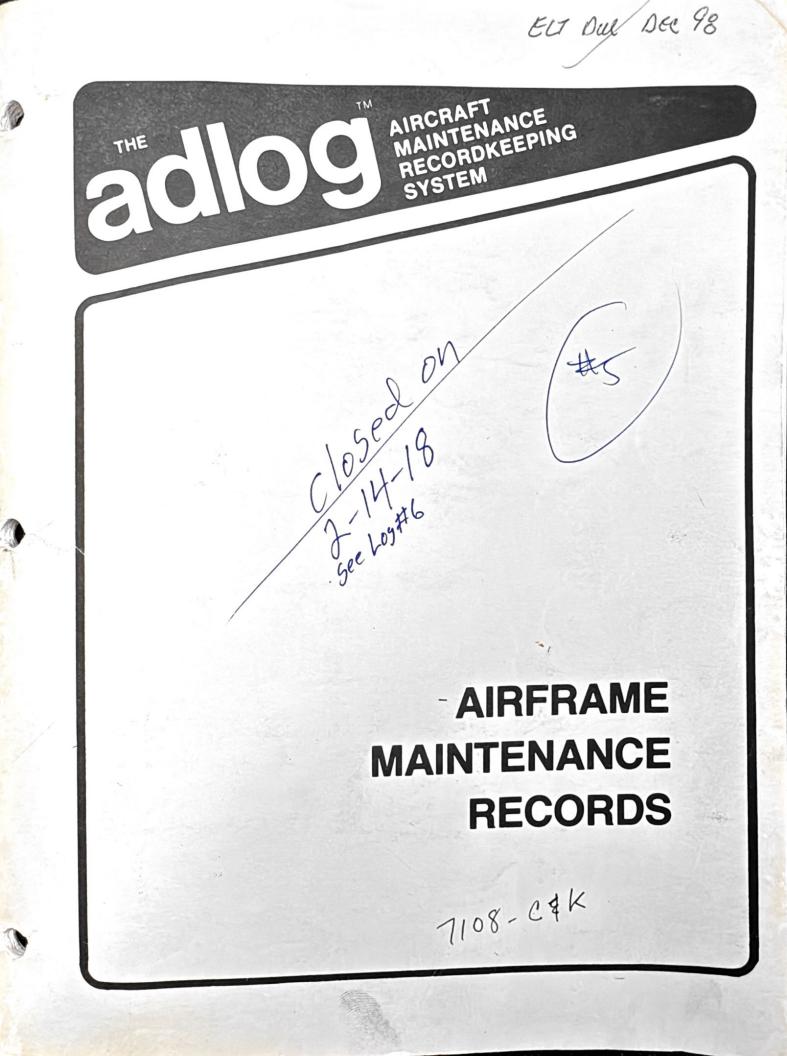
N5288J 1976 Cessna 421C

Airframe Logbook#5 of 6 1997 - 2018

MSN: 421C-0158



Prepared by the worldwide aviation specialists at RidgeAire, Inc.





AIRFRAME MAINTENANCE RECORDS

Log No. 5 Aircraft Registration No.__ 5288ン Aircraft Mfg. CESSNA Model 421C Serial No. 4216-0158 Engine Mfg. CONTINENTAL Model GTSIO - SZOL Serial No. 6/4 245802-R Engine Mfg. CONTINENTAL Model GTSIG - SZO-L Serial No. 8/4 608108 Propeller Mfg. MC CAULEY Model 3FF32C501-A Serial No. 4/H- 764541 Hub Serial No._ Hub Design No. Blade Design No. 904MB-0 Blade Serial No's. Propeller Mfg. MCCAULEY Model 3FF 326561-A Serial No. PH 820823 Hub Serial No.___ Hub Design No._ 90 UMB-0 Blade Serial No's. Blade Design No.

(All applicable information must be filled in)



Page No. DESCRIPTION OF WORK PERFORMED-TACH OR TOTAL RECORDING SIGNATURE & CERTIFICATE NO. OF PERSON PERFORMING WORK TIME METER DATE IN SERVICE TIME TOTAL brought forward from previous page 12/24/97 Permanent airframe Log entry for C421 N5288J Hourmeter: 748.6 ELT s/n 161844 inspected and tested per FAR 91.207d. Battery expires 12/98 "G" switch and signal strength satisfactory. William C. Kunder A&P 552949948 0 W/O 1024 6133 Freeport Blvd. Executive Air Maintenance 916\428-6345 Sacramento, CA 95822 12-29-97 4847.3 7.49.7 RT MAIN TIRE + REPALLED BEARINGS ROTATER RT NEN TUBE MAIN TIRE INSTALLED INSTALLED BODOVEAR 650×1078 DLY TIRE LT MAN SERVICED 111 NOSE aur AP2020357 1-72-98 760,3 VR286 FAA/PMA VOLTAGE REGULATORS 1) INSTALLED 2 AD SUSSED TO AND SPEC AP 202085 3-6-98 4878.4 780.8 RESECUCED AT WING FUEL TANK DANEL SCREWS REMOVED + REINSTALLED BOTH Pase FOR OVERHAUL REPAIRED LT COWL FASTENERS AS NSEDED INSTALLED NEW LT PROP DEICER BRUSHES (5 INSTALLED OUBRHAURED HEATER FUEL VALUE CLEARNED UNDER CABIN FLOOR 6 D REMOVED + REINSTOLLED CABIN FIRE EXTINGUISHER FOR LERTIFICATION & RECHARGE (NOTE UNIT HAD LOW PRESSURE AND SUB-TOTAL this page WAS NOT DISCHARGED IN ACET TOTAL-Carry forward to next page CONT

Page No. ___ TOTAL TIME TACH OR RECORDING DATE DESCRIPTION OF WORK PERFORMED 194 METER SERVICE SIGNATURE & CERTIFICATE NO. OF PERSON PERFORMING WORK THAT TOTAL brought forward from previous page 3-6-98 4878-4 780,8 3 SUUNIS GEAR & REREGEMED EMERGENCY BLOW DOWN OPS CHELL NORMAL SERVICED WITH 102 (MSTALLED 2 B3-5-1 VALVUM RELIEF FICTERS I CERTICY THAT THIS AIRCRAFT HAS BEEN INSPECTED IN ACCORDANCE WITH A CESSINA PROCEESSIVE CARE OPERATION # 2 AND IS PPPROKED FOR DETURN TO SERVICE Warnit fall AP 50 50 57 74 2-13-98 785.4 INSTALLY NEW TALK GENERATOR LT ENGINE P/N 22, A667 Wan's fiftely AP 2020557 INSTALLED OVER HAULED LT ALTERNATOR 796.7 4-1-98 R/N ALV 95103 Dand fipal AP 2020557 61-98 4917,2 819,6 () "JALLED ACRT TO FACILITATE LANDING CEAR WORK 3 INSTALLED NEW 650-10-10RY FUGHT CUSTON TIRE RT MAIN CLEANED + GREASED WHEEL BEARINES INSTALLED ONE NEW NESE WHEEL BEARING CLEANIED OF REGREASED NOSE BEARINGS RELINED RT BRAKE & RESEALED SLAVE CYLINDER E David Atalf AP2020357 SUB-TOTAL this page TOTAL-Carry forward to next page

DATE	TOTAL TIME IN SERVICE	TACH OR RECORDING METER TIME	DESCRIPTION OF WORK PERFORMED— SIGNATURE & CERTIFICATE NO. OF PERSON PERFORMING WORK
		TOTAL bro	ught forward from previous page
7-10-98	4925.0	827.4	2.6.431 (19,3814 780,8)
		Ø	REMOVED + REINSTOLLED LEFT ALTERNATOR FOR
	1.1	2	REPAIR
8	1943	Z	INSTALLED NEW NYD SUETIN NOSE LT ENG BAY
0.03	Service .	3	CLEAND HYD WENT MARVE
ar enable	1948 - S.	Ð	INSTALLED NEW AUTO DILOT DISCOUNCET SWITCH
	Sec. 5.15	S	INSTATUED NEW BRACKETT ELEMENTS
		6	CHANGED OIL & FILTER SERVICED W 15253
		3	WASHED 2 LUBED ENGINE & CONTROLS
		F	PLEANED, GAPDED & TESTED SPARE PLUGS
21.50	1.7 6	- Cansi	I CERTIRY TURT THIS AIRCRAFT HAS BEEN
			INSPENTRO IN ACCORDANCE WITH & CESSNA
	1.	~	PROGRESSIVE CARE OPERATION #3 AND IS
14			APPROVED FOR RETURN TO SERVICE
6.	T from Sing	NEL IN C	86-27 ING LXIN Wandfishel
	himan and	et i	HTIM BOUNDAOODA NI DETOERNI VIETOES SOZIE AN UNDERCED NI ACCORDINCE WITH
			NEER SAN THAT THAT Y THAT Y THAT AND I
7-10-98	49,25.0	827.4	->>><==================================
11010	112510	021.7	
S	A. Mark	C	DEFURLED AIRCORATS, WEIGHED AIRCRAFT
N. S. Mar	SILPS	SYX GUA	DE FURLED AIRCORATS, WEIGHED AIRCRAFT IN HONCER, COMPUTED NEW W/B INFORMATION Namit / Pall AP 20204 57
	<u> </u>	1.2. 1.	Name fifal
			AP 20204 57
		-TANKS	DIALCANCE SPEARIN WALKARE
	-364	1 6393	1 Commences and and water white
		an St.	The ment of the second of the second of the
CONTRACTORY !!	19. 14	1012	The state the price the part of the
		SUB-TOTA	L this page

Page No. TOTAL TACH OR RECORDING DATE DESCRIPTION OF WORK PERFORMED-IN SIGNATURE & CERTIFICATE NO. OF PERSON PERFORMING WORK TIME TOTAL brought forward from previous page 4953.8 10-2-98 865.2 SERVICED FUEL FILTERS REDLOCED RUSTY + 2 WORN SERENS FOR INSPECTION PANELS AS REGD. MAN SOM SOM SOM AND MAN THE (3) LUBED HINCES PULLIES + BEARINGS SERVICED TIRES BRARBS & BATTERY I CERTIRY THIS AIRCRAET HAS BEEN INSPECTED THAT N ACCORDANCE A CESSINA PROGRESSIVE CARE OPERATION WITH RETURN TO SERVICE 15 DODROVED FOR AND Name /Atal AP 2020 457 IA 12-4.98 8710 4968 800 TO B OVERUAULED TACHOMETER INSTRUMENT HEAD D IN STALLED I CERTIFY THAT THIS ELT HAS BEEN SER YELLOW JAG INSPECTED IN ACCORDANCE WITH FAR91.207(D) ATTERY A12020957 DETERMINED Malino VHTan ALLAN MARKA 2-1-99 4997.7 900.1 BRANES. BATTERY TIRES D SERVICED P. JOJ HERT, + STOL Name Askel Systems CHELLED AP2020557 SUB-TOTAL this page TOTAL—Carry forward to next page

ige No.	100		
DATE	TOTAL TIME IN SERVICE	TACH OR RECORDING METER TIME	DESCRIPTION OF WORK PERFORMED— SIGNATURE & CERTIFICATE NO. OF PERSON PERFORMING WORK
		TOTAL brou	ught forward from previous page
6-27.99	50463	948,7	12 Tax 1 Edit to a set
			USIN TAIL LIGHT LAMP
2	RESEA	LEO T	HE FOLLOWING ITEMS NOSE GEAR ACTUATOR,
	RIGHT	MAIN (EAR ACTUATOR, NOSE GEAR UPLOCK ACTUATOR
	AND	RIGHT	MAIN UPLOCK ACTUATOR.
Ð	SWON	6 68	AR CHELKED FOR LEAKS PERFORMED CANDING
2. ¹ 63.	GEAR	Bron	DOWN TEST & SERVICED SYSTEM
B	INSTAL	LED I	VEN BEARING RUDDER TRIM HORN
S	REPAC	KED 1	WHEEL BEARING INSTALED ONE NEW SEAL
	IN NO	SE N.	HEEL
G	INSTA	4120	NEW PRESSURAZATION DUCT LT WING
Ø	REPAIR	LED I	EICE LIGHT NEW LAMP
(8)	RELIN	es E	OTH BRAKES AND ROTATED BOTH TIRES
Ð	SERVI	CED 1	AC System
Ô	REMOV	02	REINSTALLED FIRE EXTINGUISHER FOR SERVICE
Ð			OG FOR AD'S
V.Z.	I CEI	TIPY THAT 1	HIS ELT HAS BEEN I CERTIFY THAT THIS AIRCRAFT
11-11	INSP	ected in a	ACCORDANCE WITH ACCORDANCE WITH AN
Sp file		T DUE A	
1.19	- 1948		AIRWORTH CONDITION Name Africk AZ0209574P
			npocosa na
0-26-99			All and the second second second to a
0	INSTAL	100	NEW CABIN DOOR SNUBBER TN 5211369-1
	OPS	OHErry	NORMAL
		Reca	
Concella Conce			Naut /Atrif AP 20205 57
		SUB-TOT	AL this page
		TOTAL-0	Carry forward to next page

Page No. TOTAL TACH OR TIME RECORDING DATE DESCRIPTION OF WORK PERFORMED-IN SERVICE SIGNATURE & CERTIFICATE NO. OF PERSON PERFORMING WORK TIME TOTAL brought forward from previous page 11-2299 5094.9 996.6 6-00 5116.0 1618 SERVICED TIRES BRAKES, + BATFERY LUSED CONTROLS + Kitor CI 18ATERS LIGHTS AP2020957 12-10-99 51 05.7 1008,1 (INSTALLED FALTORY NEN FUEL TRANSFER LT WING LOCAER P/N 476411 Pump 25 CD 5196 9 (2) CHECKED 9 OPS Y NO LEACS OPERATION TOR NORMOL APZOZOG 12-29-99 5618402-3 BOOT DEICER JIMER FACTORI NEW INSTALLED DEICER PRESSURE HOSE FROM 3 ENGINE INSTALLED 27 PUMP TO VALVE CONTROL DEICER (3 VALVY CLEANED 17 For HRORER OPERATION CHECEED ISTEM AP2020957 (MASSARD) HINGS 1-13-00 RESEALED NOSE SERVICED WITH STRUT ALFT JAURED \bigcirc FROM JACT REMODED ALFT MIL 5606 AP2020 1712 I REPUCTALE SUB-TOTAL this page TOTAL-Carry forward to next page

Page No.

DESCRIPTION OF WORK PERFORMED-TACH OR RECORDING TOTAL SIGNATURE & CERTIFICATE NO. OF PERSON PERFORMING WORK TIME DATE METER IN SERVICE TIME TOTAL brought forward from previous page 2-6-00 5116.0 1018.4 DINSTRUCED & NEW 650X10 FLIGHT CUSTOM TIRES BEARING REPAURED WHEEL INSTALLED NEW SEAL & REDALLED NOSE WHEEL BEARINGS (2)SWUND LANDING GEAR Dand PAtel AP 2020357 9-25-00 5146,9 1049:3 REMOVED ORIGINAL ELT AND INSTALLED AK450 ELT (D) FABRICATE WEN MOUNT FOR BUT AND INSTALLED REMOTE SWITCH & INDRATOR. WIRING FOR REMOTE ROUTED THRU AN EXISTING HOLE & RESEALED SWITCH WAS AT REAR PRESSURE BURK HEAD 2) RESECURED LT BAGGAGE DOOR STAY JALLED ACRT, SWUNG LANDING BEAR, DERFORMED (3)BLOW DOWN TEST, INSTALLED 2 NEW EMERGENCY SYSTEM FILTERS, SERVICED WITH MIL 5606 & NE (4) RT MAIN GRAD, AFT OF GRAD, SKIN PATCH NORN AND LOOSE. INSTALLED LARGER PATCH (5) INSTALLED NEW LT GEAR DOOR HINGE PIN (IN STALLED NEW AN SOF- 8R-8 SCREWS LT GEAR DOOR INBOARD HINGE HALF DINSTALLED NEW FAA | PMA WHEEL BEARING LT MAIN P/N LM29749-20629 + LM29710-20629 (2) RESECURED SEVERAL RIVETS LT FLAD OR REPLACED 19 RESECURED LT WING TIP TRAILING EDGE ABOUT I NEAR AILERON SUB-TOTAL this page CONT 7 TOTAL-Carry forward to next page

Page No. _ TOTAL TACH OR TIME DATE RECORDING DESCRIPTION OF WORK PERFORMED-IN METER SERVICE SIGNATURE & CERTIFICATE NO. OF PERSON PERFORMING WORK TIME 5146.9 TOTAL brought forward from previous page (10) RESECURED REAR CABIN DOOR CABLE INSPECTED ALLARON, RUDDER, & ELEVATOR TRINK TAB FREERAY PER BERUILE MANUAL ALL ARE WITHIN LIMITS REMOVED BUT FLOW VALUE AND CLEANED PER INSTRUCTION (12) ON VALVE IN STALLEO 113 NEW RIGHT SIDE OVERBOARD EXHAUST TAIL PIPZ WITH NEW BELLOWS & REIBOW (ID REMOLEO AT TURBO W/E REPAR BY FOR KNISLEY SEE TAG (15 INSTALLED 2 NEW OVERBOARD BELLOWS CLAMPS (16) INSTALLED NEW WASTE 24TE DRAIN HOSE LT BUG 17 INSTALLED NEW HOSE LT ENGINE FUEL POMP TO METERINC BODY (18 INSTALLED OVERHAULED WASTE BATE ACTUATOR LT ENG 119 INSTALLED NEW HOSE ENG TO OIL FILTER ADAPTER RIGHT ENGINE 2 NEW BRACKETT AIR CLEANER ECEMENTS 20) INSTALLED HEATER FUEL FILTER & SPARK PLUG 121 CLEANED BOTH ENGINE FUEL STRAINERS 22 CLEANED NEW EGT PROBE IT ENGINE INSTALLED SEZ Æ4 2 DA AD LISTING FOR I CERTIFY THAT THIS AIRCRAFT SILL CERTIFY THAT THIS ELT HAS BEEN INSPECTED IN ACCORDANCE WITH HAS BEEN INSPECTED IN ACCORDANCE FAR91.207(D) WITH AN AMMALINSPECTION AND WAS NEXT DUE DETERMINED TO BE IN AIRWORTHY CONDITION SUB-TOTAL this page TOTAL-Carry forward to next page

Page No. TACH OR DESCRIPTION OF WORK PERFORMED 10TAL TIME RECORDING SIGNATURE & CERTIFICATE NO. OF PERSON PERFORMING WORK DATE METER IN TIME SERVICE DATE: 08-23-02 N5288J S/N: 421C-0158 W.O. #: 12332 MODEL: 421C TOTAL TIME: 5154.4 **TACH TIME: 1056.8** CPAFT MAINTENA AIRFRAME LOG FAA APPROVED REPAIR STATION # JVR016L 2980 AIRWAY AV,, COSTA MESA, CA 92626 CIRUS DESIGN AUTHORIZED REPAIR STATION TEL: 714-433-2275 AND 714-433-2276 THE AIRCRAFT AND / OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED AND INSPECTED IN ACCORDANCE WITH CURRENT FEDERAL AIR REGULATIONS AND WAS FOUND AIRWORTHY FOR RETURN TO SERVICE, PERTINENT DETAILS OF THE REPAIR / REPCACEMENT ARE ON FILE AT THIS AGENCY. SIGNED: in 08-23-02 DATE: C/W an ANNUAL INSPECTION as per Cessna 421C Annual Inspection Guide. C/W Yearly Inspection of ELT as per 91-207(d) battery replacement due at 03-04. Installed a Ram Overhauled engine into the Left side of this aircraft on this date. C/W AD200-01-16 Exhaust Cracks, no cracks found at this time, AD next due at 1106 Tach. R & R Oxygen Bottle for Hydrostatic check done by Fire Stop. Installed JPI Engine Monitoring System P/N EDM760 as per STC# SA00729SE. FAA form 337 made out this date. Installed new Instrument Filter P/N AM1035351A. Installed new Air Filter on Right Engine P/N BA3. END 9-16-02 Seridar 250 3 3 7 SUB-TOTAL this page TOTAL-Carry forward to next page

