br>LS Lower Static I<br>0822-4D elbow fit<br>a Aircraft Compai<br>. Tested and insp<br>R 91.411 inspect<br>on, and it was deter<br>(A.69) S/N V294   | een inspected in acc<br>and static system tes<br>ed and found to con<br>A-3 (A.86) S/N 4264<br>A-1 (A.69) S/N V294<br>wonder P/N 011-0049<br>tion, found the Pilot<br>s follows: LS Lower<br>Port P/N CM2933-6<br>tting, and two new c<br>ny under WO#'s 113<br>pected the Pilot and<br>tion found the Copili<br>rimined the unit was<br>1 (ref. FAA form 81  | cordance with the requirements of FAR Section 91.207(d) and was found to b<br>its and inspections required by 14 CFR 91.411 and transponder tests includin<br>inply with 14 CFR Part 43 Appendices E and F.<br>98<br>1<br>90-00 S/N 83711103<br>and Copilot Static system to show excessive leak rate. Troubleshot problem in<br>Heated Static Port to be cracked and leaking, both #1 and #2 system Drain S<br>S/N N/A (ref. FAA form 8130-3 dated Feb. 22, 2013 completed by Cessna Ai<br>ondition Drain Sumps P/N S1258-1 S/N N/A (ref. FAA form 8130-3 dated Ser  | e in satisfactory condition. Battery<br>g data correspondence required by 14 C<br>and determined the source of leaks for b<br>umps defective and leaking. Installed Ni<br>rcraft Company under WO# 1259473) w<br>o. 20, 2011 and Dec. 27, 2012 complete<br>port plate, and re-installed per AC-43.13<br>o negative results were observed.<br>G9839 and sent to repair facility for<br>condition exchange unit P/N   |
| I certify the<br>expiration<br>I certify the<br>91.413 ha<br>Certified to<br>#1 Altim<br>#2 Altim<br>Garmin<br>During FA<br>Pilot and 0<br>condition 1<br>new MS20<br>by Cessna<br>standards<br>During FA<br>evaluation<br>5934PA-1<br>system per   | at the ELT has bi<br>date = Dec 2017<br>at the altimeter a<br>ve been perform<br>o 30,000 feet:<br>heter P/N 5934PA<br>GTX 327 Transp<br>R 91.411 inspec<br>Copilot system as<br>LS Lower Static I<br>0822-4D elbow fit<br>a Aircraft Compai<br>. Tested and insp<br>R 91.411 inspec<br>h, and it was dete<br>(A.69) S/N V294<br>er 14 CFR Part 9   | een inspected in acc<br>and static system tes<br>ed and found to con<br>A-3 (A.86) S/N 4264<br>A-1 (A.69) S/N V294<br>wonder P/N 011-0043<br>tion, found the Pilot<br>s follows: LS Lower<br>Port P/N CM2933-6<br>ting, and two new c<br>ny under WO#'s 113<br>pected the Pilot and<br>tion found the Copil<br>srmined the unit was<br>1 (ref. FAA form 813<br>1.411 paragraph (a)  | cordance with the requirements of FAR Section 91.207(d) and was found to b<br>ats and inspections required by 14 CFR 91.411 and transponder tests includin<br>mply with 14 CFR Part 43 Appendices E and F.<br>98<br>1<br>90-00 S/N 83711103<br>and Copilot Static system to show excessive leak rate. Troubleshot problem i<br>Heated Static Port to be cracked and leaking, both #1 and #2 system Drain S<br>S/N N/A (ref. FAA form 8130-3 dated Feb. 22, 2013 completed by Cessna Ai<br>ondition Drain Sumps P/N S1258-1 S/N N/A (ref. FAA form 8130-3 dated Sep<br>335245 and 1246177), removed and inspected and re-sealed the outer static<br>Copilot static system I/A/W 14 CFR Part 91.411 paragraphs (a)(2) and (3). N<br>ot Altimeter fails the calibration test. Removed unit P/N 5934PA-1 (A.69) S/N<br>30-3 dated Feb. 1, 2013 completed by Mid-Continent Instruments under WO#   | e in satisfactory condition. Battery<br>g data correspondence required by 14 C<br>and determined the source of leaks for b<br>umps defective and leaking. Installed Ne<br>rcraft Company under WO# 1259473) w<br>o. 20, 2011 and Dec. 27, 2012 completed<br>port plate, and re-installed per AC-43.13<br>o negative results were observed.<br>G9839 and sent to repair facility for<br>condition exchange unit P/N  |
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| I certify the<br>expiration<br>I certify the<br>91.413 ha<br>Certified to<br>#1 Altim<br>#2 Altim<br>Garmin<br>During FA<br>Pilot and 0<br>condition 1<br>new MS20<br>by Cessna<br>standards<br>During FA<br>evaluation<br>5934PA-1<br>system per<br>Installed n<br>Relocated   | at the ELT has bi<br>date = Dec 2017<br>at the altimeter a<br>ve been perform<br>o 30,000 feet:<br>heter P/N 5934PA<br>GTX 327 Transp<br>R 91.411 inspec<br>Copilot system as<br>LS Lower Static I<br>0822-4D elbow fil<br>a Aircraft Compai<br>. Tested and insp<br>R 91.411 inspec<br>h, and it was dete<br>(A.69) S/N V294<br>er 14 CFR Part 9<br>missing STEC Au<br>I RH aft facing IC  | een inspected in acc<br>and static system tes<br>ed and found to con<br>A-3 (A.86) S/N 4264<br>A-1 (A.69) S/N V294<br>wonder P/N 011-0049<br>tion, found the Pilot<br>s follows: LS Lower<br>Port P/N CM2933-6<br>ting, and two new c<br>ny under WO#'s 112<br>pocted the Pilot and<br>tion found the Copili<br>trimined the unit was<br>1 (ref. FAA form 813<br>1.411 paragraph (a)<br>tro-Pilot 55X FLight I<br>S jacks from emerg  | cordance with the requirements of FAR Section 91.207(d) and was found to b<br>ats and inspections required by 14 CFR 91.411 and transponder tests includin<br>nply with 14 CFR Part 43 Appendices E and F.<br>98<br>1<br>90-00 S/N 83711103<br>and Copilot Static system to show excessive leak rate. Troubleshot problem in<br>Heated Static Port to be cracked and leaking, both #1 and #2 system Drain S<br>S/N N/A (ref. FAA form 8130-3 dated Feb. 22, 2013 completed by Cessna Ai<br>condition Drain Sumps P/N S1258-1 S/N N/A (ref. FAA form 8130-3 dated Sep<br>335245 and 1246177), removed and inspected and re-sealed the outer static<br>Copilot static system I/A/W 14 CFR Part 91.411 paragraphs (a)(2) and (3). N<br>of Altmeter fails the calibration test. Removed unit P/N 5934PA-1 (A.69) S/N<br>is recommended for overhauled exchange. Procured and installed overhauled<br>30-3 dated Feb. 1, 2013 completed by Mid-Continent Instruments under WO#<br>(2) instructions. No negative results were observed.<br>Manual Supplement into the Pilot's Operating Handbook.   | e in satisfactory condition. Battery<br>g data correspondence required by 14 C<br>and determined the source of leaks for b<br>umps defective and leaking. Installed Ni<br>rcraft Company under WO# 1259473) w<br>o. 20, 2011 and Dec. 27, 2012 complete<br>port plate, and re-installed per AC-43.13<br>o negative results were observed.<br>G9839 and sent to repair facility for<br>condition exchange unit P/N<br>WRK0211066). Tested and inspected  |
| l certify the<br>expiration<br>l certify the<br>91.413 ha<br>Certified to<br>#1 Altim<br>#2 Altim<br>Garmin<br>During FA<br>Pilot and Q<br>condition<br>new MS20<br>by Cessna<br>standards<br>During FA<br>evaluation<br>5934PA-1<br>system pe<br>Installed n<br>Relocated<br>Retermina   | at the ELT has bi<br>date = Dec 2017<br>at the altimeter a<br>we been perform<br>o 30,000 feet:<br>leter P/N 5934PA<br>GTX 327 Transp<br>R 91.411 inspect<br>Copilot system as<br>LS Lower Static I<br>D822-4D elbow fit<br>a Aircraft Compai<br>. Tested and insp<br>R 91.411 inspect<br>(A 69) S/N V294<br>er 14 CFR Part 9<br>missing STEC Au<br>I RH aft facing IC<br>ated BNC connect<br>with Garmin Soft<br>On as well as the   | een inspected in acc<br>and static system tes<br>ed and found to con<br>A-3 (A.86) S/N 4264<br>A-1 (A.69) S/N V294<br>wonder P/N 011-0049<br>tion, found the Pilot<br>s follows: LS Lower<br>Port P/N CM2933-6<br>ting, and two new c<br>ny under WO#'s 113<br>bected the Pilot and<br>tion found the Copili<br>rmined the unit was<br>1 (ref. FAA form 813<br>1.411 paragraph (a)<br>ito-Pilot 55X FLight I<br>S jacks from emerge<br>ctor at marker beacc<br>tware service bulleti<br>a units GPS software  | cordance with the requirements of FAR Section 91.207(d) and was found to b<br>ats and inspections required by 14 CFR 91.411 and transponder tests includin<br>mply with 14 CFR Part 43 Appendices E and F.<br>98<br>1<br>90-00 S/N 83711103<br>and Copilot Static system to show excessive leak rate. Troubleshot problem<br>Heated Static Port to be cracked and leaking, both #1 and #2 system Drain S<br>S/N N/A (ref. FAA form 8130-3 dated Feb. 22, 2013 completed by Cessna Ai<br>condition Drain Sumps P/N S1258-1 S/N N/A (ref. FAA form 8130-3 dated Sep<br>335245 and 1246177), removed and inspected and re-sealed the outer static<br>Copilot static system I/AW 14 CFR Part 91.411 paragraphs (a)(2) and (3). N<br>of Altimeter fails the calibration test. Removed unit P/N 5934PA-1 (A.69) S/N<br>recommended for overhauled exchange. Procured and installed overhauled<br>30-3 dated Feb. 1, 2013 completed by Mid-Continent Instruments under WO#<br>(2) instructions. No negative results were observed.<br>Manual Supplement into the Pilot's Operating Handbook.   | e in satisfactory condition. Battery<br>g data correspondence required by 14 C<br>and determined the source of leaks for b<br>umps defective and leaking. Installed Ne<br>rcraft Company under WO# 1259473) w<br>b. 20, 2011 and Dec. 27, 2012 complete<br>port plate, and re-installed per AC-43.13<br>o negative results were observed.<br>G9839 and sent to repair facility for<br>condition exchange unit P/N<br>WRK0211066). Tested and inspected<br>plugs into arm rest.<br>t unit.   |