DATE	TOTAL	TACH OR		-
DATE	TIME IN SERVICE	RECORDING METER TIME	DESCRIPTION OF WORK PERFORMED— SIGNATURE & CERTIFICATE NO. OF PERSON PERFORMING WORK	
		TOTAL brou	ght forward from previous page	
				-
100 C	N5288J		8	
	S/N: 421C-	0158		
200	MODEL: A	121C	DATE: 12-18-02	
	inch m	TE: 1108.0	W.O. #: 12600 TOTAL TIME: 5205.6	
			AIRFRAME LOG	
100	2980 AIR TEL: 7	WAY AV,. COS 14-433-2275 AN	STA MESA, CA 92626 FAA APPROVED REPAIR STATION # JVR016L SD 714-433-2276 CIRUS DESIGN AUTHORIZED REPAIR STATION	
	THE ACCORD/ TO SEE	CAIRCRAFT AN ANCE WITH CU ANCE. PERTIN	ND / OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED AND INSPECTED IN URRENT FEDERAL AIR REGULATIONS AND WAS FOUND AIRWORTHY FOR RETURN NENT DETAILS OF THE REPAIR / REPLACEMENT ARE ON FUE AT THIS AGENCY.	
	DATE:	12-18-02	SIGNED: Aren and Anin and	
			The and the and the	
	R & R Attitud	de indicator for re	pair done by Global Tech-under their WO#43033. END	
				-
	-			
	5288J			
N	/N: 421C-0158 IODEL: 421C		DATE: 02-11-03	-
	ACH TIME: 11	24.9	W.O. #: 12739	
			TOTAL TIME: 5222.50	
_			AIRFRAME LOG	
	THE AIR	3-2275 AND 714	OR COMPONENT IDENTIFIED A DOUT NU COMPONENT IDENTIFIED A DOUT NU COMPONENT IDENTIFIED A DOUT NU COMPONENT	Contraction of the second
A	TEL: 714-43 THE AIR CCORDANCE TO SERVICE	3-2275 AND 714 CRAFT AND / O WITH CURRE! . PERTINENT	A433-2276 CIRUS DESIGN AUTHORIZED REPAIR STATION # JVR016L CIRUS DESIGN AUTHORIZED REPAIR STATION — OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED AND INSPECTED IN NT FEDERAL AIR REGULATIONS AND WAS FOUND AIRWORTHY FOR RETURN DETAILS OF THE REPAIR / REPLACEMENT ARE ON FILE AT THIS AGENCY.	
A	TEL: 714-43 THE AIRC CCORDANCE TO SERVICE	3-2275 AND 714 CRAFT AND / O WITH CUPPE	I-433-2276 CIRUS DESIGN AUTHORIZED REPAIR STATION	
A	TEL: 714-43 THE AIRC CCORDANCE TO SERVICE DATE <u>: 02-</u>	3-2275 AND 714 CRAFT AND / O WITH CURRE: PERTINENT 11-03	CIRUS DESIGN AUTHORIZED REPAIR STATION # JVR016L CIRUS DESIGN AUTHORIZED REPAIR STATION OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED AND INSPECTED IN NT FEDERAL AIR REGULATIONS AND WAS FOUND AIRWORTHY FOR RETURN DETAILS OF THE REPAIR / REPARCEMENT ARE ON FILE AT THIS AGENCY. SIGNED: SIGNED:	
A	TEL: 714-43 THE AIRC CCORDANCE TO SERVICE DATE <u>: 02-</u>	3-2275 AND 714 CRAFT AND / O WITH CURRE: PERTINENT 11-03	A433-2276 CIRUS DESIGN AUTHORIZED REPAIR STATION # JVR016L CIRUS DESIGN AUTHORIZED REPAIR STATION — OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED AND INSPECTED IN NT FEDERAL AIR REGULATIONS AND WAS FOUND AIRWORTHY FOR RETURN DETAILS OF THE REPAIR / REPLACEMENT ARE ON FILE AT THIS AGENCY.	
A	TEL: 714-43 THE AIRC CCORDANCE TO SERVICE DATE <u>: 02-</u>	3-2275 AND 714 CRAFT AND / O WITH CURRE: PERTINENT 11-03	CIRUS DESIGN AUTHORIZED REPAIR STATION # JVR016L CIRUS DESIGN AUTHORIZED REPAIR STATION OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED AND INSPECTED IN NT FEDERAL AIR REGULATIONS AND WAS FOUND AIRWORTHY FOR RETURN DETAILS OF THE REPAIR / REPARCEMENT ARE ON FILE AT THIS AGENCY. SIGNED: SIGNED:	
A	TEL: 714-43 THE AIRC CCORDANCE TO SERVICE DATE <u>: 02-</u>	3-2275 AND 714 CRAFT AND / O WITH CURRE: PERTINENT 11-03	CIRUS DESIGN AUTHORIZED REPAIR STATION # JVR016L CIRUS DESIGN AUTHORIZED REPAIR STATION OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED AND INSPECTED IN NT FEDERAL AIR REGULATIONS AND WAS FOUND AIRWORTHY FOR RETURN DETAILS OF THE REPAIR / REPARCEMENT ARE ON FILE AT THIS AGENCY. SIGNED: SIGNED:	
A	TEL: 714-43 THE AIR CCORDANCE TO SERVICE DATE: 02- R Vacuum pum	3-2275 AND 714 CRAFT AND / O WITH CURREN . PERTINENT 11-03 p with Overhauk	1-433-2276 CIRUS DESIGN AUTHORIZED REPAIR STATION OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED AND INSPECTED IN NT FEDERAL AIR REGULATIONS AND WAS FOUND AIRWORTHY FOR RETURN DETAILS OF THE REPAIR / REPLACEMENT ARE ON FILE AT THIS AGENCY.	
A	TEL: 714-43 THE AIR CCORDANCE TO SERVICE DATE: 02- R Vacuum pum	3-2275 AND 714 CRAFT AND / O WITH CURREN PERTINENT 11-03 p with Overhauk	1-433-2276 CIRUS DESIGN AUTHORIZED REPAIR STATION OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED AND INSPECTED IN NT FEDERAL AIR REGULATIONS AND WAS FOUND AIRWORTHY FOR RETURN DETAILS OF THE REPAIR / REPLACEMENT ARE ON FILE AT THIS AGENCY.	
A	TEL: 714-43 THE AIR CCORDANCE TO SERVICE DATE: 02- R Vacuum pum	2-2275 AND 714 CRAFT AND / O WITH CURREN . PERTINENT 11-03 p with Overhauk	1-433-2276 CIRUS DESIGN AUTHORIZED REPAIR STATION OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED AND INSPECTED IN NT FEDERAL AIR REGULATIONS AND WAS FOUND AIRWORTHY FOR RETURN DETAILS OF THE REPAIR / REPLACEMENT ARE ON FILE AT THIS AGENCY.	
A	TEL: 714-43 THE AIR CCORDANCE TO SERVICE DATE: 02- R Vacuum pum	2-2275 AND 714 CRAFT AND / O WITH CURREN . PERTINENT 11-03 p with Overhauk	1-433-2276 CIRUS DESIGN AUTHORIZED REPAIR STATION OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED AND INSPECTED IN NT FEDERAL AIR REGULATIONS AND WAS FOUND AIRWORTHY FOR RETURN DETAILS OF THE REPAIR / REPLACEMENT ARE ON FILE AT THIS AGENCY.	
A	N5288J S/N: 421C-G MODEL: 4 TACH TIM	2-2275 AND 714 CRAFT AND / O WITH CURREN . PERTINENT 11-03 p with Overhauk p with Overhauk 0158 21C [E: 1156.3	I-433-2276 CIRUS DESIGN AUTHORIZED REPAIR STATION OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED AND INSPECTED IN NT FEDERAL AIR REGULATIONS AND WAS FOUND AIRWORTHY FOR RETURN DETAILS OF THE REPAIR / REPLACEMENT ARE ON FIRE AT THIS AGENCY.	
A	N5288J S/N: 421C-0 MODEL: 4 2980 AIR 2980 AIR 2980 AIR	23-2275 AND 714 CRAFT AND / O WITH CURREE . PERTINENT 11-03 p with Overhauk p with Overhauk 21C E: 1156.3 WAY AV,. COS 14-433-2275 AN	I-433-2276 CIRUS DESIGN AUTHORIZED REPAIR STATION OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED AND INSPECTED IN NT FEDERAL AIR REGULATIONS AND WAS FOUND AIRWORTHY FOR RETURN DETAILS OF THE REPAIR / REPLACEMENT ARE ON FILE AT THIS AGENCY. SIGNED: Image: Im	
A	N5288J S/N: 421C-0 MODEL: 4 2980 AIR 2980 AIR 72980 AIR 72980 AIR 72980 AIR	2-2275 AND 714 CRAFT AND / O WITH CURREL . PERTINENT 11-03 p with Overhauk p w	I-433-2276 CIRUS DESIGN AUTHORIZED REPAIR STATION OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED AND INSPECTED IN NT FEDERAL AIR REGULATIONS AND WAS FOUND AIRWORTHY FOR RETURN DETAILS OF THE REPAIR / REPLACEMENT ARE ON FILE AT THIS AGENCY. SIGNED:	N
A	N5288J S/N: 421C-0 MODEL: 4 2980 AIR 2980 AIR 72980 AIR 72980 AIR 72980 AIR	2-2275 AND 714 CRAFT AND / O WITH CURREL . PERTINENT 11-03 p with Overhauk p w	I-433-2276 CIRUS DESIGN AUTHORIZED REPAIR STATION OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED AND INSPECTED IN NT FEDERAL AIR REGULATIONS AND WAS FOUND AIRWORTHY FOR RETURN DETAILS OF THE REPAIR / REPLACEMENT ARE ON FILE AT THIS ACENCY.	N
A	N5288J S/N: 421C-0 MODEL: 4 2980 AIR 2980 AIR 72980 AIR 72980 AIR 72980 AIR	2-2275 AND 714 CRAFT AND / O WITH CURREL . PERTINENT 11-03 p with Overhauk p w	I-433-2276 CIRUS DESIGN AUTHORIZED REPAIR STATION OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED AND INSPECTED IN NT FEDERAL AIR REGULATIONS AND WAS FOUND AIRWORTHY FOR RETURN DETAILS OF THE REPAIR / REPLACEMENT ARE ON FILE AT THIS AGENCY. SIGNED:	N
A	N5288J S/N: 421C-C MODEL: 4 TO SERVICE	2-2275 AND 714 CRAFT AND / O WITH CURRENT - PERTINENT 11-03 	Image: Additional and the state of the	

e

8

	and here i		A subscription of the last second	
DATE	E TOTAL TIME IN SERVICE	TACH OR RECORDING METER TIME	SIGNATURE & CERTIFIC	N OF WORK PERFORMED— ATE NO. OF PERSON PERFORMING WORK
		TOTAL brought f	forward from previous page	12124 Construction of the second s
			10.	A
_	N5288J	0		DATE: 00 20 02
	SN: 421C-0158		-	DATE: 09-29-03 W.O. #: 13205
	MODEL: 4210	20	the state of the state	TOTAL TIME: 5290.4
	TACH TIME: 119	2	AIRFRAME LOG	
	\cup			
	TEL: 714-433	AV., COSTA MESA, -2275 AND 714-433-	-2276 CIRUS D	PROVED REPAIR STATION # JVR016L ESIGN AUTHORIZED REPAIR STATION
2	THE AIRC ACCORDANCE TO SERVICE	CRAFT AND / OR CO WITH CURRENT F , PERTINENT DET	AILS OF THE BETAR / REPLAC	WAS REPAIRED AND INSPECTED IN D WAS FOUND AIRWORTHY FOR RETURN EMENT ARE ON FILE AT THIS AGENCY.
	DATE <u>: 09-</u> 2	29-0.3	SIGNED: Jores 0	husor
		IN SOME OF LODI	Corona 121C Annual Inspection Gu	8
Ē	CONTRACT Income	tion of FIT as ner 91	r Cessna 421C Annual Inspection Gui -207(d) battery replacement due at 03	-04.
-	C/W AD200-01-16	Exhaust Cracks, no c	cracks found at this time, AD next due	at 1292 Tach.
	R & R Nitrogen Bo	ottie for repair done by	/ S.J. Obrien.	
	R&R RH brake lin	ings (RA66-66) ium Filters P/N RAB3	-5-1 END	
L	Installed new vacu	ium riters ray RABS	-5-1, 1, 0	
				and the second of the
				the second se
102				
	N5288J S/N: 421C-0158			DATE: 08-11-04
	MODEL: 421C			W.O. #: 13949 TOTAL TIME: 5389.2
85	HOBBS TIME: 1	291.6	TACPAFT MAINTENANS	TOTAL MALL SCORE
			AIRFRAME LOG	
	2980 AIRWAY TEL: 714-43	AV., COSTA MESA 3-2275 AND 714-433		PROVED REPAIR STATION # JVYR016L DESIGN AUTHORIZED REPAIR STATION
	THE AIR	CRAFT AND / OR C	COMPONENT IDENTIFIED ABOV FEDERAL AIR REGULATIONS A TAILS OF THE REPAIR / REPLA	E WAS REPAIRED AND INSPECTED IN NO WAS FOUND AIRWORTHY FOR RETURN CENTEXT ARE ON FILM AT THIS AGENCY.
100	TO SERVIC	E. PERTINENT DE		
	TO SERVIC	E. PERTINENT DE	SIGNED: Aamen	I Show you -
	TO SERVIC	E. PERTINENT DE 3-11-04	SIGNED:	I Thomas your
	TO SERVIC DATE: 08 R&R Right main t	E. PERTINENT DE 3-11-04 tire and tube with new.	SIGNED:	A Shor your
	TO SERVIC DATE: 08 R&R Right main t	E. PERTINENT DE 3-11-04 ire and tube with new. ased bearings. Ops ch Hobbs	SIGNED: Americanov . P/N# 6.50-10-10FLCII neck ok. END 5: 1304, 0	A Shorsyaw
	TO SERVIC DATE: 08 R&R Right main t	E. PERTINENT DE 3-11-04 ire and tube with new ased bearings. Ops ch Hobbs CERTIFIED	SIGNED: American 	ATIC SYSTEMS
	TO SERVIC DATE: 08 R&R Right main t	E. PERTINENT DE 3-11-04 ire and tube with new. ased bearings. Ops ch Hobbs CERTIFIED REQUIRED	SIGNED: 	A.R. PART 43,
	TO SERVIC DATE: 08 R&R Right main t	E. PERTINENT DE 3-11-04 ire and tube with new ased bearings. Ops ch Hobbs CERTIFIED REQUIRED APPENDIX	SIGNED: 	PERFORMED
	TO SERVIC DATE: 08 R&R Right main t	E. PERTINENT DE 3-11-04 ire and tube with new. ased bearings. Ops ch Hobbs CERTIFIED REQUIRED APPENDIX A.T.C. TRA	SIGNED: 	PERFORMED IN TRED BY F.A.R.
	TO SERVIC DATE: 08 R&R Right main t	E. PERTINENT DE a-11-04 ire and tube with new. ased bearings. Ops ch Hobbs CERTIFIED REQUIRED APPENDIX A.T.C. TRA 91.413 AC	SIGNED: P/N# 6.50-10-HOFLCII ALTIMETER(S) AND ST/ BY F.A.R. 91.411 I.A.W. F E TO <u>30,000</u> FEET NSPONDER TESTS REQU COMPLISHED I.A.W. F	A.R. PART 43, PERFORMED IRED BY F.A.R. A.R. PART 43,
	TO SERVIC DATE: 08 R&R Right main t	E. PERTINENT DE 3-11-04 irre and tube with new. ased bearings. Ops ch Hobbs CERTIFIED REQUIRED APPENDIX A.T.C. TRA 91.413 AC APPENDIX	SIGNED: P/N# 6.50-105-00FLCII ALTIMETER(S) AND STA BY F.A.R. 91.411 I.A.W. F E TO <u>30,000</u> FEET NSPONDER TESTS REQU COMPLISHED I.A.W. F. F. W.O.# 44	A.R. PART 43, PERFORMED IRED BY F.A.R. A.R. PART 43, A.R. PART 43, A.R. PART 43,
	TO SERVIC DATE: 08 R&R Right main t	E. PERTINENT DE a-11-04 ire and tube with new. ased bearings. Ops ch <i>Hobbs</i> CERTIFIED REQUIRED APPENDIX A.T.C. TRA 91.413 AC APPENDIX LT. ALT. SN_	SIGNED: P/N# 6.50-10510FLCII Neck ok. END ALTIMETER(S) AND STA BY F.A.R. 91.411 I.A.W. F E TO 30 000 FEET NSPONDER TESTS REQU CCOMPLISHED I.A.W. F. F. W.O.# 4 SALSTAL #1XPDR SM	A.R. PART 43, PERFORMED IRED BY F.A.R. A.R. PART 43, A.R. PART 43, A.R. PART 43, A.R. PART 43,
	TO SERVIC DATE: 08 R&R Right main t	E. PERTINENT DE irre and tube with new. ased bearings. Ops ch Hobbs CERTIFIED REQUIRED APPENDIX A.T.C. TRA 91.413 AC APPENDIX LT. ALT. SN RT. ALT. SN	SIGNED: P/N# 6.50-10510FLCII neck ok. END ALTIMETER(S) AND ST/ BY F.A.R. 91.411 I.A.W. F E TO <u>30 000</u> FEET INSPONDER TESTS REQU CCOMPLISHED I.A.W. F. F. W.O.# 4 XISPON #1XPDR SN XISPON #1XPDR SN XISPON SI	A.R. PART 43, PERFORMED IRED BY F.A.R. A.R. PART 43, A.R. PART 43, A.R. PART 43, A.R. PART 43,
	TO SERVIC DATE: 08 R&R Right main t	E. PERTINENT DE irre and tube with new. ased bearings. Ops ch Hobbs CERTIFIED REQUIRED APPENDIX A.T.C. TRA 91.413 AC APPENDIX LT. ALT. SN RT. ALT. SN	SIGNED: P/N# 6.50-10510FLCII neck ok. END ALTIMETER(S) AND ST/ BY F.A.R. 91.411 I.A.W. F E TO <u>30,000</u> FEET INSPONDER TESTS REQU COMPLISHED I.A.W. F. W.O.# 4 ST + ST + 122PDR SI DED FAILS SN	A.R. PART 43, PERFORMED IRED BY F.A.R. A.R. PART 43, A.R. PART 44, A.R. PART
	TO SERVIC DATE: 08 R&R Right main t	E. PERTINENT DE irre and tube with new. ased bearings. Ops ch Hobbs CERTIFIED REQUIRED APPENDIX A.T.C. TRA 91.413 AC APPENDIX LT. ALT. SN RT. ALT. SN	SIGNED: P/N# 6.50-10510FLCII neck ok. END ALTIMETER(S) AND ST/ BY F.A.R. 91.411 I.A.W. F E TO <u>30,000</u> FEET INSPONDER TESTS REQU COMPLISHED I.A.W. F. W.O.# 4 ST + ST + 122PDR SI DED FAILS SN	A.R. PART 43, PERFORMED IRED BY F.A.R. A.R. PART 43, A.R. PART 43, A.R. PART 43, A.R. PART 43,

TOTAL—Carry forward to next page

Page No. TOTAL TACH OR TIME DATE RECORDING DESCRIPTION OF WORK PERFORMED-IN SERVICE METER SIGNATURE & CERTIFICATE NO. OF PERSON PERFORMING WORK TIME TOTAL brought forward from previous page N5288J S/N: 421C-0 58 2 MODEL: 42 C TACH TIME: 1192.8 DATE: 10-05-04 W.O. #: 14014 TOTAL TIME: 5401.6 AIRFRAME LOG 2980 AIRWAY AV., COSTA MESA, CA 92626 TEL: 714-433-2275 AND 714-433-2276 FAA APPROVED REPAIR STATION # JVYR016L CIRRUS DESIGN AUTHORIZED REPAIR STATION THE AIRCRAFT AND / OR COMPONENT IDENTIFIED ABOVE WAS RETAIRED AND INSPECTED IN ACCORDANCE WITH CURRENT FEDERAL AIR REGULATIONS AND WAS FOUND AIRWORTHY FOR RETURN TO SERVICE - PERTINENT DETAILS OF THE REGULATIONS AND WAS FOUND AIRWORTHY FOR RETURN TO SERVICE. PERTINENT DETAILS OF THE REPAIR / REPLACE MENT ARE ON FILE AT THIS AGENCY. DATE: SIGNED: Gema han x 20 C/W an ANNUAL INSPECTION as per Cessna 421C Annual Inspection Guide. C/W Yearly Inspection of ELT as per 91-207(d) R&R Batteries. C/W AD 2060-01-16 Exhaust Inspection, as per AD. No cracks found. p R&R #1 JPI Probe with new. R&R RH and LH brake linings (RA66-66) R&R Vacuum filter with new. P/N# B351 R&R Taxi light bulb with new. P/N# GE4594. Installed new gears in right and left landing light. P/N# A6580, A6582 R&R Left main tire and tube with new _P/N# C650-10-8TT, M650-10. Cowl light installed as per STC SA4270SW. Ground run up inspection performed, ops check ok. END 2 N5288J DATE: 12-15-04 S/N: 421C-0158 W.O. #: 14180 MODEL: 421C TOTAL TIME: 5513.6 HOBBS TIME: 1304.8 AIRFRAME LOG FAA APPROVED REPAIR STATION # JVYR016L 2980 AIRWAY AV., COSTA MESA, CA 92626 CIRRUS DESIGN AUTHORIZED REPAIR STATION TEL: 714-433-2275 AND 714-433-2276 THE AIRCRAFT AND / OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED AND INSPECTED IN ACCORDANCE WITH CURRENT FEDERAL AIR REGULATIONS AND WASYOUND AIRWORTHY FOR RETURN TO SERVICE. PERTINENT DETAILS OF THE REPAIR / REPLACEMENT ARE ON FILE AT THIS AGENCY. 0-Alla amaly SIGNED: 12-15-04 DATE: R&R Fuel Drain Valve with new. P/N# 50C4. Ground run up inspection performed, ops check ok. END SUB-TOTAL this page TOTAL-Carry forward to next page

N5288J S/N: 421C-0158 MODEL: 421C HOBBS TIME: 1327.7



DATE: 04-06-05 W.O. #: 14386 TOTAL TIME: 5536.5

2980 AIRWAY AV,. COSTA MESA, CA 92626 TEL: 714-433-2275

FAA APPROVED REPAIR STATION # JVYR016L

THE AIRCRAFT AND / OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED AND INSPECTED IN ACCORDANCE WITH CURRENT FEDERAL AIR REGULATIONS AND WAS FOUND AIRWORTHY FOR RETURN TO SERVICE. PERTINENT DETAILS OF THE REPAIR / REPLACEMENT ARE ON FILE AT THIS AGENCY.

DATE: 04-06-05

SIGNED:____

InstalledNew Light Bulb. P/N 4594 InstalledNew Light Bulb GE. P/N 305 Ground un up inspection performed, ops check ok. END

N5288J S/N: 421C-0158 MODEL: 421C HOBBS TIME: 1328.3



DATE: 04-26-05 W.O. #: 14420 TOTAL TIME: 5537.1

AIRFRAME LOG

2980 AIRWAY AV,. COSTA MESA, CA 92626 TEL: 714-433-2275

FAA APPROVED REPAIR STATION # JVYR016L

THE AIRCRAFT AND / OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED AND INSPECTED IN ACCORDANCE WITH CURRENT FEDERAL AIR REGULATIONS AND WAS FOUND AIRWORTHY FOR RETURN TO SERVICE. PERTINENT DETAILS OF THE REPAIR / REPLACEMENT ARE ON FILE AT THIS AGENCY.

DATE: 04-26-05

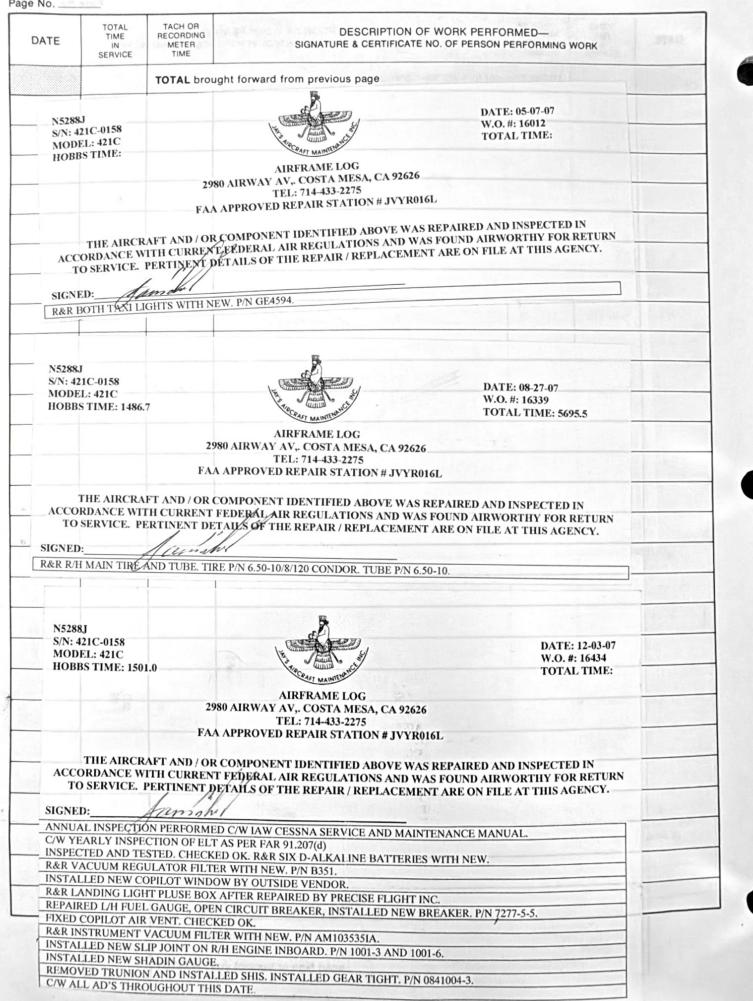
___SIGNED:__

Installed STC SA 4005NM KIT per Pulselite Installation Manual Ground run up inspection performed, ops check ok. END

DATE	TOTAL TIME IN SERVICE	TACH OR RECORDING METER TIME		DESCRIF GNATURE & CERT	PTION OF WOR	K PERFORMED— PERSON PERFORM	ING WORK	- 47.80
		TOTAL bro	ught forward from	previous page				
				cf.	B			
	N528 S/N:	421C-0158		1-	S.		DATE: 05-03-05 W.O. #: 14435	
	MOL	DEL: 421C	0.3	13.5.10	Cuntin Car		TOTAL TIME:	5539.1
	HOB	BS TIME: 133	0.5	CRAFT	MAINTEN			
				AIRFRAM	,, COSTA MESA	CA 92626		
				TEL	: 714-433-2275			
			FAA	APPROVED R	EPAIR STATION	N # JVYR016L	1.	
5	ACC	THE AIRCH CORDANCE W TO SERVICE.	RAFT AND / OR CO VITH CURRENT FI PERTINENT DET/	MPONENT IDE EDERAL AIR RI AILS OF THE RI	INTIFIED ABOV EGULATIONS A EPAIR / REPLAC	E WAS REPAIRE ND WAS FOUND CEMENT AME OF	AND INSPECT ARWORTHY F FILE AT THIS	TED IN OR RETURN AGENCY.
)		TE <u>: 05-0</u>		SIGNED:	Au	min		
				avad Vinke				
4	Re-ad	justed Turbo In	let Temp. Probe, Rem	loved KINKS.				
	1	1			*			
	N5288	3J 121C-0158					DATE: 10-10-05	5
		EL: 421C		The of			W.O. #: 14766	5502 2
		BS TIME: 138-	1.5	AIRCRAFT	MAINTENANC		TOTAL TIME:	5593.3
				AIRFRAM	E LOG			
	1		29		COSTA MESA : 714-433-2275	, CA 92626		
	24		FAA	APPROVED RE	EPAIR STATION	# JVYR016L		
		THE AIRCR CORDANCE W O SERVICE.	AFT AND / OR CO ITH CURRENT FE PERTINENT DETA	MPONENT IDE DERAL AIR RE AILS OF THE RE	NTIFIED ABOV CGULATIONS AN EPAIR / REPLAC	E WAS REPAIR ND WAS FOUND CEMENY ARE ON	D AND INSPECT ARWORTHY F FILE AT THIS	TED IN OR RETURN AGENCY.
	DA	TE <u>: 10-10</u>	0-05	SIGNED:	Aa	maky		
2 Mars		(0) (with	11		24	
	R&R	LIH Landing Ligh	ght Bulb. P/N 1524 b at with new. P/N 4594	4 lamp	0			
1					-			
<u>.</u>	N528	8J		da.	A THE STATE		DATE: 01-18-0	6
		421C-0158		12	Y		W.O. #: 14897 TOTAL TIME	
		EL: 421C BS TIME: 140	0.6 .	S ALACRAFT	MAINTENAME		TOTAL IIME.	
		" Acq		AIRFRAM	IE LOG		1	1 2
			25	980 AIRWAY AV TEL	. COSTA MESA	A, CA 92626		
1.1.1.1				APPROVED R	EPAIR STATIO	N # JVYR016L	19.2	
1							IND INSPEC	TED IN
1		THE AIRCI CORDANCE V TO SERVICE.	RAFT AND / OR CO VITH CURRENT FI PERTINENT DET/	OMPONENT IDE EDERAL AIR RI AILS OF THE R	ENTIFIED ABOV EGULATIONS A EPAIR / REPLA	ND WAS YOUND CENTRY ARE OF	AIRWORTHY F N FILE AT THIS	OR RETURN AGENCY.
			8-06	SIGNED:	for	where the	3	
	Instal	led New Gill A	cid Lead Battery. P/N	11-01680 MODI	G-246	4		
			L. this page		- Alf			
	Г		arry forward to ne					
		I TAL-C	any lorward to ne	xt page				

ATE		TACH OR							
	TOTAL TIME IN SERVICE	RECORDING METER TIME	DESCRIPTION OF WORK PERFORM SIGNATURE & CERTIFICATE NO. OF PERSON PERI	IED- FORMING WORK					
		TOTAL brough	t forward from previous page	LATOT					
	N5288J S/N: 421C								
	MODEL:	-0158 421C		DATE: 11-07-06					
	HOBBS TI	ME: 1429.1		W.O. #: 15432					
				TOTAL TIME: 5637.9					
			AIRFRAME LOG	AND -					
1			2980 AIRWAY AV,. COSTA MESA, CA 92626 TEL: 714-433-2275						
				1					
100			FAA APPROVED REPAIR STATION # JVYR016L	1					
	THE	AIRCRAFT AND	OR COMPONENT IDENTIFICATION OF						
	ACCORDA TO SER	NCE WITH CUR VICE. PERTINE) / OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED A RENT FEDERAL AIR REGULATIONS AND WAS FOUND AIR NT DETAILS OF THE REPAIR / REPLACEMENT ARE ON FI	ND INSPECTED IN RWORTHY FOR RETURN LE AT THIS AGENCY.					
	DATE <u>:</u>		SIGNED:						
	C/W an ANNI	AL INSPECTION	N in second se						
14		· and Cooning Sint	N, in accordance with Cessna 421C Service and Maintenance Manual. pud as per STC SA3721SW. P/N 1221-1						
	- C/W Tearly In	spection of ELT as	per FAR 91.207 (d) R&R Batteries with name size "D"						
	Thoi Static Ira	insponder Test Perfo	ormed by: Western Avionics						
	C/W AD95-09	-13 Fuel Float Valve	e Functional Test, by Inspection.						
	Elbow Exhaust	P/N K9910300-01	ection, in accordance with Cessna SB MEB 99-6-9-12. Replaced L/H	Tail Pipe Overboard and					
	C/W AD2005-2	20-26 Avionics Bus	Switch Inspection/ Replacement Circuit Breaker in accordance with	Cesena ME.S.B. MED.05.1					
	Rock Cheult D	caker Switch with r	new.	Cessia ME-3.D. MED 03-1,					
H	C/w with all A	D's Throughout 200	06-22 AD Compliance Record.						
— F	Installed Out	R&R R/H and L/H Brake Linings with new, P/N RA66-66							
	Installed Overhauled L/H wing Locker Fuel Cell By: Aero Tech, under W.O.# 23484-2. P/N 99-10241, S/N 76-3597 R&R Battery with new. P/N G-246								
E	R&R Battery w	th new. P/N G-246		8/N 76-3597					
	R&R Battery will R&R Fuel Line	th new. P/N G-246 Divider Line with n	ew. P/N 5100108-73	S/N 76-3597					
	R&R Battery w R&R Fuel Line R&R Hydraulic	th new. P/N G-246 Divider Line with n Pump Gasket with 1	ew. P/N 5100108-73 new. P/N 5154056-2	8/N 76-3597					
	R&R Battery w R&R Fuel Line R&R Hydraulic	th new. P/N G-246 Divider Line with n Pump Gasket with 1	ew. P/N 5100108-73	S/N 76-3597					
	R&R Battery w R&R Fuel Line R&R Hydraulic R&R Alternator	th new. P/N G-246 Divider Line with n Pump Gasket with 1	ew. P/N 5100108-73 new. P/N 5154056-2	S/N 76-3597					
	R&R Battery w R&R Fuel Line R&R Hydraulic R&R Alternator	th new. P/N G-246 Divider Line with n Pump Gasket with 1 Following after Out	ew. P/N 5100108-73 new. P/N 5154056-2 tside repair, performed by: Aero Accessories.	S/N 76-3597 DATE: 01-19-07					
	R&R Battery w R&R Fuel Line R&R Hydraulic R&R Alternator N5288J S/N: 421C-01	th new. P/N G-246 Divider Line with n Pump Gasket with n Following after Out	ew. P/N 5100108-73 new. P/N 5154056-2 tside repair, performed by: Aero Accessories.	DATE: 01-19-07 V.O. #: 15768					
	R&R Battery w R&R Fuel Line R&R Hydraulic R&R Alternator	th new. P/N G-246 Divider Line with n Pump Gasket with n Following after Out	ew. P/N 5100108-73 new. P/N 5154056-2 tside repair, performed by: Aero Accessories.	DATE: 01-19-07					
	R&R Battery w R&R Fuel Line R&R Hydraulic R&R Alternator N5288J S/N: 421C-01 MODEL: 42	th new. P/N G-246 Divider Line with n Pump Gasket with n Following after Out	ew. P/N 5100108-73 new. P/N 5154056-2 tside repair, performed by: Aero Accessories.	DATE: 01-19-07 V.O. #: 15768					
	R&R Battery w R&R Fuel Line R&R Hydraulic R&R Alternator N5288J S/N: 421C-01 MODEL: 42	th new. P/N G-246 Divider Line with n Pump Gasket with n Following after Out	ew. P/N 5100108-73 new. P/N 5154056-2 tside repair, performed by: Aero Accessories.	DATE: 01-19-07 V.O. #: 15768					
	R&R Battery w R&R Fuel Line R&R Hydraulic R&R Alternator N5288J S/N: 421C-01 MODEL: 42	th new. P/N G-246 Divider Line with n Pump Gasket with n Following after Out	ew. P/N 5100108-73 new. P/N 5154056-2 tside repair, performed by: Aero Accessories.	DATE: 01-19-07 V.O. #: 15768					
	R&R Battery w R&R Fuel Line R&R Hydraulic R&R Alternator N5288J S/N: 421C-01 MODEL: 42 HOBBS TIM	th new. P/N G-246 Divider Line with n Pump Gasket with n Following after Out	ew. P/N 5100108-73 new. P/N 5154056-2 tside repair, performed by: Aero Accessories.	DATE: 01-19-07 V.O. #: 15768 TOTAL TIME: 5651.2					
	R&R Battery w R&R Fuel Line R&R Hydraulic R&R Alternator N5288J S/N: 421C-01 MODEL: 42 HOBBS TIM	th new. P/N G-246 Divider Line with n Pump Gasket with n Following after Out	ew. P/N 5100108-73 new. P/N 5154056-2 tside repair, performed by: Aero Accessories.	DATE: 01-19-07 N.O. #: 15768 FOTAL TIME: 5651.2					
	R&R Battery w R&R Fuel Line R&R Hydraulic R&R Alternator N5288J S/N: 421C-01 MODEL: 42 HOBBS TIM	th new. P/N G-246 Divider Line with n Pump Gasket with n Following after Out 58 1C E: 1442.4	ew. P/N 5100108-73 new. P/N 5154056-2 tside repair, performed by: Aero Accessories. AirFRAME LOG 2980 AIRWAY AV., COSTA MESA, CA 92626 TEL: 714-433-2275 FAA APPROVED REPAIR STATION # JVYR016L OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED A	DATE: 01-19-07 N.O. #: 15768 FOTAL TIME: 5651.2 ND INSPECTED IN					
	R&R Battery w R&R Fuel Line R&R Hydraulic R&R Alternator N5288J S/N: 421C-01 MODEL: 42 HOBBS TIM	th new. P/N G-246 Divider Line with n Pump Gasket with n Following after Out 58 1C E: 1442.4	ew. P/N 5100108-73 new. P/N 5154056-2 tside repair, performed by: Aero Accessories. AirFRAME LOG 2980 AIRWAY AV., COSTA MESA, CA 92626 TEL: 714-433-2275 FAA APPROVED REPAIR STATION # JVYR016L OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED A	DATE: 01-19-07 N.O. #: 15768 FOTAL TIME: 5651.2 ND INSPECTED IN					
	R&R Battery w R&R Fuel Line R&R Hydraulic R&R Alternator S/N: 421C-01 MODEL: 42 HOBBS TIM THE A ACCORDAN TO SERV	th new. P/N G-246 Divider Line with n Pump Gasket with n Following after Out 58 1C E: 1442.4 AIRCRAFT AND / NCE WITH CURR TCE. PERTINEN	ew. P/N 5100108-73 new. P/N 5154056-2 tside repair, performed by: Aero Accessories.	DATE: 01-19-07 N.O. #: 15768 FOTAL TIME: 5651.2 ND INSPECTED IN					
	R&R Battery wi R&R Fuel Line R&R Hydraulic R&R Alternator N5288J S/N: 421C-01 MODEL: 42 HOBBS TIM THE A ACCORDAN TO SERV DATE:	th new. P/N G-246 Divider Line with n Pump Gasket with n Following after Out 158 IC E: 1442.4 AIRCRAFT AND / CE WITH CURR ICE. PERTINEN 01-19-07	ew. P/N 5100108-73 new. P/N 5154056-2 tside repair, performed by: Aero Accessories. Aireframe LOG 2980 AIRWAY AV, COSTA MESA, CA 92626 TEL: 714-433-2275 FAA APPROVED REPAIR STATION # JVYR016L OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED A EENT FEDERAL AIR REGULATIONS AND WAS FOUND AIR T DETAILS OF THE REPAIR / REPLACEMENT ARE ON FI SIGNED:	DATE: 01-19-07 N.O. #: 15768 FOTAL TIME: 5651.2 ND INSPECTED IN					
	R&R Battery wi R&R Fuel Line R&R Hydraulic R&R Alternator N5288J S/N: 421C-01 MODEL: 42 HOBBS TIM THE A ACCORDAN TO SERV DATE:	th new. P/N G-246 Divider Line with n Pump Gasket with n Following after Out 158 IC E: 1442.4 AIRCRAFT AND / CE WITH CURR ICE. PERTINEN 01-19-07	ew. P/N 5100108-73 new. P/N 5154056-2 tside repair, performed by: Aero Accessories. Aireframe LOG 2980 AIRWAY AV, COSTA MESA, CA 92626 TEL: 714-433-2275 FAA APPROVED REPAIR STATION # JVYR016L OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED A EENT FEDERAL AIR REGULATIONS AND WAS FOUND AIR T DETAILS OF THE REPAIR / REPLACEMENT ARE ON FI SIGNED:	DATE: 01-19-07 N.O. #: 15768 FOTAL TIME: 5651.2 ND INSPECTED IN					
	R&R Battery w R&R Fuel Line R&R Hydraulic R&R Alternator N5288J S/N: 421C-01 MODEL: 42 HOBBS TIM THE A ACCORDAN TO SERV DATE: Installed New S	th new. P/N G-246 Divider Line with n Pump Gasket with n Following after Out 158 1C E: 1442.4 AIRCRAFT AND / CE WITH CURR TCE. PERTINEN 01-19-07 eals on R/H Wasteg	ew. P/N 5100108-73 new. P/N 5154056-2 tside repair, performed by: Aero Accessories.	DATE: 01-19-07 N.O. #: 15768 FOTAL TIME: 5651.2 ND INSPECTED IN					
	R&R Battery wi R&R Fuel Line R&R Hydraulic R&R Alternator N5288J S/N: 421C-01 MODEL: 42 HOBBS TIM THE A ACCORDAN TO SERV DATE:	th new. P/N G-246 Divider Line with n Pump Gasket with n Following after Out 158 1C E: 1442.4 AIRCRAFT AND / CE WITH CURR TCE. PERTINEN 01-19-07 eals on R/H Wasteg	ew. P/N 5100108-73 new. P/N 5154056-2 tside repair, performed by: Aero Accessories. Aireframe LOG 2980 AIRWAY AV, COSTA MESA, CA 92626 TEL: 714-433-2275 FAA APPROVED REPAIR STATION # JVYR016L OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED A EENT FEDERAL AIR REGULATIONS AND WAS FOUND AIR T DETAILS OF THE REPAIR / REPLACEMENT ARE ON FI SIGNED:	DATE: 01-19-07 N.O. #: 15768 FOTAL TIME: 5651.2 ND INSPECTED IN					
	R&R Battery w R&R Fuel Line R&R Hydraulic R&R Alternator N5288J S/N: 421C-01 MODEL: 42 HOBBS TIM THE A ACCORDAN TO SERV DATE: Installed New S	th new. P/N G-246 Divider Line with n Pump Gasket with n Following after Out 158 1C E: 1442.4 AIRCRAFT AND / CE WITH CURR TCE. PERTINEN 01-19-07 eals on R/H Wasteg	ew. P/N 5100108-73 new. P/N 5154056-2 tside repair, performed by: Aero Accessories. Aireframe LOG 2980 AIRWAY AV, COSTA MESA, CA 92626 TEL: 714-433-2275 FAA APPROVED REPAIR STATION # JVYR016L OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED A EENT FEDERAL AIR REGULATIONS AND WAS FOUND AIR T DETAILS OF THE REPAIR / REPLACEMENT ARE ON FI SIGNED:	DATE: 01-19-07 N.O. #: 15768 FOTAL TIME: 5651.2 ND INSPECTED IN					
	R&R Battery w R&R Fuel Line R&R Hydraulic R&R Alternator N5288J S/N: 421C-01 MODEL: 42 HOBBS TIM THE A ACCORDAN TO SERV DATE: Installed New S	th new. P/N G-246 Divider Line with n Pump Gasket with n Following after Out 158 1C E: 1442.4 AIRCRAFT AND / CE WITH CURR TCE. PERTINEN 01-19-07 eals on R/H Wasteg	ew. P/N 5100108-73 new. P/N 5154056-2 tside repair, performed by: Aero Accessories. Aireframe LOG 2980 AIRWAY AV, COSTA MESA, CA 92626 TEL: 714-433-2275 FAA APPROVED REPAIR STATION # JVYR016L OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED A EENT FEDERAL AIR REGULATIONS AND WAS FOUND AIR T DETAILS OF THE REPAIR / REPLACEMENT ARE ON FI SIGNED:	DATE: 01-19-07 N.O. #: 15768 FOTAL TIME: 5651.2 ND INSPECTED IN					
	R&R Battery w R&R Fuel Line R&R Hydraulic R&R Alternator N5288J S/N: 421C-01 MODEL: 42 HOBBS TIM THE A ACCORDAN TO SERV DATE: Installed New S	th new. P/N G-246 Divider Line with n Pump Gasket with n Following after Out 158 1C E: 1442.4 AIRCRAFT AND / CE WITH CURR TCE. PERTINEN 01-19-07 eals on R/H Wasteg	ew. P/N 5100108-73 new. P/N 5154056-2 tside repair, performed by: Aero Accessories. Aireframe LOG 2980 AIRWAY AV, COSTA MESA, CA 92626 TEL: 714-433-2275 FAA APPROVED REPAIR STATION # JVYR016L OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED A EENT FEDERAL AIR REGULATIONS AND WAS FOUND AIR T DETAILS OF THE REPAIR / REPLACEMENT ARE ON FI SIGNED:	DATE: 01-19-07 N.O. #: 15768 FOTAL TIME: 5651.2 ND INSPECTED IN					
	R&R Battery w R&R Fuel Line R&R Hydraulic R&R Alternator S/N: 421C-01 MODEL: 42 HOBBS TIM THE A ACCORDAN TO SERV DATE: Installed New S Installed Sun Vi	th new. P/N G-246 Divider Line with n Pump Gasket with n Following after Out 158 1C E: 1442.4 AIRCRAFT AND / ICE WITH CURR TCE. PERTINEN 01-19-07 eals on R/H Wasteg sor.	ew. P/N 5100108-73 new. P/N 5154056-2 tside repair, performed by: Aero Accessories. AIRFRAME LOG 2980 AIRWAY AV, COSTA MESA, CA 92626 TEL: 714-433-2275 FAA APPROVED REPAIR STATION # JVYR016L OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED A ENT FEDERAL AIR REGULATIONS AND WAS FOUND AFF T DETAILS OF THE REPAIR / REPLACEMENT ARE ON FI 	DATE: 01-19-07 N.O. #: 15768 FOTAL TIME: 5651.2 ND INSPECTED IN RWORTHY FOR RETURN LE AT THIS AGENCY.					
	R&R Battery w R&R Fuel Line R&R Hydraulic R&R Alternator S/N: 421C-01 MODEL: 42 HOBBS TIM THE A ACCORDAN TO SERV DATE: Installed New S Installed Sun Vi	th new. P/N G-246 Divider Line with n Pump Gasket with n Following after Out 158 1C E: 1442.4 AIRCRAFT AND / CE WITH CURR TCE. PERTINEN 01-19-07 eals on R/H Wasteg	ew. P/N 5100108-73 new. P/N 5154056-2 tside repair, performed by: Aero Accessories. AIRFRAME LOG 2980 AIRWAY AV, COSTA MESA, CA 92626 TEL: 714-433-2275 FAA APPROVED REPAIR STATION # JVYR016L OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED A ENT FEDERAL AIR REGULATIONS AND WAS FOUND AFF T DETAILS OF THE REPAIR / REPLACEMENT ARE ON FI 	DATE: 01-19-07 N.O. #: 15768 FOTAL TIME: 5651.2 ND INSPECTED IN RWORTHY FOR RETURN LE AT THIS AGENCY.					