| l certify the<br>expiration<br>l certify the<br>91.413 ha<br>Certified to<br>#1 Altim<br>#2 Altim<br>Garmin<br>During FA<br>Pilot and C<br>by Cessna<br>standards<br>During FA<br>evaluation<br>5934PA-1<br>system per<br>Installed n<br>Relocated<br>Retermina<br>Complied<br>version 5.<br>190-00350  | at the ELT has bi<br>date = Dec 2017<br>at the altimeter a<br>ve been perform<br>o 30,000 feet:<br>heter P/N 5934PA<br>GTX 327 Transp<br>R 91.411 inspect<br>Copilot system as<br>LS Lower Static I<br>D822-4D elbow fil<br>a Aircraft Compai<br>. Tested and insp<br>R 91.411 inspect<br>A.699 S/N V294<br>er 14 CFR Part 9<br>nissing STEC Au<br>I RH aft facing IC<br>ated BNC connect<br>with Garmin Soff<br>00, as well as the<br>6-02, Rev. G, Da<br>with Garmin Soff<br>00, as well as the   | een inspected in acc<br>and static system tes<br>ed and found to con<br>A-3 (A.86) S/N 4264<br>A-1 (A.69) S/N V294<br>ionder P/N 011-004<br>tion, found the Pilot<br>s follows: LS Lower<br>Port P/N CM293-6<br>ting, and two new co<br>ny under WO#'s 112<br>pected the Pilot and<br>tion found the Copili<br>rmined the unit was<br>1 (ref. FAA form 81:<br>1.411 paragraph (a)<br>ito-Pilot 55X FLight I<br>S jacks from emerg<br>ctor at marker beacc<br>tware service bulleti<br>a units GPS software<br>ted: December, 200<br>tware service bulleti<br>a units GPS software   | cordance with the requirements of FAR Section 91.207(d) and was found to b<br>the and inspections required by 14 CFR 91.411 and transponder tests includin<br>inply with 14 CFR Part 43 Appendices E and F.<br>98<br>1<br>90-00 S/N 83711103<br>and Copilot Static system to show excessive leak rate. Troubleshot problem is<br>Heated Static Port to be cracked and leaking, both #1 and #2 system Drain S<br>S/N N/A (ref. FAA form 8130-3 dated Feb. 22, 2013 completed by Cessna Ai<br>nordition Drain Sumps P/N S1258-1 S/N N/A (ref. FAA form 8130-3 dated Sep<br>335245 and 1246177), removed and inspected and re-sealed the outer static<br>Copilot static system I/A/W 14 CFR Part 91.411 paragraphs (a)(2) and (3). N<br>ot Altimeter fails the calibration test. Removed unit P/N 5934PA-1 (A.69) S/N<br>is recommended for overhauled exchange. Procured and installed overhauled<br>30-3 dated Feb. 1, 2013 completed by Mid-Continent Instruments under WO#<br>(2) instructions. No negative results were observed.<br>Manual Supplement into the Pilot's Operating Handbook.<br>pency exit window to lower panel below window and installed cover plate with pon<br>antenna. Marker beacon function check satisfactory using the NAV401 test<br>in No. 1225 Rev A, Dated: October 19, 2012. Upgraded Garmin GNS 430W N<br>e from 3.2 to version 5.0. Performed operational check, in accordance with G<br>8. System checks satisfactory on ground.   | e in satisfactory condition. Battery<br>g data correspondence required by 14 C<br>and determined the source of leaks for b<br>umps defective and leaking. Installed Ni<br>rcraft Company under WO# 1259473) w<br>o. 20, 2011 and Dec. 27, 2012 complete<br>port plate, and re-installed per AC-43.13<br>o negative results were observed.<br>G9839 and sent to repair facility for<br>condition exchange unit P/N<br>WRK0211066). Tested and inspected<br>plugs into arm rest.<br>t unit.<br>Main software from version 3.30 to softwa<br>armin GNS 430W Installation Manual P/I  |
| l certify the<br>expiration<br>l certify the<br>91.413 ha<br>Certified to<br>#1 Altim<br>#2 Altim<br>Garmin<br>During FA<br>Pilot and 0<br>condition<br>new MS20<br>by Cessna<br>standards<br>During FA<br>evaluation<br>5934PA-1<br>system pe<br>Installed in<br>Relocated<br>Retermina<br>Complied<br>version 5.<br>190-00350<br>Complied<br>version 2.   | at the ELT has bi<br>date = Dec 2017<br>at the altimeter a<br>we been perform<br>o 30,000 feet:<br>leter P/N 5934PA<br>GTX 327 Transp<br>R 91.411 inspec<br>Copilot system as<br>LS Lower Static I<br>D822-4D elbow fil<br>a Aircraft Compai<br>. Tested and insp<br>R 91.411 inspec<br>(A.69) S/N V294<br>er 14 CFR Part 9<br>missing STEC Au<br>I RH aft facing IC<br>ated BNC connect<br>with Garmin Soff<br>00, as well as the<br>6-02, Rev. G, Da<br>with Garmin Soff<br>10, Performed op  | een inspected in acc<br>and static system tes<br>ed and found to con<br>A-3 (A.86) S/N 4264<br>A-1 (A.69) S/N V294<br>wonder P/N 011-0049<br>tion, found the Pilot<br>s follows: LS Lower<br>Port P/N CM2933-6<br>ting, and two new c<br>ny under WO#'s 113<br>bected the Pilot and<br>tion found the Copili<br>rmined the unit was<br>1 (ref. FAA form 813<br>1.411 paragraph (a)<br>ito-Pilot 55X FLight I<br>S jacks from emergent<br>tor at marker beacce<br>tware service bulleti<br>e units GPS software<br>ted: December, 200<br>tware service bulleti<br>e units GPS software<br>ted: December, 200<br>tware service bulleti  | cordance with the requirements of FAR Section 91.207(d) and was found to b<br>ats and inspections required by 14 CFR 91.411 and transponder tests includin<br>mply with 14 CFR Part 43 Appendices E and F.<br>98<br>1<br>90-00 S/N 83711103<br>and Copilot Static system to show excessive leak rate. Troubleshot problem<br>Heated Static Port to be cracked and leaking, both #1 and #2 system Drain S<br>S/N N/A (ref. FAA form 8130-3 dated Feb. 22, 2013 completed by Cessna Ai<br>ondition Drain Sumps P/N S1258-1 S/N N/A (ref. FAA form 8130-3 dated Sep<br>335245 and 1246177), removed and inspected and re-sealed the outer static<br>Copilot static system I/AW 14 CFR Part 91.411 paragraphs (a)(2) and (3). N<br>ot Altimeter fails the calibration test. Removed unit P/N 5934PA-1 (A.69) S/N<br>recommended for overhauled exchange. Procured and installed overhauled<br>30-3 dated Feb. 1, 2013 completed by Mid-Continent Instruments under WO#<br>(2) instructions. No negative results were observed.<br>Manual Supplement into the Pilot's Operating Handbook.<br>Itency exit window to lower panel below window and installed cover plate with p<br>on antenna. Marker beacon function check satisfactory using the NAV401 tes<br>in No. 1225 Rev A, Dated: October 19, 2012. Upgraded Garmin GNS 430W M<br>e from 3.2 to version 5.0. Performed operational check, in accordance with Gi<br>8. System checks satisfactory on ground.<br>in No. 1225 Rev A, Dated: October 19, 2012. Upgraded Garmin GNS 530W M  | e in satisfactory condition. Battery<br>g data correspondence required by 14 C<br>and determined the source of leaks for b<br>umps defective and leaking. Installed Nu<br>rcraft Company under WO# 1259473) w<br>5. 20, 2011 and Dec. 27, 2012 completed<br>port plate, and re-installed per AC-43.13<br>o negative results were observed.<br>G9839 and sent to repair facility for<br>condition exchange unit P/N<br>WRK0211066). Tested and inspected<br>plugs into arm rest.<br>t unit.<br>Main software from version 3.30 to softwa<br>armin GNS 430W Installation Manual P/I<br>Main software from version 3.30 to softwa<br>armin GNS 530W Installation Manual P/I  |