Page No._



Page No. TOTAL TACH OR TIME DATE RECORDING DESCRIPTION OF WORK PERFORMED-IN METER SIGNATURE & CERTIFICATE NO. OF PERSON PERFORMING WORK SERVICE TIME N5288J S/N: 421C-0158 DATE: 09-10-08 MODEL: 421C W.O. #: 17153 HOBBS TIME: 1551.3 TOTAL TIME: AIRFRAME LOG 2980 AIRWAY AV,, COSTA MESA, CA 92626 TEL: 714-433-2275 FAA APPROVED REPAIR STATION # JVYR016L THE AIRCRAFT AND / OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED AND INSPECTED IN ACCORDANCE WITH CURRENT FOR REFURAL AIR REGULATIONS AND WAS FOUND AIRWORTHY FOR RETURN TO SERVICE. PERTINENT DETAILS OF THE REPAIR / REPLACEMENT ARE ON FILE AT THIS AGENCY. SIGNED: amo SECURED WIRDING AND INSPECTED FOR FULL TRAVEL OF YOKE. TRAVEL CLEARANCE CHECKED OK. N5288J S/N: 421C-0158 DATE: 10-01-08 W.O. #: 17086 MODEL: 421C TOTAL TIME: HOBBS TIME: 1574.3 AIRFRAME LOG 2980 AIRWAY AV,. COSTA MESA, CA 92626 TEL: 714-433-2275 FAA APPROVED REPAIR STATION # JVYR016L THE AIRCRAFT AND / OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED AND INSPECTED IN ACCORDANCE WITH CURRENT FEDERAL AIR REGULATIONS AND WAS FOUND AIRWORTHY FOR RETURN TO SERVICE. PERTINENT DETAILS OF THE REPAIR / REPLACEMENT ARE ON FILE AT THIS AGENCY. lon SIGNED: 21 SERVICED AIR CONDITIONING AS PER AMM. P/N R 12 FREON. R&R RIGHT SIDE TACH GENERATOR WITH NEW. P/N 22A667EX R&R R/H PULSE LIGHT WITH NEW. P/N GE4594. R&R LEFT MAIN TIRE AND TUBE WITH NEW. P/N 650-10. N5288J DATE: 04-25-09 S/N: 421C-0158 W.O. #: 17438 MODEL: 421C TOTAL TIME: HOBBS TIME: 1580.6 AIRFRAME LOG 2980 AIRWAY AV,. COSTA MESA, CA 92626 TEL: 714-433-2275 FAA APPROVED REPAIR STATION # JVYR016L THE AIRCRAFT AND / OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED AND INSPECTED IN ACCORDANCE WITH CURRENT FEDERAL AIR REGULATIONS AND WAS FOUND AIRWORTHY FOR RETURN TO SERVICE. PERTINENT BETAILS OF THE REPAIR / REPLACEMENT ARE ON FILE AT THIS AGENCY. SIGNED: C/W ANNUAC INSPECTION IAW CESSNA SERVICE AND MAINTENANCE MANUAL. FAR 91-207(#) COMPLIED WITH YEARLY INSPECTION OF ELT BATTERY. BY PHYSICAL TEST OPS. CHECKED OK. REMOVED AND REINSTALLED OXYGEN BOTTLE AFTER HYDROSTATIC TEST PERFORMED BY AERO ACCESSORIES UNDER INV # 09-9474. SERVICED W/OXYGEN. REMOVED AND REINSTALLED BLOWDOWN BOTTLE AFTER HYDROSTATIC TEST PERFORMED BY AERO ACCESSORIES UNDER INV # 09-9474. SERVICED W/NITROGEN R&R RH RECOGNITION LIGHT W/NEW. P/N GE4594. INSTALLED SERVICEABLE NOSE CABIN BAGGAGE LATCH R&R VACUUM REGULATOR FILTERS W/NEW. P/N B3-5-1. AD95-09-13 FUEL FLOAT VALVE FUNCTIONAL TEST DUE AT HOBBS TIME 2029.1 C/W ALL ADs THROUGHOUT THIS DATE.

CERTIFIED ALTIMETER(S) AND STATIC SYSTEMS REQUIRED BY FAR 91.411 I.A.W. FAR PART 43, APPENDIX E TO SOUD FEET. -PERFORMED A.T.C. TRANSPONDER TESTS REQUIRED BY FAR 91.413 ACCOMPLISHED I.A.W. FAR PART 43, APPENDIX F.

TAIL NUMBER: N52885	WO#	222417
LT. ALT S/N:	#1 XPDR _	<u> </u>
RT. ALT S/N:	#2 XPDR _	NA
ADC / SDC / BLD. ENC S/N:	~	
INSPECTOR	DATE: _	1/14/09
WESTERN AVIONICS, INC.	FAA REPAIR	STATION # YH3R997L
Hobbs 4580.6		

DATE	TOTAL TIME IN SERVICE	TACH OR RECORDING METER TIME	DESCRIPTION OF WORK PERFORMED— SIGNATURE & CERTIFICATE NO. OF PERSON PERFORMING WORK	
		TOTAL bro	ught forward from previous page	
N	5288J			The lose
S	N: 421C-0158		DATK: 08-06-10	
	ODEL: 421C OBBS TIME:		W.O. #: 18298	
	Obto mana	10100	TOTAL TIME: 5822.1	1 1.00
			AIRFRAME LOG	
			2980 AIRWAY AV,, COSTA MESA, CA 92626 TEL: 714-433-2275	
			FAA APPROVED REPAIR STATION # JVYR016L	
	CORDANCE	CRAFT AND / S WITH CURR	OR COMPONENT IDENTIFIED ABOVE WAS REPAIRED AND INSPECTED IN ENT FEDERAL AIR REGULATIONS AND WAS FOUND AIRWORTHY FOR RETURN	
	TO SERVIC	E. PERTINEN	T DETAILS OF THE REPAIR / REPLACEMENT ARE ON FILE AT THIS AGENCY.	1.1.1
SIG	NED:	F		
		SPECTION IAV	V CESSNA SERVICE AND MAINTENANCE MANUAL	
REM	IOVED & REF	PLACED ELT B	BATTERIES W/NEW. P/N D-CELL	
- FAR	91-207(d) CO	MPLIED WITH	YEARLY INSPECTION OF ELT BATTERY BY PHYSICAL TEST. OPS CHECKED OK	1
EL1	BATTERIES I	EXPIRE JUNE 2	2010. TEST DUE DATE APRIL 2011.	1 1 1 1 1
R&R	VACUUM RI	EGULATOR FI	TEST DUE DATE APRIL 2011. LTERS W/NEW, P/N B3-5-1,	1
HEA	TER INSPECT	FED AND TEST	TED OK.	
REM	OVED AND F	REINSTALLED	LH ALTERNATOR AFTER BENCH TESTED BY OUTSIDE VENDOR, TESTED OK.	1111111111
1851	ALLED I WO	STRIPS AT AF	T BULKHEAD. NEW. P/N 600-6 8 PLY CO & TU 600-6 ACL (RESPECTIVELY).	
DISA	SSEMBLED A	AND ROTATED	D RH MAIN TIRE,	
SERV	ICED STRUT	S WITH NITRO	OGEN.	
TIGH	ALTERNATO	DRAIN FITTI	NG. S W/NEW, P/N MS25171-1S.	
- C/W	ALL ADS THR	ROUGHOUT TH	IS DATE	1.1
		Hobbs	1634.2	Sugar was
		CERTIFI	ED ALTIMETER(S) AND STATIC SYSTEMS	
			ED BY F.A.R. 91.411 I.A.W. F.A.R. PART 43,	
			RANSPONDER TESTS REQUIRED BY F.A.R.	
100			ACCOMPLISHED I.A.W. F.A.R. PART 43,	
		APPEND		
	al and a second	LT. ALT.		
		RT. ALT.	SN #2XPDR SN N14	
	2. 1 2 5 3	ADCIE	D.C.BLDENC BN	
	and the second			
		INSPECT	DATE 2/18/11	
		WESTERN	AVIONICS, INC. N 52 885 F.A.A. YH3R997L	
		WESTERN	AVIONICS, INC. N 52835 F.A.A. YH3R997L	
		1000		
the second s				
1997.3				
		SUB-TOTAL	this page	i da

	The construction of the second			
DATE	TOTAL TIME IN SERVICE	TACH OR RECORDING METER TIME	DESCRIPTION OF WORK PERFORME SIGNATURE & CERTIFICATE NO. OF PERSON PERFO	
		AA	Aircraft Maintenance	
	ach TE 09-08-1	7070 Ho	Aircraft Maintenance Summit Valley Rd. Hesperia, Ca. 92345 (760) 947-0807 Dbbs 1647.7 5288J S/N 421C-0158 MODEL 421C	
Ser	viced A/C sy ertify this Air	d airframe and i stem. Inspected rcraft has been nined to be in a	attery NEXT DUE 09-2013. Opened aircraft for annual flight controls, Replaced both main tires and tubes. d cabin components. Post service engine run checks good. inspected in Accordance With Annual/100 hr inspection n Airworthy condition.	
	Garry Abbo	ott	AP564570639IA	
Cle	Coil filter an/ Gap plug filter x	gs x Check	x Exhaust system x Oil service x x fuel x Lube x checked x	
MAKE: CES MODEL: 42 S/N: 421C0 REG. NO: N WORK ORI	1C 158	AVIONICE	Hangar One Avionics Repair Station No. 1HGR058B 2026 Palomar Airport Road Carlsbad, CA 92011 Phone: 760-929-2270	DATE: 4/ A/C TSN: HOBBS:
Avionics	s Entries		Filone. 100-929-2210	
Removed antenna a	Ameriking AK-45 nd coax supplied	50 121.5 / 243.0 MH	Iz ELT unit S/N 453168 and installed new Ameriking AK450 121.5 / 243.0 MH	
	ing open opphed	with new AK-450 u	init.	Iz ELT unit with S/N 500619. Replaced
I certify the		een inspected in acc	cordance with the requirements of FAR Section 91.207(d) and was found to b	
I certify the expiration	at the ELT has be date = Dec 2017 at the altimeter a	een inspected in acc 7. nd static system tes	Init.	e in satisfactory condition. Battery
I certify the expiration I certify the 91.413 ha Certified to #1 Altim #2 Altim	at the ELT has b date = Dec 2017 at the altimeter a ve been perform o 30,000 feet: heter P/N 5934PA	een inspected in acc , nd static system tes ed and found to con A-3 (A.86) S/N 4264 A-1 (A.69) S/N V294	cordance with the requirements of FAR Section 91.207(d) and was found to b ots and inspections required by 14 CFR 91.411 and transponder tests includin nply with 14 CFR Part 43 Appendices E and F. 98	e in satisfactory condition. Battery
I certify the expiration I certify the 91.413 ha Certified to #1 Altim #2 Altim Garmin During FA Pilot and 0 condition 1 new MS20 by Cessna	at the ELT has bi date = Dec 2017 at the altimeter a ve been perform o 30,000 feet: heter P/N 5934PA GTX 327 Transp R 91.411 inspec Copilot system at LS Lower Static I 0822-4D elbow fit a Aircraft Compar	een inspected in acc and static system tes ed and found to con A-3 (A.86) S/N 4264 A-1 (A.69) S/N V294 wonder P/N 011-0049 tion, found the Pilot s follows: LS Lower Port P/N CM2933-6 tting, and two new c ny under WO#'s 113	cordance with the requirements of FAR Section 91.207(d) and was found to b ets and inspections required by 14 CFR 91.411 and transponder tests includin inply with 14 CFR Part 43 Appendices E and F. 98 1 90-00 S/N 83711103 and Copilot Static system to show excessive leak rate. Troubleshot problem in Heated Static Port to be cracked and leaking, both #1 and #2 system Drain S S/N N/A (ref. FAA form 8130-3 dated Feb. 22, 2013 completed by Cessna Ai condition Drain Sumps P/N S1258-1 S/N N/A (ref. FAA form 8130-3 dated Sep 355245 and 1246177), removed and inspected and re-sealed the outer static.	e in satisfactory condition. Battery g data correspondence required by 14 C and determined the source of leaks for t umps defective and leaking. Installed Ni rcraft Company under WO# 1259473) w 0. 20, 2011 and Dec. 27, 2012 complete port plate and re-installed per 4C-43 13
l certify the expiration l certify the 91.413 ha Certified to #1 Altim #2 Altim #2 Altim Garmin During FA Pilot and 0 condition new MS20 by Cessna standards During FA evaluation 5934PA-1	at the ELT has bi date = Dec 2017 at the altimeter a we been perform o 30,000 feet: leter P/N 5934PA GTX 327 Transp R 91.411 inspect Copilot system as LS Lower Static I 0822-4D elbow fit a Aircraft Compai . Tested and insp R 91.411 inspect on, and it was deter (A.69) S/N V294	een inspected in acc and static system tes ed and found to con A-3 (A.86) S/N 4264 A-1 (A.69) S/N V294 wonder P/N 011-0049 tion, found the Pilot s follows: LS Lower Port P/N CM2933-6 tting, and two new c ny under WO#'s 113 pected the Pilot and tion found the Copili rimined the unit was 1 (ref. FAA form 81	cordance with the requirements of FAR Section 91.207(d) and was found to b its and inspections required by 14 CFR 91.411 and transponder tests includin inply with 14 CFR Part 43 Appendices E and F. 98 1 90-00 S/N 83711103 and Copilot Static system to show excessive leak rate. Troubleshot problem in Heated Static Port to be cracked and leaking, both #1 and #2 system Drain S S/N N/A (ref. FAA form 8130-3 dated Feb. 22, 2013 completed by Cessna Ai ondition Drain Sumps P/N S1258-1 S/N N/A (ref. FAA form 8130-3 dated Ser	e in satisfactory condition. Battery g data correspondence required by 14 C and determined the source of leaks for b umps defective and leaking. Installed Ni rcraft Company under WO# 1259473) w o. 20, 2011 and Dec. 27, 2012 complete port plate, and re-installed per AC-43.13 o negative results were observed. G9839 and sent to repair facility for condition exchange unit P/N
I certify the expiration I certify the 91.413 ha Certified to #1 Altim #2 Altim Garmin During FA Pilot and 0 condition 1 new MS20 by Cessna standards During FA evaluation 5934PA-1 system per	at the ELT has bi date = Dec 2017 at the altimeter a ve been perform o 30,000 feet: heter P/N 5934PA GTX 327 Transp R 91.411 inspec Copilot system as LS Lower Static I 0822-4D elbow fit a Aircraft Compai . Tested and insp R 91.411 inspec h, and it was dete (A.69) S/N V294 er 14 CFR Part 9	een inspected in acc and static system tes ed and found to con A-3 (A.86) S/N 4264 A-1 (A.69) S/N V294 wonder P/N 011-0043 tion, found the Pilot s follows: LS Lower Port P/N CM2933-6 ting, and two new c ny under WO#'s 113 pected the Pilot and tion found the Copil srmined the unit was 1 (ref. FAA form 813 1.411 paragraph (a)	cordance with the requirements of FAR Section 91.207(d) and was found to b ats and inspections required by 14 CFR 91.411 and transponder tests includin mply with 14 CFR Part 43 Appendices E and F. 98 1 90-00 S/N 83711103 and Copilot Static system to show excessive leak rate. Troubleshot problem i Heated Static Port to be cracked and leaking, both #1 and #2 system Drain S S/N N/A (ref. FAA form 8130-3 dated Feb. 22, 2013 completed by Cessna Ai ondition Drain Sumps P/N S1258-1 S/N N/A (ref. FAA form 8130-3 dated Sep 335245 and 1246177), removed and inspected and re-sealed the outer static Copilot static system I/A/W 14 CFR Part 91.411 paragraphs (a)(2) and (3). N ot Altimeter fails the calibration test. Removed unit P/N 5934PA-1 (A.69) S/N 30-3 dated Feb. 1, 2013 completed by Mid-Continent Instruments under WO#	e in satisfactory condition. Battery g data correspondence required by 14 C and determined the source of leaks for b umps defective and leaking. Installed Ne rcraft Company under WO# 1259473) w o. 20, 2011 and Dec. 27, 2012 completed port plate, and re-installed per AC-43.13 o negative results were observed. G9839 and sent to repair facility for condition exchange unit P/N
l certify the expiration l certify the 91.413 ha Certified to #1 Altim #2 Altim Garmin During FA Pilot and 0 condition new MS20 by Cessna standards During FA evaluation 5934PA-1 system per Installed n	at the ELT has bi date = Dec 2017 at the altimeter a we been perform o 30,000 feet: heter P/N 5934PA GTX 327 Transp R 91.411 inspec LS Lower Static I D822-4D elbow fil a Aircraft Compai . Tested and insp R 91.411 inspec h, and it was dete (A.69) S/N V294 er 14 CFR Part 9 missing STEC Au	een inspected in acc and static system tes ed and found to con A-3 (A.86) S/N 4264 A-1 (A.69) S/N V294 conder P/N 011-0045 tion, found the Pilot s follows: LS Lower Port P/N CM2933-6 tting, and two new c ny under WO#'s 113 pected the Pilot and tion found the Copilier rmined the unit was 1 (ref. FAA form 81: 1.411 paragraph (a) ito-Pilot 55X FLight	cordance with the requirements of FAR Section 91.207(d) and was found to b ats and inspections required by 14 CFR 91.411 and transponder tests includin mply with 14 CFR Part 43 Appendices E and F. 98 1 90-00 S/N 83711103 and Copilot Static system to show excessive leak rate. Troubleshot problem in Heated Static Port to be cracked and leaking, both #1 and #2 system Drain S S/N N/A (ref. FAA form 8130-3 dated Feb. 22, 2013 completed by Cessna Ai ondition Drain Sumps P/N S1258-1 S/N N/A (ref. FAA form 8130-3 dated Sep 335245 and 1246177), removed and inspected and re-sealed the outer static Copilot static system I/A/W 14 CFR Part 91.411 paragraphs (a)(2) and (3). N ot Altimeter fails the calibration test. Removed unit P/N 5934PA-1 (A.69) S/N is recommended for overhauled exchange. Procured and installed overhauled 30-3 dated Feb. 1, 2013 completed by Mid-Continent Instruments under WO# (2) instructions. No negative results were observed.	e in satisfactory condition. Battery g data correspondence required by 14 C and determined the source of leaks for b umps defective and leaking. Installed Ne roraft Company under WO# 1259473) w 0. 20, 2011 and Dec. 27, 2012 completed port plate, and re-installed per AC-43.13 o negative results were observed. G9839 and sent to repair facility for condition exchange unit P/N WRK0211066). Tested and inspected
I certify the expiration I certify the 91.413 ha Certified to #1 Altim #2 Altim Garmin During FA Pilot and 0 condition 1 new MS20 by Cessna standards During FA evaluation 5934PA-1 system per Installed n Relocated	at the ELT has bi date = Dec 2017 at the altimeter a ve been perform o 30,000 feet: heter P/N 5934PA GTX 327 Transp R 91.411 inspec Copilot system as LS Lower Static I 0822-4D elbow fil a Aircraft Compai . Tested and insp R 91.411 inspec h, and it was dete (A.69) S/N V294 er 14 CFR Part 9 missing STEC Au I RH aft facing IC	een inspected in acc and static system tes ed and found to con A-3 (A.86) S/N 4264 A-1 (A.69) S/N V294 wonder P/N 011-0049 tion, found the Pilot s follows: LS Lower Port P/N CM2933-6 ting, and two new c ny under WO#'s 112 pocted the Pilot and tion found the Copili trimined the unit was 1 (ref. FAA form 813 1.411 paragraph (a) tro-Pilot 55X FLight I S jacks from emerg	cordance with the requirements of FAR Section 91.207(d) and was found to b ats and inspections required by 14 CFR 91.411 and transponder tests includin nply with 14 CFR Part 43 Appendices E and F. 98 1 90-00 S/N 83711103 and Copilot Static system to show excessive leak rate. Troubleshot problem in Heated Static Port to be cracked and leaking, both #1 and #2 system Drain S S/N N/A (ref. FAA form 8130-3 dated Feb. 22, 2013 completed by Cessna Ai condition Drain Sumps P/N S1258-1 S/N N/A (ref. FAA form 8130-3 dated Sep 335245 and 1246177), removed and inspected and re-sealed the outer static Copilot static system I/A/W 14 CFR Part 91.411 paragraphs (a)(2) and (3). N of Altmeter fails the calibration test. Removed unit P/N 5934PA-1 (A.69) S/N is recommended for overhauled exchange. Procured and installed overhauled 30-3 dated Feb. 1, 2013 completed by Mid-Continent Instruments under WO# (2) instructions. No negative results were observed. Manual Supplement into the Pilot's Operating Handbook.	e in satisfactory condition. Battery g data correspondence required by 14 C and determined the source of leaks for b umps defective and leaking. Installed Ni rcraft Company under WO# 1259473) w o. 20, 2011 and Dec. 27, 2012 complete port plate, and re-installed per AC-43.13 o negative results were observed. G9839 and sent to repair facility for condition exchange unit P/N WRK0211066). Tested and inspected
l certify the expiration l certify the 91.413 ha Certified to #1 Altim #2 Altim Garmin During FA Pilot and Q condition new MS20 by Cessna standards During FA evaluation 5934PA-1 system pe Installed n Relocated Retermina	at the ELT has bi date = Dec 2017 at the altimeter a we been perform o 30,000 feet: leter P/N 5934PA GTX 327 Transp R 91.411 inspect Copilot system as LS Lower Static I D822-4D elbow fit a Aircraft Compai . Tested and insp R 91.411 inspect (A 69) S/N V294 er 14 CFR Part 9 missing STEC Au I RH aft facing IC ated BNC connect with Garmin Soft On as well as the	een inspected in acc and static system tes ed and found to con A-3 (A.86) S/N 4264 A-1 (A.69) S/N V294 wonder P/N 011-0049 tion, found the Pilot s follows: LS Lower Port P/N CM2933-6 ting, and two new c ny under WO#'s 113 bected the Pilot and tion found the Copili rmined the unit was 1 (ref. FAA form 813 1.411 paragraph (a) ito-Pilot 55X FLight I S jacks from emerge ctor at marker beacc tware service bulleti a units GPS software	cordance with the requirements of FAR Section 91.207(d) and was found to b ats and inspections required by 14 CFR 91.411 and transponder tests includin mply with 14 CFR Part 43 Appendices E and F. 98 1 90-00 S/N 83711103 and Copilot Static system to show excessive leak rate. Troubleshot problem Heated Static Port to be cracked and leaking, both #1 and #2 system Drain S S/N N/A (ref. FAA form 8130-3 dated Feb. 22, 2013 completed by Cessna Ai condition Drain Sumps P/N S1258-1 S/N N/A (ref. FAA form 8130-3 dated Sep 335245 and 1246177), removed and inspected and re-sealed the outer static Copilot static system I/AW 14 CFR Part 91.411 paragraphs (a)(2) and (3). N of Altimeter fails the calibration test. Removed unit P/N 5934PA-1 (A.69) S/N recommended for overhauled exchange. Procured and installed overhauled 30-3 dated Feb. 1, 2013 completed by Mid-Continent Instruments under WO# (2) instructions. No negative results were observed. Manual Supplement into the Pilot's Operating Handbook.	e in satisfactory condition. Battery g data correspondence required by 14 C and determined the source of leaks for b umps defective and leaking. Installed Ne rcraft Company under WO# 1259473) w b. 20, 2011 and Dec. 27, 2012 complete port plate, and re-installed per AC-43.13 o negative results were observed. G9839 and sent to repair facility for condition exchange unit P/N WRK0211066). Tested and inspected plugs into arm rest. t unit.
l certify the expiration l certify the 91.413 ha Certified to #1 Altim #2 Altim Garmin During FA Pilot and C by Cessna standards During FA evaluation 5934PA-1 system per Installed n Relocated Retermina Complied version 5. 190-00350	at the ELT has bi date = Dec 2017 at the altimeter a ve been perform o 30,000 feet: heter P/N 5934PA GTX 327 Transp R 91.411 inspect Copilot system as LS Lower Static I D822-4D elbow fil a Aircraft Compai . Tested and insp R 91.411 inspect A.699 S/N V294 er 14 CFR Part 9 nissing STEC Au I RH aft facing IC ated BNC connect with Garmin Soff 00, as well as the 6-02, Rev. G, Da with Garmin Soff 00, as well as the	een inspected in acc and static system tes ed and found to con A-3 (A.86) S/N 4264 A-1 (A.69) S/N V294 ionder P/N 011-004 tion, found the Pilot s follows: LS Lower Port P/N CM293-6 ting, and two new co ny under WO#'s 112 pected the Pilot and tion found the Copili rmined the unit was 1 (ref. FAA form 81: 1.411 paragraph (a) ito-Pilot 55X FLight I S jacks from emerg ctor at marker beacc tware service bulleti a units GPS software ted: December, 200 tware service bulleti a units GPS software	cordance with the requirements of FAR Section 91.207(d) and was found to b the and inspections required by 14 CFR 91.411 and transponder tests includin inply with 14 CFR Part 43 Appendices E and F. 98 1 90-00 S/N 83711103 and Copilot Static system to show excessive leak rate. Troubleshot problem is Heated Static Port to be cracked and leaking, both #1 and #2 system Drain S S/N N/A (ref. FAA form 8130-3 dated Feb. 22, 2013 completed by Cessna Ai nordition Drain Sumps P/N S1258-1 S/N N/A (ref. FAA form 8130-3 dated Sep 335245 and 1246177), removed and inspected and re-sealed the outer static Copilot static system I/A/W 14 CFR Part 91.411 paragraphs (a)(2) and (3). N ot Altimeter fails the calibration test. Removed unit P/N 5934PA-1 (A.69) S/N is recommended for overhauled exchange. Procured and installed overhauled 30-3 dated Feb. 1, 2013 completed by Mid-Continent Instruments under WO# (2) instructions. No negative results were observed. Manual Supplement into the Pilot's Operating Handbook. pency exit window to lower panel below window and installed cover plate with pon antenna. Marker beacon function check satisfactory using the NAV401 test in No. 1225 Rev A, Dated: October 19, 2012. Upgraded Garmin GNS 430W N e from 3.2 to version 5.0. Performed operational check, in accordance with G 8. System checks satisfactory on ground.	e in satisfactory condition. Battery g data correspondence required by 14 C and determined the source of leaks for b umps defective and leaking. Installed Ni rcraft Company under WO# 1259473) w o. 20, 2011 and Dec. 27, 2012 complete port plate, and re-installed per AC-43.13 o negative results were observed. G9839 and sent to repair facility for condition exchange unit P/N WRK0211066). Tested and inspected plugs into arm rest. t unit. Main software from version 3.30 to softwa armin GNS 430W Installation Manual P/I
l certify the expiration l certify the 91.413 ha Certified to #1 Altim #2 Altim Garmin During FA Pilot and 0 condition new MS20 by Cessna standards During FA evaluation 5934PA-1 system pe Installed in Relocated Retermina Complied version 5. 190-00350 Complied version 2.	at the ELT has bi date = Dec 2017 at the altimeter a we been perform o 30,000 feet: leter P/N 5934PA GTX 327 Transp R 91.411 inspec Copilot system as LS Lower Static I D822-4D elbow fil a Aircraft Compai . Tested and insp R 91.411 inspec (A.69) S/N V294 er 14 CFR Part 9 missing STEC Au I RH aft facing IC ated BNC connect with Garmin Soff 00, as well as the 6-02, Rev. G, Da with Garmin Soff 10, Performed op	een inspected in acc and static system tes ed and found to con A-3 (A.86) S/N 4264 A-1 (A.69) S/N V294 wonder P/N 011-0049 tion, found the Pilot s follows: LS Lower Port P/N CM2933-6 ting, and two new c ny under WO#'s 113 bected the Pilot and tion found the Copili rmined the unit was 1 (ref. FAA form 813 1.411 paragraph (a) ito-Pilot 55X FLight I S jacks from emergent tor at marker beacce tware service bulleti e units GPS software ted: December, 200 tware service bulleti e units GPS software ted: December, 200 tware service bulleti	cordance with the requirements of FAR Section 91.207(d) and was found to b ats and inspections required by 14 CFR 91.411 and transponder tests includin mply with 14 CFR Part 43 Appendices E and F. 98 1 90-00 S/N 83711103 and Copilot Static system to show excessive leak rate. Troubleshot problem Heated Static Port to be cracked and leaking, both #1 and #2 system Drain S S/N N/A (ref. FAA form 8130-3 dated Feb. 22, 2013 completed by Cessna Ai ondition Drain Sumps P/N S1258-1 S/N N/A (ref. FAA form 8130-3 dated Sep 335245 and 1246177), removed and inspected and re-sealed the outer static Copilot static system I/AW 14 CFR Part 91.411 paragraphs (a)(2) and (3). N ot Altimeter fails the calibration test. Removed unit P/N 5934PA-1 (A.69) S/N recommended for overhauled exchange. Procured and installed overhauled 30-3 dated Feb. 1, 2013 completed by Mid-Continent Instruments under WO# (2) instructions. No negative results were observed. Manual Supplement into the Pilot's Operating Handbook. Itency exit window to lower panel below window and installed cover plate with p on antenna. Marker beacon function check satisfactory using the NAV401 tes in No. 1225 Rev A, Dated: October 19, 2012. Upgraded Garmin GNS 430W M e from 3.2 to version 5.0. Performed operational check, in accordance with Gi 8. System checks satisfactory on ground. in No. 1225 Rev A, Dated: October 19, 2012. Upgraded Garmin GNS 530W M	e in satisfactory condition. Battery g data correspondence required by 14 C and determined the source of leaks for b umps defective and leaking. Installed Nu rcraft Company under WO# 1259473) w 5. 20, 2011 and Dec. 27, 2012 completed port plate, and re-installed per AC-43.13 o negative results were observed. G9839 and sent to repair facility for condition exchange unit P/N WRK0211066). Tested and inspected plugs into arm rest. t unit. Main software from version 3.30 to softwa armin GNS 430W Installation Manual P/I Main software from version 3.30 to softwa armin GNS 530W Installation Manual P/I
l certify the expiration I certify the 91.413 ha Certified th #1 Altim #2 Altim Garmin During FA Pilot and C condition new MS20 by Cessna standards During FA evaluation 5934PA-1 system per Installed in Relocated Retermina Complied version 5. 190-00351 Complied version 2. satisfactor	at the ELT has bi date = Dec 2017 at the altimeter a ve been perform o 30,000 feet: heter P/N 5934PA GTX 327 Transp GTX 327 Transp R 91.411 inspect Copilot system as LS Lower Static I 3822-4D elbow fil a Aircraft Compai . Tested and insp . R 91.411 inspect (A.69) S/N V294 er 14 CFR Part 9' nissing STEC Au I RH aft facing IC ated BNC connect with Garmin Soff 00, as well as the 6-02, Rev. G, Da with Garmin Soff 10. Performed og ry on ground.	een inspected in acc 7. Ind static system tes ed and found to con A-3 (A.86) S/N 4264/ A-1 (A.69) S/N V294 ionder P/N 011-0049 tion, found the Pilot s follows: LS Lower Port P/N CM293-6 ting, and two new co ting, and two new co ting, and two new co port P/N CM293-6 ting, and two new co s follows: LS Lower Port P/N CM293-6 ting, and two new co s follows: LS Lower Port P/N CM293-6 ting, and two new co ting, and two new co tion, the pilot and tion found the Copility tion found the copility t	 cordance with the requirements of FAR Section 91.207(d) and was found to b ats and inspections required by 14 CFR 91.411 and transponder tests includin nply with 14 CFR Part 43 Appendices E and F. 98 1 90-00 S/N 83711103 and Copilot Static system to show excessive leak rate. Troubleshot problem in Heated Static Port to be cracked and leaking, both #1 and #2 system Drain S S/N N/A (ref. FAA form 8130-3 dated Feb. 22, 2013 completed by Cessna Ai condition Drain Sumps P/N S1258-1 S/N N/A (ref. FAA form 8130-3 dated Sep. 335245 and 1246177), removed and inspected and re-sealed the outer static Copilot static system I/AW 14 CFR Part 91.411 paragraphs (a)(2) and (3). N ot Altimeter fails the calibration test. Removed unit P/N 5934PA-1 (A.69) S/N recommended for overhauled exchange. Procured and installed overhauled is 30-3 dated Feb. 1, 2013 completed by Mid-Continent Instruments under WO# (2) instructions. No negative results were observed. Manual Supplement into the Pilot's Operating Handbook. gency exit window to lower panel below window and installed cover plate with pon antenna. Marker beacon function check satisfactory using the NAV401 test in No. 1225 Rev A, Dated: October 19, 2012. Upgraded Garmin GNS 430W N e from 3.2 to version 5.0. Performed operational check, in accordance with Gills. System checks satisfactory on ground. in No. 1225 Rev A, Dated: October 19, 2012. Upgraded Garmin GNS 530W N e from 3.2 to version 5.0. Performed operational check, in accordance with Gills. System checks satisfactory on ground. in No. 1225 Rev A, Dated: October 19, 2012. Upgraded Garmin GNS 530W N e from 3.2 to version 5.0. Performed operational check, in accordance with Gills. System checks satisfactory on ground. in No. 1225 Rev A, Dated: October 19, 2012. Upgraded Garmin GNS 530W N e from 3.2 to version 5.0. Performed operational check, in accordance with Gills. System checks satisfactory on ground. 	e in satisfactory condition. Battery g data correspondence required by 14 C and determined the source of leaks for b umps defective and leaking. Installed Ne rcraft Company under WO# 1259473) w 0. 20, 2011 and Dec. 27, 2012 completed port plate, and re-installed per AC-43.13 o negative results were observed. G9839 and sent to repair facility for condition exchange unit P/N WRK0211066). Tested and inspected plugs into arm rest. t unit. Main software from version 3.30 to softwa armin GNS 430W Installation Manual P/I Main software from version 3.30 to softwa armin GNS 530W Installation Manual P/I
l certify the expiration I certify the 91.413 ha Certified to #1 Altim #2 Altim Garmin During FA Pilot and C condition new MS20 by Cessna standards During FA evaluation 5934PA-1 system pe Installed in Relocated Retermina Complied version 5. 190-00350 Complied version 2. satisfactol Updated C service pr	at the ELT has bi date = Dec 2017 at the altimeter a we been perform o 30,000 feet: leter P/N 5934PA GTX 327 Transp R 91.411 inspec Copilot system as LS Lower Static I D822-4D elbow fil a Aircraft Compai . Tested and insp R 91.411 inspec t, and it was dette (A.69) S/N V294 er 14 CFR Part 9 missing STEC Au I RH aft facing IC ated BNC connect with Garmin Soft 00, as well as the 6-02, Rev. G, Da with Garmin Soft 10. Performed op ry on ground. Garmin GDL69A ocedure in accor	een inspected in acc and static system tes ed and found to con A-3 (A.86) S/N 4264 A-1 (A.69) S/N V294 wonder P/N 011-0043 tion, found the Pilot s follows: LS Lower Port P/N CM2933-6 ting, and two new c ny under WO#'s 113 bected the Pilot and tion found the Copile rmined the unit was 1 (ref. FAA form 813 1.411 paragraph (a) ito-Pilot 55X FLight (a) ito-Pilot 55X FLight (a) ito-Pilot 55X FLight (a) ito-Pilot 55X FLight (a) to a marker beacc tware service bulleti e units GPS softwar ted: December, 200 tware service bulleti perational check, in software from versic dance with service to the service bulleti perational check, in	 cordance with the requirements of FAR Section 91.207(d) and was found to b the and inspections required by 14 CFR 91.411 and transponder tests includin nply with 14 CFR Part 43 Appendices E and F. 98 1 90-000 S/N 83711103 and Copilot Static system to show excessive leak rate. Troubleshot problem in Heated Static Port to be cracked and leaking, both #1 and #2 system Drain S S/N N/A (ref. FAA form 8130-3 dated Feb. 22, 2013 completed by Cessna A and topilot Static system to show excessive leak rate. Troubleshot problem in Heated Static Port to be cracked and leaking, both #1 and #2 system Drain S S/N N/A (ref. FAA form 8130-3 dated Feb. 22, 2013 completed by Cessna A and ondition Drain Sumps P/N S1258-1 S/N N/A (ref. FAA form 8130-3 dated Sep 335245 and 1246177), removed and inspected and re-sealed the outer static Copilot static system I/A/W 14 CFR Part 91.411 paragraphs (a)(2) and (3). N ot Altimeter fails the calibration test. Removed unit P/N 5934PA-1 (A 69) S/N is recommended for overhauled exchange. Procured and installed overhauled 30-3 dated Feb. 1, 2013 completed by Mid-Continent Instruments under WO# (2) instructions. No negative results were observed. Manual Supplement into the Pilot's Operating Handbook. gency exit window to lower panel below window and installed cover plate with pon antenna. Marker beacon function check satisfactory using the NAV401 test in No. 1225 Rev A, Dated: October 19, 2012. Upgraded Garmin GNS 430W M e from 3.2 to version 5.0. Performed operational check, in accordance with Gi8. System checks satisfactory on ground. in No. 1225 Rev A, Dated: October 19, 2012. Upgraded Garmin GNS 530W M e from 3.2 to version 5.0. Performed operational check, in accordance with Gi8. System checks satisfactory on ground. in No. 0809 Rev. A, Dated: October 19, 2012. Upgraded Garmin GNS 537 Main si accordance with Garmin GTX 327 Installation Manual P/N: 190-00187-02	e in satisfactory condition. Battery g data correspondence required by 14 C and determined the source of leaks for b sumps defective and leaking. Installed Ne rcraft Company under WO# 1259473) w o. 20, 2011 and Dec. 27, 2012 complete port plate, and re-installed per AC-43.13 o negative results were observed. G9839 and sent to repair facility for condition exchange unit P/N WRK0211066). Tested and inspected plugs into arm rest. t unit. Main software from version 3.30 to softwa armin GNS 430W Installation Manual P/N Alain software from version 3.30 to software armin GNS 530W Installation Manual P/N oftware from version 2.02 to software v. Q, Dated: March, 2012. System check oftware verification procedure and return
l certify the expiration l certify the 91.413 ha Certified to #1 Altim #2 Altim Garmin During FA Pilot and C by Cessna standards During FA evaluation 5934PA-1 system pe Installed in Relocated Retermina Complied version 5. 190-00350 Complied version 2. satisfactor Updated C service pr	at the ELT has bi date = Dec 2017 at the altimeter a we been perform o 30,000 feet: leter P/N 5934PA GTX 327 Transp R 91.411 inspec Copilot system as LS Lower Static I D822-4D elbow fil a Aircraft Compai . Tested and insp R 91.411 inspec (A.69) S/N V294 er 14 CFR Part 9 missing STEC Au I RH aft facing IC ated BNC connect with Garmin Soft 00, as well as the 6-02, Rev. G, Da with Garmin Soft 10. Performed op ry on ground. Garmin GDL69A ocedure in accor	een inspected in acc acc acc acc acc acc acc acc	 cordance with the requirements of FAR Section 91.207(d) and was found to b this and inspections required by 14 CFR 91.411 and transponder tests includin inply with 14 CFR Part 43 Appendices E and F. 98 1 90-00 S/N 83711103 and Copilot Static system to show excessive leak rate. Troubleshot problem in theated Static Port to be cracked and leaking, both #1 and #2 system Drain S S/N N/A (ref. FAA form 8130-3 dated Feb. 22, 2013 completed by Cessna Ai condition Drain Sumps P/N S1258-1 S/N N/A (ref. FAA form 8130-3 dated Sep 335245 and 1246177), removed and inspected and re-sealed the outer static Copilot static system I/A/W 14 CFR Part 91.411 paragraphs (a)(2) and (3). Ni of Altimeter fails the calibration test. Removed unit P/N 5934PA-1 (A 69) S/N recommended for overhauled exchange. Procured and installed overhauled 30-3 dated Feb. 1, 2013 completed by Mid-Continent Instruments under WO# (2) instructions. No negative results were observed. Manual Supplement into the Pilot's Operating Handbook. tency exit window to lower panel below window and installed cover plate with go on antenna. Marker beacon function check satisfactory using the NAV401 test in No. 1225 Rev A, Dated: October 19, 2012. Upgraded Garmin GNS 430W M e from 3.2 to version 5.0. Performed operational check, in accordance with Gi 8. System checks satisfactory on ground. in No. 1225 Rev A, Dated: October 19, 2012. Upgraded Garmin GNS 530W M e from 3.2 to version 5.0. Performed operational check, in accordance with Gi 8. System checks satisfactory on ground. in No. 0809 Rev. A, Dated: May 19, 2008. Upgraded Garmin GTX 327 Main si accordance with Garmin GTX 327 Installation Manual P/N: 190-00187-02, Re on 3.10 to 4.01 in accordance with service bulletin 1120 Rev. A. Performed so bulletin 1120 Rev. A. GDL69A checks satisfactory on ground. we has been repaired and/or inspected in accordance with current Regulations of th ervice. 	e in satisfactory condition. Battery g data correspondence required by 14 C and determined the source of leaks for b sumps defective and leaking. Installed Ne rcraft Company under WO# 1259473) wi o. 20, 2011 and Dec. 27, 2012 completed port plate, and re-installed per AC-43.13 o negative results were observed. G9839 and sent to repair facility for condition exchange unit P/N WRK0211066). Tested and inspected plugs into arm rest. t unit. Main software from version 3.30 to softwa armin GNS 430W Installation Manual P/N Main software from version 3.30 to softwa armin GNS 530W Installation Manual P/N poftware from version 2.02 to software w. Q, Dated: March, 2012. System check oftware verification procedure and return