| l certify the<br>expiration<br>I certify the<br>91.413 ha<br>Certified th<br>#1 Altim<br>#2 Altim<br>Garmin<br>During FA<br>Pilot and C<br>condition<br>new MS20<br>by Cessna<br>standards<br>During FA<br>evaluation<br>5934PA-1<br>system per<br>Installed in<br>Relocated<br>Retermina<br>Complied<br>version 5.<br>190-00351<br>Complied<br>version 2.<br>satisfactor                           | at the ELT has bi<br>date = Dec 2017<br>at the altimeter a<br>ve been perform<br>o 30,000 feet:<br>heter P/N 5934PA<br>GTX 327 Transp<br>GTX 327 Transp<br>R 91.411 inspect<br>Copilot system as<br>LS Lower Static I<br>3822-4D elbow fil<br>a Aircraft Compai<br>. Tested and insp<br>. R 91.411 inspect<br>(A.69) S/N V294<br>er 14 CFR Part 9'<br>nissing STEC Au<br>I RH aft facing IC<br>ated BNC connect<br>with Garmin Soff<br>00, as well as the<br>6-02, Rev. G, Da<br>with Garmin Soff<br>10. Performed og<br>ry on ground.                                      | een inspected in acc<br>7.<br>Ind static system tes<br>ed and found to con<br>A-3 (A.86) S/N 4264/<br>A-1 (A.69) S/N V294<br>ionder P/N 011-0049<br>tion, found the Pilot<br>s follows: LS Lower<br>Port P/N CM293-6<br>ting, and two new co<br>ting, and two new co<br>ting, and two new co<br>port P/N CM293-6<br>ting, and two new co<br>s follows: LS Lower<br>Port P/N CM293-6<br>ting, and two new co<br>s follows: LS Lower<br>Port P/N CM293-6<br>ting, and two new co<br>ting, and two new co<br>tion, the pilot and<br>tion found the Copility<br>tion found the copility<br>t | <ul> <li>cordance with the requirements of FAR Section 91.207(d) and was found to b</li> <li>ats and inspections required by 14 CFR 91.411 and transponder tests includin nply with 14 CFR Part 43 Appendices E and F.</li> <li>98 1 90-00 S/N 83711103 and Copilot Static system to show excessive leak rate. Troubleshot problem in Heated Static Port to be cracked and leaking, both #1 and #2 system Drain S S/N N/A (ref. FAA form 8130-3 dated Feb. 22, 2013 completed by Cessna Ai condition Drain Sumps P/N S1258-1 S/N N/A (ref. FAA form 8130-3 dated Sep. 335245 and 1246177), removed and inspected and re-sealed the outer static Copilot static system I/AW 14 CFR Part 91.411 paragraphs (a)(2) and (3). N ot Altimeter fails the calibration test. Removed unit P/N 5934PA-1 (A.69) S/N recommended for overhauled exchange. Procured and installed overhauled is 30-3 dated Feb. 1, 2013 completed by Mid-Continent Instruments under WO# (2) instructions. No negative results were observed.</li> <li>Manual Supplement into the Pilot's Operating Handbook.</li> <li>gency exit window to lower panel below window and installed cover plate with pon antenna. Marker beacon function check satisfactory using the NAV401 test in No. 1225 Rev A, Dated: October 19, 2012. Upgraded Garmin GNS 430W N e from 3.2 to version 5.0. Performed operational check, in accordance with Gills. System checks satisfactory on ground.</li> <li>in No. 1225 Rev A, Dated: October 19, 2012. Upgraded Garmin GNS 530W N e from 3.2 to version 5.0. Performed operational check, in accordance with Gills. System checks satisfactory on ground.</li> <li>in No. 1225 Rev A, Dated: October 19, 2012. Upgraded Garmin GNS 530W N e from 3.2 to version 5.0. Performed operational check, in accordance with Gills. System checks satisfactory on ground.</li> <li>in No. 1225 Rev A, Dated: October 19, 2012. Upgraded Garmin GNS 530W N e from 3.2 to version 5.0. Performed operational check, in accordance with Gills. System checks satisfactory on ground.</li> </ul>   | e in satisfactory condition. Battery<br>g data correspondence required by 14 C<br>and determined the source of leaks for b<br>umps defective and leaking. Installed Ne<br>rcraft Company under WO# 1259473) w<br>0. 20, 2011 and Dec. 27, 2012 completed<br>port plate, and re-installed per AC-43.13<br>o negative results were observed.<br>G9839 and sent to repair facility for<br>condition exchange unit P/N<br>WRK0211066). Tested and inspected<br>plugs into arm rest.<br>t unit.<br>Main software from version 3.30 to softwa<br>armin GNS 430W Installation Manual P/I<br>Main software from version 3.30 to softwa<br>armin GNS 530W Installation Manual P/I  |
| l certify the<br>expiration<br>I certify the<br>91.413 ha<br>Certified to<br>#1 Altim<br>#2 Altim<br>Garmin<br>During FA<br>Pilot and C<br>condition<br>new MS20<br>by Cessna<br>standards<br>During FA<br>evaluation<br>5934PA-1<br>system pe<br>Installed in<br>Relocated<br>Retermina<br>Complied<br>version 5.<br>190-00350<br>Complied<br>version 2.<br>satisfactol<br>Updated C<br>service pr | at the ELT has bi<br>date = Dec 2017<br>at the altimeter a<br>we been perform<br>o 30,000 feet:<br>leter P/N 5934PA<br>GTX 327 Transp<br>R 91.411 inspec<br>Copilot system as<br>LS Lower Static I<br>D822-4D elbow fil<br>a Aircraft Compai<br>. Tested and insp<br>R 91.411 inspec<br>t, and it was dette<br>(A.69) S/N V294<br>er 14 CFR Part 9<br>missing STEC Au<br>I RH aft facing IC<br>ated BNC connect<br>with Garmin Soft<br>00, as well as the<br>6-02, Rev. G, Da<br>with Garmin Soft<br>10. Performed op<br>ry on ground.<br>Garmin GDL69A<br>ocedure in accor | een inspected in acc<br>and static system tes<br>ed and found to con<br>A-3 (A.86) S/N 4264<br>A-1 (A.69) S/N V294<br>wonder P/N 011-0043<br>tion, found the Pilot<br>s follows: LS Lower<br>Port P/N CM2933-6<br>ting, and two new c<br>ny under WO#'s 113<br>bected the Pilot and<br>tion found the Copile<br>rmined the unit was<br>1 (ref. FAA form 813<br>1.411 paragraph (a)<br>ito-Pilot 55X FLight (a)<br>ito-Pilot 55X FLight (a)<br>ito-Pilot 55X FLight (a)<br>ito-Pilot 55X FLight (a)<br>to a marker beacc<br>tware service bulleti<br>e units GPS softwar<br>ted: December, 200<br>tware service bulleti<br>perational check, in<br>software from versic<br>dance with service to<br>the service bulleti<br>perational check, in  | <ul> <li>cordance with the requirements of FAR Section 91.207(d) and was found to b</li> <li>the and inspections required by 14 CFR 91.411 and transponder tests includin nply with 14 CFR Part 43 Appendices E and F.</li> <li>98</li> <li>1</li> <li>90-000 S/N 83711103</li> <li>and Copilot Static system to show excessive leak rate. Troubleshot problem in Heated Static Port to be cracked and leaking, both #1 and #2 system Drain S</li> <li>S/N N/A (ref. FAA form 8130-3 dated Feb. 22, 2013 completed by Cessna A</li> <li>and topilot Static system to show excessive leak rate. Troubleshot problem in Heated Static Port to be cracked and leaking, both #1 and #2 system Drain S</li> <li>S/N N/A (ref. FAA form 8130-3 dated Feb. 22, 2013 completed by Cessna A</li> <li>and ondition Drain Sumps P/N S1258-1 S/N N/A (ref. FAA form 8130-3 dated Sep 335245 and 1246177), removed and inspected and re-sealed the outer static Copilot static system I/A/W 14 CFR Part 91.411 paragraphs (a)(2) and (3). N</li> <li>ot Altimeter fails the calibration test. Removed unit P/N 5934PA-1 (A 69) S/N is recommended for overhauled exchange. Procured and installed overhauled 30-3 dated Feb. 1, 2013 completed by Mid-Continent Instruments under WO# (2) instructions. No negative results were observed.</li> <li>Manual Supplement into the Pilot's Operating Handbook.</li> <li>gency exit window to lower panel below window and installed cover plate with pon antenna. Marker beacon function check satisfactory using the NAV401 test in No. 1225 Rev A, Dated: October 19, 2012. Upgraded Garmin GNS 430W M e from 3.2 to version 5.0. Performed operational check, in accordance with Gi8. System checks satisfactory on ground.</li> <li>in No. 1225 Rev A, Dated: October 19, 2012. Upgraded Garmin GNS 530W M e from 3.2 to version 5.0. Performed operational check, in accordance with Gi8. System checks satisfactory on ground.</li> <li>in No. 0809 Rev. A, Dated: October 19, 2012. Upgraded Garmin GNS 537 Main si accordance with Garmin GTX 327 Installation Manual P/N: 190-00187-02</li></ul>  | e in satisfactory condition. Battery<br>g data correspondence required by 14 C<br>and determined the source of leaks for b<br>sumps defective and leaking. Installed Ne<br>rcraft Company under WO# 1259473) w<br>o. 20, 2011 and Dec. 27, 2012 complete<br>port plate, and re-installed per AC-43.13<br>o negative results were observed.<br>G9839 and sent to repair facility for<br>condition exchange unit P/N<br>WRK0211066). Tested and inspected<br>plugs into arm rest.<br>t unit.<br>Main software from version 3.30 to softwa<br>armin GNS 430W Installation Manual P/N<br>Alain software from version 3.30 to software<br>armin GNS 530W Installation Manual P/N<br>oftware from version 2.02 to software<br>v. Q, Dated: March, 2012. System check<br>oftware verification procedure and return |
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MAKE: CESSNA MODEL: 421C S/N: 421C0158 REG. NO: N5288J WORK ORDER: 20121008-1M Hangar One Avionics Repair Station No. 1HGR0588 2026 Palomar Airport Road Carlsbad, CA 92011 Phone: 760-929-2270 DATE: 4/1/2013 A/C TSN: 5820.7 HOBBS: 1723.3

### **Airframe Entries**

Completed Annual Inspection IAW the Cessna Inspection Check list as applicable and FAR 43 Appendix "D"

VIONICS

Completed General Inspection of all aftermarket installations not listed with specific instructions for continued airworthiness---Inspect IAW AC43-13-1B Chapter 5 Section 2.

CW AD# 81-09-09 (Combustion Heater 250 Insp) - by inspection. Found combustion heater circulation air blower motor inoperative. Removed blower P/N 721984 S/N WO#0116134 and sent to Aircraft Heating and Electrical, Inc. Unit was returned with a work status of "Overhauled" - See FAA form 8130-3 WO# 0116134 dated 01-16-2013. Reinstalled same P/N and S/N, Replaced 3" duct P/N CEET12, 3 3/4" duct P/N SCEET15 (05-30615), 4 1/2" duct P/N SCEET18 (05-30618) and dust P/N CM3211-18A. System function check satisfactory. Inspection per AD due again in 250 Heater Hrs. Overhaul Due: 1169.0, Heater Hobbs Currently at 615.9 Hrs.

CW AD# 84-26-02 (Paper Induction Air Filter) for LH and RH engines by replacement of Induction Air Filters 2ea P/N P10-8421. Replacement mandatory ea 500hrs.

CW AD# 2000-01-16 (Exhaust System Inspection) for the LH and RH engines by inspection per para (d) and replacement of the following new exhaust components which restart the clock for paragraphs (b)(c)(e) and (g): 2ea Turbo Wyes P/N 8294-15C - LH S/N IRY75-0001 and RH S/N IRY75-0003 using 2ea gaskets P/N 002957006 with 8ea new bolts P/N MS20036-10 and 8ea new nuts P/N MS20500-624. 4ea Slip Joint Assemblies P/N K5355108-4 LH S/Ns 221636 and 221638, RH S/Ns 221646 and 229643 using new Springs 40ea P/N 1298119-4, Bolts 20ea P/N AN3C13 and 20ea P/N AN3C15, nuts 40ea P/N AN310C3 and as required washers for proper spring compression P/N AN960C10 with new stainless cotter pins. 2ea Flex Elbows P/N K9910300-03 LH S/N 220995 and RH S/N 220998 using new clamps 2ea P/N U10C050-240M and 4ea new wastegate gaskets P/N 9910306-1 with 16ea new bolts P/N AN4C11A and 16ea new nuts P/N MS20500-428. 2ea Tail Pipes P/N K9910300-01 LH S/N 220391 using 6ea flange Gaskets P/N 652458 and 24ea nuts P/N MS20500-428. 2ea Exhaust Stacks LH Engine P/N K9910295-15 LH S/N 222330 and RH S/N 222391 using 6ea flange Gaskets P/N 652458 and 24ea nuts P/N MS20500-428. 2ea Exhaust Stacks LH Engine P/N K9910295-16 LH S/N 222395 and RH S/N 222396 using 6ea flange Gaskets P/N 652458 and 24ea nuts P/N MS20500-428. Relocated, drilled and reinstalled existing EGT probes IAW JPI installation instructions Report No. 760. Paragraph (b) next due in 50 hours, paragraphs (c) and (e) next due in 5 years and then annually thereafter, paragraphs (d) next due in 500 hours, paragraph (f) not applicable at this time due to no multi-segment clamps installed and paragraph (g) next due in 2500

Removed expired fire extinguisher and bracket assembly and installed new C352TS 2.5lbs Halon 1211 fire extinguisher and associated bracket.

Fabricated and installed various missing placards throughout aircraft interiror and exterior. Installed new turn limits placard P/N 5100181-36.

Aircraft re-weighed this maintenance event and new center of gravity computed, see W&B form this date. General Electrodynamics Corporation Scales Cal Date Dec 2012, P/N DLI0000P-6 S/N I4139. Scale calibration due December 31st 2013. Inserted new weight and balance reflecting this event into the POH and aircraft permanent record.

Executed FAA form 337 due to a lack of documentation for the Removal of Brackett Induction Air Filters - reference FAA form 337 dated 09-19-1996,

Executed FAA form 337 due to a lack of documentation for the installation of Shadin digital Fuel Flow System in accordance with STC# SA607GL.

Executed FAA form 337 due to a lack of documentation for the installation of RAM Vacuum Pump Cooling Shrouds in accordance with STC#SA3721AW.

Executed FAA form 337 due to the lack of documentation for installation of Rosen sun visors in accordance with STC# SA5137NM.

Removed pipe fitting from LH Engine fuel flow transducer due to a blue stain. Cleaned fitting and applied pipe thread sealant and reinstalled. Leak and operational check satisfactory.

Fabricated new fuel pressure regulator bracket on the left engine.

Replaced LH alternate air door magnet P/N 9910016-1.

Repaired upper deck pressure to fuel limiter tube by replacement of aluminum tube segment using new b-nuts 2ea P/N AN818-6D and Sleeves 2ea P/N MS20819-6D.

Removed leaking LH wastegate actuator P/N 480164-9001 and installed an overhaul exchange unit P/N 481064-9001 S/N THR0108 - See Approved Turbo Components, Inc. FAA form 8130-3 WO# 32913 dated 03-26-2013. Replaced inlet fitting 1ea P/N AN833-6 with nut P/N AN924-6 and o-ring P/N MS28778-6 and outlet fitting 1ea P/N AN833-4 with nut P/N AN924-4 and o-ring P/N MS28778-4. Ground run, operational and leak check satisfactory.

Reset LH and RH Propeller Governor Controls for adequate travel and cockpit cushion per the Cessna 421C maintenance manual.

Adjusted LH and RH propeller governors for proper governing and feathering operation per the Cessna 421C maintenance manual.

Reset LH and RH Throttle Controls for adequate travel and cockpit cushion per the Cessna 421C maintenance manual.

Reset LH and RH Mixture Controls for adequate travel and cockpit cushion per the Cessna 421C maintenance manual.

Freed stiff RH alternate air control.

Freed stiff LH alternate air control.

Cleaned LH and RH engines and engine compartments.

Removed Propeller Anti-Ice amp gauge, found wiring reversed, cleaned contacts, lengthen and reterminated wires. Replaced gauge to panel seal P/N M83248-2-21 and reinstalled. Found one boot on the LH propeller with wires reversed, corrected wiring and checked system for proper operation - system function check on the ground was satisfactory.

Removed Co-Pilot's Turn Coordinator P/N 1394T100-7Z S/N 9007-3 and installed new exchange Mid-Continent Turn Coordinator P/N 1394T1007B S/N L12-11164. Leveled aircraft and secured Turn Coordinator.

Removed Pilot's Attitude gyro P/N 25-501-031-1 S/N T42292 and installed a Mid-Continent overhaul exchange unit P/N 23-501-032-2 S/N T42501 - See FAA form 8130-3 WO# WRK0211124 dated 01-17-2013 with a work status of " Modified." Leveled aircraft and connected Airborne vacuum test kit #343. Adjusted attitude gyro for level at 5 inches and torqued gyro to panel with existing retaining screws. Vacuum system and associated instruments function test satisfactory.

Removed Cabin Altimeter/Differential Pressure Gauge P/N C668516-0104 S/N 12683 and sent to Mid-Continent Instruments for repair. Unit was returned with a work status of "Repaired." See FAA form 8130-3 WO# WRK0210963 dated 01-17-2013. Aircraft needs check flight for verification of instrument functionality.

Removed Glareshield to determine problem with RH 3 in 1 gauge. Found wire chaffed through and hanging be a strand. Splice repaired wire with a Raychem environmental splice. Reinstalled Glareshield. Function check of RH 3 in 1 gauge satisfactory.

Removed existing LH and RH Cleveland Wheels and Brakes kit P/N 199-76 and installed new Cleveland Wheels and Brakes Kit P/N 199-76. Flushed brake system,





bled and serviced master cylinders using MIL-PRF-5606H hydraulic fluid. Completed Cleveland "Brake Lining Conditioning Procedure" per Cleveland Document No. PRM14A Rev A dated 07-01-1995.

Removed all 3 Landing gear retract system hydraulic actuators and sent to Aerospace Products / D-J Engineering in Augusta, KS. for overhaul. LH Actuator P/N 9910136-2 S/N 1512 was returned with a work status of "Overhauled" - See FAA form 8130-3 WO# 1512 dated 02-06-2013. Unit S/N 1512 was reinstalled in the LH position with new upper bolt P/N NAS464-10-24, nut P/N MS17826-10 and washer P/N AN960-1016, replaced lower nut P/N MS17826-12. RH Actuator P/N 9910136-2 S/N 1519 was returned with a work status of "Overhauled" - See FAA form 8130-3 WO# 1519 dated 02-06-2013. Unit S/N 1519 was reinstalled in the RH position with new upper bolt P/N NAS464-10-24, nut P/N MS17826-10 and washer P/N AN960-1016. Nose Actuator P/N 9910139-3 S/N 1513 was returned with a work status of "Overhauled" - See FAA form 8130-3 WO# 1513 dated 02-06-2013. Unit S/N 1513 was reinstalled in the nose position with new bolts upper P/N NAS6204-11D and lower bolt P/N NAS6204-15D - installed new 2ea AN833-4D fittings and 2ea AN924-4D nuts to nose gear actuator. Landing gear system checks performed per the Cessna 421C maintenance manual and Cessna Service Bulletin MEB91-12. Performed several fault free cycles including an emergency "Blow down" Re-Serviced blow down bottle with N2.

Rigged landing gear and gear doors as required after overhaul of actuators per the Cessna 421C maintenance manual ...

Fabricated new support for hydraulic line in RH gear wheel well.

Reduced play in Main Landing Gear Torque Links by installing 1ea S1450-4-152-032 shims and applying proper torque to center knee of both LH and RH links.

Replaced Main Landing Gear Hydraulic System Filters 2ea P/N AN6235-2A and 2ea O-rings P/N MS28775-222.

Removed LH Hydraulic pump P/N 24194-RAA S/N 172 and installed an overhaul exchange unit by Quality Aircraft Accessories, Inc. Unit was received with a work status of "Overhauled" See FAA form 8130-3 WO# WO1170006915 dated 02-06-2013. Installed unit P/N 24194-RAA S/N 43 using a new P/N S3346-1 gasket. Run-up of LH engine revealed that the Hydraulic annunciator extinguishes appropriately and no leaks were noted.

Hydraulic fluid leak in RH engine compartment. Cleaned engine and engine compartment, replaced seals in hydraulic filter assembly. Engine run up revealed no leaks.

Removed lower Nose Strut assembly and replaced the following parts: 1ea P/N 0841200-19 retainer ring, 1ea P/N 0841200-25 retainer ring, 1ea P/N A2748 polypack seal, 1ea P/N MS28775-034 o-ring, 1ea P/N MS28775-333 o-ring and 1ea S3376 scraper ring. reassembled strut assembly and serviced with MIL-PRF-5606H Hydraulic fluid and nitrogen per the Cessna 421C Maintenance Manual.

Removed RH main landing gear, cleaned and repacked strut assembly using the following new parts: 1ea A2749 poly pack seal, 1ea MS28778-015 o-ring, 1ea MS28775-228 o-ring, 1ea MS28776M2-22 wiper ring and 1ea MS28869-2 valve. Reinstalled RH main landing gear and serviced strut with MIL-PRF-5606H Hydraulic fluid and N2. All work done per the Cessna 421C maintenance manual.

Removed landing gear blow down bottle due to an undetermined leak. Isolated leak to bottle valve. Removed existing valve and adapter and replaced with an MS28889-2 serviceable valve and new adapter. Reinstalled bottle and serviced with N2 per the Cessna 421C maintenance manual. Leak check satisfactory.