ß

1

Page No. __

MAKE: CESSNA MODEL: 421C S/N: 421C0158 REG. NO: N5288J WORK ORDER: 20121008-1M Hangar One Avionics Repair Station No. 1HGR0588 2026 Palomar Airport Road Carlsbad, CA 92011 Phone: 760-929-2270 DATE: 4/1/2013 A/C TSN: 5820.7 HOBBS: 1723.3

Airframe Entries

Completed Annual Inspection IAW the Cessna Inspection Check list as applicable and FAR 43 Appendix "D"

VIONICS

Completed General Inspection of all aftermarket installations not listed with specific instructions for continued airworthiness---Inspect IAW AC43-13-1B Chapter 5 Section 2.

CW AD# 81-09-09 (Combustion Heater 250 Insp) - by inspection. Found combustion heater circulation air blower motor inoperative. Removed blower P/N 721984 S/N WO#0116134 and sent to Aircraft Heating and Electrical, Inc. Unit was returned with a work status of "Overhauled" - See FAA form 8130-3 WO# 0116134 dated 01-16-2013. Reinstalled same P/N and S/N, Replaced 3" duct P/N CEET12, 3 3/4" duct P/N SCEET15 (05-30615), 4 1/2" duct P/N SCEET18 (05-30618) and dust P/N CM3211-18A. System function check satisfactory. Inspection per AD due again in 250 Heater Hrs. Overhaul Due: 1169.0, Heater Hobbs Currently at 615.9 Hrs.

CW AD# 84-26-02 (Paper Induction Air Filter) for LH and RH engines by replacement of Induction Air Filters 2ea P/N P10-8421. Replacement mandatory ea 500hrs.

CW AD# 2000-01-16 (Exhaust System Inspection) for the LH and RH engines by inspection per para (d) and replacement of the following new exhaust components which restart the clock for paragraphs (b)(c)(e) and (g): 2ea Turbo Wyes P/N 8294-15C - LH S/N IRY75-0001 and RH S/N IRY75-0003 using 2ea gaskets P/N 002957006 with 8ea new bolts P/N MS20036-10 and 8ea new nuts P/N MS20500-624. 4ea Slip Joint Assemblies P/N K5355108-4 LH S/Ns 221636 and 221638, RH S/Ns 221646 and 229643 using new Springs 40ea P/N 1298119-4, Bolts 20ea P/N AN3C13 and 20ea P/N AN3C15, nuts 40ea P/N AN310C3 and as required washers for proper spring compression P/N AN960C10 with new stainless cotter pins. 2ea Flex Elbows P/N K9910300-03 LH S/N 220995 and RH S/N 220998 using new clamps 2ea P/N U10C050-240M and 4ea new wastegate gaskets P/N 9910306-1 with 16ea new bolts P/N AN4C11A and 16ea new nuts P/N MS20500-428. 2ea Tail Pipes P/N K9910300-01 LH S/N 220391 using 6ea flange Gaskets P/N 652458 and 24ea nuts P/N MS20500-428. 2ea Exhaust Stacks LH Engine P/N K9910295-15 LH S/N 222330 and RH S/N 222391 using 6ea flange Gaskets P/N 652458 and 24ea nuts P/N MS20500-428. 2ea Exhaust Stacks LH Engine P/N K9910295-16 LH S/N 222395 and RH S/N 222396 using 6ea flange Gaskets P/N 652458 and 24ea nuts P/N MS20500-428. Relocated, drilled and reinstalled existing EGT probes IAW JPI installation instructions Report No. 760. Paragraph (b) next due in 50 hours, paragraphs (c) and (e) next due in 5 years and then annually thereafter, paragraphs (d) next due in 500 hours, paragraph (f) not applicable at this time due to no multi-segment clamps installed and paragraph (g) next due in 2500

Removed expired fire extinguisher and bracket assembly and installed new C352TS 2.5lbs Halon 1211 fire extinguisher and associated bracket.

Fabricated and installed various missing placards throughout aircraft interiror and exterior. Installed new turn limits placard P/N 5100181-36.

Aircraft re-weighed this maintenance event and new center of gravity computed, see W&B form this date. General Electrodynamics Corporation Scales Cal Date Dec 2012, P/N DLI0000P-6 S/N I4139. Scale calibration due December 31st 2013. Inserted new weight and balance reflecting this event into the POH and aircraft permanent record.

Executed FAA form 337 due to a lack of documentation for the Removal of Brackett Induction Air Filters - reference FAA form 337 dated 09-19-1996,

Executed FAA form 337 due to a lack of documentation for the installation of Shadin digital Fuel Flow System in accordance with STC# SA607GL.

Executed FAA form 337 due to a lack of documentation for the installation of RAM Vacuum Pump Cooling Shrouds in accordance with STC#SA3721AW.

Executed FAA form 337 due to the lack of documentation for installation of Rosen sun visors in accordance with STC# SA5137NM.

Removed pipe fitting from LH Engine fuel flow transducer due to a blue stain. Cleaned fitting and applied pipe thread sealant and reinstalled. Leak and operational check satisfactory.

Fabricated new fuel pressure regulator bracket on the left engine.

Replaced LH alternate air door magnet P/N 9910016-1.

Repaired upper deck pressure to fuel limiter tube by replacement of aluminum tube segment using new b-nuts 2ea P/N AN818-6D and Sleeves 2ea P/N MS20819-6D.

Removed leaking LH wastegate actuator P/N 480164-9001 and installed an overhaul exchange unit P/N 481064-9001 S/N THR0108 - See Approved Turbo Components, Inc. FAA form 8130-3 WO# 32913 dated 03-26-2013. Replaced inlet fitting 1ea P/N AN833-6 with nut P/N AN924-6 and o-ring P/N MS28778-6 and outlet fitting 1ea P/N AN833-4 with nut P/N AN924-4 and o-ring P/N MS28778-4. Ground run, operational and leak check satisfactory.

Reset LH and RH Propeller Governor Controls for adequate travel and cockpit cushion per the Cessna 421C maintenance manual.

Adjusted LH and RH propeller governors for proper governing and feathering operation per the Cessna 421C maintenance manual.

Reset LH and RH Throttle Controls for adequate travel and cockpit cushion per the Cessna 421C maintenance manual.

Reset LH and RH Mixture Controls for adequate travel and cockpit cushion per the Cessna 421C maintenance manual.

Freed stiff RH alternate air control.

Freed stiff LH alternate air control.

Cleaned LH and RH engines and engine compartments.

Removed Propeller Anti-Ice amp gauge, found wiring reversed, cleaned contacts, lengthen and reterminated wires. Replaced gauge to panel seal P/N M83248-2-21 and reinstalled. Found one boot on the LH propeller with wires reversed, corrected wiring and checked system for proper operation - system function check on the ground was satisfactory.

Removed Co-Pilot's Turn Coordinator P/N 1394T100-7Z S/N 9007-3 and installed new exchange Mid-Continent Turn Coordinator P/N 1394T1007B S/N L12-11164. Leveled aircraft and secured Turn Coordinator.

Removed Pilot's Attitude gyro P/N 25-501-031-1 S/N T42292 and installed a Mid-Continent overhaul exchange unit P/N 23-501-032-2 S/N T42501 - See FAA form 8130-3 WO# WRK0211124 dated 01-17-2013 with a work status of " Modified." Leveled aircraft and connected Airborne vacuum test kit #343. Adjusted attitude gyro for level at 5 inches and torqued gyro to panel with existing retaining screws. Vacuum system and associated instruments function test satisfactory.

Removed Cabin Altimeter/Differential Pressure Gauge P/N C668516-0104 S/N 12683 and sent to Mid-Continent Instruments for repair. Unit was returned with a work status of "Repaired." See FAA form 8130-3 WO# WRK0210963 dated 01-17-2013. Aircraft needs check flight for verification of instrument functionality.

Removed Glareshield to determine problem with RH 3 in 1 gauge. Found wire chaffed through and hanging be a strand. Splice repaired wire with a Raychem environmental splice. Reinstalled Glareshield. Function check of RH 3 in 1 gauge satisfactory.

Removed existing LH and RH Cleveland Wheels and Brakes kit P/N 199-76 and installed new Cleveland Wheels and Brakes Kit P/N 199-76. Flushed brake system,





bled and serviced master cylinders using MIL-PRF-5606H hydraulic fluid. Completed Cleveland "Brake Lining Conditioning Procedure" per Cleveland Document No. PRM14A Rev A dated 07-01-1995.

Removed all 3 Landing gear retract system hydraulic actuators and sent to Aerospace Products / D-J Engineering in Augusta, KS. for overhaul. LH Actuator P/N 9910136-2 S/N 1512 was returned with a work status of "Overhauled" - See FAA form 8130-3 WO# 1512 dated 02-06-2013. Unit S/N 1512 was reinstalled in the LH position with new upper bolt P/N NAS464-10-24, nut P/N MS17826-10 and washer P/N AN960-1016, replaced lower nut P/N MS17826-12. RH Actuator P/N 9910136-2 S/N 1519 was returned with a work status of "Overhauled" - See FAA form 8130-3 WO# 1519 dated 02-06-2013. Unit S/N 1519 was reinstalled in the RH position with new upper bolt P/N NAS464-10-24, nut P/N MS17826-10 and washer P/N AN960-1016. Nose Actuator P/N 9910139-3 S/N 1513 was returned with a work status of "Overhauled" - See FAA form 8130-3 WO# 1513 dated 02-06-2013. Unit S/N 1513 was reinstalled in the nose position with new bolts upper P/N NAS6204-11D and lower bolt P/N NAS6204-15D - installed new 2ea AN833-4D fittings and 2ea AN924-4D nuts to nose gear actuator. Landing gear system checks performed per the Cessna 421C maintenance manual and Cessna Service Bulletin MEB91-12. Performed several fault free cycles including an emergency "Blow down" Re-Serviced blow down bottle with N2.

Rigged landing gear and gear doors as required after overhaul of actuators per the Cessna 421C maintenance manual ...

Fabricated new support for hydraulic line in RH gear wheel well.

Reduced play in Main Landing Gear Torque Links by installing 1ea S1450-4-152-032 shims and applying proper torque to center knee of both LH and RH links.