Removed LH and RH 3D2352-01 wing deice flow valves S/Ns N808 and N2369 and sent to Quality Aircraft Accessories/AMETEK B&S Aircraft Parts & Accessories for overhaul. Unit S/N N808 was returned with a work status of "Overhauled" See FAA form 8130-3 WO# R13-00249 dated 02-28-2013, Reinstalled S/N N808 into the LH wing. Unit S/N N2369 was returned with a work status of "Overhauled" See FAA form 8130-3 WO# R13-00250 dated 02-28-2013, Reinstalled S/N N2369 Into the RH wing. Reterminated wiring in both LH and RH wings. Connected Airborne vacuum test kit #343, system function check as stated below.

Removed Surface De-ice annunciator switch, cleaned and bench tested unit. Switch opens and closes appropriately with pressure applied and/or removed. Pressures values used as listed in the Cessna 421C maintenance manual. Unit reinstalled and wiring re-terminated

Patch repaired aircraft de-ice boots as required using BFGoodrich patch kit and sealed pin holes in various locations with sealant. Connected Airborne vacuum test kit #343 and found boots to be weathered beyond repair. Surface Deice system deactivated and placarded in accordance with 14 CFR Part 91.213.

Found wiring on Heated windshield disconnected, connected wiring and performed operational check of windshield heat - System function check satisfactory.

Replaced Propeller Deice brush assemblies. Cleaned slip rings and readjusted clearances per the Cessna 421C maintenance manual. System function check satisfactory

Performed a bench functional test of air conditioning hydraulic system annunciator pressure switch. Switch body leaks hydraulic fluid. Removed discrepant switch assembly. Installed serviceable 9910287-11 (ITT 98087-1106P61) air conditioning hydraulic pressure switch S/N 4614 (A3Q96) and repaired air conditioning system wiring in the Right nacelle area by removal of deteriorated wiring and connectors and reterminated using new wiring segments and new connectors. Removed Air Conditioning compressor, inspected, cleaned, flushed and drained oil. Compressor serviced with 7oz of 5GS Suniso Refrigeration Oil and reinstalled. Disconnected various lines, removed receiver dryer and flushed all Air Conditioning system lines, including condenser. Removed LH and RH evaporator/motor assemblies, disassembled plastic ducting and squirrel fans, cleaned out debris from cooling fins, fans and flushed units, reassembled units and sealed ducting assemblies Installed new expansion valves LH and RH 2ea P/N ES26101-4 using new adapters 2ea P/N 1523447-3 with new M83248/1-908 o-rings. Reinstalled evaporator/motor assemblies and reterminated electrical connections. Replaced receiver dryer using P/N ES343029-3 and installed new adapter fittings. Reconnected and torqued all refrigerant lines and fittings. Pressurized Air-Conditioning system with N2 and found no leaks. Evacuated system. Ran RH engine and serviced system with R-12 refrigerant. Air-Conditioning system function check satisfactory.

Cleaned outflow and safety valve seals with alcohol.

Replaced 6ea pressurization belly drain seals P/N CM3575-4, replaced 4ea retainers P/N 5111339-3.

Replaced various nut plates for inspection panels as required. No fuel panel nut plates were replaced at this time.

Removed stinger and examined tail tie-down attach point due to tie down being detached. Secured tie-down with new bolt P/N AN3-6A, washer P/N NAS1149F0363P, and nut P/N MS21044N3. No significant damage noted. Skin deformation minor.

Reformed nose baggage door to open and close with proper clearance.

Secured lower cabin door actuator on lower cabin door.

Secured Lower cabin door interior trim.

Checked all airframe and engine fuel screens and found clean. Reused seals for LH and RH fuel strainer bowls. Leak checks satisfactory.

Cleaned fuel stain from LH wing outboard fuel drain, actuated drain valve and monitored for leaks. No leaks noted.

Replace LH and RH outer fuel cap o-rings and installed new LH and RH required fuel placarding using 2ea P/N 5026015-15 placards.

Disconnected cross feed shut-off cable at fuel transfer valve, cleaned, lubricated, re-secured and reinstalled. System function check satisfactory.

Removed Stall Warning Transmitter P/N 186-8 (No S/N) and installed and overhauled exchange transmitter P/N 186-8 O/H S/N 1337 See FAA form 8130-3 WO# R44741 dated 1-21-2012. On ground function check satisfactory, unit ready for flight check adjustments.

Removed elevator trim actuator, aileron trim actuator and rudder trim actuators, disassembled, cleaned and replaced the following parts: ELEVATOR TRIM: new Removed elevator trim actuator, and on the bound of the rest of the state of the state of the rest of the following parts. ELEVATOR TRI housing P/N 0310296, new drive end bearing P/N 0310297-1, new screw end bearing P/N 0310298-1, new seal P/N MS28775-010, 1ea new snap ring P/N MS16624-1081 and new groove pins 4ea P/N MS35675-13 and 2ea P/N GP3H094x0375-14. RUDDER TRIM: new drive end bearing P/N 0310297-1, new screw 2 of 3

end bearing P/N 0310298-1, new seal P/N MS28775-010 and new groove pins 4ea P/N MS35675-13 and 2ea P/N GP3H094x0375-14. AlLERON TRIM: new bearing P/N MS27641-5 (S3952-5), new seal P/N MS28932C02-05, new Snap rings 2ea P/N MS16624-4112 and new roll pins 2ea P/N MS35675-10. All units assembled using Molykote 33 grease. Alleron and rudder trim tab attach bolts replaced using 2ea new P/N NAS464P3-9 bolts, all 3 trim actuator tab attach bolts installed using new P/N MS17826-3 nuts. All units reinstalled and trim systems rigged per the Cessna 421C maintenance manual.

Repositioned fuel line beneath cabin floor to avoid flap cable chaffing. Noted cable to be clear of flap motor.

Retensioned rudder trim cables per the Cessna 421C maintenance manual.

Removed incorrect cable segments from LH aft elevator trim system and installed OEM cable assembly P/N 520008-43 with new trim control cable master link P/N S2294-25. Elevator trim cable tensions and deflections reset per the Cessna 421C maintenance manual.

Cleaned RH flap return cable and found serviceable after clean-up, reset cable tensions per the Cessna 421C maintenance manual.

Deactivated and placarded "Inop", the propeller synchronization system in accordance with 14 CFR Part 91.213.

Removed RH retractable landing light assembly P/N 45-0148-3 (No S/N) and sent to Quality Aircraft Accessories/Aerospace Turbine Rotables, Inc. Unit was returned with a S/N of 75603 and a work status of "Overhauled." See FAA form 8130-3 WO# WO33276 dated 01-28-2013. Unit reinstalled in RH position, function check satisfactory.

Replaced RH cowling landing light using lamp P/N 4591.

T&B and spare annunciators do not light when tested, checked lamps and found no fault. Troubleshooting revealed the annunciator module had a broken wire for spare power and the turn coordinator wire was not terminated at the turn coordinator. Removed annunciator module and solder repaired broken spare power. Reterminated Pilot's turn coordinator connector. Function check of annunciator module satisfactory.

Replaced Deice light lens P/N 0820501-11.

Inspected tail position light and found lamp filament broken. Checked voltage, has 28 volts at lamp. Replaced tail nav lamp P/N 305.

Pulselite system deactivated and placarded in accordance with 14 CFR Part 91.213.

Rerouted Tach generator wires and connectors in LH and RH engines for better service loop.

Capped and stowed unused wiring in RH engine compartment.

Wiring harnesses for LH and RH cowling lights not secured, have no connector as needed for servicing-maintenance. Installed Mate and Lock harness disconnects and secured wire bundles in accordance with AC 43-13.1b standards. No further action taken.

Replaced Alternator on/off switches 2ea P/N MS24524-23. Replaced Emergency Field switch P/N MS35059-23. Replaced LH and RH field fuses 2ea P/N HKP and fuses 2ea P/N S3037A3. Cleaned rotary selector switch with contact cleaner and adjusted paralleling per the Cessna 421C maintenance manual.

LH "Alternator Out" light remains illuminated with switch on and charging system operational - troubleshooting revealed that the Alternator out sensor was defective. Installed new Sensor P/N 9910094-3 S/N 80181814. System function check satisfactory.

Removed battery box, cleaned corrosion and reinstalled.

Replaced LH and RH Vacuum relief filters 2ea P/N RAB3-5-1.

Cleaned FOD and debris below floor boards as practicable.

Installed 2ea new S2237-3 friction bushings to Pilot and Co-Pilot seat belt.

SIGNED.

ONT

Annual / I certify that this AIRCRAFT has been inspected in accordance with an Annual Inspection and was determined to be in airworthy condition -end-

This aircraft, airframe, or appliance identified above has been repaired and/or inspected in accordance with current Regulations of the Federal Aviation Administration and Manufacturer data and is approved for return to service.

DATE: 4/1/2013

Mike Clingerman Certified Repair Station No. 1HGR058B Work Order: 20121008-1M

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MAKE: CESSNA MODEL: 421C S/N: 421C0158 REG. NO: N5288J WORK ORDER: 20130515-1M Hangar One Avionics Repair Station No. 1HGR0588 2026 Palomar Airport Road Carlsbad, CA 92011 Phone: 760-929-2270 DATE: 6/6/2013 A/C TSN: 5824.5 HOBBS: 1727.1

### **Airframe Entries**

Removed cabin door perimeter seal and horizontal seal, then installed new perimeter seal P/N 9910007-3 and new horizontal seal P/N 9910007-4. All work IAW Cessna 421C Service Manual. Aircraft ready for check flight.

Removed RH engine gauge (3-in-1: Oil PSI, CHT, Oil Temp) P/N C662019-0101 S/N A91290 and installed overhauled exchange gauge P/N C662019-0101RX S/N T2143. Reference Aero-Mach Labs, Inc. FAA Form 8130-3 and Work Order 0406598 dated March 11, 2013 for status work "OVERHAULED." System function check satisfactory.

Cleaned RH engine and performed engine run-up. Suspected leak was residual only. Check again after next flight.

Removed vertical stabilizer pneumatic de-icer ("boot") and both horizontal stabilizer pneumatic de-icers IAW Cessna 421C Service Manual. Installed new vertical stabilizer pneumatic de-icer 1 ea. P/N SMR5070-14 and new horizontal stabilizer pneumatic de-icers 2 ea. P/N SMR5070-11 (RH S/N 1819, LH S/N 1819,). Reference B/E Aerospace Inc. dba SMR Technologies Inc. FAA Forms 8130-3 for Work Order (SMR Order No.) 73859 dated May 24, 2013, Work Order (SMR Order (SMR Order No.) 73158 dated March 1, 2013, and Work Order (SMR Order No.) 61726 dated May 4, 2010. All Status/Work "NEW." Installation of all pneumatic deicers IAW Cessna 421C Service Manual and B/E Aerospace Installation and Maintenance Manual (SMR Report No. 97-33-047 Rev. 10 dated March 26, 2010). System function check revealed tail deice shuttle valve sticking. Removed tail deice shuttle valve P/N 3D2352-01 S/N N970 and sent to B&S for overhaul. Stowed air inlet fitting and wiring and connected tail deice system vacuum source using AN adapters. System remains deactivated per FAR 91.213.

This aircraft, airframe, or appliance identified above has been repaired and/or inspected in accordance with current Regulations of the Federal Aviation Administration and Manufacturer data and is approved for return to service.

DATE: 6/6/2013

SIGNED: David Corish Certified Repair Station No. 1HGR058B Work Order: 20130515-1M

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| DATE  | TOTAL<br>TIME<br>IN<br>SERVICE   | TACH OR<br>RECORDING<br>METER<br>TIME  | DESC<br>SIGNATURE & C   | CRIPTION OF WORK PERFORMED-<br>ERTIFICATE NO. OF PERSON PERFORMING WORK  |   |
| IAKE: CESSNA<br>IODEL: 421C<br>/N: 421C0158<br>EG. NO: N5288<br>/ORK ORDER: 2   | 20130515-1M  |  | Hangar One Avionics<br>Repair Station No. 1HGR058B<br>2026 Palomar Airport Road<br>Carlsbad, CA 92011<br>Phone: 760-929-2270  |  | DATE: 6/6//<br>A/C TSN: 58<br>HOBBS: 17 |
| ilot reports ma<br>evision L date   | TT switch P/N 9<br>arker beacon inc<br>d June 2003, Sy                                     | op. Performed func<br>ystem test okay. C   | wheel. Operational check okay.<br>tion test of marker beacon system pe<br>ould not duplicate the discrepancy.   | r the Garmin GMA340 Audio Installation Manual Document   |   |
|   |  |  | - percenta production   | ertical speed mode." Performed system test of auto-pilot, in   |   |
| nis aircraft, airfi<br>lanufacturer dai<br>ATE: 6/6/2013  |  |  | X   | ccordance with current Regulations of the Federal Aviation Adm   | ninistration and                        |
|   |  | SIGNED<br>David Cor  | ish   |  | ork Order: 20130515                     |
|   |  | Certified F  | Repair Station No. 1HGR058B   | Printed by E   | Bis 3 (datcomedia.c                     |
|   | н  |  |   |  |   |
| MAKE: CESSNA<br>MODEL: 421C<br>S/N: 421C0158<br>REG. NO: N528<br>WORK ORDER:<br>20130703-2M                               | 8J   | Re<br>202<br>Car   | ngar One Avionics<br>bair Station No. 1HGR058B<br>6 Palomar Airport Road<br>Isbad, CA 92011<br>ine: 760-929-2270  | DATE: 9/5/2013<br>A/C TSN: 5831.7<br>HOBBS: 1734.3   |   |
| AMETEK B&S A  | er flow valve P/N 31<br>52-01 S/N N970, ii<br>iircraft Parts & Acco<br>frame, or appliance | in fuselage, IAW Cess<br>cessories FAA Form 8  | na Service Many/al, for operation of emper<br>130-3 and Work Order #R13-02915 dated<br>been repaired and/or inspected in accorda  | aluation / repair, then installed overhauled de-icer flow<br>nage surface-mounted de-ice boots. (Reference<br>June 11, 2013, for Status/Work "OVERHAULED." |   |
| DATE: 9/5/20  |  | SIGNED:  | kr  | Work Order: 20130703-2M  |   |
|   |  | David Corish   |   | Printed by EBis 3 (datcomedia.com)   |   |
|   |  | Certified Repair S   | tation No. 1HGR058B   |  | Second Charles                          |
| MAKE: CESSNA<br>MODEL: 421C<br>S/N: 421C0158<br>REG. NO: N5288<br>WORK ORDER:<br>00130607-2M                              | -  | Certified Repair S   | ation No. 1HGR058B<br>ngar One Avionics<br>air Station No. 1HGR058B<br>3 Palomar Airport Road<br>sbad, CA 92011<br>ne: 760-92-9270  | DATE: 9/5/2013<br>A/C TSN: 5831.7<br>HOBBS: 1734.3   |   |
| MODEL: 421C<br>S/N: 421C0158<br>REG: NO: N5288<br>WORK ORDER:<br>20130607-2M<br>Airframe En                               | BJ<br>Itries   | Certified Repair S<br>Ha<br>Control Control Contro | ngar One Avionics<br>air Station No. 1HGR058B<br>8 Palomar Airport Road<br>sbad, CA 92011<br>ne: 760-929-2270   | A/C TSN: 5831.7  |   |
| MODEL: 421C<br>S/N: 421C0158<br>REG NO: N5288<br>WORK ORDER:<br>20130607-2M<br>Airframe En<br>Replaced cracke<br>Manual.  | BJ<br>Itries<br>Id LH glass heated<br>rame, or appliance<br>Id Manufacturer di             | Certified Repair S<br>Ha<br>Rep<br>202<br>Car<br>Pho<br>d windshield with new  | ngar One Avionics<br>air Station No. 1HGR058B<br>3 Palomar Airport Road<br>sbad, CA 92011<br>ne: 760-929-2270<br>LH glass heated windshield P/N 9910049-5<br>been repaired and/or inspected in accorda                          | 9 S/N 13200H3137. All work IAW Cessna Service  |   |
| MODEL: 421C<br>S/N: 421C0158<br>REG. NO: N5288<br>WORK ORDER:<br>20130607-2M<br>Airframe En<br>Replaced cracke<br>Manual. | BJ<br>Itries<br>Id LH glass heated<br>rame, or appliance<br>Id Manufacturer di             | Certified Repair S<br>Ha<br>Rep<br>202<br>Car<br>Pho<br>d windshield with new<br>e identified above has<br>ata and is approved<br>SIGNED:<br>David Corish  | ngar One Avionics<br>air Station No. 1HGR058B<br>3 Palomar Airport Road<br>sbad, CA 92011<br>ne: 760-929-2270<br>LH glass heated windshield P/N 9910049-5<br>been repaired and/or inspected in accorda                          | A/C TSN: 5831.7<br>HOBBS: 1734.3   |   |
| MODEL: 421C<br>S/N: 421C0158<br>REG NO: N5288<br>WORK ORDER:<br>20130607-2M<br>Airframe En<br>Replaced cracke<br>Manual.  | BJ<br>Itries<br>Id LH glass heated<br>rame, or appliance<br>Id Manufacturer di             | Certified Repair S<br>Ha<br>Rep<br>202<br>Car<br>Pho<br>d windshield with new<br>e identified above has<br>ata and is approved<br>SIGNED:<br>David Corish  | ngar One Avionics<br>air Station No. 1HGR058B<br>5 Palomar Airport Road<br>sbad, CA 92011<br>ne: 760-929-2270<br>LH glass heated windshield P/N 9910049-1<br>been repaired and/or inspected in accorda<br>or return to service. | A/C TSN: 5831.7<br>HOBBS: 1734.3   |   |
| MODEL: 421C<br>S/N: 421C0158<br>REG NO: N5288<br>WORK ORDER:<br>20130607-2M<br>Airframe En<br>Replaced cracke<br>Manual.  | BJ<br>Itries<br>Id LH glass heated<br>rame, or appliance<br>Id Manufacturer di             | Certified Repair S<br>Ha<br>Rep<br>202<br>Car<br>Pho<br>d windshield with new<br>e identified above has<br>ata and is approved<br>SIGNED:<br>David Corish  | ngar One Avionics<br>air Station No. 1HGR058B<br>5 Palomar Airport Road<br>sbad, CA 92011<br>ne: 760-929-2270<br>LH glass heated windshield P/N 9910049-1<br>been repaired and/or inspected in accorda<br>or return to service. | A/C TSN: 5831.7<br>HOBBS: 1734.3   |   |
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| MODEL: 421C<br>S/N: 421C0158<br>REG NO: N5288<br>WORK ORDER:<br>20130607-2M<br>Airframe En<br>Replaced cracke<br>Manual.  | BJ<br>Itries<br>Id LH glass heated<br>rame, or appliance<br>Id Manufacturer di             | Certified Repair S<br>Ha<br>Rep<br>202<br>Car<br>Pho<br>d windshield with new<br>e identified above has<br>ata and is approved<br>SIGNED:<br>David Corish  | ngar One Avionics<br>air Station No. 1HGR058B<br>sbad, CA 92011<br>ne: 760-929-2270<br>LH glass heated windshield P/N 9910049-0<br>been repaired and/or inspected in accorda<br>or return to service.                           | A/C TSN: 5831.7<br>HOBBS: 1734.3   |   |