Replaced Main Landing Gear Hydraulic System Filters 2ea P/N AN6235-2A and 2ea O-rings P/N MS28775-222.

Removed LH Hydraulic pump P/N 24194-RAA S/N 172 and installed an overhaul exchange unit by Quality Aircraft Accessories, Inc. Unit was received with a work status of "Overhauled" See FAA form 8130-3 WO# WO1170006915 dated 02-06-2013. Installed unit P/N 24194-RAA S/N 43 using a new P/N S3346-1 gasket. Run-up of LH engine revealed that the Hydraulic annunciator extinguishes appropriately and no leaks were noted.

Hydraulic fluid leak in RH engine compartment. Cleaned engine and engine compartment, replaced seals in hydraulic filter assembly. Engine run up revealed no leaks.

Removed lower Nose Strut assembly and replaced the following parts: 1ea P/N 0841200-19 retainer ring, 1ea P/N 0841200-25 retainer ring, 1ea P/N A2748 polypack seal, 1ea P/N MS28775-034 o-ring, 1ea P/N MS28775-333 o-ring and 1ea S3376 scraper ring. reassembled strut assembly and serviced with MIL-PRF-5606H Hydraulic fluid and nitrogen per the Cessna 421C Maintenance Manual.

Removed RH main landing gear, cleaned and repacked strut assembly using the following new parts: 1ea A2749 poly pack seal, 1ea MS28778-015 o-ring, 1ea MS28775-228 o-ring, 1ea MS28776M2-22 wiper ring and 1ea MS28869-2 valve. Reinstalled RH main landing gear and serviced strut with MIL-PRF-5606H Hydraulic fluid and N2. All work done per the Cessna 421C maintenance manual.

Removed landing gear blow down bottle due to an undetermined leak. Isolated leak to bottle valve. Removed existing valve and adapter and replaced with an MS28889-2 serviceable valve and new adapter. Reinstalled bottle and serviced with N2 per the Cessna 421C maintenance manual. Leak check satisfactory.

Removed LH and RH 3D2352-01 wing deice flow valves S/Ns N808 and N2369 and sent to Quality Aircraft Accessories/AMETEK B&S Aircraft Parts & Accessories for overhaul. Unit S/N N808 was returned with a work status of "Overhauled" See FAA form 8130-3 WO# R13-00249 dated 02-28-2013, Reinstalled S/N N808 into the LH wing. Unit S/N N2369 was returned with a work status of "Overhauled" See FAA form 8130-3 WO# R13-00250 dated 02-28-2013, Reinstalled S/N N2369 Into the RH wing. Reterminated wiring in both LH and RH wings. Connected Airborne vacuum test kit #343, system function check as stated below.

Removed Surface De-ice annunciator switch, cleaned and bench tested unit. Switch opens and closes appropriately with pressure applied and/or removed. Pressures values used as listed in the Cessna 421C maintenance manual. Unit reinstalled and wiring re-terminated

Patch repaired aircraft de-ice boots as required using BFGoodrich patch kit and sealed pin holes in various locations with sealant. Connected Airborne vacuum test kit #343 and found boots to be weathered beyond repair. Surface Deice system deactivated and placarded in accordance with 14 CFR Part 91.213.

Found wiring on Heated windshield disconnected, connected wiring and performed operational check of windshield heat - System function check satisfactory.

Replaced Propeller Deice brush assemblies. Cleaned slip rings and readjusted clearances per the Cessna 421C maintenance manual. System function check satisfactory

Performed a bench functional test of air conditioning hydraulic system annunciator pressure switch. Switch body leaks hydraulic fluid. Removed discrepant switch assembly. Installed serviceable 9910287-11 (ITT 98087-1106P61) air conditioning hydraulic pressure switch S/N 4614 (A3Q96) and repaired air conditioning system wiring in the Right nacelle area by removal of deteriorated wiring and connectors and reterminated using new wiring segments and new connectors. Removed Air Conditioning compressor, inspected, cleaned, flushed and drained oil. Compressor serviced with 7oz of 5GS Suniso Refrigeration Oil and reinstalled. Disconnected various lines, removed receiver dryer and flushed all Air Conditioning system lines, including condenser. Removed LH and RH evaporator/motor assemblies, disassembled plastic ducting and squirrel fans, cleaned out debris from cooling fins, fans and flushed units, reassembled units and sealed ducting assemblies Installed new expansion valves LH and RH 2ea P/N ES26101-4 using new adapters 2ea P/N 1523447-3 with new M83248/1-908 o-rings. Reinstalled evaporator/motor assemblies and reterminated electrical connections. Replaced receiver dryer using P/N ES343029-3 and installed new adapter fittings. Reconnected and torqued all refrigerant lines and fittings. Pressurized Air-Conditioning system with N2 and found no leaks. Evacuated system. Ran RH engine and serviced system with R-12 refrigerant. Air-Conditioning system function check satisfactory.

Cleaned outflow and safety valve seals with alcohol.

Replaced 6ea pressurization belly drain seals P/N CM3575-4, replaced 4ea retainers P/N 5111339-3.

Replaced various nut plates for inspection panels as required. No fuel panel nut plates were replaced at this time.

Removed stinger and examined tail tie-down attach point due to tie down being detached. Secured tie-down with new bolt P/N AN3-6A, washer P/N NAS1149F0363P, and nut P/N MS21044N3. No significant damage noted. Skin deformation minor.

Reformed nose baggage door to open and close with proper clearance.

Secured lower cabin door actuator on lower cabin door.

Secured Lower cabin door interior trim.

Checked all airframe and engine fuel screens and found clean. Reused seals for LH and RH fuel strainer bowls. Leak checks satisfactory.

Cleaned fuel stain from LH wing outboard fuel drain, actuated drain valve and monitored for leaks. No leaks noted.

Replace LH and RH outer fuel cap o-rings and installed new LH and RH required fuel placarding using 2ea P/N 5026015-15 placards.

Disconnected cross feed shut-off cable at fuel transfer valve, cleaned, lubricated, re-secured and reinstalled. System function check satisfactory.

Removed Stall Warning Transmitter P/N 186-8 (No S/N) and installed and overhauled exchange transmitter P/N 186-8 O/H S/N 1337 See FAA form 8130-3 WO# R44741 dated 1-21-2012. On ground function check satisfactory, unit ready for flight check adjustments.

Removed elevator trim actuator, aileron trim actuator and rudder trim actuators, disassembled, cleaned and replaced the following parts: ELEVATOR TRIM: new Removed elevator trim actuator, and on the bound of the rest of the state of the state of the rest of the following parts. ELEVATOR TRI housing P/N 0310296, new drive end bearing P/N 0310297-1, new screw end bearing P/N 0310298-1, new seal P/N MS28775-010, 1ea new snap ring P/N MS16624-1081 and new groove pins 4ea P/N MS35675-13 and 2ea P/N GP3H094x0375-14. RUDDER TRIM: new drive end bearing P/N 0310297-1, new screw 2 of 3

end bearing P/N 0310298-1, new seal P/N MS28775-010 and new groove pins 4ea P/N MS35675-13 and 2ea P/N GP3H094x0375-14. AlLERON TRIM: new bearing P/N MS27641-5 (S3952-5), new seal P/N MS28932C02-05, new Snap rings 2ea P/N MS16624-4112 and new roll pins 2ea P/N MS35675-10. All units assembled using Molykote 33 grease. Alleron and rudder trim tab attach bolts replaced using 2ea new P/N NAS464P3-9 bolts, all 3 trim actuator tab attach bolts installed using new P/N MS17826-3 nuts. All units reinstalled and trim systems rigged per the Cessna 421C maintenance manual.

Repositioned fuel line beneath cabin floor to avoid flap cable chaffing. Noted cable to be clear of flap motor.

Retensioned rudder trim cables per the Cessna 421C maintenance manual.

Removed incorrect cable segments from LH aft elevator trim system and installed OEM cable assembly P/N 520008-43 with new trim control cable master link P/N S2294-25. Elevator trim cable tensions and deflections reset per the Cessna 421C maintenance manual.

Cleaned RH flap return cable and found serviceable after clean-up, reset cable tensions per the Cessna 421C maintenance manual.

Deactivated and placarded "Inop", the propeller synchronization system in accordance with 14 CFR Part 91.213.

Removed RH retractable landing light assembly P/N 45-0148-3 (No S/N) and sent to Quality Aircraft Accessories/Aerospace Turbine Rotables, Inc. Unit was returned with a S/N of 75603 and a work status of "Overhauled." See FAA form 8130-3 WO# WO33276 dated 01-28-2013. Unit reinstalled in RH position, function check satisfactory.

Replaced RH cowling landing light using lamp P/N 4591.

T&B and spare annunciators do not light when tested, checked lamps and found no fault. Troubleshooting revealed the annunciator module had a broken wire for spare power and the turn coordinator wire was not terminated at the turn coordinator. Removed annunciator module and solder repaired broken spare power. Reterminated Pilot's turn coordinator connector. Function check of annunciator module satisfactory.

Replaced Deice light lens P/N 0820501-11.

Inspected tail position light and found lamp filament broken. Checked voltage, has 28 volts at lamp. Replaced tail nav lamp P/N 305.

Pulselite system deactivated and placarded in accordance with 14 CFR Part 91.213.

Rerouted Tach generator wires and connectors in LH and RH engines for better service loop.

Capped and stowed unused wiring in RH engine compartment.

Wiring harnesses for LH and RH cowling lights not secured, have no connector as needed for servicing-maintenance. Installed Mate and Lock harness disconnects and secured wire bundles in accordance with AC 43-13.1b standards. No further action taken.

Replaced Alternator on/off switches 2ea P/N MS24524-23. Replaced Emergency Field switch P/N MS35059-23. Replaced LH and RH field fuses 2ea P/N HKP and fuses 2ea P/N S3037A3. Cleaned rotary selector switch with contact cleaner and adjusted paralleling per the Cessna 421C maintenance manual.

LH "Alternator Out" light remains illuminated with switch on and charging system operational - troubleshooting revealed that the Alternator out sensor was defective. Installed new Sensor P/N 9910094-3 S/N 80181814. System function check satisfactory.

Removed battery box, cleaned corrosion and reinstalled.

Replaced LH and RH Vacuum relief filters 2ea P/N RAB3-5-1.

Cleaned FOD and debris below floor boards as practicable.

Installed 2ea new S2237-3 friction bushings to Pilot and Co-Pilot seat belt.

SIGNED.

ONT

Annual / I certify that this AIRCRAFT has been inspected in accordance with an Annual Inspection and was determined to be in airworthy condition -end-

This aircraft, airframe, or appliance identified above has been repaired and/or inspected in accordance with current Regulations of the Federal Aviation Administration and Manufacturer data and is approved for return to service.

DATE: 4/1/2013

Mike Clingerman Certified Repair Station No. 1HGR058B Work Order: 20121008-1M

Printed by EBis 3 (datcomedia.com)

MAKE: CESSNA MODEL: 421C S/N: 421C0158 REG. NO: N5288J WORK ORDER: 20130515-1M Hangar One Avionics Repair Station No. 1HGR0588 2026 Palomar Airport Road Carlsbad, CA 92011 Phone: 760-929-2270 DATE: 6/6/2013 A/C TSN: 5824.5 HOBBS: 1727.1

Airframe Entries

Removed cabin door perimeter seal and horizontal seal, then installed new perimeter seal P/N 9910007-3 and new horizontal seal P/N 9910007-4. All work IAW Cessna 421C Service Manual. Aircraft ready for check flight.

Removed RH engine gauge (3-in-1: Oil PSI, CHT, Oil Temp) P/N C662019-0101 S/N A91290 and installed overhauled exchange gauge P/N C662019-0101RX S/N T2143. Reference Aero-Mach Labs, Inc. FAA Form 8130-3 and Work Order 0406598 dated March 11, 2013 for status work "OVERHAULED." System function check satisfactory.

Cleaned RH engine and performed engine run-up. Suspected leak was residual only. Check again after next flight.

Removed vertical stabilizer pneumatic de-icer ("boot") and both horizontal stabilizer pneumatic de-icers IAW Cessna 421C Service Manual. Installed new vertical stabilizer pneumatic de-icer 1 ea. P/N SMR5070-14 and new horizontal stabilizer pneumatic de-icers 2 ea. P/N SMR5070-11 (RH S/N 1819, LH S/N 1819,). Reference B/E Aerospace Inc. dba SMR Technologies Inc. FAA Forms 8130-3 for Work Order (SMR Order No.) 73859 dated May 24, 2013, Work Order (SMR Order (SMR Order No.) 73158 dated March 1, 2013, and Work Order (SMR Order No.) 61726 dated May 4, 2010. All Status/Work "NEW." Installation of all pneumatic deicers IAW Cessna 421C Service Manual and B/E Aerospace Installation and Maintenance Manual (SMR Report No. 97-33-047 Rev. 10 dated March 26, 2010). System function check revealed tail deice shuttle valve sticking. Removed tail deice shuttle valve P/N 3D2352-01 S/N N970 and sent to B&S for overhaul. Stowed air inlet fitting and wiring and connected tail deice system vacuum source using AN adapters. System remains deactivated per FAR 91.213.

This aircraft, airframe, or appliance identified above has been repaired and/or inspected in accordance with current Regulations of the Federal Aviation Administration and Manufacturer data and is approved for return to service.

DATE: 6/6/2013

SIGNED: David Corish Certified Repair Station No. 1HGR058B Work Order: 20130515-1M

Printed by EBis 3 (datcomedia.com)

			Constant Annual Constant of the second second		Page No.
DATE	TOTAL TIME IN SERVICE	TACH OR RECORDING METER TIME	DESC SIGNATURE & C	CRIPTION OF WORK PERFORMED- ERTIFICATE NO. OF PERSON PERFORMING WORK	
IAKE: CESSNA IODEL: 421C /N: 421C0158 EG. NO: N5288 /ORK ORDER: 2	20130515-1M		Hangar One Avionics Repair Station No. 1HGR058B 2026 Palomar Airport Road Carlsbad, CA 92011 Phone: 760-929-2270		DATE: 6/6// A/C TSN: 58 HOBBS: 17
ilot reports ma evision L date	TT switch P/N 9 arker beacon inc d June 2003, Sy	op. Performed func ystem test okay. C	wheel. Operational check okay. tion test of marker beacon system pe ould not duplicate the discrepancy.	r the Garmin GMA340 Audio Installation Manual Document	
			- percenta production	ertical speed mode." Performed system test of auto-pilot, in	
nis aircraft, airfi lanufacturer dai ATE: 6/6/2013			X	ccordance with current Regulations of the Federal Aviation Adm	ninistration and
		SIGNED David Cor	ish		ork Order: 20130515
		Certified F	Repair Station No. 1HGR058B	Printed by E	Bis 3 (datcomedia.c
	н				
MAKE: CESSNA MODEL: 421C S/N: 421C0158 REG. NO: N528 WORK ORDER: 20130703-2M	8J	Re 202 Car	ngar One Avionics bair Station No. 1HGR058B 6 Palomar Airport Road Isbad, CA 92011 ine: 760-929-2270	DATE: 9/5/2013 A/C TSN: 5831.7 HOBBS: 1734.3	
AMETEK B&S A	er flow valve P/N 31 52-01 S/N N970, ii iircraft Parts & Acco frame, or appliance	in fuselage, IAW Cess cessories FAA Form 8	na Service Many/al, for operation of emper 130-3 and Work Order #R13-02915 dated been repaired and/or inspected in accorda	aluation / repair, then installed overhauled de-icer flow nage surface-mounted de-ice boots. (Reference June 11, 2013, for Status/Work "OVERHAULED."	
DATE: 9/5/20		SIGNED:	kr	Work Order: 20130703-2M	
		David Corish		Printed by EBis 3 (datcomedia.com)	
		Certified Repair S	tation No. 1HGR058B		Second Charles
MAKE: CESSNA MODEL: 421C S/N: 421C0158 REG. NO: N5288 WORK ORDER: 00130607-2M	-	Certified Repair S	ation No. 1HGR058B ngar One Avionics air Station No. 1HGR058B 3 Palomar Airport Road sbad, CA 92011 ne: 760-92-9270	DATE: 9/5/2013 A/C TSN: 5831.7 HOBBS: 1734.3	
MODEL: 421C S/N: 421C0158 REG: NO: N5288 WORK ORDER: 20130607-2M Airframe En	BJ Itries	Certified Repair S Ha Control Control Contro	ngar One Avionics air Station No. 1HGR058B 8 Palomar Airport Road sbad, CA 92011 ne: 760-929-2270	A/C TSN: 5831.7	
MODEL: 421C S/N: 421C0158 REG NO: N5288 WORK ORDER: 20130607-2M Airframe En Replaced cracke Manual.	BJ Itries Id LH glass heated rame, or appliance Id Manufacturer di	Certified Repair S Ha Rep 202 Car Pho d windshield with new	ngar One Avionics air Station No. 1HGR058B 3 Palomar Airport Road sbad, CA 92011 ne: 760-929-2270 LH glass heated windshield P/N 9910049-5 been repaired and/or inspected in accorda	9 S/N 13200H3137. All work IAW Cessna Service	
MODEL: 421C S/N: 421C0158 REG. NO: N5288 WORK ORDER: 20130607-2M Airframe En Replaced cracke Manual.	BJ Itries Id LH glass heated rame, or appliance Id Manufacturer di	Certified Repair S Ha Rep 202 Car Pho d windshield with new e identified above has ata and is approved SIGNED: David Corish	ngar One Avionics air Station No. 1HGR058B 3 Palomar Airport Road sbad, CA 92011 ne: 760-929-2270 LH glass heated windshield P/N 9910049-5 been repaired and/or inspected in accorda	A/C TSN: 5831.7 HOBBS: 1734.3	
MODEL: 421C S/N: 421C0158 REG NO: N5288 WORK ORDER: 20130607-2M Airframe En Replaced cracke Manual.	BJ Itries Id LH glass heated rame, or appliance Id Manufacturer di	Certified Repair S Ha Rep 202 Car Pho d windshield with new e identified above has ata and is approved SIGNED: David Corish	ngar One Avionics air Station No. 1HGR058B 5 Palomar Airport Road sbad, CA 92011 ne: 760-929-2270 LH glass heated windshield P/N 9910049-1 been repaired and/or inspected in accorda or return to service.	A/C TSN: 5831.7 HOBBS: 1734.3	
MODEL: 421C S/N: 421C0158 REG NO: N5288 WORK ORDER: 20130607-2M Airframe En Replaced cracke Manual.	BJ Itries Id LH glass heated rame, or appliance Id Manufacturer di	Certified Repair S Ha Rep 202 Car Pho d windshield with new e identified above has ata and is approved SIGNED: David Corish	ngar One Avionics air Station No. 1HGR058B 5 Palomar Airport Road sbad, CA 92011 ne: 760-929-2270 LH glass heated windshield P/N 9910049-1 been repaired and/or inspected in accorda or return to service.	A/C TSN: 5831.7 HOBBS: 1734.3	
MODEL: 421C S/N: 421C0158 REG NO: N5288 WORK ORDER: 20130607-2M Airframe En Replaced cracke Manual.	BJ Itries Id LH glass heated rame, or appliance Id Manufacturer di	Certified Repair S Ha Rep 202 Car Pho d windshield with new e identified above has ata and is approved SIGNED: David Corish	ngar One Avionics air Station No. 1HGR058B 5 Palomar Airport Road sbad, CA 92011 ne: 760-929-2270 LH glass heated windshield P/N 9910049-1 been repaired and/or inspected in accorda or return to service.	A/C TSN: 5831.7 HOBBS: 1734.3	
MODEL: 421C S/N: 421C0158 REG NO: N5288 WORK ORDER: 20130607-2M Airframe En Replaced cracke Manual.	BJ Itries Id LH glass heated rame, or appliance Id Manufacturer di	Certified Repair S Ha Rep 202 Car Pho d windshield with new e identified above has ata and is approved SIGNED: David Corish	ngar One Avionics air Station No. 1HGR058B sbad, CA 92011 ne: 760-929-2270 LH glass heated windshield P/N 9910049-0 been repaired and/or inspected in accorda or return to service.	A/C TSN: 5831.7 HOBBS: 1734.3	