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| DATE     | TOTAL<br>TIME<br>IN<br>SERVICE                | TACH OR<br>RECORDING<br>METER<br>TIME   | DESCRIPTION OF WORK PERFORMED—<br>SIGNATURE & CERTIFICATE NO. OF PERSON PERFORMING WORK |              |  |  |  |  |
|----------|---|---|---|--------------|--|--|--|--|
|          |   | Aviation 1  | nc  | Sec.         |  |  |  |  |
|          | 7000 Merri                                    | ll Ave. Hanga   | ur A-340 Chino CA 91710   |              |  |  |  |  |
|          | N52881 HO                                     | bbs: 1753 4   | 11/09/2013  |              |  |  |  |  |
| 1996     | ail and filte                                 | r changed bot   | h engines, re-connected loose hose on oil separator on right                            | A New States |  |  |  |  |
|          | engine rem                                    | oved Left tac   | h. Generator, then re-installed with new gaskets and new                                |              |  |  |  |  |
|          | cannon nlu                                    | o after bench   | check done by Precision instruments. Installed new hobbs                                |              |  |  |  |  |
| 3        | meter on pi                                   | lot's instrume  | nt panel and got it wire to master switch circuit per customer's                        |              |  |  |  |  |
|          | request. Re                                   | moved throttl   | e quadrants and center console to access panel lighting                                 |              |  |  |  |  |
|          | rheostats, re                                 | eplaced 3 bro   | ken rheostats with new. Removed left landing light and                                  |              |  |  |  |  |
|          | replaced me                                   | otor, gears an  | d reinstalled back under the wing, replaced wing de-ice light                           |              |  |  |  |  |
|          | bulb, instal                                  | led new LED   | taxi light on nose gear.  |              |  |  |  |  |
|          | Agustin Ce                                    | ndejas AP260  | 06564 delenan   |              |  |  |  |  |
|          | U   | 0   | egt and   |              |  |  |  |  |
|          | Chino   | Aviation  | Inc   |              |  |  |  |  |
|          |   |   | ar A-340 Chino CA 91710   |              |  |  |  |  |
|          |   | obbs: 1765 1  |   |              |  |  |  |  |
|          | Replaced 1                                    | laced main door, nose baggage compartment and wing lockers' locks,  |   |              |  |  |  |  |
|          | Replaced 1                                    | Replaced nose tire and tube and greased nose wheel bearings.  |   |              |  |  |  |  |
|          |   | endejas AP26  |   |              |  |  |  |  |
|          | 1h  | he olis   | 21  |              |  |  |  |  |
|          | t p   | Cean  |   |              |  |  |  |  |
|          | Chino A                                       | viation l   | nc  |              |  |  |  |  |
|          | 7000 Merri                                    | ll Ave. Hanga   | ar A-340 Chino CA 91710   |              |  |  |  |  |
|          | N5288J Ho                                     | -   | 04/10/2014 Aircraft   |              |  |  |  |  |
|          | Troublesho                                    | ot starting iss   | ue at Yuca valley airport, found INOP fuel pump on right side,                          |              |  |  |  |  |
| 1        | de-fueled a                                   | ircraft and rep   | placed electric fuel pump with P.n.2B7-29. Wired newer pump in                          |              |  |  |  |  |
|          | accordance                                    | with Cessna   | service manual. Serviced with fuel and released to service.                             |              |  |  |  |  |
|          |   |   | nd tube and reinforced arm rest on aft left seat.                                       |              |  |  |  |  |
| _        |   | ndejas AP260  |   |              |  |  |  |  |
|          | +   | +   | Ja wegin  |              |  |  |  |  |
|          |   |   |   |              |  |  |  |  |
|          | Chino Aviation Inc                            |   |   |              |  |  |  |  |
|          | 7000 Merrill Ave. Hangar A-340 Chino CA 91710 |   |   |              |  |  |  |  |
| in (123) | N5288J Hobbs: 1765 12/03/2013                 |   |   |              |  |  |  |  |
|          | Replaced                                      | INOP hobbs  | meter on heater, Old hobbs meter was reading 615.9 hours. On                            |              |  |  |  |  |
| - See a  | heater an                                     | nd New hobbs  | meter hours 0000. complied with AD-2000-01-16 par. B by                                 |              |  |  |  |  |
| 6.65 (P  | inspectio                                     | on of exhaust   | system per customer's request.  |              |  |  |  |  |
| -        | Agustin                                       | Cendejas AP   | 2606564 Mandalan  |              |  |  |  |  |
|          | **  |   | upp - constraint  |              |  |  |  |  |
|          |   |   |   |              |  |  |  |  |
|          |   |   |   | -            |  |  |  |  |
|          |   | SUB-TOTAL th  | lis page  |              |  |  |  |  |
|          |   | TOTAL-Carry   | forward to next page  |              |  |  |  |  |
|          |   | and the second se |   |              |  |  |  |  |

DATE 6/18/2014 A/C TSN: 5888.7 HOBBS: 1791.3

MAKE CESSNA MODEL: 421C S/N: 421C0158 REG. NO: N5288J WORK ORDER: 20140505-2M



Hangar One Avionics Repair Station No. 1HGR058B 2026 Palomar Airport Road Carlsbad, CA 92011 Phone: 760-929-2270

### Airframe Entries

Completed Annual Inspection IAW the Cessna Inspection Check list as applicable and FAR 43 Appendix "D"

Completed General Inspection of all aftermarket installations not listed with specific instructions for continued airworthiness---Inspected IAW AC43-13-1B Chapter 5 Section 2

CW AD# 84-26-02 (Paper Induction Air Filter) for LH and RH engines by replacement of Induction Air Filters 2ea P/N P10-8421. Replacement mandatory ea 500hrs.

CW AD# 2000-01-16 (Exhaust System Inspection) IAW para b -- no defects, due again the latter of 50 hours or 30 days.

Performed visual inspection on Halon fire extinguisher and reweighed. Weight: 4lbs 12.5oz

Replaced fire sleeve on Right Engine breather with 3 feet Aero quip AS102-12.

Replace vacuum hoses on Right Engine with 8 feet of MIL-H-6000 hose, 8 feet of Aero Quip AS102-16 fire sleeve and 6 Breeze 200-12H clamps.

Removed and replaced broken exhaust tail pipe bracket with new P/N 5155186-1 on RH engine, outboard location.

Installed new cotter pin and clevis pin P/N MS20392-2C11 on RH alternate air control at air filter assembly.

Rerouted RH propeller control cable and secured to avoid exhaust.

Replaced adel clamp for LH prop cable with new P/N MS21919WCH12

Freed stiff LH and RH alternate air controls

Verified Propeller synchronization system still properly deactivated and placarded in accordance with FAR 91.213.

Replaced Main Landing Gear Hydraulic System Filters 2ea P/N AN6235-2A and 2ea O-rings P/N MS28775-222.

Removed and replaced RH brake phenolic shim with new P/N 068-03100

Installed new Cam loc fastener P/N S3275-102R in nose baggage.

Installed Flush head rivet P/N CR3214-4-2 in cabin door with sealant and painted to match finish.

Replace LH and RH inner and outer fuel cap o-rings with new 2ea P/N MS29513-010, 2ea P/N MS29513-338.

Installed new center button and circlip on LH fuel selector using selector kit P/N B9-61-1

Removed broken lanyards from LH, RH and Aux fuel caps and replaced with new 3ea P/N MS27380-1

Inspected LH and RH main fuel strainers, cleaned and reassembled with new seals 2ea P/N MS29513-233

Cleaned corrosion near base of 3 GPS antennas on cabin roof, sealed antennas, and painted white to match.

Cleaned corrosion near base of Comm antenna on cabin roof, sealed antennas, and painted white to match.

Capped and stowed unused wiring in LH engine compartment.

Removed and replaced static wick on LH wing with new 1ea P/N 16165

Removed and replaced static wicks on RH wing with new 2ea P/N 16165

Radio lights deactivated and placarded in accordance with 14 CFR Part 91.213.

Repaired right tachometer wiring by installation of new MS3456L10SL-3s connector and M85049/52-1-10N strain relief to RH tachometer generator position.

LH landing light deactivated and placarded in accordance with 14 CFR Part 91.213.

Verified Pulselite system still properly deactivated and placarded in accordance with FAR 91.213.

Replaced LH and RH Vacuum relief filters 2ea P/N RAB3-5-1.

Vacuum / Replaced central vacuum filter P/N AM103535-IA

Removed RH evaporator cover, trimmed to fit properly, and secured using new screws.

Removed and replaced cabin door aft cable with new P/N 5111508-71CR and Fwd cable with new P/N 5111561-27CR using new roll pin in fwd bracket 1 ea P/N NAS561CF6-18.

Removed oxygen cylinder, sent for sublet hydrostatic testing (reference S.J. O'Brien Ent., Inc. work order HY 39723 dated 5/29/2014), then reinstalled IAW Cessna Service Manual. Next Hydro due 5/2019

Installed new O2 overboard discharge indicator disc from assembly P/N CM3578-1

SIGNED

Annual / I certify that this AIRCRAFT has been inspected in accordance with an Annual Inspection and was determined to be in airworthy condition --end--

This aircraft, airframe, or appliance identified above has been repaired and/or inspected in accordance with current Regulations of the Federal Aviation Administration and Manufacturer data and is approved for return to service.

DATE: 6/18/2014

Mike Clingerman Certified Repair Station No. 1HGR058B

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Work Order: 20140505-2M

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| IAKE: CESSNA<br>IODEL: 421C<br>/N: 421C0158<br>EG: NO: N5288J<br>/ORK ORDER: 20 | 140505-2M  |   | Hangar One Avionics<br>Repair Station No. 1HGR058B<br>2026 Palomar Airport Road<br>Carisbad, CA 92011  | DATE: 6/18/2014<br>A/C TSN: 5888.7<br>HOBBS: 1791.3   |
| vionics Entr  | ios  |   | Phone: 760-929-2270  |   |
|   | ELT has been in  | spected in accord   | dance with the requirements of FAR Section 91.207(d)   | ) and was found to be in satisfactory condition. Battery  |
| e-terminated ca   | able at ELT with   | new RJ11 conne  | ctor. Performed ops check via ELT remote no discrep  | pancies noted at this time.   |
| his aircraft, airfra<br>anufacturer data  | me, or appliance<br>and is approved                      | e identified above h<br>I for return to servi                                     | has been repaired and/or inspected in accordance with cu<br>ce.  | rrent Regulations of the Federal Aviation Administration and  |
| ATE: 6/18/2014  | Ļ  | SIGNED  | . 1911   | Work Order: 20140505-2M   |
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| KE CESSNA<br>DEL 421C<br>421C0158   | T  | ONV   | Hangar One Avionics<br>Repair Station No. 1HGR058B<br>2026 Palomar Airport Road  | DATE: 6/18/2014<br>A/C TSN: 5888.7<br>HOBBS: 1791.3   |
| ORK ORDER: 201  | 40618-2M   |   | Carlsbad, CA 92011<br>Phone: 760-929-2270  |   |
| ted Feb 12, 201<br>lance negligible<br>publeshot No TI<br>111 to LH engin       | 4 completed by<br>T indication on a<br>e and left the su | Hartzell Engine T<br>JPI indicator on R<br>wapped operation<br>dentified above ha | H engine by swapping TIT probes LH to RH and foun<br>al TIT probe on the RH engine, ops check satisfactor<br>s been repared and/or inspected in accordance with curr | nator P/N ALV-9610, S/N H-O020366, (see FAA form 8130-3<br>nber H-O020366). Ops check satisfactory. Weight and<br>and TIT probe to be inoperative. Installed new TIT probe P/N<br>y.<br>rent Regulations of the Federal Aviation Administration and |
| nufacturer data a   | nd is approved f   | or return to service  | Ch   |   |
| TE: 6/18/2014   |  | SIGNED  | 10/  | Work Order: 20140618-2M   |
|   |  | Mike Clinge<br>Certified Re   | man<br>pair System No. 1HGR058B  | Printed by EBis 3 (datcomedia.com)  |
| WART SALE   |  |   |  |   |
|   | 7000 Me<br>N5288J<br>Installatio<br>Faa 337              | Aircraft ho<br>on of Micro<br>form filed,   | Innary A-340 Chino CA 91710<br>bbs: 1825 02/13/2015<br>o vortex generator kit on wings per<br>no change on airspeed indicator of<br>6606564IA                        | er STC. # SA5193NM<br>or systems.   |
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|   | 7000 M<br>N5288J<br>Inspecte<br>Opened<br>jacks an<br>bearings<br>replaced<br>balance<br>on exha<br>for reco<br>I certify<br>Inspecti               | Aircraft he<br>ed this aircr<br>inspection<br>d performe<br>s, tested and<br>d, is next du<br>weight, no<br>ust system,<br>gnition ligh<br>that this Ai<br>on and four | #33 Hangar A-340 Chino CA 91710<br>obbs:1805 05/17/2015 Aircraft TT:<br>aft in accordance with FAR 43 Appendix D for annual inspection<br>panels, lubricated control cable pulleys and rod ends, set aircraft on<br>d landing gear tests, serviced hydraulic reservoirs, greased all wheel<br>d inspected ELT in accordance with FAR 91.207, ELT battery<br>the December 2017, complied with AD82-26-05 by inspection of rudder<br>cracks found, complied with AD 200-01-16 by inspection per par©,<br>replaced recognition light retainers on cowlings, replaced flasher box<br>its. |
|   | Agustili  |  | AP2606564IA Color   |
| >   | S   |  | NOTES   |
| FAR   | 411-413   | ~  | Fels. 23, 2016 APR 411 + 413 Try  |
|   | 7124  | -  | Approvable Avionics Under-<br>Nr. Or 2487.  |
| -   |   |  |   |
| - 7<br>N<br>Ir<br>O<br>ja<br>be<br>re<br>ba<br>or | 000 Merrill<br>5288J Air<br>spected this<br>pened insp<br>cks and per<br>earings, tes<br>placed, is r<br>alance weign<br>exhaust sy<br>certify that | craft hobbs<br>is aircraft in<br>ection pane<br>rformed lan<br>ted and insp<br>next due De<br>tht, no crack<br>ystem, repla<br>this Aircraft                           | Hangar A-340 Chino CA 91710   |

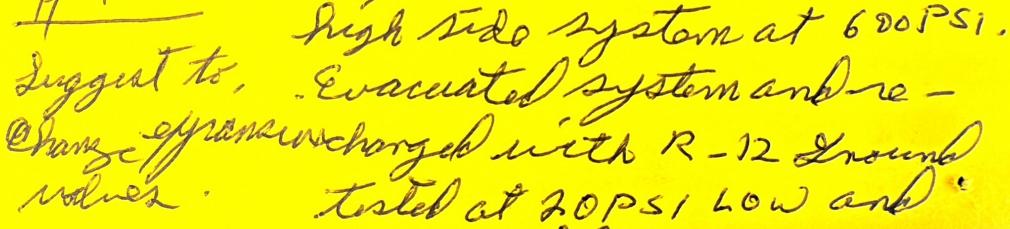
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Executed N/c system. Tourf





210 PSI high 1 Don Alhson ALP 1481667 I-A,

Affordable Avionics Inc. 7000 Merrill Ave # 18 Chino, CA 91710 8AAR592Y LOG ID# 2317 23-February-2016 WO# 2482 AC TT 171.0 Pg 1/1 S/N 421C0158 CESSNA 421C N5288J

1.Removed existing GTX 327 Transponder. Installed GTX 330 ES Transponder. FAA form 337 filed. Weight and Balance and Equipment list updated. Post installation ground check carried out satisfactory.The Transponder System was inspected and was found to comply with FAR 91.413 and Appendix F of FAR 43. Test Equipment used : ATC 601-2 S.N. 6616 Date of Cal 01/16/2016 2.Upon inspection found heading flag does not go away on Glideslope indicator. Swapped connectors from indicators 1 & 2 leading form gyros for troubleshooting purposes. When connector harness for Gyro 1 was switch to Glideslope indicator 2, indicator worked. Connector harness from gyro 2 moved to indicator 1, indicator didnt work. Suspect issue with wiring harness leading from Gyro 2 or Gyro

2 itself. Recommend customer have gyro #2 benchchecked. 3. Upon inspection found Static leak of 300 Ft per minute at 4,000 Ft. Isolated leaks to CoPilots side. All static instruments had push on hose without clamps. Installed six hose clamps and checked for leaks.

The Altimeter(s), static system, Transponder(S) and encoder were inspected upto 30,000 Ft. and were found to comply with FAR 91.411 and 91.413 and Appendix E and F of FAR 43. Test Equipment used DPST-8000M S.N. 040115-96 Date of Cal 06/05/2015

ATC 601-2 S.N. 6616 Date of Cal 01/16/2016

Transponder P.N. GTX 330ES S.N. 84143078

Fame W. Monin 02/23/2016 Signed Date

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|      | 7000 Merr<br>N5288J H<br>Replaced I  | lobbs:1878.6.  | Hangar A-340 Chino CA 91710and 191.0 on panel.12/14/2016ump P.N. 2B7-29 on right side.   |  |
|      |  |  |  |  |
| 7    | 2000 Merr<br>N5288J H<br>Replaced y<br>Replaced lo<br>over on co   | obbs: 1886 1<br>acuum pump<br>eft main tire a  | Angar A-340 Chino CA 91710<br>/10/2017 Airframe<br>on right engine with P.N. AA441CC Serial no. OBSP08<br>nd tube. Removed magnetic compass to Re covered plastic<br>hield with new fabric material, then re installed.  |  |
|      |  |  |  |  |
|      |  |  | 11, in .   |  |
|      | 7000 Merril<br>N5288J Ai<br>Inspected th<br>Opened insp<br>jacks and po<br>bearings, te<br>replaced, is<br>balance we<br>on exhaust<br>I certify tha<br>Inspection | rcraft hobbs: 192<br>nis aircraft in acc<br>pection panels, h<br>erformed landing<br>ested and inspecto<br>next due Decem<br>ight, no cracks for<br>system,<br>at this Aircraft ha | gar A-340 Chino CA 91710<br>20 07/09/2017 Aircraft TT:6017.4<br>Fordance with FAR 43 Appendix D for annual inspection<br>ubricated control cable pulleys and rod ends, set aircraft on<br>g gear tests, serviced hydraulic reservoirs, greased all wheel<br>ed ELT in accordance with FAR 91.207, ELT battery<br>aber 2017, complied with AD82-26-05 by inspection of rudder<br>bound, complied with AD 200-01-16 by inspection per par C,<br>as been inspected In accordance with an Annual<br>borthy at this time. |  |
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Page No. TOTAL TACH OR DATE TIME DESCRIPTION OF WORK PERFORMED-RECORDING IN METER SIGNATURE & CERTIFICATE NO. OF PERSON PERFORMING WORK SERVICE TIME TOTAL brought forward from previous page AIR IMPRESSIONS, INC. 7929 Karl May Dr. Waco, Texas 76708 CRS No. AK6R541N Date: 07/20/2017 Hobbs: 1928.4 Cessna S/N 421C 0158 N5288J On above referenced aircraft, troubleshot air conditioner system and checked for Freon leaks, no leaks found. Serviced air conditioner system. Checked operation of system and found both expansion valves have been updated to the new manually adjustable versions without adjustments being made. Performed expansion valve adjustment procedures. Air conditioner system operational tested satisfactory after adjustments made. Repaired plastic air conditioner duct behind co-pilot seat at sidewall location. Repaired cabin door to allow for proper step deployment by installed new link assembly P/N 5111508-63. (See 8130-3.) Adjusted elevator servo cable tensions. Adjusted all landing gear struts with nitrogen. Fabricated and installed new lower entry door forward lower cable attachment due to excessive wear. Aircraft is approved for return to service for maintenance performed. Pertinent details on file under work order 4496. Air Impressions, Inc. CRS# AK6R541N Advantage Aviation, Inc. 7000 Merrill Ave. Box 20 Hangar A-240 Chino, CA 91710 VZKR081Y LOG ID# 2672 17-November-2017 WO# 7851 AC TT 261.0 HOBBS 261.0 CESSNA N5288J S/N 421C0158 421C Pg 1/1 Removed from aircraft: GNS 430w, GNS 530w, VSI, HSI System, Turn & Bank, Inverter, GDL 49, Radar System, M811B, Airspeed Ind., EDM 760, Attitude Gyro, Directional Gyro, Engine Gauges, Shadin Fuel Flow, Altimeter Installed in aircraft: G600 System (QTY 2), GTN 750, GTN 650, GTX 345, WX-500, GTS 800, GWX 70, EDM 960, Flightstream 510, New Flight Panels I CERTIFY THAT ENCODER, ALTIMETER, AND STATIC SYSTEM TESTS REQUIRED BY FAR 91.411 HAVE BEEN PERFORMED AND FOUND TO MEET THE STANDARDS OF FAR PART 43 APPENDIX E. THE ALTIMETER HAS BEEN TESTED TO 30,000 FEET. PERFORMED ATC TRANSPONDER TEST AND INSPECTION. UNIT FOUND TO COMPLY WITH FAR 91.413 AND APPENDIX F OF FAR 43. For more information, see FAA form 337. Dated 11/17/17 Mark Krueger VZKR081Y SUB-TOTAL this page TOTAL-Carry forward to next page

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