g

DATE	TOTAL TIME IN SERVICE	TACH OR RECORDING METER TIME	DESCRIPTION OF WORK PERFORMED— SIGNATURE & CERTIFICATE NO. OF PERSON PERFORMING WORK					
		Aviation 1	nc	Sec.				
	7000 Merri	ll Ave. Hanga	ur A-340 Chino CA 91710					
	N52881 HO	bbs: 1753 4	11/09/2013					
1996	ail and filte	r changed bot	h engines, re-connected loose hose on oil separator on right	A New States				
	engine rem	oved Left tac	h. Generator, then re-installed with new gaskets and new					
	cannon nlu	o after bench	check done by Precision instruments. Installed new hobbs					
3	meter on pi	lot's instrume	nt panel and got it wire to master switch circuit per customer's					
	request. Re	moved throttl	e quadrants and center console to access panel lighting					
	rheostats, re	eplaced 3 bro	ken rheostats with new. Removed left landing light and					
	replaced me	otor, gears an	d reinstalled back under the wing, replaced wing de-ice light					
	bulb, instal	led new LED	taxi light on nose gear.					
	Agustin Ce	ndejas AP260	06564 delenan					
	U	0	egt and					
	Chino	Aviation	Inc					
			ar A-340 Chino CA 91710					
		obbs: 1765 1						
	Replaced 1	laced main door, nose baggage compartment and wing lockers' locks,						
	Replaced 1	Replaced nose tire and tube and greased nose wheel bearings.						
		endejas AP26						
	1h	he olis	21					
	t p	Cean						
	Chino A	viation l	nc					
	7000 Merri	ll Ave. Hanga	ar A-340 Chino CA 91710					
	N5288J Ho	-	04/10/2014 Aircraft					
	Troublesho	ot starting iss	ue at Yuca valley airport, found INOP fuel pump on right side,					
1	de-fueled a	ircraft and rep	placed electric fuel pump with P.n.2B7-29. Wired newer pump in					
	accordance	with Cessna	service manual. Serviced with fuel and released to service.					
			nd tube and reinforced arm rest on aft left seat.					
_		ndejas AP260						
	+	+	Ja wegin					
	Chino Aviation Inc							
	7000 Merrill Ave. Hangar A-340 Chino CA 91710							
in (123)	N5288J Hobbs: 1765 12/03/2013							
	Replaced	INOP hobbs	meter on heater, Old hobbs meter was reading 615.9 hours. On					
- See a	heater an	nd New hobbs	meter hours 0000. complied with AD-2000-01-16 par. B by					
6.65 (P	inspectio	on of exhaust	system per customer's request.					
-	Agustin	Cendejas AP	2606564 Mandalan					
	**		upp - constraint					
				-				
		SUB-TOTAL th	lis page					
		TOTAL-Carry	forward to next page					
		and the second se						

DATE 6/18/2014 A/C TSN: 5888.7 HOBBS: 1791.3

MAKE CESSNA MODEL: 421C S/N: 421C0158 REG. NO: N5288J WORK ORDER: 20140505-2M



Hangar One Avionics Repair Station No. 1HGR058B 2026 Palomar Airport Road Carlsbad, CA 92011 Phone: 760-929-2270

Airframe Entries

Completed Annual Inspection IAW the Cessna Inspection Check list as applicable and FAR 43 Appendix "D"

Completed General Inspection of all aftermarket installations not listed with specific instructions for continued airworthiness---Inspected IAW AC43-13-1B Chapter 5 Section 2

CW AD# 84-26-02 (Paper Induction Air Filter) for LH and RH engines by replacement of Induction Air Filters 2ea P/N P10-8421. Replacement mandatory ea 500hrs.

CW AD# 2000-01-16 (Exhaust System Inspection) IAW para b -- no defects, due again the latter of 50 hours or 30 days.

Performed visual inspection on Halon fire extinguisher and reweighed. Weight: 4lbs 12.5oz

Replaced fire sleeve on Right Engine breather with 3 feet Aero quip AS102-12.

Replace vacuum hoses on Right Engine with 8 feet of MIL-H-6000 hose, 8 feet of Aero Quip AS102-16 fire sleeve and 6 Breeze 200-12H clamps.

Removed and replaced broken exhaust tail pipe bracket with new P/N 5155186-1 on RH engine, outboard location.

Installed new cotter pin and clevis pin P/N MS20392-2C11 on RH alternate air control at air filter assembly.

Rerouted RH propeller control cable and secured to avoid exhaust.

Replaced adel clamp for LH prop cable with new P/N MS21919WCH12

Freed stiff LH and RH alternate air controls

Verified Propeller synchronization system still properly deactivated and placarded in accordance with FAR 91.213.

Replaced Main Landing Gear Hydraulic System Filters 2ea P/N AN6235-2A and 2ea O-rings P/N MS28775-222.

Removed and replaced RH brake phenolic shim with new P/N 068-03100

Installed new Cam loc fastener P/N S3275-102R in nose baggage.

Installed Flush head rivet P/N CR3214-4-2 in cabin door with sealant and painted to match finish.

Replace LH and RH inner and outer fuel cap o-rings with new 2ea P/N MS29513-010, 2ea P/N MS29513-338.

Installed new center button and circlip on LH fuel selector using selector kit P/N B9-61-1

Removed broken lanyards from LH, RH and Aux fuel caps and replaced with new 3ea P/N MS27380-1

Inspected LH and RH main fuel strainers, cleaned and reassembled with new seals 2ea P/N MS29513-233

Cleaned corrosion near base of 3 GPS antennas on cabin roof, sealed antennas, and painted white to match.

Cleaned corrosion near base of Comm antenna on cabin roof, sealed antennas, and painted white to match.

Capped and stowed unused wiring in LH engine compartment.

Removed and replaced static wick on LH wing with new 1ea P/N 16165

Removed and replaced static wicks on RH wing with new 2ea P/N 16165

Radio lights deactivated and placarded in accordance with 14 CFR Part 91.213.

Repaired right tachometer wiring by installation of new MS3456L10SL-3s connector and M85049/52-1-10N strain relief to RH tachometer generator position.

LH landing light deactivated and placarded in accordance with 14 CFR Part 91.213.

Verified Pulselite system still properly deactivated and placarded in accordance with FAR 91.213.

Replaced LH and RH Vacuum relief filters 2ea P/N RAB3-5-1.

Vacuum / Replaced central vacuum filter P/N AM103535-IA

Removed RH evaporator cover, trimmed to fit properly, and secured using new screws.

Removed and replaced cabin door aft cable with new P/N 5111508-71CR and Fwd cable with new P/N 5111561-27CR using new roll pin in fwd bracket 1 ea P/N NAS561CF6-18.

Removed oxygen cylinder, sent for sublet hydrostatic testing (reference S.J. O'Brien Ent., Inc. work order HY 39723 dated 5/29/2014), then reinstalled IAW Cessna Service Manual. Next Hydro due 5/2019

Installed new O2 overboard discharge indicator disc from assembly P/N CM3578-1

SIGNED

Annual / I certify that this AIRCRAFT has been inspected in accordance with an Annual Inspection and was determined to be in airworthy condition --end--

This aircraft, airframe, or appliance identified above has been repaired and/or inspected in accordance with current Regulations of the Federal Aviation Administration and Manufacturer data and is approved for return to service.

DATE: 6/18/2014

Mike Clingerman Certified Repair Station No. 1HGR058B

TOTAL-Carry forward to next page

Printed by EBis 3 (datcomedia.com)

Work Order: 20140505-2M

DATE	TOTAL TIME IN SERVICE	TACH OR RECORDING METER TIME		F WORK PERFORMED— NO. OF PERSON PERFORMING WORK
	2.2	TOTAL bro	ught forward from previous page	and a state of the second state of the
	1.19	1.		
IAKE: CESSNA IODEL: 421C /N: 421C0158 EG: NO: N5288J /ORK ORDER: 20	140505-2M		Hangar One Avionics Repair Station No. 1HGR058B 2026 Palomar Airport Road Carisbad, CA 92011	DATE: 6/18/2014 A/C TSN: 5888.7 HOBBS: 1791.3
vionics Entr	ios		Phone: 760-929-2270	
	ELT has been in	spected in accord	dance with the requirements of FAR Section 91.207(d)) and was found to be in satisfactory condition. Battery
e-terminated ca	able at ELT with	new RJ11 conne	ctor. Performed ops check via ELT remote no discrep	pancies noted at this time.
his aircraft, airfra anufacturer data	me, or appliance and is approved	e identified above h I for return to servi	has been repaired and/or inspected in accordance with cu ce.	rrent Regulations of the Federal Aviation Administration and
ATE: 6/18/2014	Ļ	SIGNED	. 1911	Work Order: 20140505-2M
		Mike Cline Certified F	erman tepair Station No. 1HGR058B	Printed by EBis 3 (datcomedia.com)
		124022245	[•
KE CESSNA DEL 421C 421C0158	T	ONV	Hangar One Avionics Repair Station No. 1HGR058B 2026 Palomar Airport Road	DATE: 6/18/2014 A/C TSN: 5888.7 HOBBS: 1791.3
ORK ORDER: 201	40618-2M		Carlsbad, CA 92011 Phone: 760-929-2270	
ted Feb 12, 201 lance negligible publeshot No TI 111 to LH engin	4 completed by T indication on a e and left the su	Hartzell Engine T JPI indicator on R wapped operation dentified above ha	H engine by swapping TIT probes LH to RH and foun al TIT probe on the RH engine, ops check satisfactor s been repared and/or inspected in accordance with curr	nator P/N ALV-9610, S/N H-O020366, (see FAA form 8130-3 nber H-O020366). Ops check satisfactory. Weight and and TIT probe to be inoperative. Installed new TIT probe P/N y. rent Regulations of the Federal Aviation Administration and
nufacturer data a	nd is approved f	or return to service	Ch	
TE: 6/18/2014		SIGNED	10/	Work Order: 20140618-2M
		Mike Clinge Certified Re	man pair System No. 1HGR058B	Printed by EBis 3 (datcomedia.com)
WART SALE				
	7000 Me N5288J Installatio Faa 337	Aircraft ho on of Micro form filed,	Innary A-340 Chino CA 91710 bbs: 1825 02/13/2015 o vortex generator kit on wings per no change on airspeed indicator of 6606564IA	er STC. # SA5193NM or systems.
		····		
		SUB-TOTAL	this page	

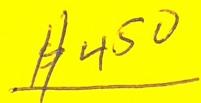
DATE	TOTAL TIME IN SERVICE	TACH OR RECORDING METER TIME	DESCRIPTION OF WORK PERFORMED— SIGNATURE & CERTIFICATE NO. OF PERSON PERFORMING WORK
		TOTAL broug	ght forward from previous page
	7000 M N5288J Inspecte Opened jacks an bearings replaced balance on exha for reco I certify Inspecti	Aircraft he ed this aircr inspection d performe s, tested and d, is next du weight, no ust system, gnition ligh that this Ai on and four	#33 Hangar A-340 Chino CA 91710 obbs:1805 05/17/2015 Aircraft TT: aft in accordance with FAR 43 Appendix D for annual inspection panels, lubricated control cable pulleys and rod ends, set aircraft on d landing gear tests, serviced hydraulic reservoirs, greased all wheel d inspected ELT in accordance with FAR 91.207, ELT battery the December 2017, complied with AD82-26-05 by inspection of rudder cracks found, complied with AD 200-01-16 by inspection per par©, replaced recognition light retainers on cowlings, replaced flasher box its.
	Agustili		AP2606564IA Color
>	S		NOTES
FAR	411-413	~	Fels. 23, 2016 APR 411 + 413 Try
	7124	-	Approvable Avionics Under- Nr. Or 2487.
-			
- 7 N Ir O ja be re ba or	000 Merrill 5288J Air spected this pened insp cks and per earings, tes placed, is r alance weign exhaust sy certify that	craft hobbs is aircraft in ection pane rformed lan ted and insp next due De tht, no crack ystem, repla this Aircraft	Hangar A-340 Chino CA 91710

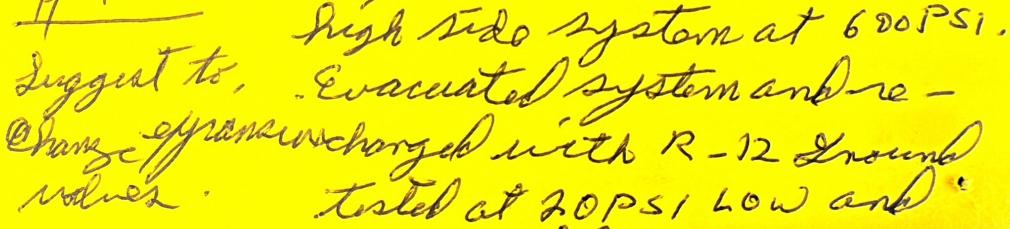
SUB-TOTAL this page

TOTAL—Carry forward to next page

N52885 C-421 6-29-15 freen Slez = #200 John - served \$250 Change

Executed N/c system. Tourf





210 PSI high 1 Don Alhson ALP 1481667 I-A,

Affordable Avionics Inc. 7000 Merrill Ave # 18 Chino, CA 91710 8AAR592Y LOG ID# 2317 23-February-2016 WO# 2482 AC TT 171.0 Pg 1/1 S/N 421C0158 CESSNA 421C N5288J

1.Removed existing GTX 327 Transponder. Installed GTX 330 ES Transponder. FAA form 337 filed. Weight and Balance and Equipment list updated. Post installation ground check carried out satisfactory.The Transponder System was inspected and was found to comply with FAR 91.413 and Appendix F of FAR 43. Test Equipment used : ATC 601-2 S.N. 6616 Date of Cal 01/16/2016 2.Upon inspection found heading flag does not go away on Glideslope indicator. Swapped connectors from indicators 1 & 2 leading form gyros for troubleshooting purposes. When connector harness for Gyro 1 was switch to Glideslope indicator 2, indicator worked. Connector harness from gyro 2 moved to indicator 1, indicator didnt work. Suspect issue with wiring harness leading from Gyro 2 or Gyro

2 itself. Recommend customer have gyro #2 benchchecked. 3. Upon inspection found Static leak of 300 Ft per minute at 4,000 Ft. Isolated leaks to CoPilots side. All static instruments had push on hose without clamps. Installed six hose clamps and checked for leaks.

The Altimeter(s), static system, Transponder(S) and encoder were inspected upto 30,000 Ft. and were found to comply with FAR 91.411 and 91.413 and Appendix E and F of FAR 43. Test Equipment used DPST-8000M S.N. 040115-96 Date of Cal 06/05/2015

ATC 601-2 S.N. 6616 Date of Cal 01/16/2016

Transponder P.N. GTX 330ES S.N. 84143078

Fame W. Monin 02/23/2016 Signed Date

DATE	TOTAL TIME IN SERVICE	TACH OR RECORDING METER TIME	DESCRIPTION OF WORK PERFORMED— SIGNATURE & CERTIFICATE NO. OF PERSON PERFORMING WORK	
		TOTAL brought	t forward from previous page	
	7000 Merr N5288J H Replaced I	lobbs:1878.6.	Hangar A-340 Chino CA 91710and 191.0 on panel.12/14/2016ump P.N. 2B7-29 on right side.	
7	2000 Merr N5288J H Replaced y Replaced lo over on co	obbs: 1886 1 acuum pump eft main tire a	Angar A-340 Chino CA 91710 /10/2017 Airframe on right engine with P.N. AA441CC Serial no. OBSP08 nd tube. Removed magnetic compass to Re covered plastic hield with new fabric material, then re installed.	
			11, in .	
	7000 Merril N5288J Ai Inspected th Opened insp jacks and po bearings, te replaced, is balance we on exhaust I certify tha Inspection	rcraft hobbs: 192 nis aircraft in acc pection panels, h erformed landing ested and inspecto next due Decem ight, no cracks for system, at this Aircraft ha	gar A-340 Chino CA 91710 20 07/09/2017 Aircraft TT:6017.4 Fordance with FAR 43 Appendix D for annual inspection ubricated control cable pulleys and rod ends, set aircraft on g gear tests, serviced hydraulic reservoirs, greased all wheel ed ELT in accordance with FAR 91.207, ELT battery aber 2017, complied with AD82-26-05 by inspection of rudder bound, complied with AD 200-01-16 by inspection per par C, as been inspected In accordance with an Annual borthy at this time.	
1223		SUB-TOTAL thi	S Dage	

1

Page No. TOTAL TACH OR DATE TIME DESCRIPTION OF WORK PERFORMED-RECORDING IN METER SIGNATURE & CERTIFICATE NO. OF PERSON PERFORMING WORK SERVICE TIME TOTAL brought forward from previous page AIR IMPRESSIONS, INC. 7929 Karl May Dr. Waco, Texas 76708 CRS No. AK6R541N Date: 07/20/2017 Hobbs: 1928.4 Cessna S/N 421C 0158 N5288J On above referenced aircraft, troubleshot air conditioner system and checked for Freon leaks, no leaks found. Serviced air conditioner system. Checked operation of system and found both expansion valves have been updated to the new manually adjustable versions without adjustments being made. Performed expansion valve adjustment procedures. Air conditioner system operational tested satisfactory after adjustments made. Repaired plastic air conditioner duct behind co-pilot seat at sidewall location. Repaired cabin door to allow for proper step deployment by installed new link assembly P/N 5111508-63. (See 8130-3.) Adjusted elevator servo cable tensions. Adjusted all landing gear struts with nitrogen. Fabricated and installed new lower entry door forward lower cable attachment due to excessive wear. Aircraft is approved for return to service for maintenance performed. Pertinent details on file under work order 4496. Air Impressions, Inc. CRS# AK6R541N Advantage Aviation, Inc. 7000 Merrill Ave. Box 20 Hangar A-240 Chino, CA 91710 VZKR081Y LOG ID# 2672 17-November-2017 WO# 7851 AC TT 261.0 HOBBS 261.0 CESSNA N5288J S/N 421C0158 421C Pg 1/1 Removed from aircraft: GNS 430w, GNS 530w, VSI, HSI System, Turn & Bank, Inverter, GDL 49, Radar System, M811B, Airspeed Ind., EDM 760, Attitude Gyro, Directional Gyro, Engine Gauges, Shadin Fuel Flow, Altimeter Installed in aircraft: G600 System (QTY 2), GTN 750, GTN 650, GTX 345, WX-500, GTS 800, GWX 70, EDM 960, Flightstream 510, New Flight Panels I CERTIFY THAT ENCODER, ALTIMETER, AND STATIC SYSTEM TESTS REQUIRED BY FAR 91.411 HAVE BEEN PERFORMED AND FOUND TO MEET THE STANDARDS OF FAR PART 43 APPENDIX E. THE ALTIMETER HAS BEEN TESTED TO 30,000 FEET. PERFORMED ATC TRANSPONDER TEST AND INSPECTION. UNIT FOUND TO COMPLY WITH FAR 91.413 AND APPENDIX F OF FAR 43. For more information, see FAA form 337. Dated 11/17/17 Mark Krueger VZKR081Y SUB-TOTAL this page TOTAL-Carry forward to next page

DATE	TOTAL TIME IN SERVICE	TACH OR RECORDING METER TIME	DESCRIPTION OF WORK PERFORMED— SIGNATURE & CERTIFICATE NO. OF PERSON PERFORMING WORK			
		TOTAL brought forward from previous page				
				-		
				_		
				_		
	1					
				-		
				-		
				-0		
			l l lb			
			() ()			
			1			
			U	-		
	and the second second			-		
				-		
				-		
	/	1		-		
	(_		
				1		
		SUB-TOTAL	this page			

.

